



TECHNOLOGY TRANSFER PROGRAM [TTP]

FINAL REPORT

ENGINEERING & DESIGN

# ENGINEERING & DESIGN VOLUME 2 APPENDICES

Prepared by:

Levingston Shipbuilding Company in conjunction with: IHI Marine Technology, Inc.

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# APPENDIX A BRIEF EXPLANATION OF IHICS



# BRIEF EXPLANATION OF

Integrated Hull Information Control System



OCT. 1978

## Ishikawajima-Harima

Heavy industries Co., Ltd. TOKYO JAPAN

REF. No.

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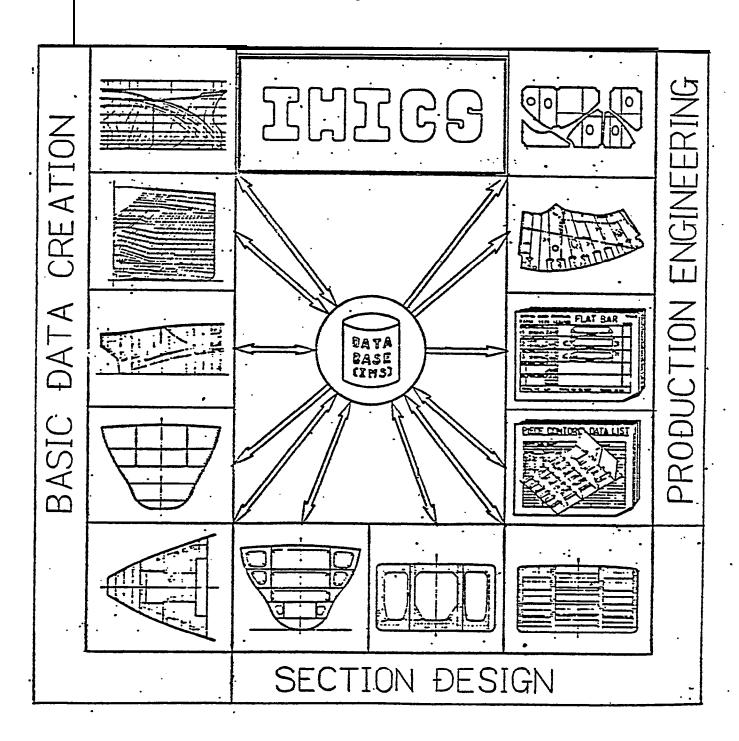
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#### INTRODUCTION

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#### INTRODUCTION

THICS (<u>Intergrated Hull Information Control System</u>) is a series of program packages which assists engineers in the fields of design and production engineering of hull construction, and also furnishes them with all information, necessary for the execution of production.



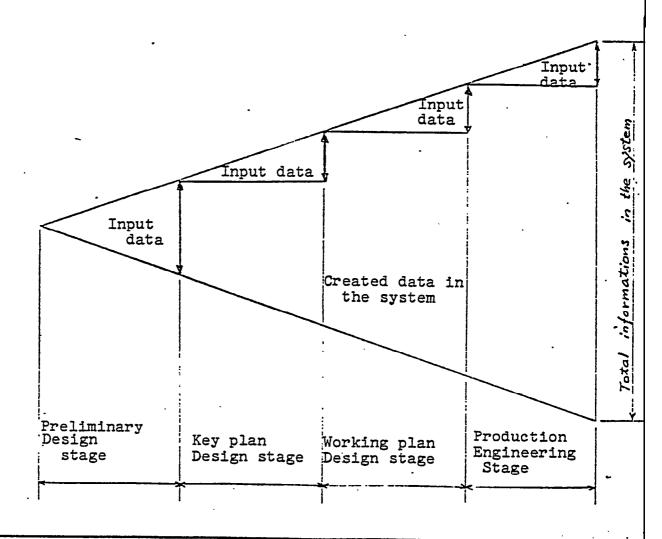
- 1. Problems in Manufacturing Division.
  - There are many problems in the manufacturing division of shipbuilding today.
  - (A) Increase of ship's type to be constructed which has been caused by ship's market.
  - (B) The method of transmission of the large volume information and data from the design division to the manufacturing division, for instance,
    - -Numerical Control Data
    - -Production Engineering Data
    - -Production Control Data
  - (C) Requirements of highly precise data and information.
  - (D) Delivery in short time
- 2. IHICS solves the above problems.
  - (A) Generates the engineering and production data from a small volume of input data prepared by engineers.
  - (B) Assists engineers in the design and production engineering activity.
  - (C) Creates the full part data base which supplies following information to the manufacturing division.

Information of production engineering.

Numerical Control Data.

Piece list for each stage.

- 3. Scope of the System.
  - \* Covers the detailed design and production engineering for hull.
  - \* Excludes the functions of origination of design concept nor structural analysis.
  - \* Maximum output from minimum input.
  - \* Illustration of the proportion of the input required at each stage to the whole information in the system is shown in following figure.



4. Ships having been applied in IHI (as of May, 1978)

\* 470,000 DWT Tankers

\* 250,000 DWT Tankers

\* 100,000-150,000 DWT Tankers

\* 15,000-100,000 DWT Bulk carriers

\* 20,000-100,000 DWT Combination Carriers

(Ore/Oil,Bulk/Oil)

\* 20,000- 30,000 DWT Containers

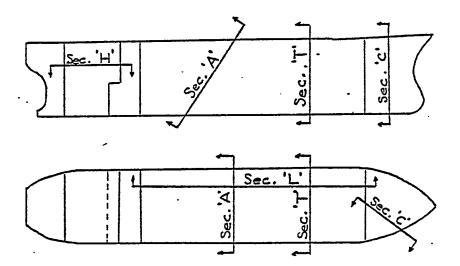
\* 15,000- 30,000 DWT cargos

- \* Floating docks
- \* Platfrom-mounted pulp plant
- \* Large derric barge

5. Specific Features of the System.

Specific features of IHICS are the following.

- (A) System is based on data base concept.
  System data base is under the control of
  IBM Information Management System (IMS).
- (B). 3-D Process and Functional Offset Data. (Theree dimensional Process)
  This technique allows to retrieve geometric data in any position and any section.



Transverse Sec.

Sec. 'T'

Horizontal Sec.

Sec. 'H'

Longitudinal Sec.

Sec. 'L'

Any other cut sec.

Sec.'A','C'

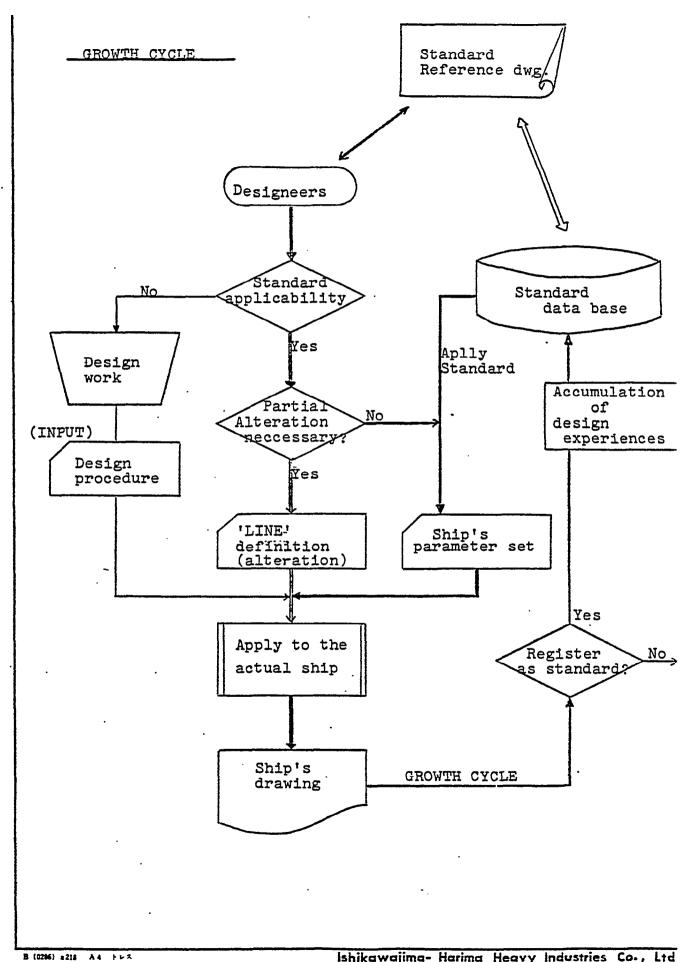
These section data can be obtained through 'Cut plane program.'

- (C) "LINE" Statements.
  - \* "LINE" Statements (Language for IHI Numerical control Engineering) are developed for means of <a href="intercomunication">intercomunication</a> between designers and the system.
  - \*-Easy expression of ship's design **figures** and descriptions of design standard data are possible by use of these statements.
- (D) Isolation of Technology and Accumulation of Standards.
  - \* The system is isolated from design technology.
  - \* The "STANDATD" is <u>maintained</u> by engineers

    followed by the progress of the design technology.
  - \* These standards will be accumulated in the system and will grow towards a high technical design system with time in the same way as the accumulation of a designer's experiences would make him an expert.

#### STANDARD

- Shape standard:slot (Longitudinal cut out),
   hole, scallop Bracket, Stiffener
- Standard how to select/apply standards
- Fabrication standard
   excess, edge preparation



#### (E) Relative expression of Data format

"LINE" is designed to describe objective figures to be designed in relative expression as far *as* possible so as to minimize corrections caused by the alteration of design.

This concept is coherent in all subsystems.

#### (F) Flexible operation

IHICS can be selected the most convenient usage of each subsystem againt the given circumstances such as ;

- Applied ship
  - \* Newly designed ship
  - \* Sister ship
  - \* Repaired or reconstructed ship
- Allowable designing period
- Computer Hardwares & Machinary for Fabrication

  Typical selection can be seen in the following table.

#### 2. Composition and usage of Subsystems

case Ŝybsystem	A	В	C	D	Notes
Basic data creation subsystem	*	*	*	*	* Program language for
Section design Subsystem		*	*	*	applications : PL/I : Optimizing
Production engineering subsys.	* .	*	*	*	compiler. * Data base
Data base control system	* *		*	*	control: IMS DB/DC
D/C-I Character display			*	*	* Required core size : 512KB
D/C-II Graphic display				*	

#### (G) Online Capability (Option)

IHICS supports many online terminals under IMS/DB.DC capability.

#### 6. Background of IHICS

IHICS is based on the total hull design systems developed by IHI in 1971.

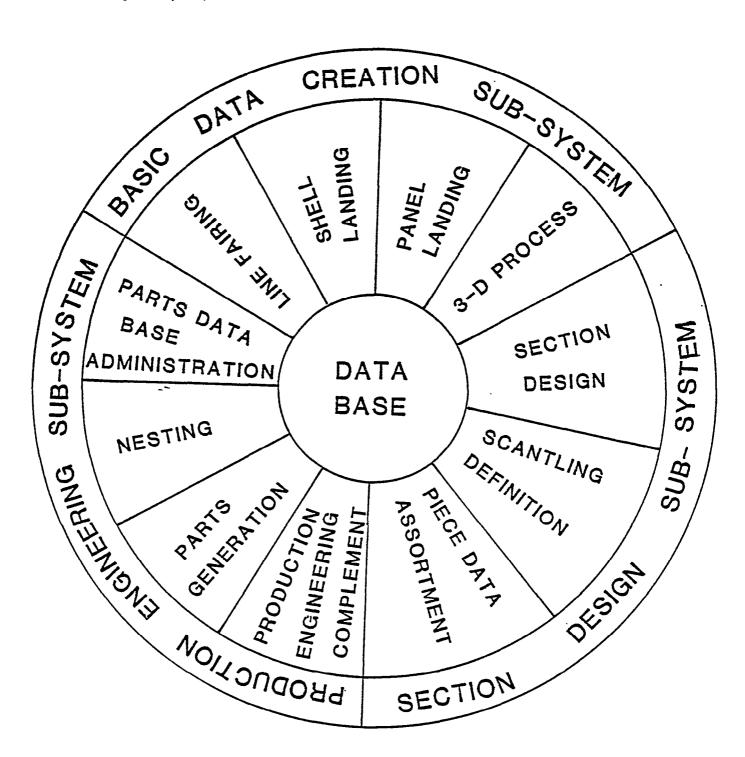
The new IHICS includes the suggestions obtained from the past ten year's experience.

#### IHICS PERSPECTIVE

IHICS is composed of the following three sub-systems:

- \* Basic Data Creation Sub-system
- \* Section Design Sub-system
- \* Production Engineering Sub-system

The system perspective and outline are shown in the following figure.



#### BASIC DATA CREATION SUB SYSTEM

#### FAIRING OF A SHIP'S HULL LINES FAIRING PROGRAM

#### :\* SHELL LANDING PROGRAM

- CONSISTS OF SEAM/BUTT LANDING PROGRAM
  - .LONGITUDINALS LANDING PROGRAM
    - .SCANTLING DEFINITION PROGRAM
- \* PANEL LANDING PROGRAM

- CONSTSTS OF SEAM/BUTT LANDING PROGRAM
  - LONGITUDINALS LANDING PROGRAM
  - SCANTLING DEFINITION PROGRAM
- \* 3-D PROSESS PROGRAM

- CONSTSTS OF . PANEL DEFINITION PROGRAM
  - COMPARTMENT DEFINITION PROGRAM
  - .CUT PLANE PROGRAM
  - . PANEL COMPOSITION PROGRAM.

#### ...OUTPUT...

- .GEOMETRY DATA BASE, PANEL DATA BASE, SCANTLING DATA BASE
- A COMPLETE DRAWING OF ANY DESIRED PORTION OF LINES DRAWING
- THE BOOK OF MOLD LOFT OFFSETS
- STRUCTURAL BODY PLAN(1/10, 1/50)
- .SHELL EXPANSION PLAN
- PANEL PLAN DECK/BULKHEAD/FLAT/.....

#### SECTION DESIGN SUB SYSTEM

- \* SECTION DESIGN PROGRAM

  - CONSISTS OF .WEB'S FIGURE DEFINITION
    - •STIFFEMER & JOINT ARRANGEMENT ON A WEB
- \* SCANTLING DEFINITION PROGRAM
  - CONSISTS OF • WEB/FACE PLATE SCANTLING DEFINITION PROGRAM
    - STIFFENER SCANTLING DEFINITION PROGRAM
- \* PIECE DATA ASSORTMENT PROGRAM
  - CONSISTS OF ·PIECE DATA ASSORTMENT PROGRAM
  - · · · OUTPUT · · ·
    - SECTION PLAN(1/10, 1/50)
    - .PIECE CONTROL DATA LIST
      - .PRESENTS PIECE LISTS EACH ASSEMBLY UNIT.
      - INCLUDES PIECE NAME, QUANTITY, SCANTLING, WEIGHT, PIECE DWG FORMAT, FABRICATION PROCESS, AND OTHER PRODUCTION CONTROL DATA.
      - AFFORDS FACILITIES FOR DATA CORRECTION.

#### PRODUCTION ENGINEERING SUB SYSTEM

\* EDITTING PROGRAM

CONSISTS OF PART PROGRAM GENERATOR

PLATE EDGE MODIFIER PROGRAM

\* PART GENERATION PROGRAM

CONSTSTS OF SHELL PLATE DEVELOPMENT AND ASSEMBLING

DATA CALCULATION (SHELL)

·LONGITUDINAL/TRANSVERSE FRAME DEVELOPMENT

PROGRAM (LODACS)

·INTERNAL STRUCTURE DEVELOPMENT (LINE

SYSTEM)

(WEB PLATE, FACE PLATE, STIFFENERS

AND OTHERS)

\* NESTING PROGRAM

CONSISTS OF \*MANUAL NESTING PROGRAM

· INTERACTIVE NESTING PROGRAM BY CADS

• POST PROCESSOR FOR NUMERICAL CONTROL

MACHINE

\* PART DATA BASE

ADMINISTRATIVE PROGRAM

CONSISTS OF .PART DATA BASE HANDLER

• PIECE LIST EDITTING PROGRAM

FOR FABRICATION

SUB-ASSEMBLY

**ASSEMBLY** 

ERECTION

#### ··· OUTPUT ···

- .PIECE DRAWING (INCLUDING TABLE FORMAT)
- .NUMERICAL CONTROL DATA/TAPE
- .PIECE LIST FOR EACH STAGE
- TEMPLATE FOR BENDING (SHELL PLATE AND LONGITUDINAL

FRAME)

- .BLOCK MARKING DATA FOR SHELL
- .JIG HEIGHT FOR ASSEMBLING CURVED SHELL BLOCK

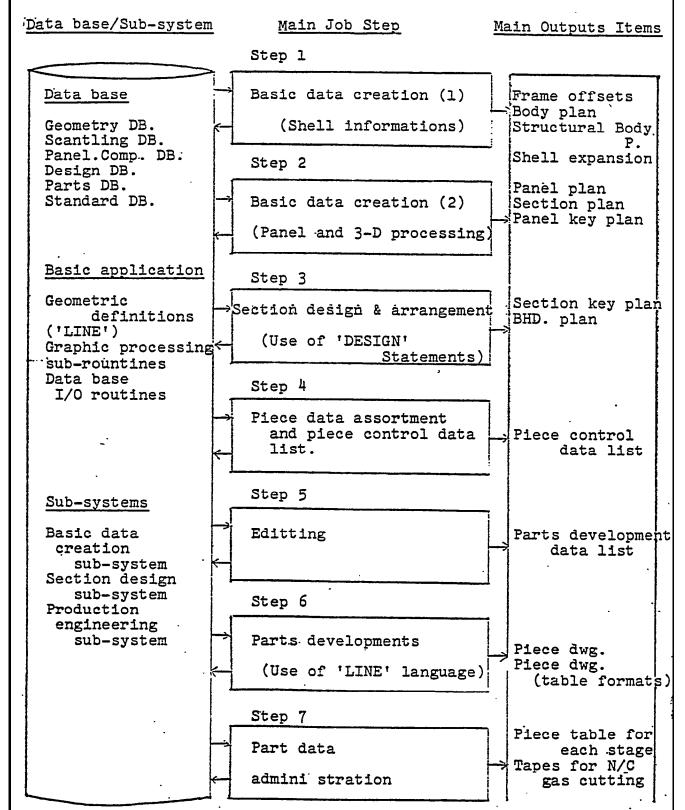
#### 'APPLICATION AREA'

STAG	E	ITEM.	PROGRAM	DATA BASE
PRELIMINARY	DESIGN	GENERAL ARRANGEMENT & PLANNING SCANTLING CALCULATION LINES		C=====
DETAILED	DESIGN	SHELL & PANEL LANDING SECTION DESIGN LOCAL SCANTLING CALCULATION WORKING PLAN		
PRODUCTION	ENGINEERING	LINES FAIRING PIECE LIST ISSUE PARTS GENERATION BLOCK ASSENMBLY NESTING N/C DATA		
PRODUCTION	CONTROL	SCHEDULING & OTHER PRODUCTION CONTROL		

#### GENERAL SYSTEM OPERATION FLOW

The diagrammatic general system operation flow is shown in the following figure, in which main job steps, main output and main program modules are indicated.

The detaied explanations of each step are to be referred to the following sections.



#### FUNCTIONS OF "LINE" STATEMENT

1. Objection for 'LINE' statements

'LINE' statement is developed for common language of the geometric expression of hull form and structures.

It is a kind of problem oriented language of which format is free.

#### 2. Functions of 'LINE' statements

#### (A) Geometric definitions

Geometric expressions such as point, straight line, circle, tabcyl and thier composed line can be defined at any stage.

The definition method adopts rather relative expression.

- (B) Auto-reference function to standard data
- (C) <u>Contouring definition</u> (Web plate definition)
  This gives contouring of figure
  (it is called as 'Motion'). Cutter location data of plates are created through this program.
- (D) Opening definition

Man holes, drain holes, etc. are defined by this statement

(E) Plate edge information

Edge preparation for welding and margin amount of material are indicated.

(F) Data base reference

Easy access to or reference to geometry, design and data base is available.

(G) Marking line definition

For the definition of marking line only (water line etc.)

- (H) Stiffner development (Bracket, Flat bar, Inverted Angle) exact shape of a stiffener takitig account of **plate** thickness
  - Precise shape at its both end.
  - Bevel angle calculation.
  - Calculation for many types of stiffeners(more than 400) are built in program.
  - Any new type of stiffener can be easily registered.
- (I) Face plate development.
- (J) Collar plates and others.
- (K) Weight, Area, Marking length calculation.

Example expressions are shown in the next page.

#### 3. An example of 'LINE' descriptions

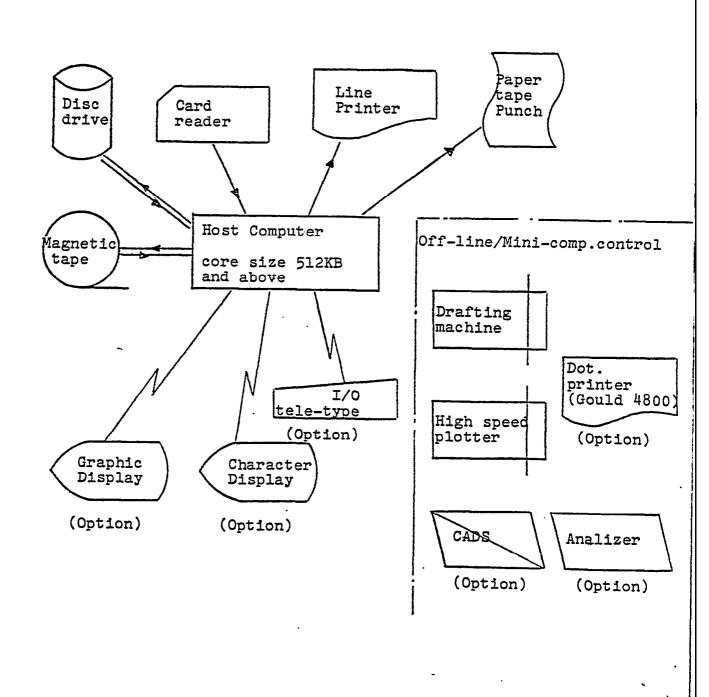
(An example for corner part of transv. section)

```
T
         F60
      T1=F60----GEOMETRY DB REFERENCE
                                              UPP.
      PA1=PA-UD, ML---PANEL DB REFERENCE
                                                             L21
                                                                   PII
      Pll=OUT(SL,UD,1)----DESIGN DB REFERENCE
      Sl=PAl,Pll
      S11=PR-SL,L=2000,D
Definition
      LONG=SL,L40,L59
                         GEOMETRY, SCANTLING
                                                                  L59
      LONG=UD,L20,L21
                         DB REFERENCE
                                                                L58
      C1=TD-S11,T1,-S2
      P1=CP-C1
Geometric
      P2=SLOT-PC2,UD-L20
      S5=PT-P2,PT-P1
      P3=INT(S5-C1),U
    P4=ON-C1,FROM-P3,GL=150,D
      P5=UD-L20, TOP
      Al=P100,S11,C1,S2,-----COMPLEX SURFACE
      S24=PT-P4, PT-P5, SCS=UD-L20, ECS=A1
      S21=SL-L57,
                      SCS=SL-L57,ECS=A1
      S25=UD-L21,
                       SCS=UD-L21,ECS=S23
 C
           -,Tl,MSL(SL,L40,L59),Tl,MSC(100),PAl,
                                                         ] CONTOURING
            MSL(UD, L21, L21), PA1, MBS(S5, UD, L20, P3, 21), C1) DEFINITION
      MH(1,P51,150)
 H
                                                          OPENING
                                                          DEFINITION
..X
      S21=F,FIT=A,PD=D,MRK=U,TYP+FC1S1,NAM=F15
 X
      S22=B,FIT=A,PD=D,MRK=U,TYP=BClSlSl,NAM=B16 PARTS EXPANSION
·E
                                             END OF PART PROGRAM
```

#### COMPOSITION OF AVAILABLE HARDWARES AND SUBSYSTEMS.

Required hardwares and subsystems for smooth operations of this system are as follows.

 Available hardwares for this system
 Facility of host computer should be determined taking any other system's applications into consideration.



#### APPENDIX B

<u> IHICS - ACTUAL OUTPUT EXAMPLES</u>



# IHICS

**Actual** Input/output Examples



OCT. 1978

# Ishikawajima-Harima

Heavy industries Co., Ltd. TOKYO JAPAN

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′	****	E Language	1	****					
	*****			*****			Example	of partmy	yam
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1	INPUT_NO	DEFINITION	alcomment.	, and the same of			w De	sigm-Eubsysta	<b></b> ,
		- 1						_	
,	TITLE		, W1	P160F /	APFF23	1ER121LE00050		_	
r	0001	CO09 = 1N(CO09+R2				1ND00100			
	2000	5014 = 1N(S014,R2				1ND00150			
1	0003 0004	CO10 = JN(CO10,R2 S005 = IN(S005,R1		•		1ND00200 1ND00250			
	0004	C003 = IN(C003,RI				1ND00230	Flamout	surface of a	wah trans
	0066	\$006 = IN(\$006,R1				1ND00350		, ,	, ,
7	0007	5013 = IN(5013,R2	)			IND00400	(Ratriau	ed from the D	45:am-DB)
	9000	' C004 = IN(C004-R1				1ND00450		. *	
(	0009	LONG = SL , F23		318		L%G00500	Longitudio	nal data razrie	vel
•	0010 0017	LONG = EFO2, F23	, L6 ,L	10		LNG00550	_ ' .		
•	0012	T011 = F23				AUX00650		ne retrieved	
•	0017	5506 = PT-P1 757	7.7. 10849.91. PT	-Pt 6579.2. 100	96-01	AUX00750		at entire on 146.	
	0314	\$509 = PT-P1 455				D080010L			
	0015	S512 = PT-P( 54?			38.8)	JU100850			
,	0016	P150=1N1 (5509-500)					_	- 0	
	0017 _	P151=IN(5006,C07			<b>.</b>		. Ринч	Point	definition
1	001a 001¥	SOLL = SL-L30D, FCS=S04				STF00950		ea	
	0020	\$022 =\$L-L30,EC\$=P300 \$023 =\$L-L30A,EC\$=\$00		•		S1F01050	S x 44	2XAMILIN	t line definition
	0021	P300=INT(S22-C3) +0	TYDOS-SE ESON	•		31101030	C # **	Circle	definition
(,	0022	\$024 =\$L-L306, ECS=\$0	io,SCS=SL -L30G	-		STF01100	<b>G</b>	٠١١٠٠١	a-Jiman
	0023	S025 =SL-L31,ECS=S020	S,SCS=SLL31			STF01150	PA ** _	Inner p	and surface
<b>c</b> .	0024	\$030 =EF02-L10.ECS=5				STF01300			it kion
۲۰	0025 0026	\$039 =\$L-L308,EC\$=A1				STF01450		1	
	0027	\$041 =\$L-L29,EC\$=A11 \$044=PR-\$(CL),L=5		S=5022	•	\$1F01500	Tuna	manner Command	line definitio
(	·0028	S046 =SL-L30E.ECS=S00		J	* *	• •	THAM	Carver	ine dejining
	0056	S049=PR-S(CL),L=5		S=\$03 <u>9</u>			AANE	Grange	L. surface
_	0030	\$050 =FF02-L8, L=0,0,	CS=S055 .ECS=\$10	2		STF01700		1.	inition
٠	0031	S021=IN(S014,R2,P) _			- • • •	STF01750		,,,,,	
•	0032	CO22=IN(CO10,R2,P)		•		S1F02200			
<i>(</i>	0034	\$010=1N(\$005;R1;P) \$017=1N(\$012;R2;P)		·		STF02300 _ STF02350		••	
-	C035	P030=SLOT-PC2(1),EF0	!L9			STF02400			
	0036	PA02=EFQ2				STF02500			<del></del>
<b>(</b>	0037	\$003×EF02-L8,L=0,0			/= ····	\$1F02550	•		,
	0038	COZO=1N(COO9,R2,P)	0774 0			S1F02600			,
r	0039	. P504 = 5731.0, 1		A44 244 X X		S1F02700			
· .	0240 0641	P505 = 3194.2, P023=INT(SG49SG39_)				STF02750 STF02150			
	0042	\$009 =PR-SC21,L=15,0		11		\$TF00900			
(	0743	P025=3NT(S009 -S011		<del>-</del>		\$102000			
	0044	\$004=PA02,\$003				STF02250			
,	0045	S054 =PT-P(CP-C022)+	'R-\$004,\$C\$=\$010,E	ECS=SO17		STF02050			
,	0046	P031=CP-C020	. FFE-0030 FFE-11	•		STF02450			
	0047 0049	\$055 =PT-P130 • PT-P031				, STF02100			•
(	0049	P033=DN-S024 , FRJM-P( S026 =PT-P(EF02-LY,S1				STF01800 . STF01200			
	0050	P03Z=DN-S026 .FR04-P1				\$1F01850			
	0051	SO27 =PT-PISL-L31A.ST				STF01250			
	0052	\$100=FQ-5050				02601			

.

	0054 0755 0056 0057 0053 0059 0060 0061	\$107=*Q-\$046 A11=P150,\$095,C003, P034=1NT(\$U44 -5022) \$033 =PT-P034,PP-\$022, P024=ON-\$039 ,FROM-P023, \$039 =PT-P024,PT-P025, \$333=PR-\$(CL),L=\$100,0,\$ \$336=PR-\$333,L=\$45,0 \$334=\$L-L305,\$C\$=\$333,E6	,SCS=S022 ,ECS=S041 8,GL=500,0 5CS=S011 ,ECS=S039 5CS=S509,ECS=S22 		02603 \$7F01400 \$7F01350 \$7F01450 \$7F01400			<b>?</b> , (
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     ECS=P300
          SCS_AYGLE=__67.4
                              ECS_ANGLE= . 90.0 ... EXP_ANGLE= ...79.7
          SCS_THETA= 90.9
                              ECS_THETA= 90.0
          E_XS= 5.4686
E_XE= 4.4562
                              E_YS=
                                       5.7359
                              E_YE=
                                       6.1746
          SYMS(1)=S INDXS(1) = 246 SYMS(2) = L INDXS(2) = 6
          SYME(1)=S INDXE(1)=247 SYME(2)=
                                              INDXE(2) = 0
          SCS_PLANE
                          0.3815
                                     <u>-0.8427</u>
                                                  _0.3799_
          ECS_PLANE =
                          0.0000
                                      0.9336
                                                 -0.3584
                                                             -1.9304
         EXP_PLANE ...
                          0-1790
                                     -0.3526
                                                 -0.9185
                                                              3.7772 ..
          SCANTLING=
                       150.0
                              12.5
                                         0.0
         LENGTH=
                        1.0844
         WEIGHT=
                        13.9523
          EXP_NAME= F2
X 5039 =8,TYP=8C2C6F1,PD=U,NAH=124/81.F1
                                                           .MRK=D
                                                                        PEXP 03900
                                                                                       Stiffener (bracket) definition
    F1T=A,DATA(C=150,R=300)
                                                                         EXP03950
                                                                                             [ Bracket development built in program
---- SKT CL_DATA COUNT =
                                                                                            Results of a stiffener above defined
      0.0000
                  0.0000 20001.0000
                                           1.5821
                                                       0.0000 20002.0000
                                                                               1-5821
                                                                                          -0-1769 20011-0000
                                                                                                                   0.5396
     -0.4457 30012.0000 405000.0000
                                          0.6273
                                                      -0.7326
                                                                   0.2999
                                                                               0.3361
                                                                                          -0.6605 20013.0000
                                                                                                                   0.3226
```

```
-0.6670 20004.0066
                                       0.0234
                                                  -0.0492 402200.0000 406000.0000
                                                                                       0-0149
                                                                                                   0.0000
                                                                                                               0-0499
                                                                                                                           0.0649
              0-0000 402210-0000
                                     977.0000
                                                 -0.0492 402200.0000 406000.0000
                                                                                       0.0149
                                                                                                   0.0000
                                                                                                               0.0499
                                                                                                                           6-0449
        --- BKT MK_DATA COUNT =
         4050 )0-0000
                          0.2846
                                     -0.5080 405100.0000
                                                               1-5471
                                                                          -0-1220 40500D-0000 40410D-0000
                                                                                                               0.9541
                                                                                                                          -0-2533
              0.9158
                         -0.2650
                                    999.0000 405100.0000
                                                               1.5471
                                                                          -0.1220 405000.0000 404100.0000
                                                                                                               0-9541
                                                                                                                          -0.2533
        --- BKT CH_DATA COUNT =
        Fì
           ....- CL_DATA STORE END --
           ----- MK_DATA STORE END -
                                                                                         Stiffener development
             --- UP_DATA STORE END -----
                                                                                           * parts generation
           ---- VI_DATA STORE END ---
                                                                                              - graphical data and dimension
             --- CH_DATA STORE FND ---
                                                                                              -rtore results into DB
             ***** EXPANSION SURFACE (SO39) *****
                                                                                           * adje preparation (Baval & excess)
                SCS=SL-L309
                                     ECS=All
                SCS_ANGLE= 64.2
                                     ECS_ANGLE= 90.0 ... EXP_ANGLE=
                 SCS_THETA= 89.2
                                     ECS_THETA= 86.9
                                                       STR_THETA=
                E_XS=
                         6.1358
                                     E_YS=
                                             7-1020
                                                                                            marking data of a stiffener
                E_XE=
                         4-5560
                                     E_YF=
                                              7-1876
                SYMS(1)=5 INDXS(1)=246 SYMS(2)=L
                                                     INDXS(2)= 8
                SYME(1)=S INDXE(1)= 2 SYME(2)=F
                                                     INDXE(2)= 1
               . SCS_PLANE =
                               0.4354
                                            -0.8981
                                                         0.0623 __
                                                                                                direction of plate thickness.
                ECS_PLANE=
                                0.0000
                                             1.0000
                                                         0.0000
                                                                    -4.5560
                EXP_PLANE=
                                0.0000
                                            -0.0541
                                                        -0.9985
                                                                     7.4235
                SCANTLING=
                               12.5 100.0
                                              11.0
                LENGTH=
                               1-5821
                WEIGHT=
                               50.8340
                EXP_NAME= YZ
                                       . 81
       X SO41 =B, TYP=9C2C6F1, PD=U, NAM=13Y/B1.F1
                                                                  .MRK=D
                                                                               • EXP 0403Ó
(
           FIT=A,DATA(C=150-R=300)
C.
        --- BKT CL_DATA COUNT ×
             0.0000
                         0.0000 20001.0000
                                                  1.1209
                                                              0.0000 20002.0000
                                                                                      1-1206
                                                                                                 -0.1289 20011.0000
                                                                                                                          0.4406
            _0.3367__30012.0000_406000.0000
                                                 0.5283
                                                                                      0.2365
                                                                                                 20.5542 20013.0000
                                                                                                                          0.2226
            -0.5597 20004.0000
                                     0.0197
                                                -0.0497 402200.0000 406C00.0000
                                                                                      0.0149
                                                                                                  0.0000
                                                                                                              0.0479
                                                                                                                          0.0649
             0-0000 402210-0300
                                   999-0000
                                                -0.0497 402200.0000 406000.0000
                                                                                      0.0149
                                                                                                  0.0000
                                                                                                             0.0499
                                                                                                                          0.0649
       --- BKT MK_DATA COUNT =
        405000.0000
                         0.1952
                                    -0.3961 405100.0000
                                                              1.0856
                                                                        -0.1239 405000.0000 404100.0000
                                                                                                             0.6786
                                                                                                                        -0.1983
             0.6404
                        -0-2100
                                   999.0000 405100.0000
                                                                        -0-1239 405000.0000 404100.0n)0
                                                              1-0856
                                                                                                             0.6756
                                                                                                                        -0.1983
```

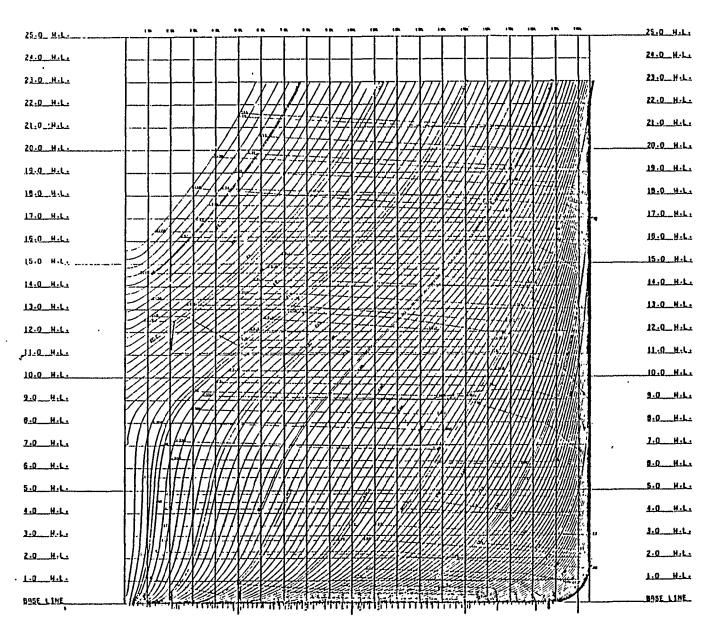
·							Ð	. •		4 1.4	•	_		
1 2715	2F35	1 7r/06		-+		-	rr	ocessak		parts list	- m	Su	mmary	_
SUB	PUZ_AG	PIECE	ı	LENGTH	1 (	CUT_LNG	ı	M¤K_Li+G	ı	WEIGHT	ER_COD		SURFAC	-
23	1	F2	1	1.0844	1	0.000	1	0.000	1		)	1	2055	-
23	j 72	81	H	1.5821	ĺ	3.878	1	1.320	1	50.8	)	- 1	5039	
23	Y2	Fl		1-3202	i	0.000	ĺ	0.000	Ĺ	11.4	J	- 1	S039	
23	1 Y1	<b>41</b>	- 1	1.1209	ĺ	2.903	1	0.931	1	30.3	1	- 1	5041	
23	Y1	rı .		0.9311	1	0.000	ŧ	0.000	ł	0.0	ı	- 1	5041	
23	1	F11 , ,	. I	0.7508	Ì,	0.000	1.	0.000	. 1	9.3	l .	. 1	S033	
23	1	<del>3</del> 12	- 1	0.0000	1	0.000	1	0.000	1	0.0	)	- 1	5044	
23	i	F3	- 1	0.6022	Ì	0.000	ı	0.000	Ĺ	8.0	1	- 1	5023	
23	j	A14	- i	1.9105	İ	0.000	İ	0.000	ĺ	60.4	]	Ì	\$049	
23	i i	F4	- 1	1.6884	İ	0.000	Ĺ	0.000	1	24.2	l	- 1	SOIL	
1 23	J	, F13	1	1.1400	Ĺ	0.000	1	0.000	ì	13.6	t	- 1	5036	
23	1	F16	1	1.5148	1	0.000	1	0.000	1	30.7	I	- 1	5004	
23	i	A15	1	1.3428	1	0.000	t	0.000	1	30.9	l	- 1	\$050	
23	1	F6	- 1	0.9827	1	0.000	1	0.000	ı	12.4	1	1	5024	
23	1	1 45	- 1	0.0000	Ì	2-286	1	0.000	1	22.2	Ì	ı	5046	
23	İ	1 1-9	Ì	1.3062	Ì	0-000	ı	0.000	1	17.6	)	1	5026	
23	ì	j F7	ĺ	0.7938	ĺ	0.000	1	0.000	1	9.6	ì	1	5025	
23	Ì	F8	i	1.1156	1.	0.000	Ĺ	0.000	Ì	14.3	Ι.	-1	5027	
23	ĺ	210	j	0.3243	ì	1.092	İ	0.000	Ĺ	5.8	l	Ì	\$030	
23	ì	F18	i	3.2304	İ	0.000	Ĺ	0.000	1	277.0	ł	Ì	\$333	

E23. 18. DET -34-2---FW 1/ WE B-A SEC 11111-1 - 23) £1001-11A> PILL E TIK OF 130,30-E. 24V 別日 TANK 19 505 8" CZKJIA-MES) B-D'SECT MINITER . MINITER (3032\_23A)@ F24 (P) 1 1 1 2 SNº 2715 MOT. (D32-23A) DAGN'S KIIII ZOZ < 032 -110 # 11 F23 SEC

193/34 1

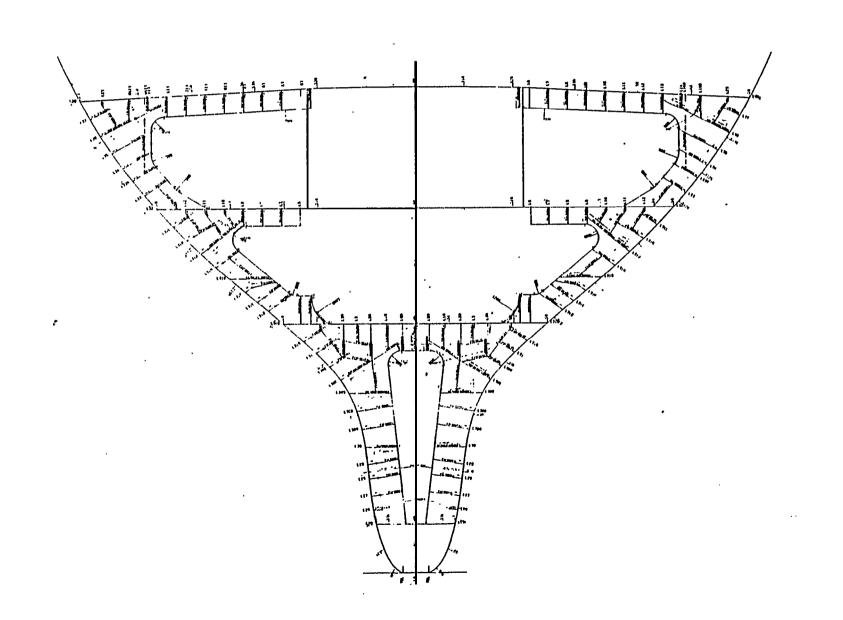
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## STRUCTURAL BODY FLAN

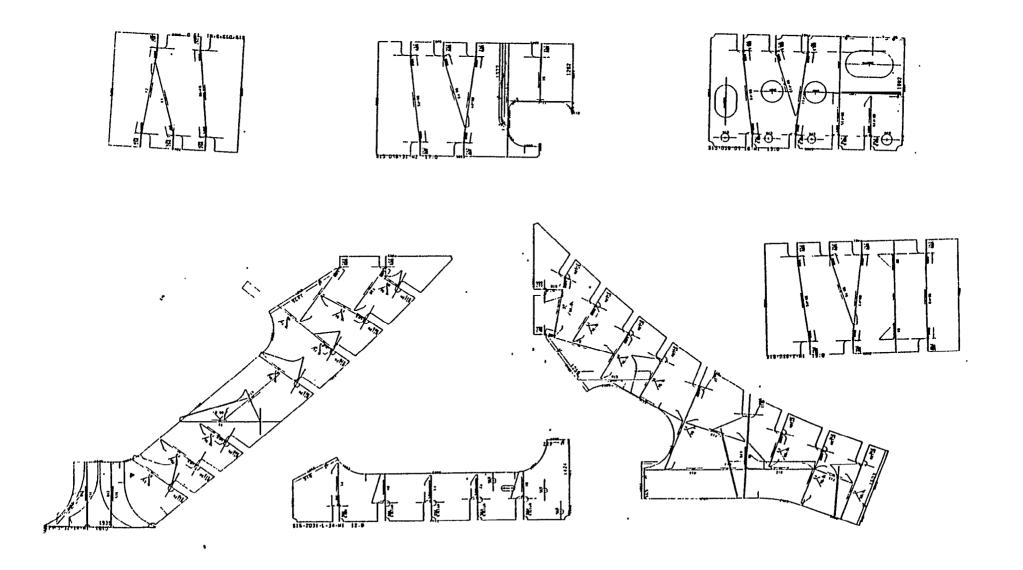


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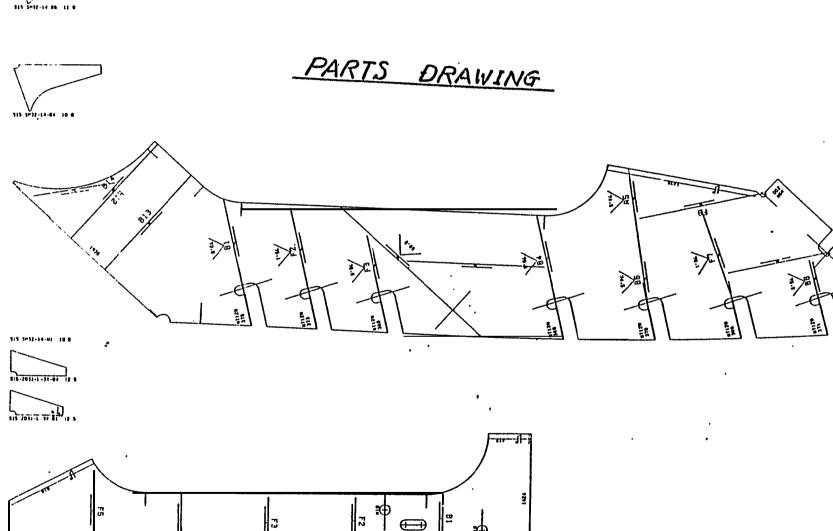
# SECTION DRAWING

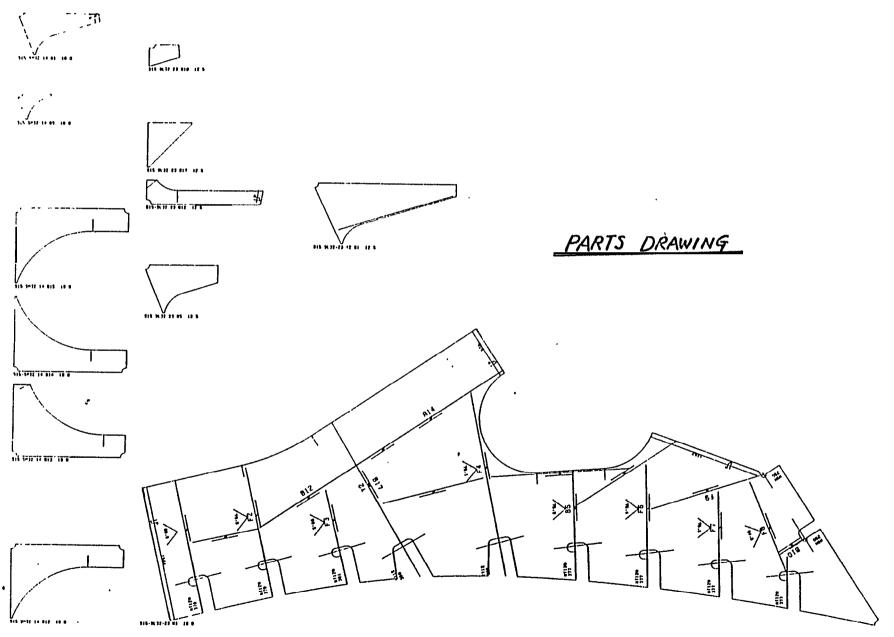


## DRWING FOR SUB-ASSEMBLY









SHIP NO PANE		FAB. SHOP FACE	PLATE
2715 DS HOPIRT 1 CODE DS-T1	150 ×13.0	SHOWN SCHMILI P-ON 150 Xi	ONTE 78.5.29 PAGE 1 NG W.LENGTH WEIGHT C.DE GINOTES 3.0 2314 35.0 1157
18.5 Killing	\$130		9130 (OT)
P 1 2 C 3	3 4 5 6 7	8 9 10 11 12 13 L3A	13F 1010L
NOPHRT DS-12		SHUHN SCANTLI P-DN 150 X13	NG H.LENGTH RELGHT C. OF C NOTES 3.0 2315   35,0   1158
EB 2 13.0	S130 ====		5130 (01)
1 2 P C S	3 4 5 6 7	8 9 10 11 12 13 13A	13F torm.
TOTAL HEIGHT	U.U	FOTAL LENGTH 0.0	PANEL DS

3H1P NO 2715	BLOCK · DS6	SCANTLIN 150 ×13.0			PLATE	PACE	
T CODE	-2-11		P C S SH	DN 150 X13.	133.3	11157	HOTES
NOPAHT 2 CODE		\$130	151818181		2315	(01)	
Z CODE	-3-11	3130 ====				I C. OF 6	NOTES
MUIDORY I		3130	,		5130	(01)	
NOPORT 3 CODE	•	·	P C S SHO	WN SCANTLING	H.LENGTINHETGH	C.OF G	NOTES
UDPART UCODE	•		PCSSII	ON SCOUTLING	H-TENGIN HEIGH	COF G	NOTES

SHIP NO PANEL SCAN	ITLING FAB	FLAT B	AR PRGE 1 DATE 78. 5.23 NOTES
NO PART CODE	SHOWN		Unic 70. 3 .E3 AUTES
1 DS-F14	P-UP	S	S (FH)
1 2 3 4 5		9   10   11   12   13   13A   13F	TOTAL
P	39		
	39		
NO PART CODE	SHOUN	<del></del>	<u> </u>
2 DS-F14AxS	P-UP	3	S (FH)
1 2 3 4 5		9 10 11 12 13 138 138	FOTAL
	-		
	<del>                                     </del>		
NOT PART CODE	SHOHN		
3 DS-F148×S	P-UP.	S	S (FH)
1 2 3 4 5	6 7 8	9 10 11 12 13 13A13F	TOTAL
P			
C	<del>                                     </del>		
TOTAL HEIGHT 0.0		LENGTH Q.Q	PANEL DS

SHIP NO. BLOCK SCA	ITLING FAB. SHOP FLAT	r BAR 🔲	
<u>2715 056 150</u>	X13.0	PAG DATE 78.5.23	E J NOT
NO PART CODE	P C S I I SHOWN S-01	45×35R ~ (UP)	
NO PART CUDE	P C S	A STATE OF THE STA	
2	SHOWN		
NO PART CODE	PCS		
3	SHOWN		
NO PART CODE	PCS	•	
lu	SHOWN		
NO PART CODE	PCS		
5	SHOWN		
NO PART CODE	PCS		
6	SHOWN		
TOTAL HEIGHT 32.0	TOTAL LENGTH 2424.0	BLOCK DS	56

	IIP 1	40 	PAI DS	NEL			TĻ]N <u>X 13.</u>		F(	1B.	SHO	P =	F	Li	A 7	<del>-</del>	В	A		= *F •	•	PAG	
NO		PAR	Ţ	COD	E		SHO	ни	<u> </u>										<u>h</u> n	15_	<u> </u>	5 . 23	NOTES
1	DS-	FI					5-0	r				<b>6</b> 5×35	n							135 <b>A</b>		(UP)	
	PC	1	2	3	Ц	5	6 16	7	8	9	10	11	12	13	130	13F						TOTAL	
1	S						16							-	1	<u> </u>		1-	1 -	1	╁	1	
140		PAR	T	COD	E		SHO	111							*		<u> </u>		<u> </u>		<u></u>		
2	05-	F2		•			S-0	т		•	<	3							_65 <u>+</u>		5 H	(UP)	
Ì		1	5	3	4	5	6	7	8	9	10	11	12	13	130	13F		Т	1	T	1	TOTAL	
ł	P		<u> </u>	L			16																1 1
	C S	ļ	- <del></del>			ļ	-				ļ	<b> </b>	<u> </u>					<b> </b>	<u> </u>				
NO	*	PANT	<u> </u>	CODE	<u> </u>	-	116 5HO	<u> </u>	<u></u>		<u> </u>	<u> </u>	<u> </u>	<u> </u>	<u> </u>		<u></u>	<u></u>	<u> </u>	<u> </u>	<u></u>	<u></u>	
3	DS-1				-		S-0					ISH3SI	1						#5H _160		1	(UP)	
		1	5	3	4	<u>5</u>	6	7	8	9	10	11	12	13	13A	13F						TOTAL	
	F C						16												<u> </u>	_			
	S						16											<u> </u>	<u> </u>		<u> </u>		
r	<u> </u>	WE	GHT	0.0	<u></u>		1 10 1		TOTA		ENG	IH C	3.0		لبيبا			<u> </u>	<u> </u>	<u></u>	PAN	EL DS	

	SH [ F		PANEL		ANTLI		FA	в. 5	HOP	•	FI	LF	TF	•	В	AR				
_	10				X 13					===						0	ATE	70. 5	PAG 5.23	E 1 Notes
1		<u> </u>	r coi	UE.	SHO S-				<u>_</u>	5+35N					•		5435R 608		(UP)	
		P C S	2 3	ц	5 6 5 5	7	8	9	10	11		13	130	1 3F					FOTAL	
<u>N</u>		PANT 9-F6	·	E	SHO	NN			1			,							<u> </u>	
		1 P C S	2 3	il	5 6 4	7	8	9	10	11	12	13	เรก	13F				-N	TOTAL	
3 		PART 5-F6A	Caa	E	3-C			গ্ৰ	38	<u>.</u> J							₩\! 5×35/1 /	5 	(UP)	
	9	5	2 3		5 6 1 1	7	8					13	3A	13F					TOTAL	
	1011	HE MEI	GHT O.	U			τστη	L LE	NGT	H 0	.0					-		PANE	L OS	<del></del>

SHIP NO	J BLOCK	* FAB. SI	10P					, <del></del>	
2715	056		· · · · · ·	ANGLE					
NOPART	<u> </u>	<del></del>	P C S SHOWN	SCANTILING   1	IOLE   GRADE   S	DATE 78. PRM WEIGHT C	5.23 PAGE	1	
1 CODE	-NL1	······································		250 X 90 X 10 / 15 P:	35 AM	360.0	<u>FH N+3</u>		
			• ,					<u>"</u>	
		uk	FR 136 F	7N138 FR141	FAIUU		FN147	H	(FW)
	NSL				0	O NSU	NSU		(FW)
		150 800	<b>ा ३०० २५०० २</b> ३०	<u> </u>	200 <u>  7709  </u> 8	1600 <u>  10000  1</u>	510 <b>d  1</b> 050d	15403	
NO PART 2 CODE	-BL2		P C S SHOWN 1 1 1 S-IN 2			PAM NEIGHT C	OF G		
-1-00-1			111 111 2-111 12	250 X 90 X 10 / 15 P3	35 AM	226.0			
		<del></del>							j
1		FR136	FRI	20					i
	<del>- "</del>	1	i	38		141	FALUU	. 🔄	(FW)
	L_ NOL	<u> </u>	ol	<del></del>	0	L	0		l
	159	100 <u>  1300                                  </u>	Trod saud		<u> 4809  5300</u>		7200  7700	7800	
NO PRAT 3 CODE	-AL3		P C S SHOWN			PRM WEIGHT C.	OF G		
		······				1177 0 1			
li .				<u> </u>	SU AM	133.0			
İ	Г—н	·			SU JAM [	[133.0 ]		и 1	
,		·				[133.0]			
,	<u></u> _	·		FAJ47	. <u>  AM                                       </u>	[133.0]		<u> </u>	(FH)
			P3:	FA147	SU	[133.0 ]		<u>#</u>	(FH)
NOIPARTI			P3 O 1804	FA147 5   2204   2404			OF CI	₩ <u></u> ₩ <u></u> 	(ғы)
NO PART 4 CODE			P3 O 1804   P   C   S   SHØNN	FA147 5   2204   2404	OLE  GRADE S.	PRM   HE [ GHT   C.   293.0	OF G	₩ <u></u>	(FH)
NO PART 4 CODE			P3 O 1804   P   C   S   SHØNN	FR147 5   2204   2404   3404	OLE  GRADE S.	PRM   HE I GHT   C.	OF G	# <u></u>	(FH)
NO PART 4 CODE			P3 O 1804   P   C   S   SHØNN	FR147 5   2204   2404   3404	OLE  GRADE S.	PRM   HE I GHT   C.	OF G		(FH)
NO PART 4 CODE		Fn) 36	P3 O 1804   P   C   S   SHØNN	FR147 5   2204   2404   3404	OLE  GRADE S.	PRM   HE I GHT   C.	OF 6	4606 S	
NO PART 4 CODE	-ALU	FRI 36	P3: O 1804 P C S SHONN 1 1 S-IN 2	FR147 5   2204  2304  2404  SCANTLINC   H 50 X 90 ,X10 / 15   P3	OLE  GRADE S.	PRM   NE IGHT   C.	OF G	S	(FH)
NO PART 4 CODE	HST HK  -UTA  -UTA	0	P3: O 1804 P C S SHONN 1 1 S-IN 2	FR147 5 2204 2304 2404 H SCANTLINC H 50 X 90 X 10 / 15 P3	OLE   GRADE   S. 5 AM	PRM   HE [ GHT   C.	0		

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SHIP NO		BLOCK	FAB.	SHOP			•		•										
2715	!	056						AN	GLE	<del>-</del>									
MAIGABLE					<u> </u>	01000	·						D	ATE 78	. 5 .23 C.OF G	PAGE	2		
NO PART 1 CODE	-AL5	<del></del>	<del></del>		P C 1	S S110 1 S-1		SCANT X 90	L1NG X10 / 1	HOL 5 P35	E GAA	DE IS.	HIMAY.	EIGHI 86.0	C.OF G				
	<del></del>																		
1																		S	
		нк		FR136	F	A130	FA	41	FRIUU		FBJ4	17		1	FAJU9				
	- NSL		0		0	1	2	0		O N	เบ	изи	0	0		0	:	35 A C	(FH)
		150	. 80g <b>  1</b> 3	विदी इस	0 <u>05.   po</u>	जी तबठरी	5300	720q	2700  96	वर्गे ग्वयवर्गे	10103 1	០ឧ០ឦ 1	050g 11	300 117	0011100	13200	13300		
NO PART	-AL7		,		PC	SSHOI	IN .	SCANT	LING	HOL	E GRA		PRM H	EIGHT	C.OF G		·		
z [CODE]			<u>t</u>	l	3	3 5-1	N 1250	X 90 X	K10 / 1	5 P35	IAM	l_	2;	26.0				·	
	[ <u></u>	<del></del>																	
															<del></del>				
	<b> </b>		FR136	)			FR138					FR	41			FALL	d F	<u>-</u>	(FH)
	<u>⊢ NSL</u>	0	<u>_</u>			0	l		<del></del>		0				0		·		,,,,,,
10000000						••	<b></b>		······································		<b>r</b> abd	_5300			7200	_720g _	7800		
NO PAAT				}	PC	SSHOL	IN	SCANT	LING	HOL	GRAI	DE S.		ELGHT	C.OF G				
		<del></del>	<del></del>	l.			L	···	•	<u></u>	l			1	i		<del></del>		
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											•		•	•					
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						•													
VO PART					PICIS	SISHON		COOUT	7110	1 1144 6	10000								
CODE						) Sunu	<del>4</del>	SCANTI	בומט	HOLE	: ICHHI	<u>15   3 •</u>	PRM INE	IGHT	C.OF G				
	•							•		<del></del>			L.		<u>-</u>				
		:					•												
• •																			]
							-			-			,						
TOTAL WE	IGHT	2128.0	<del></del>		Ti	OTAL I	ENGTH	79400	1.1	<del></del>	·					DI.	ack ne	20	
TOTAL HE	IGHT	2128.0	<del></del>		T	OTAL L	ENGTH	73400	. 1	<del></del>	·					AL	OCK D	88	I

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船 殼 HULL PARTICLIST 部 品 表

for Assembly

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# # 1978 D5 23

ASS UST -D1-

プロック名 DS5 委 船 2715

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船 殼 注1] 1サブ内弧数 Δ 上段ワP玄 £ ٨ HULL PARTS LIST 部品景 下段ワS玄 Δ 往2) 製 造 数 忢 for Assembly 4444 上段7所異玄 Δ ٨ 下段ワ存在玄 反远远 标 (ke) 197 図画页 データ 名 改 早品重量 6 PC (MM) 物質 (14) (MM) S 正 22 叮 ĸ 01 1 02 03 200 - 12. 18. 1 7 1 1 Ç 04 05 06 07 08 09 10, B1 10 11 12 31993 \*\* \*\* SUB ÇA 1 1 168. 1 92. 3 1 1 1266. 31 528. 1 32 527. 1 33 192. 1 1 34 1 3613. 41 1 3449. 42 3401. 43 1 37P2 44 2 2 501 DS-1 1 502 DS-1A 1 502 DS-1B 3 3 503. D5-2 504 1 D'5-2A 4 503. DS-3 (4730) 1978 55 23 ≨ 総 2715 プロック名

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中栖

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**DS6** 

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船 殼 HULL PARTS LIST 部 品 表 tr Sub assembly

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 注2) 製造数
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## APPENDIX C SUMMARY OF IHI SHELL

# **SHELL**



## lshikawajima-Karima

Heavy Industries Co., Ltd.

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1. Ge	eneral aspect of SHPLL system	. 3
1-1	Purpose of the development of SHELL system	3
1-2	Cherecteristics of SHELL system	4
1-3	Scope of application of SHELL system	12
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2. Sy	stem composition and outline of input/output	
of	the system	. 18
2-1	Main computer and its terminal equipments	18
2-2	System composition	
2-3	Input/output system	25
•		•

#### 1-1 Purpose of the development of SHELL system

This system constitutes an integrated, computerised data processing system that provides various highly accurated and utilizable informations pertaining to all the process in production of curved shell blocks from the shell expansion in design section or mould loft to the block assembly in shop.

In the conventional system, shell expansion, jig calculation of the curved shell blocks, etc., are also carried out by numerical calculation by the medium of an offsets data bank (as original data file) and they were improving and brushing up passing through a lot of trials and errors based on the fed back data from the production field. However, the system was merely constituted single-purpose programs being concerned to the jobs in mould loft.

Purpose of the new development of SHELL system is to obtain the most optimum informations to meet the needs in the respective process in production. For the purpose of it, a certain number of standardized production technologies and application know-hows are stored in the system and the planners are able to put their option into the system by means of inputs according to the situation of the shop. The standardization of production technologies and analysis of application know-hows in SHELL system have been established by mobilizing of the engineering power in IHI's five shipyards.

significant fruit of the new development of SHELL system.

(化松衣形式)

### 1-2 Charactaristics of SHELL system

## 1-2-1 Designing policy of SHELL system

The following considerations were put into

- l)Lines on the optional cut plane at designer's
  direction to be used for the calculations, in
   the system.
  - 2) Lines data to be-stored in the bank by a certain concurrence of points on the respective line which is approximated by straight lines.
  - 3) A common offsets data bank to be installed in the system, to which required data to be shifted from the respective data base in each subsystem passing through a certain conversion program.
  - 4) The templetes to check the curveture of shell plates are to be designed standing at right angle against the mean level of the curved plate when it is on the bending slab.
  - 5) Figure of a expanded shell plate to be calculated as a part of the ship's surface including surounded
  - 6) Geodestic line method and rolling method to de adopted in the system as the developing logics for shell expansion.
  - 7) A corrective routine to be installed in the system to modify the shape of expanded shell plate after running

## 1-2-2 Charactaristics of SHELL system

1) SEELL system is a composite system for the geometri calculation and data processing system *relevant* to the production of curved shell blocks in ship

#### follows:

- 2) The calculating logics are of simple and higher level of accuracies are uniformally displayed in the outputs by the system, since the lines in the offsets bank are dram by the concurrences points approximated by a certain supplementary straight lines.
- b) The higher accurated shell expansions are achieved easily by this system, since the optional cut plane method is fully adopted for the expansions.

( Refer to Fig.1 )

- co Unified accuracy on the peripheral parts of expanded shell plate is ensured, since the desired plate is cut out of a larger expanded plan including the surounded area of the plate.
- formal **check** of the curved plate d) The geometrical by the templetes becomes easier to ensure the accuracy, since the templetes are set up at the right angle against the mean level of the curved plate. ( Refer to Fig. 2 & 3 )
- 4) A remarkable improvement on the workability and accuracy in the assembly stage can be expected, since the various working practices in the Shop are taking into considered from the first step of the system running, as follows:
- 2) SINGLE PANEL ASSMBLY system is available.

Refer to Fig. 4 & 5 )

b) Since the intersections of datum planes in the supporting jig lines and shell plates are marked on each shell plate and, in addition, the datum planes are orthogonal to the platform surface, the accuracy of angles between contiguous seam and butt can be maintained easily.

( Refer Fig. 6 & 7 )

- c) **Instructions can** be given in connection with the *position* for plate setting, position of stopper and the height of additional supporting jigs, if required.
- d) Data for accuracy control relating to- the dimension of black, diagonal dimension, date of curveture on seam and butt, etc., are output by the system.
- e) The availability of automatic welding on the block can be checked by the system, if required.
- 5) Easy to maintenance of the system.

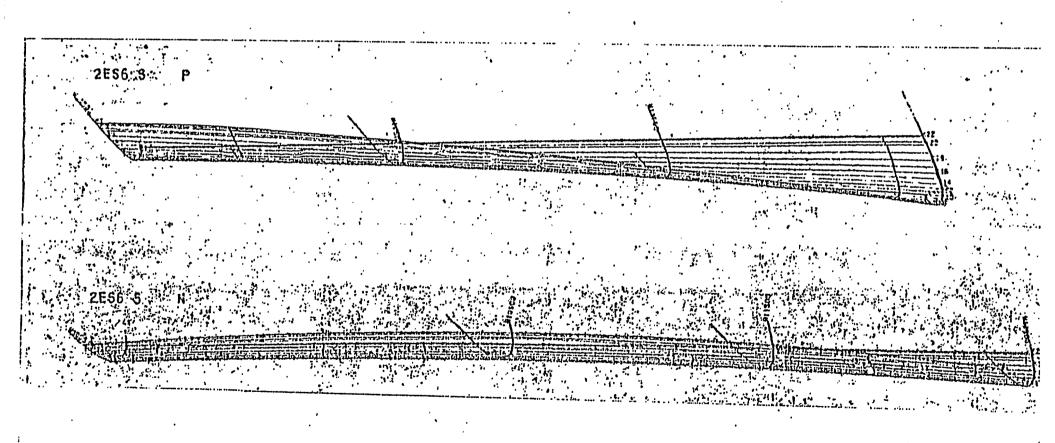
  The partial revision of the system, rather easy,
  since the system of workinig with modules is adopted
  and the logical constitution of the system is simpler.
- 6) Easy to replacement of data in the data files.'

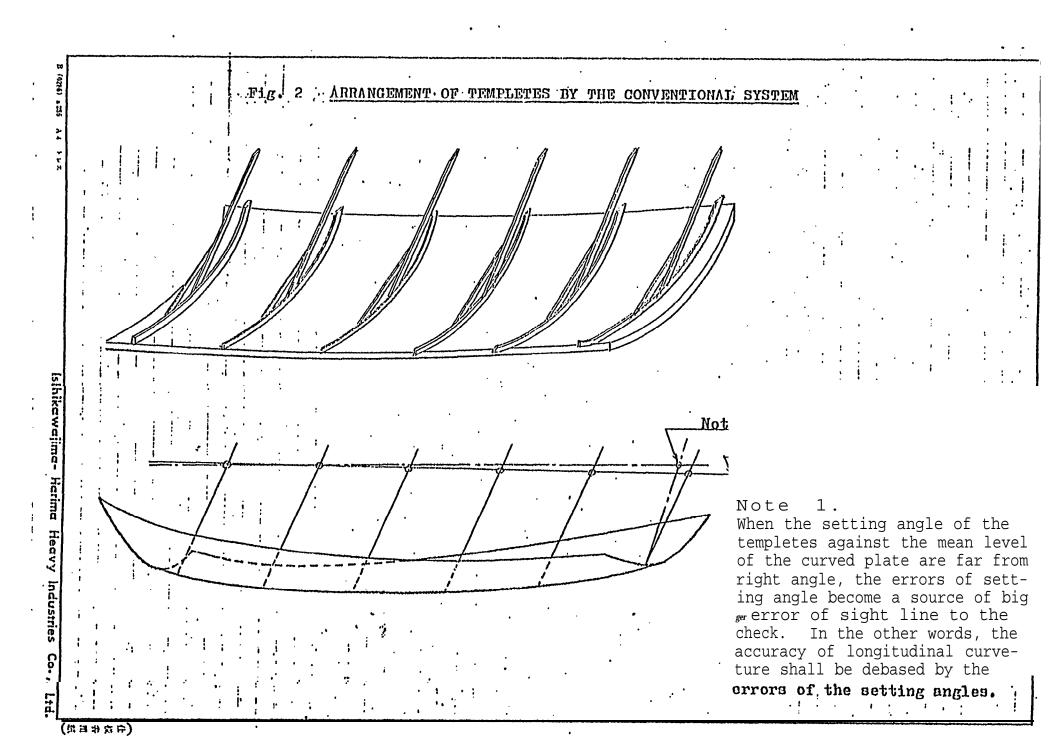
  Since the system has 2 main offsets data bank and peripheral data files separatory, replacement of data can be made indipendently when it becomes necessary.-
- Partial manual correction of the outputs.

  Partial manual c orrection for the output of shell expansions available in accordance with the changement of working process and/or expansion method.
- 8) Easy recording of fed back data.

  Quantative data such as **distortion caused** by lockedin stress due to press or heating, in particular,
  can be fed into the system from-time to time and
  recorded them-as reference data for system.
  improvement in-future.

Fig. 1 ONE EXAMPLE OF OPTIONAL CUT PLANE PROCESS





(年校矽田県)

- 1. The datum planes of the transverse and longitudinal supporting jig lines are mutually orthogonal and also are orthogonal to the platform plane.
- 2. The inter section of the datum planes and shell plates are the datum lines for the block marking.
- 3. Those datum lines of block marking are marked on each shell plate in the fabrication shop.
- 4. Since both datum planes are crthogonal to the platform plane, the check of datum lines will be easy. In addition, these planes become the vital medium to maintain the accuracy of block marking. Accordingly, the accuracy as well as the workability are improved in the field.

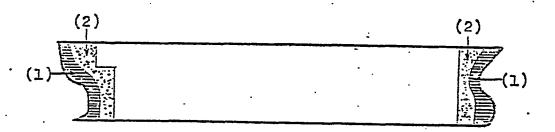
(世经会所或)

## 1-3 Scope of application of SHELL system

According to the current version, SHELL system is applicable to all shell plates other than stern plates and stem plates. Preparation are being advanced, however, to include even stern and stem plates in the scope of application of SHELL system in the near future.

#### 1) Functional limitations

- a) SHELL system can not be applied to the processing of stern and stem plates. ———— (1)
- b) As for the plates adjacent to stern and stem plates, it is required to supplement the offsets data to process. ( PAN valet system ) ———— (2)



#### 2) Operational limitations

- a) When the offsets data bank is not available to the processing, a cortain supplemental offsets data will be required by the manual input.
- b) When the system is applied for a part of the ships, such as repair ships, etc., PAN valet system will be available to supplement the incomplete offsets data.
- c) The operational limitations are also depending on the function of the gas cutting machines to be used for.

(仕任会用紙

#### 1-4 Operation of SHELL system

#### 1-4-1 Establishment of the operation system

SHELL system incorporates a wide scope of the application know-how in production and is designed for working with all kinds of related facility and production methods. For this very reason, its most optimum operation system be displayed only by the establishment of a suitable system for inducing it into the ship yards taking into consideration of computor or numerical control system and also working process in production.

Namely, sufficient preliminary deliberation will be necessary to coordinate functional aptitude with the fabrication and the assembly stage, as follows:

- \* Preparation of base data ----- Functional design
- \* Formal geometrial development ----- Applied design
- \* Post processor --Outpost of production data To elaborate, 2 slient characteristics of SHELL system lies in the fact that system demands preliminary instituition of a comprehensive production process engineering prior to its application.

#### 1-4-2 Type of the application system

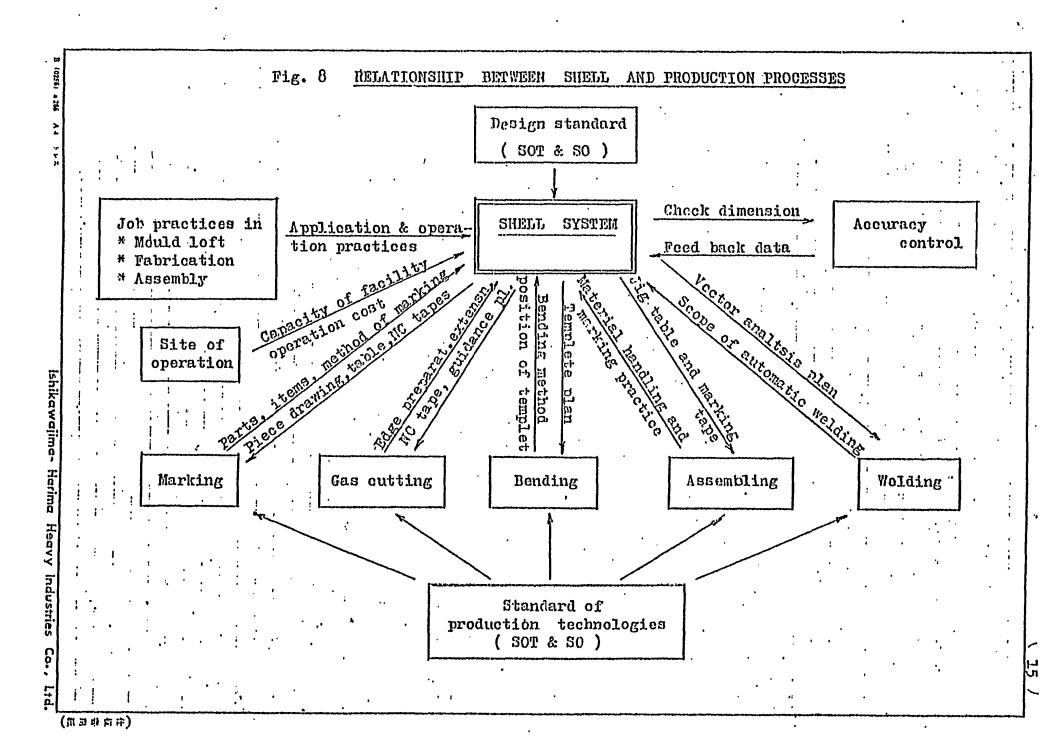
The the of the application systems may roughly classified into the followings:

- a) Hand marking oriented type
- b) Electro-photo marking oriented type
- c) NC cutting oriented type
- 2) Hybrid electro-photo marking and NC cutting oriented
- e) SINGLE PANEL ASSEMBLY system-oriented type
  SHELL system provides highly accurated output data
  coducing to excellent workmanship for the all type

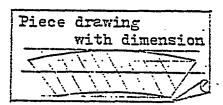
of application systemslisted in the above listed.

1-4-3 Standardization of production technologies and ifs application know-hows

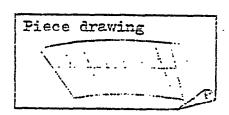
SHELL system is an advanced and fruitful system constructed on the basis of the systematic standardized production technologies and higher leveled application know-hows in the each stages of production, such as cutting, bending, assembling, welding, etc., showing in the following Figs..



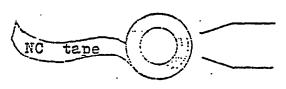
A. Marking and gas cutting in fabrication stage.



Manual marking



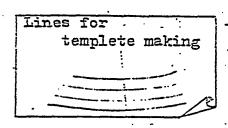
Electro-photo marking



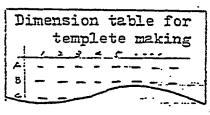
NC marking

NC gas cutting

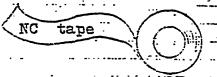
B. Plale bending



Wooden templete



Universal templete



Supporting jigs in SINGLE PANEL assembly

C. Supporting jigs in assembly stage. ( Dimension plan - Jig heights ) Fixed position jigs Seam line jigs Frame line jigs SINGLE PANEL ASS. Sclid jigs

- 2 System composition and Out line input/output of the system
- 2-1. Main computer and its terminal ecuipment
- 1) Main computer
  - a) IBM s/370 135/158 are available.
  - b) vs 1, 2 to be used for the CS.
  - c) FORTRAN-G is available for the programming.
  - a) 3330-1 (N=1) to be used for the disk handling.
- 2) Terminal equipments

The terminal equipments for the outputs primarily consists of the drafting devices, for which the users are free to select the desired terminal equipments, which may consist of any of the followings:

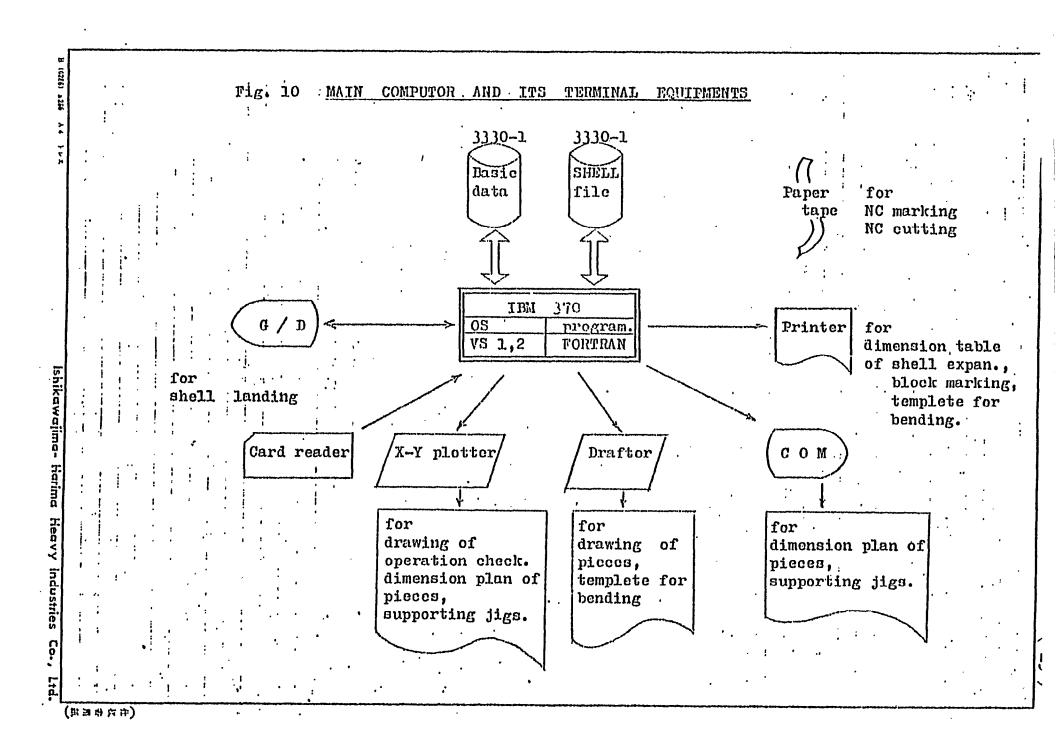
- a) Graphic displayAvailable for shell landing.
- b) X-Y plotter

Available for the piece drawing, dimension table in piece drawing, dimension table of the supporting jigs in, assembly stage, check drawing, etc..

- c) Drafting machine

  Available for drawings, to which higher accuracy of
  lines will be required such as a certain kinds of
  piece drawing, lines for templete making, etc.
- d) COM:

Available for the dimension table of pieces, supporting jigs, etc...



## 2-2 System Composition

#### 2 - 2 - 1 Data base

The most effective application of SHELL system in the processes of ship building field will primarily depend on how well the original offsets data file of the system can be prepared in the shortest time possible.—Shell landing to generate the offsets data should be advanced by the computor processing wherever possible, \*&me the data base of the system requires the most accurated offsets data as an original data. However, when the computer processing shell landing is unavailable for the preparation of the data base, SHELL system so designed as to permit the system operation even through manual preparation of the offsets data.

# 1) Kinds of preparation methods

#### a) FAIRLAND

FAIRLAND system (fairing and landing system) is available for the preparation of the offsets data.

# b) DACSSI/ S

The base data can be provided by the output from the offsets data file of DACSSI/S in which the offsets data are generated by means of conversational inputs with graphic display and/or input cards.

c) Card inputs from offsets table

The base data are provided directly by input cards in this system when it is required to process for the repaire ships and so on.

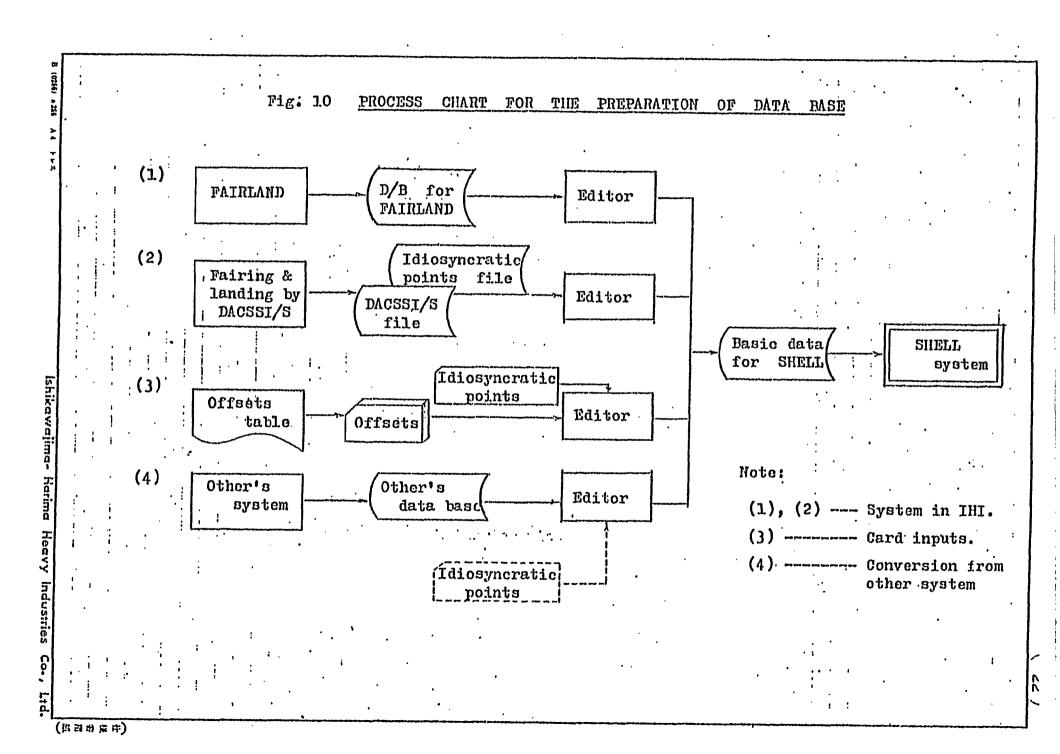
- d) Conversion from the other systems
  The base data are provided by the outputs of other systems through 2 certain conversion program.
- 2) Editting of base data

a) Editting method.

Editting on the sequential file in 80 columns card

c) Data format

X---, Y---, Z---, pen, element, structural points,
nature of points,



NC data for production

B (0226) .			Table 1. STEPS	involved in s	HELL SYSTEM	, <u>                                    </u>	1
216 A 4	Step	Base data	Optional cut plane	Expansion	Calculations of Ass. data	Shape formation	Post processor
	Input	*Station offset *Landing data *Idiosyncratic points data *Fabrication data (bevel, extension, etc)	*Block name *Option for cut plane	*Method of shell expansion	*Type of suppor- ting jigs	*Correction of fabrication data	*Piece name
	Data process	landing Storing of	*Preparation of concurrence of points *Proparation of data file on the cut plane	*Calculation of shell expansion *Calculation of lines for bend. templetes	*Height of supp- orting jigs *Dimension plan for block mark. *Vector analysis plan	shell expansion *Cut out of shell plates	nog post processor
kawajima- I	e		*Baso data file	*Data file on the cut plane	*Data file on the cut plane	*Shell expansion data file *Fabrication data file	*Nesting data file
larima Keav	Fil.		points file	*Shell expansion file *Bending templete file		*Nesting data #: file	
ry Industries Co., Ltd.		Offsets table	Rough block arrangement Body plan by optional cut plane		*Dimension plan of support. jig *Block mark.plan *Sectional plan of support. jig *Assembly plan	*Dimension plan of piece *Lines for temp-	*NC tape

# 2-3 Input/Output system

## 2-3-1 Input data

The input data system in SHEEL is so designed as to conserve the labour by avoiding deplication of input data. Principal items of inputs are 2s follows:

- 1) Name of block and structure
- 2) Instructive data relating to the fabrication practics.
- 3) Instructive data relating to the assembly practice.

# 2-3-2 Output data ( Other than NC tape for marking & cutting

- 1) Check drawing of the operation
- a) check drawing fo the data base (fig.11A)
- b) Check drawing of body plan the optional 'cut plane (Fig. 1 l B )
- c) Sectional plan of supporting jigs. (fig. 12 )

  The check drawings in the above are using the check of the operation conditions of the are in option controled by the inputs,
  - 2) Output of shell expansion
  - a) Dimention plan of shell plate

    Shell plate marking can be made directly by this output without any full scale marking tape the the manual marking. (Fig. 13)
  - b) Expanded plan of shell plate negatives for the electrophoto marking are output by the system by means of drafting machine as the expanded plan of shell plate, (Fig. 14)

Note: Outputs of the dimension table for the preparation of full scale marking tapes are also provided by this system, if required.

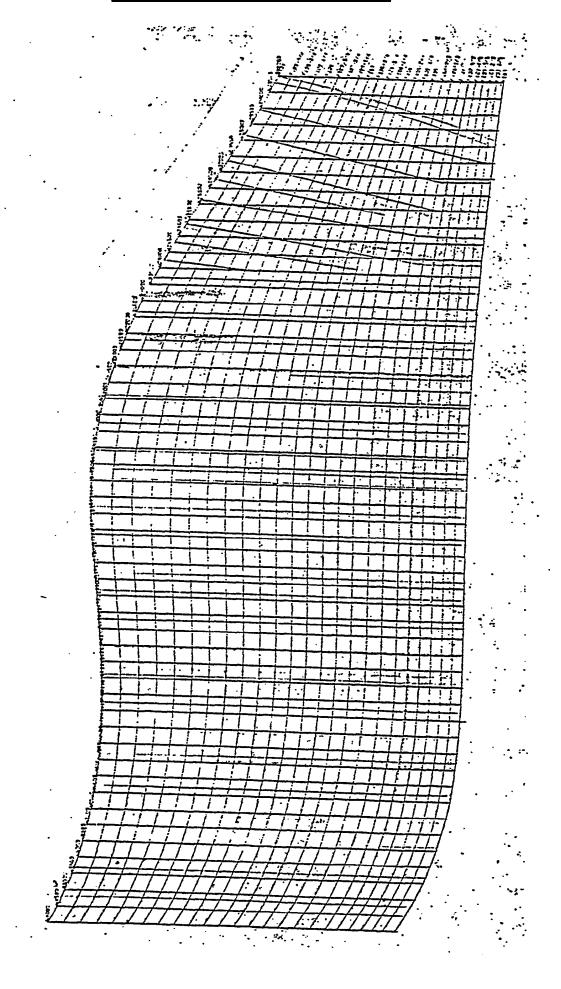


Fig. 11 B CHECK DRAWING OF OPTIONAL CUT PLANE PROCESS

in attitution abious

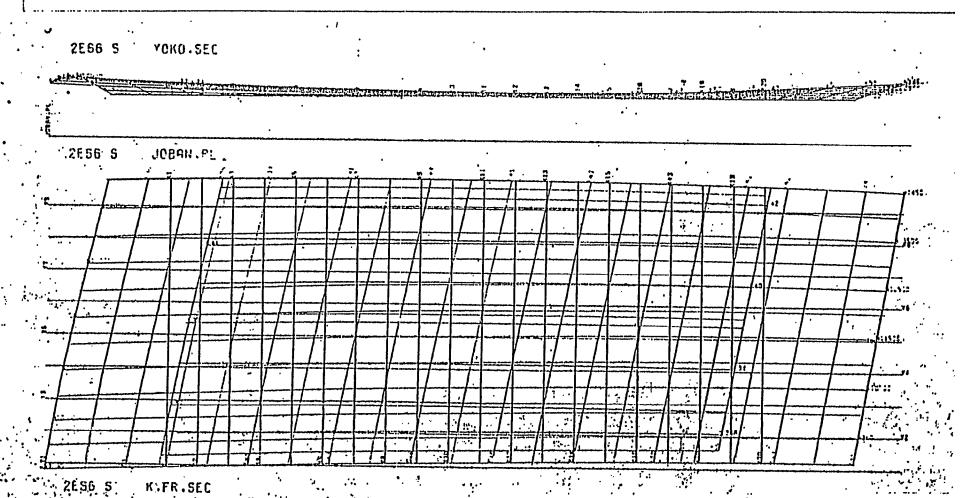


Fig. 12 SECTIONL PLAN OF SUPPORTING JIG LINE

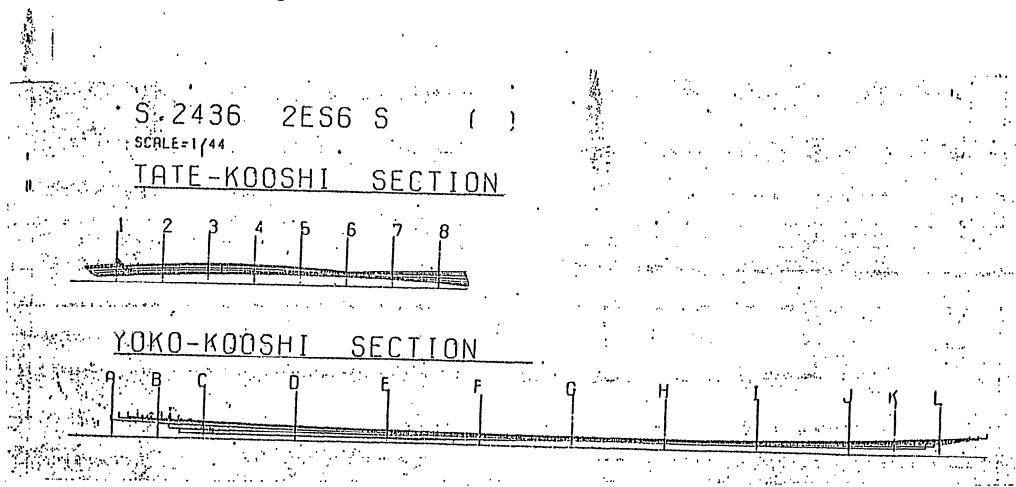


Fig. 13 DIMENSION PLAN OF SHELL PLATE

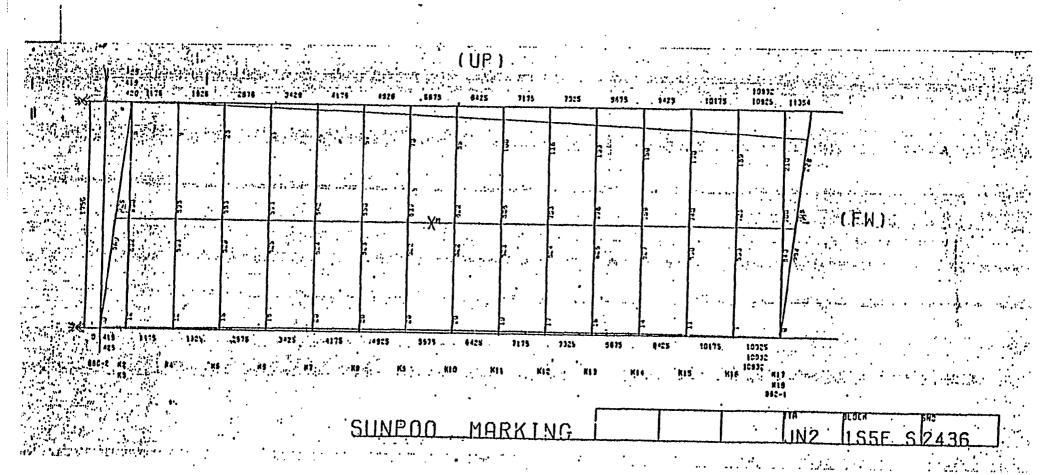


Fig. 14 EXPANDED PLAN OF SHELL PLATE

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- 3) Output of templete for bending
  - a) Lines for templetes marking

    The most optimum position determined by a given logic as well as the shape (lines) of templetes at 2 designated position are prepared by means of lines drawn by draftor. (Fig.15)
  - b) Dimension table for universal templete When a certain type of universal templete are available in the shop, suitable dimension table can be provided by the system to set the figure of the templete. ( Table 2 )
- 4) output data for supporting jigs in assembly

  The output mode of Sheel system is diversified into

  various modes owing to the inclusion of data relat

  ing to the type of supporting jigs as well as the

  data relating to the methods of block marking.

  The users can therefore, 'select any desirable

  output met to the working practices to be done.
  - a) Dimension plan. for block supporting jigs.
    - \* For fixed position jigs \* (fig. 16)
    - \* For seam line jigs (fig. 17)
    - \* For frame line jigs (Fig.18)
    - \* For SINGLE system jigs (fig.19)
    - \* For solid tape jigs

While the five kinds of dimension plans, listed above, are available, they can be used also in combination, if necessary.

b) Sectional plan of jig lines (Fig.20)

Sectional plan of jig lines can be drafted t 0

indicate the curvetures at the transverse or longitudinal section of jig lines including the ends height of the block at the cross points of jig line; and seam or butt line of the block, if required.

Fig. 15 Lines for bending templete making

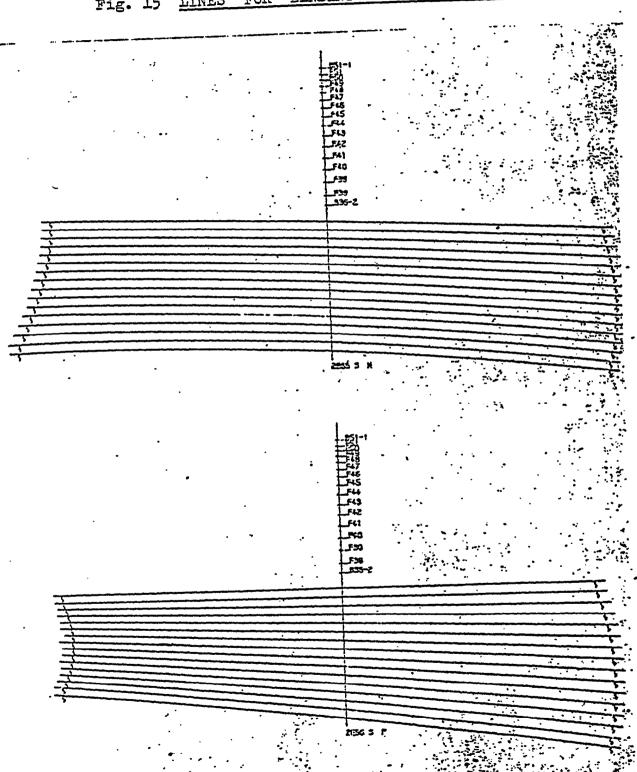


Table 2. DIMENSION TABALE FOR UNIVERSAL TEMPLETE

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# Fig. 16 DIMENSION PLAN FOR FIXED POSITION JIGS

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Fig. 17 LINE JIGS PLAN FOR SEAM DIMENSION

Š - 2436 scale=1/50 **256** S

SEAM-JIGU PLAN

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H=277 1=3471 105 C-721		¥=378	₩., 1160 ,	)=? §9 }==	#=184. %=3513	H=162 Y=3522 0=0	H=149 H=3525	#=145 #=3525	7=1814 7=1814	U2166 V23600 D21	##1781#/ 1/190 Part Se	H=193 Y=3479	•
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Fig. 18 DIMENSION PLAN FOR FRAME LINE JIGS

5.2436 2ES6 S SCALE=1/64

Fig. 19 DIMENSION PLAN FOR SINGLE PANEL SYSTEM

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Fig. 20 SECTIONAL PLAN OF JIG LINES

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The dimension table for the block marking is printed out by the system and full scale marking tape can be made from the output in mould loft. In the dimension table, length and with of the block are shown on the frame lines either of ordinally and optional cut plane.

Block marking is made using-the transverse and long-itudinal datum lines on the block. Since those datum lines are marked on the each shell plate of the block in beforehand, easier working procedures will be expected to maintain the accuracy, so far as the previous stages are keeping their accuracy in the torelance.

- 6) Check data for the automatic welding

  The maximum slope of the seam lines against ground
  level are output in the vectoral indication to check
  the availability of automatic welding on the block.
- 7) Accuracy control data
  - a) Geometrical check of block

    Dimension plan is provided for the accuracy control of geometrical form of the block.
  - b) Positionning of materials

    Data relating to the size of shell plate and location of the positionning jig for the shell plate are indicated in the dimension table of supporting jigs. These data will be available to secure the right position of Shell plate on the platform and it ensure the geometrical form of block.
  - c) Datum planes The datum planes for the block marking are orthonal ly mutually and also to the platform plane. It means that working condition is better for the ensuring of block accuracy.

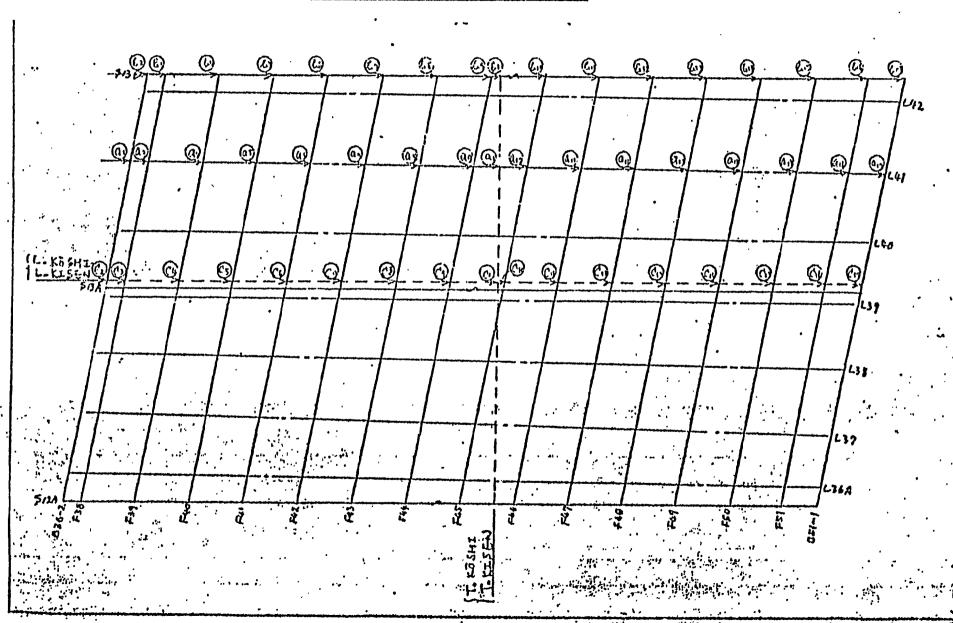
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Table 3 <u>DIMENSION TABLE FOR BLOCK MARKING</u>

( Ordinary frame )

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Ļ	56	1	7621	7621	398	5 6 2	1541	2517	3493	4466	5438	<b>ሉ</b> ፋታባ	• 7377	9244	anna	10273
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l.	364	1	7621	7621	67	251	1230	2208	3164	4159	5132	6104	7914	9043	4017	9978
S	150	. 1	<b>@7621</b>	<b>@</b> 7621	<b>(</b> )1329	<b>@1490</b>	<b>(</b> ;) 2459	@3426	£ 4390	<b>6</b> 95354	(j) 57 15	<b>@7775</b>	(J) 9224	(An. 1)	(h) () ( 65	(1) 10 cs
5	134	11	7621	7621	637	801	1,779	2755	3731	4764	5675,	4444	7611	9577	9541	1:0503
S	124	1	7621	7621	. 0	164-	114.	2122	, <u>50</u> 68	407.5	50-0	6053	4795	7044	4244	9904
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Fig. 21 EXPLANETORY PLAN OF Table 3.



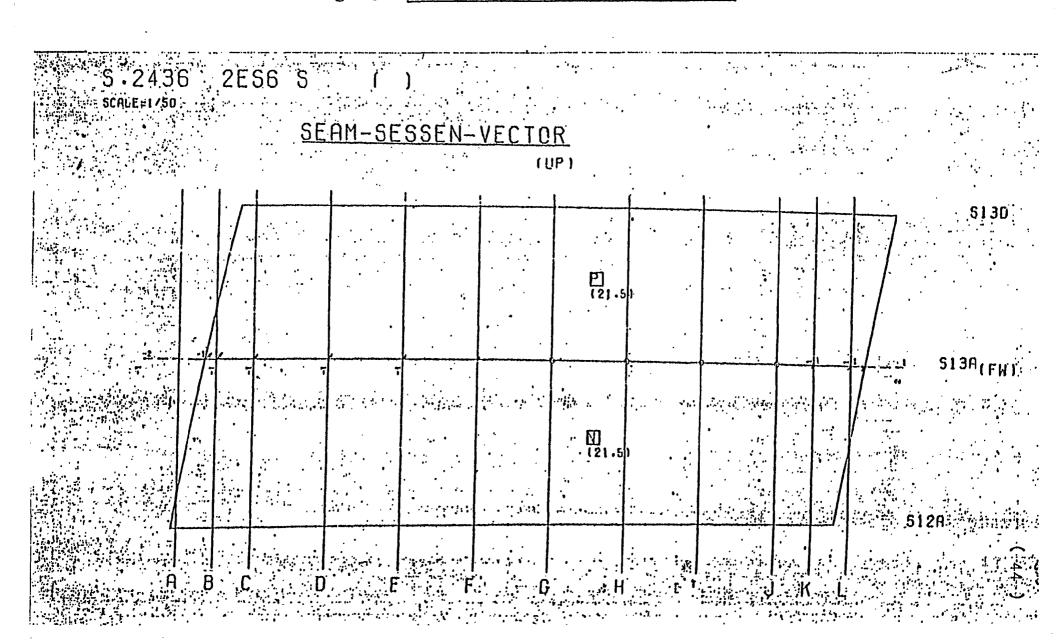
DIMENSION TABLE FOR BLOCK MARKING Table 4

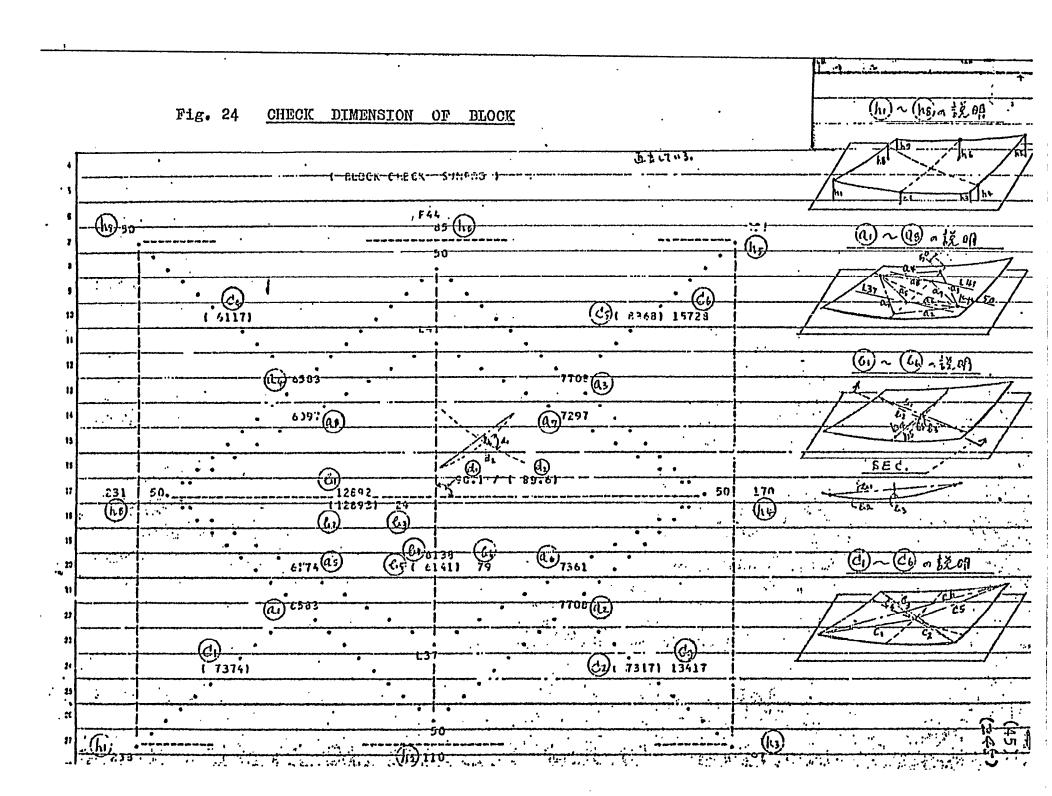
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Fig. 22 EXPLANETORY PLAN OF Table 4.

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Fig. 23 CHECK DATA FOR AUTOMATIC WELDING





#### APPENDIX D

LODACS - SHIP FRAME DATA PROCESSING SYSTEM

# Development of LODACS, a Ship Frame Data Processing System

By: Sumio Kohtake and Hidehiko Matsubara Ishikawajima-Harima Heavry Industries Co., Ltd.

Almost all of the previous ship framing computer programs which have been developed and utilized individually at each IHI shipyard not only contained several weak points such as troublesome operation, difficult expression of output showing graphical images by a line printer, and insufficient functions for data processing, but also could not keep pace with the recent progress in the Numerical Control Machines such as steel marking, flame cutting and hydrauric bending equipment. To cope with these circumstances, by standardizing the design and the fabrication procedures and rationalizing graphical output by the dot-printer, X-Y plotter and COM, we have developed a newly integrated system called LODACS (LOngitudinal frame Developing And Conducting System), a system applicable to all shipbuilding facilities and equipment of our shipyards. LODACS, a ship frame data processing system, is now contributing to the improvement of quality and accuracy of framing parts and to the saving of manpower in the design and the fabrication stages. This paper provides an outline of LODACS with actual examples of its outputs.

#### 1. Introduction

Computerization in the field of mold loft work for shipbuilding was initiated in the area of curved shell plate expansion calculation. The storing of the hull offset data into the computer file as an input in such calculations contributed much to the subsequent speedy acceleration of the development of the hull lofting system and promoted the establishment of a comprehensive fundamental system that became an integral part of the hull design system. What is more, the establishment of this hull offset file has had the effect of establishing the method of the longitudinal frame development as the next phase of computerization. Nine years have elapsed since this system was first developed, and seven years have passed since the techniques were universally adopted by our shipyards. The tremendous effects in the areas of work speed-up and labor saving that have resulted from the replacement of the conventional manual lofting system by the computer-zided lofting system can hardly be overstressed.

Following the development of the above method, a twisting mold program for longitudinal members was developed as an associated technique with the primary system, and its extended application to the internal members including end-bracket development was attempted as the component module of the integrated hull design system which, in fact, has been partly implemented.

These, however, involved the following problems: .

- 1. There was a lack of mutuality among the programs, due to differences in the time of development, and operational difficulties were observed in the management of the files and program maintenance with their inherent complexities. Such is considered attributable to the difference in their development stages. Further, in the case of the component module of the integrated system, the partial system operation independent from the entire system was not readily obtainable due to the system structural restrictions.
- 2. The part dimension plans of the longitudinal frames which are the main product of the system were produced in the form of the line printer output, thus the statements had to be expressed only with the code and symbols limited by alpha numeric. As a result, special knowledge was required to read off the statements one of the causes that lead to erroneous productional operation.

In the meantime, development in the area of NC cutters, NC marking devices and NC frame benders was aggressively advanced in association with the application and implementation of the main system. To keep up with the development in such NC techniques, a basic integrated system has become necessary. The above is the background that lies behind the development of the LODACS (Longitudinal frame Development And Conducting System). In accordance with the changing demands

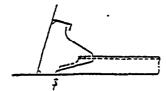


Fig. 1 Example of Frame-end Having Data
Common to the End-bracket

at present for ship forms ranging from tankers to cargo boats, efforts are being made to incorporate the transverse frame development program, and practical fields of application of this system have been widened regardless of the limits implied by the original naming of the system.

#### 2. Development

In the system development, emphasis was placed on the following points:

- 1. The system should be capable of processing normal curveless frames, those to be fitted to the shell plates, and also frame-end brackets.
- 2. The part dimension plans should be output in the form of X-Y plotter, dot printer and COM (Computer Output Microfilm), and be visualized from the dimension tables into dimension plans. No special knowledge or experience is required for reading off the information. English letter or code are used as they provide better access to external users including overseas users.
- To allow simple operation, each program should be well systematized and unified.
- 4. The output required for common manual marking should be obtained as basic information, and it should be designed in such a way that the control information can be provided to all the NC devices through the postprocessors.

The features of this system thus developed provide for easy access to the part dimension plan, and outstanding, flexibility allowing applications of the system in any type of shipyard production facilities. Fairly long time was required before determining the suitable specification of a part dimension plan; extensive process standardization was developed by analyzing the marking procedure. A system development should always be based on the end user's requirements; and it is important that the user should not compromise his needs, reasoning that the system producer is limited by restrictions in computer hardware composition or complications in programing. The hull-shop production facilities of our five shipyards are-varied in operation systems, -ranging from the all NC system to the all manual system; the flexibility of this system has been realized through its application to all the facilities. This system uses the IBM model 370/135 ~ 158 computers at present. The program language employed is principally FORTRAN, and ASSEMBLER is partly used.

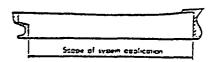


Fig. 2 System Application Range

#### 3. Function

This system is designed to deal mainly with the development processing of the hull panel stiffener such as the transverse frames, longitudinal frames and their end-brackets with the added function of a variety of associated application programs. (Hereinafter the transverse frame and longitudinal frame are simply called frames; the frame-end brackets are simply called end-members.) The added capability to handle the end-members was provided mainly because these members are always connected to the frames, and they have elements determining the cutout shapes of frames (Fig. 1). Also, it not only enables the system to simultaneously process the data common to the frames and end-members by a single input operation but also provides ability to process the frame or end-member alone independently.

The scope of application of this system covers the frames and end-members to be fitted to the side shell, upper deck, longitudinal bulk-heads, and various flat decks except for those highly complicated parts at the end of the bow and stern (Fig. 2). As shipbuilding materials, not only the angles and built-up sections, but also slabs and bulb angles are taken into consideration.

# 4. Setting Up

As shown in Fig. 3, the development program is implemented by producing the input data from the hull design, and taking the offset information from the hull base-data file. The results of such a development operation will then be put into the storage of the dimension data file in the case of the frames and standard type endmembers, and those off-standard end-members will be stored in the cutter location data file. Subsequent system flow will be processed into the output program groups adaptable to the production facilities being set up in the shipyard under consideration.

## 4.1 Setting Up of System

#### 4.1.1 Development Program

The development program is the vital part of the system and handles the development processing of the frames and end-members as mentioned above. To simplify the input data as much as possible, the system was elaborated to yield the following functions:

 According to the kind of section type, dimmension, and bending direction (i.e., outward bending or inward bending), different bending neutral axis is calculated automatically<sup>15</sup>.

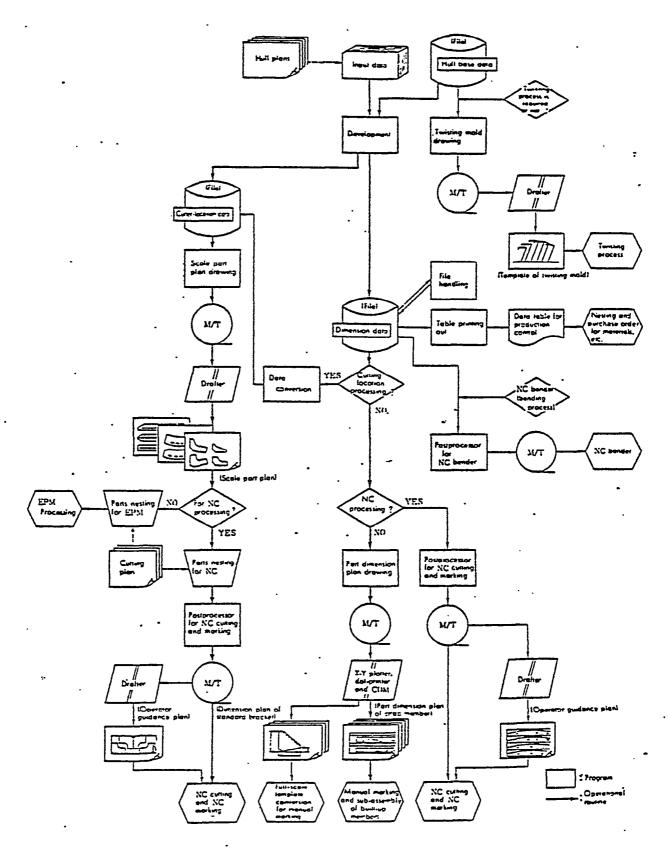


Fig. 3 Flow Diagram

- 2. Inclined fitting angles of frames, their beveling of edge preparation, and marking surface arc determined automatically.
- 3. Shapes of frames in detail (cut-off dimension (a), dimension (b). scallop (r) and bevel angles to be profiled (\$\theta\_1\$) \( \beta\_2\$tc.) are decided automatically (Fig. 4).

In addition to the above, automatic generation of the data for the secondary members such as face plates and flat bars is mad: possible; the computing function of weight, cutting length, and center of gravity are also provided, and their results are put into the storage of the file. The frame bending lines (Fig. 5) to be made straight after the bending process are computed to a maximum of three lines on judging the scantlings of frames and their bending requirements.

#### 4.1.2 File Handling program

The programs that control the dimension data File are:

- 1. The initializing program that is set when the system operation is initiated.
- 2. ship to ship data file transfer program to be used partially or entirely for a sister ship or a ship of similar construction and-dimensions.
- 3. Corrective or cancellative program to be used for partial modification when the design is to be changed in part.

All of the above help to facilitate system operations.

#### 4.1.3 Production Control Data printing Our program

To grasp and verify the contents of the dimension data file, and check adequacy of material size for a purchase order and fabrication work scheme this program prints out the parts data such as the block name, code, length, scantling and weight.

#### 4.1.4 Part Dimension Plan Drawing Program

A part dimension plan is drawn by the X-Y plotter (Fig. 6), dot printer (Fig. 7), or by COM with the required output from the dimension data file for the manual marking and built-up guidance plans. In this process, data are not only assorted according to groups of classified materials such as slabs, angles and built-ups or bend pieces and straight pieces but also assorted based on the fabrication lines, scantlings and the order -of lengths to suit the intended purposes.

All the shapes are unified in figures and/or terms of actual *images* and the title code and the like are stated in English in consideration of possible license agreement with shipyards in foreign nations (Fig. 8 - Fig. 11).

Namely:

PART CODE Code of parts
- SCATLING Sectional dimensions -SHOWN Surface shown on the plan
s . PRM Preliminary surface treatment &
Shop prima painting

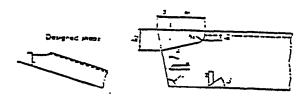


Fig. 4 Example of Fram-end Decided Decided Automatically

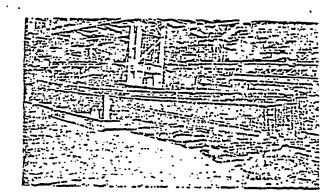


Fig. 5 Frame Bending Line Straightened after Bending.

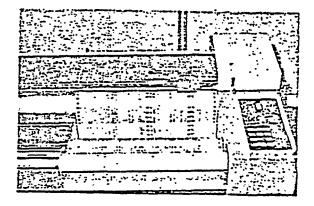


Fig. 6 Drawing View of Pan Dimension Plan by the X-Y Plotter

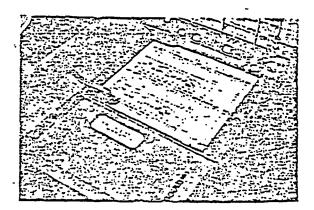


Fig. 7 Drawing View of Part Dimension Plan, by the Dot Printer

BEND Bending method

TWIST Twisting in fabrication shop [nese represent only a few examples. To facilitate that marking, due considerations were taken on the owing points:

1. Marking length is shown on the & basis of the cumulative dimensions, and in particular the end Cut out sequence is made identical to those in manual marking procedure.

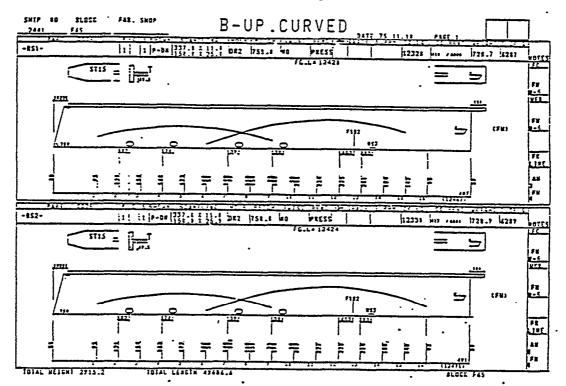


Fig. 8 Part Dimension Plan of Curved Built-up Frame (by COM)

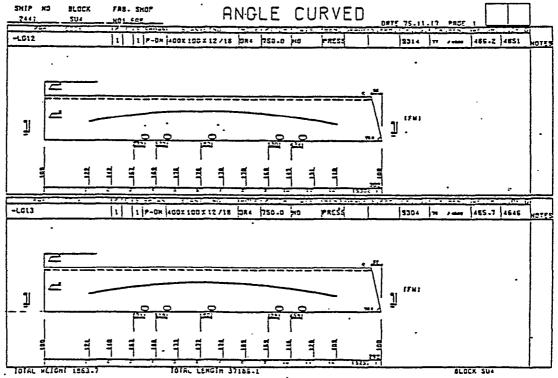


Fig. 9 Part Dimension Plan of Curved Angle Frame (by Plotter)

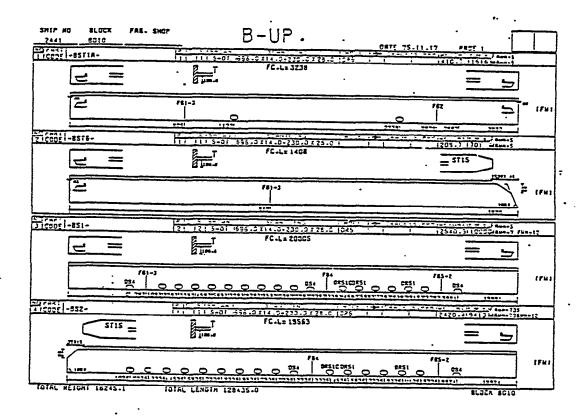


Fig. 10 Part Dimension Plan of Straight Built-up Frame (by Plotter)

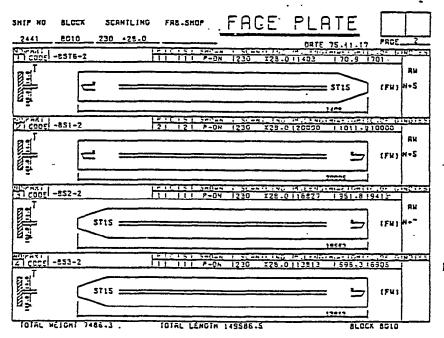


Fig. 11 Part Dimension Plan of Face Plats of Built-up Frame (by Plotter)

- 2. Drawing is made with the same side view as marking
- 3. End shapes and bending line curves are expressed in a somewhat exaggemted manner to improve presentation of images.

The part dimension plan (Fig. 12) of the curve cutout frame differs from that in the ordinary bending process

effected by referring to the frame bending the in that an analogous pattern requiring no bending process car be restored by marking through the dimensional con from one base line. In the case of the standard brackers the program also provides the dimension output aSSum possible requirements for the marking with a given of dimensions or full-scale template conversion.

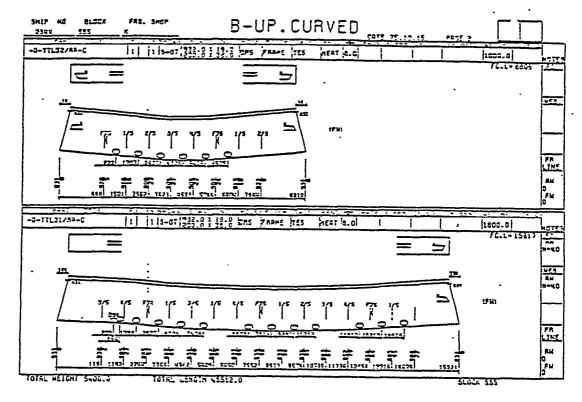


Fig. 12 Parr Dimension Plan of Cut-out Frame [by Dot-printer)

# 4.1.5 Posiprocessor for Numercally Controlled Frame curring and Marking

This program outputs NC tape, by which the cutting of drain holes on the frame and frame-ends, the marking of the frame bending line and other necessary lines are conducted by the NC frame cutter (Fig. 13). It also carries out parts nesting automatically with the given information from the dimension data file as required; output in the form of magnetic tape or punched tape is provided. This punched tape will then be fed into the drafter capable of checking the contents of the information to draw a plan in a 1/20 scale which will further be supplied to an operator of the NC cutter as a guidance plan (Fig. 14).

#### 4.1.6 Posrprocessor for Frame NC Bender

This is a postprocessing program that acts *to* subject the cut frames to the automatic bending process by the *NC* frame bender (Fig. 15). The data necessary for frame bending is obtained from the dimension data file to determine the distance between the points of press action in accordance with the bending requirements, and the computed magnitude of bending action will be output to magnetic tape as the control numerical data in terms of angular quantity.

#### 4.1.7 Production Program of Twisting Mord

When twisting of the frame is judged necessary by the design criteria, the essential data are obtained from the hull base data file in accordance with the name of longitudinals, and designations specifying the range. In the case of manual Operation, necessary information should be obtained from the ship's body plan. Thus the twisting mold is made so as to be set on the transverse

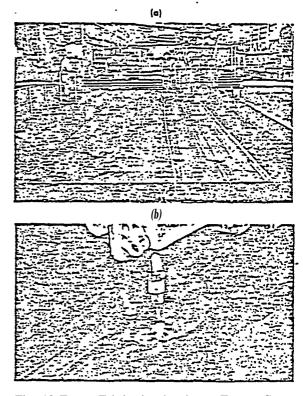


Fig. 13 Frame Fabrication by the NC Frame Cutter

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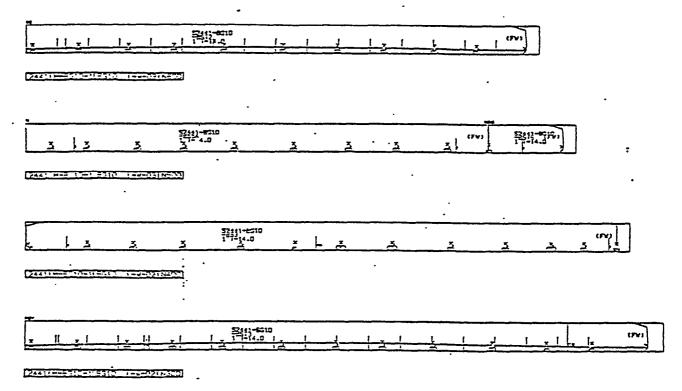


Fig. 14 Operator Guidance Plan for Frame NC Cutting

frame lines; further, it involves a certain degree of angle to the web surface of the longitudinal. This program provides the output of twisting mold at a right angle to the shell line and web surface. With the processing of this program, not only can the marking of transverse frame lines be omitted but also the twisting operation itself has improved significantly (Fig. 16).

#### 4.1.8 Data Conversion Program

When the standard brackets, cutout frames and face plates being stored in the dimension data file are to be drawn as output in the form of EPM film or full-scale contour templates, the dimensional data can be expressed as cutter location data which are stored in the cutter location data file.

#### 4.1.9 1/10 Scale Part Plan Drawing Program.

This program outputs magnetic tape for drawing by designating the block name or the part code from the cutter location data file. This magnetic tape will then be used to draw a 1/10 scale part plan film (Fig. 17) by the drafter. This part plan is used for multiple purposes such as EPM, film for projector marking, and parts nesting for NC marking or NC cutting.

#### 4.1.10 Postprocessor for NC Cutting

The parts nesting produced by the above part plans will be channelled through this postprocessor to output data in the form of magnetic tape or punched tape per every unit of steel plates. As with the case in the postprocessor (4.1.5) for frame NC cutting and marking,

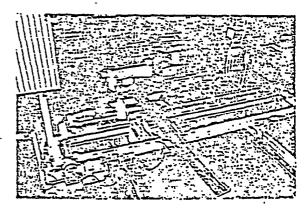


Fig. 15 Frame Bending by the NC Bender

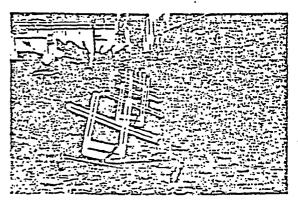


Fig. 16 Frame Twisting

Japan Shipbuilding & Marine Engineering

such tape will be supplied to the work site together with the operator guidance plan drawn by the drafter. In other words, this program can follow up either the NC cutter or NC marking device (Fig. 18).

#### 4.2 Setting Up of File

#### 4.2.1 Hull Base Data File

The Direct Access Method type file is used wherein the hull offsets, scantlings of the longitudinals, those section types, and specific data of the longitudinals such as the thickness side of webs against mold line and the direction of the flanges are contained. Furthermore, this file is not only utilized for the LODACS but also for the shell plates expansion and internal members development processes.

#### 4.2.2 Dimension Data File

This Direct Access Method type dimension data file containes the indices of block names. Here, the developed

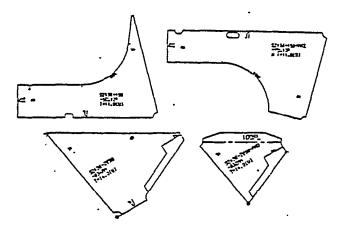


Fig. 17 Part Plan of Frame End-bracket (by Drafter)

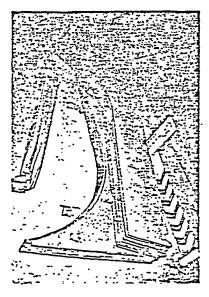


Fig. 18 Frame End-bracket by NC Cutting

frames. face plates and flat bars known as the strap members, and standard type brackets and cutout frames are included in the dimensional forms. Each component member has its corresponding record, and the information consists of the general items such as the part cod; santling. length and control parameter for production activity, the common graphical items such as end shape and scallops, bending information. data regarding drain holes, and marking information.

#### 4.2.3 Cutter Location Data File.

This Direct Access Method type curter location data file contains the cutter location data which arc to be output in the form of the analog patterns. In addition to the end-members, the standard type brackets, face plates and cutout frames data converted from the above dimension data file can be put into the storage of this file- as required. This file, as with the cases above, can be utilized also for other systems covering the expansion

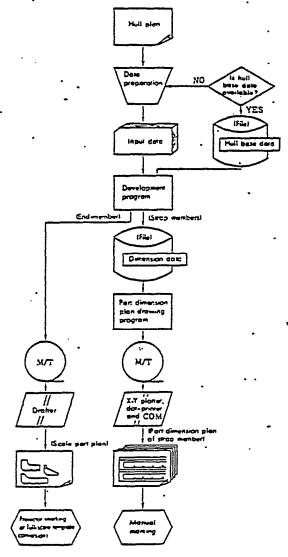


Fig. 19 Example of System Operation Flow Chart

curved shell plates and development of internal members.

#### 5. Operation

5.1 Preparation for System Operation and Its Simplified Operation

With respect to the flow diagram shown in Fig. 3, a variety of operational methods an be taken Namely, only those necessary programs can be selected after development programs.

The preparation of the dimension data file is indispensable for the control of the files, but in an emergency as: as in the quick repairs of a vessel, specific data can be input even when the hull base data file is not available. When the preparation of EPM films or full-scale templates is intended, the cutter location data file may be bypassed, and magnetic tape tape or punched tape for NC drafting can be output directly from the development program. Shown in Fig. 19 is an example Of the system operation without utilizing the hardware techniques of numerical control fabrication.

#### 5.2 Standardization

It goes without saying that computerization and standardization go hand in hand. Extensive standardization which resulted in distinct simplification both in design and in input operation was developed in this system. In particular, five types of frame-end shapes and more than ten types of end-members have been standardized Consequently, the necessity to write down all the details of end-members in the design phase has been eliminated. All that is required is to note the designated type number of end-member shown in the standard specifications. Similarly, only the designated type number of end-member shown on the relevant design plan is required to be input. The newly provided access permitting the input of those principal dimensions related to design functions allows added system flexibility to deal with the hull members of special types, and contributes significantly to even greater labor saving.

# 5..3 Input

Input items of the system, aside from the development program, con&m the block names, part codes and selective machine code, hence description on these items is omitted her:. In this article the input for the development program will be shown. Generally, processing is made for every building block where data are broadly classified as those data common to blocks and those concerning each frame member.

#### 5.3.1 Data Common 10 Blocks

These data cover the ship number, block name, and drain holes which require only one input operation at the initial stage of each block processing. Of these, the input operation for drain holes is cumbersome; however, simplification has been made in such a way that the

type of drain holes can be omitted where they are identical to those on the immediately preceding member.

#### 5.3.2 Data Specific to Each frame

To minimize the input operation frequency, the name of the longitudinal. type number of the end-member, range of frames processed and specific data other than the standard type are required to be input for the first member only, but all such data for the subsequent members can be largely omitted, except for the name of the longitudinal and part cod: name. providing that they are dimensionally identical to the first one.

#### 6. Conclusion

With comprehensive coverage and sophisticated functions, LODACS has, become an extremely useful system that meets the multiple requirements of the shipbuildiig design and production system. As already mentioned, this system it a longitudinal frame development and conducting system incorporating the groups of programs developed and implemented by MI with added functions later developed. This system does, however, have some unsolved problems in dealing with the transverse frame end-brackets, beams and beam-knees etc. Other problem areas include such items as the automatic drawing of the sectional view of frames in the course of hull plan preparation, supplying those control data to the production control system, and the like, to name but a few. A greater corporate effort will thesefore be made in the future to solve these remaining problems.

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- (4) S. Kohtake, H. Hamajima and K. Shimizu IHI High Speed NC Marking System, Ishikawajima-Harima Engineering Review Vol. 16 No. 1 Jan. 1976 pp. 100-106
- (5) I. Tsuji: On the Position of Neutral Axis of Section during Bending. Transactions West-Japan Society of Naval Architects No. 27 My. 1964 pp. 147-460

## APPENDIX E

SPECS - SHIP'S PRELIMINARY AND EXACT CALCULATION SYSTEM



# SPECS

SHIP'S PRELIMINARY AND EXACT CALCULATION SYSTEM



OCT. 1978

# Ishikawajima-Harima

Heavy Industries co, Ltd. TOKYO JAPAN REF. No.

## (1) ABSTRACT OF 'SPECS' SYSTEM

This system shall be applied to detailed design in ship's preliminary calculation on Hull Form and its capacities. The system design concepts and aims are as below.

- \* Expansion of application by widely used programs
- \* Easy Maintenance of both programs and Data.
- \* Reconsideration of design system and Pursuit of man power saving.

This is a totalized system which can make designers possible to prevent from duplicate input by registration the results of each step in the Data Bank through which next programs shall retrieve them.

## 1) Applied ship's type

Applied tankers and carriers up to D/W 1,000,000 In addition to the ordinary ships, following type can be also applied.

Ship's type ----- naval vessels, patrol ships, fishing boats

Hull form ----- Initial trim, Knuckled shell

Inner hull structure -----

Almost all types of ship shall be applied using MAP-method developed by IHI.

## 2) Other restrictions

ORDINATE ----- Max, 50 (w.L = 60)

FRAME ----- Max, 500 (W.L = 150)

TANK / HOLD ----- Max, 70 block

Loading condition --- Max, 100

4

#### (3). FEATURE OF THE SYSTEM

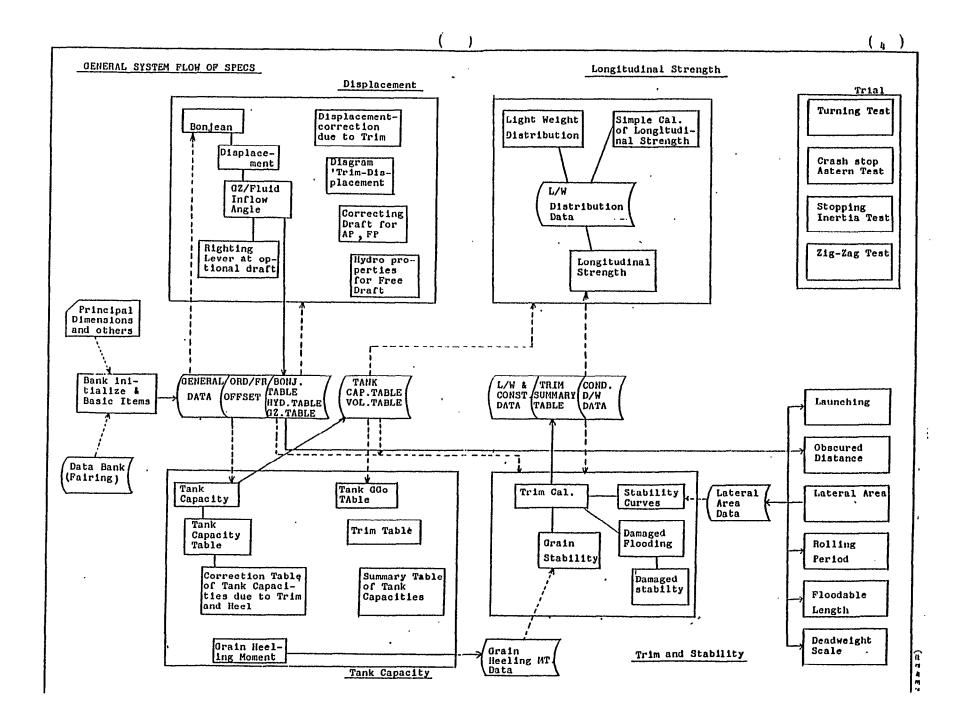
- 1. Operating methods
  - \* Using as total system which can save input data by retrieval the preceding results from the data Bank.
  - \* Independent RUN method
    Related data to be INPUT directly for each program.

    Both running method are available.
- 2. Drawing shall be generated as output.
- 3. Calculated results in the Bank can be applied to another sister ship for effective process.
- 4. FRAME & ORDINATE OFFSETS shall be automatically applied, in case that LINES FAIRING has been complexed.
- 5. FRAME OFFSETS can be provided from ORDINATE OFFSETS' by OPTION FUNCTION.

Various studies in initial design stage can be proceded by this function.

- 6. MAP- method developed by IHI shall make it easy to apply the system for the complexed shapes of ship.
- 7. New Rules of IMCO are applied.

4



# (4) FUNCTION AND OUTPUT OF EACH PROGRAM

Program name	Function of Program	Output
BANK INITIALIZE & Basic items	Preparation for I/O process of all calculations shall be done. Basic items such as principal dimensions, FRAME SPACE and etc. are stored.	
Bonjean Cal.	Bonjean calculation of MLD and EXT are executed.	Bonjean (EXT) Bonjean (MLD) Bonjean Curves
Hydrostatic Properties Cal.	HYDRO PROTERTIES TABLE is gene rated.	MAP APPEN
GZ and FLUID INFLOW ANGLE	Calculation of GZ (Righting Lever) at any angle.	Cross Curve of stability
Cal.	Calculation of the displace- ment at the start of INFLOW of sea water.	Curve of INFLO
Righting Lever at optional. draft	GZ TABLE at optional draft is generated.	GZ for opera- tional infor- mation
		GZ-Healing Curves
Displacement- correction Cal. due to Trim	Displacement correction Table in optional draft-Trim is generated.	Displacement correction table due to Trim
'Trim-Displace- ment' Diagram	Generating Diagram 'Trim-dis- placement' for finding draft at bow and stern.	Diagram 'Trim- Displacement'
Draft Correc- tion Table at the point of draft Mark	Values for correction which shall be used for calculating drafts at the both of AP and FP from the appearent drafts at the Draft-Mark position.	Draft correction table for Draft-Mark
HYDRO Proper- ties Cal. for Free Draft	Calculation of HYDROSTATIC PROPERTIES at any part of the ship, at Trim-status and at HOG & SAG status.	HYDRO PROPER- TIES for Free Draft
Read in frame offsets	Frame offsets read by eyes.	Frame distance table
•	·	Frame offsets

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Program name	Function of Program	Output
Tank Capacity Cal.	Capacity, center of Gravity INERTIA etc. are calculated.	Tank capacity table Tank capacity curves
Tank Capacity Table Cal.	Tank capacity table is gene- rated due to SOUNDING and ULLAGE.	Tank capacity table
Correction Table of Tank capaci- ties due to Trim and Heel	EVEN KEEL at TRIM HEEL is	Correction value table
Grain Heeling Moment Cal.	TRANSVERSE & VERTICAL HEELING Moment are calculated.	Grain Heeling Moment table
GGO Table for Operational Information	Decrease of GM by free surface effect is calculated.	Table .
TRIM Table	Estimation of the change of Trim when a load of 100t added to optional Tank.	Trim Table
Summary of capacities of Tanks and etc.	Final drawing of Tank Capcity.	Tank Capacity Summary Table
Trim Cal.	HOMOGENEOUS LOADING at option- al Tank and Trim adjustment by two fixed tank.	Loading condi- tion Drawing
Lateral area Cal.	Lateral area and center of Lateral area are calculated.	Curve of Lateral Area
Stability Curves Cal.	GZ angle of vanishing stability and GZ maximum. Coefficient C1, C2 considered DYNAMICAL STABILITY of IMCO Rule, maximum allowable helling angle, wind and oscillation.	'GZ-Heeling
Grain Stabili- ty Cal.	Residual Area, Heeling angle, Allowable Heeling Moment etc. are calculated.	Table of Heeling Mo- ment Diagram 'GZ-Heeling Angle'

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Program name	Function of Program	output
Damaged flooding and damaged stability Cal.	Final balanced CONDITION after damaged by method of lost Buoyancy is generated. DYNAMICAL STABILITY considered wind and oscillation can be also calculated.	Diagram GZ- Heeling Angle
Light Weight	Light Weight, center of Gravity	Light Weight
distribution	are calculated and distributed	Distribution
Cal.		Curve
Longitudinal strength Cal.	SF. BM DEFLECTION in case of STILL, SAG, HOG are calculated.	Comparison Table for Bend- ing Moment Diagram Bend- ing Moment- Shearing Force Envelope Curve of Shearing Force Envelope of Bending Moment
Simple Cal. of Longitudinal Strength	BASE VALUE such as Coefficient of Weight Distribution Shearing Force etc. at the point of Bending Moment are calculated.	
Launching Cal.	Launching particulars, Float condition, launching speed etc. are calculated.	Launching Curves Launching Speed and Travel Curves
Obscured distance in relation to various Draft and Trim	Table of Obscured Distant at the designated draft and Trim is generated.	Table of Ob- scured Distance in relation to various Draft and Trim
Rolling period Cal.	Table of GoM at every designated draft and oscillation period is generated. BY Kato's equation or results of experiment.	

Program name	Function of <i>Program</i>	Output
Floodable length Cal.	Floodable and Permissible length at optional points are calculated. By the standard of Ministry of Transport or Floodable length detailed Bonjean Method.	Floodable Length Curve
Dead Weight- Scale Drawing	D/W SCALE unit in METRIC or FEET.	Dead weight- Scale
Turning Test	Measured results such as Speed and Turning Angle at the Trial are analyzed and calculated.	Drawing on Turning Test
Crash Stop Astern Test and stopping Inertia Test	Results of Crash Stop Astern Test and stopping inertia test are analyzed. Its wake and speed curve are drawn.	Results of crash stop Astern test Cource of Crash stop Astern Test Result of Stop- ping Inertia Test Course of Stop-, ing Inertia Test
Zig-Zag Test	Results of Zig-Zag steering Test are analyzed.	Table of Maneu- vability

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### APPENDIX F

SPECS - ACTUAL OUTPUT EXAMPLE

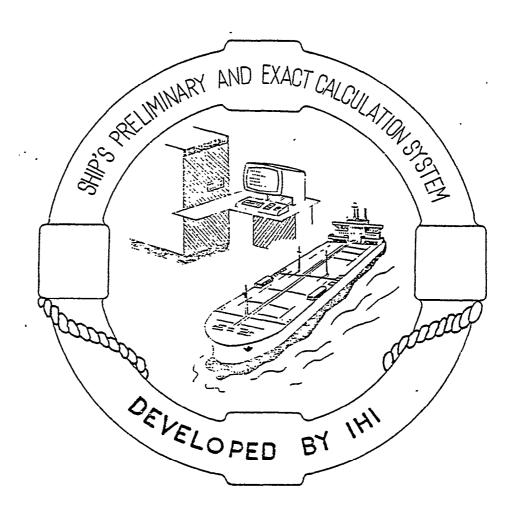


ACTUAL OUTPUT EXAMPLE

OF

SHIP'S PRELIMINARY AND EXACT

CALCULATION SYSTEM



1011 石川島播房重互業株式会社

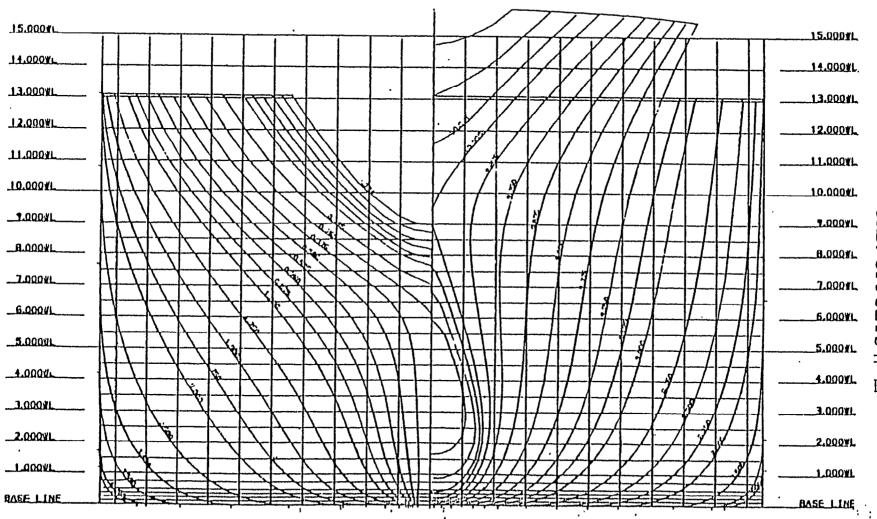
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LENGTH SETHEEN PERPENDICULARS  DREADTH (MOULDED)  DEPTH (MOULDED)  DRAFT (DESIGNED HOULDED)  DRAFT (DESIGNED HOULDED)  DRAFT (DESIGNED EXTREME)  DRAFT (DESIGNED EXTREME)  O.O H  DISTANCE AFLEND FAON A.P. DISTANCE FELEND FAON A.P.  ATSE OF FLOOR AT HISSIPP  STARTING POINT OF RISE OF FLOOR  ADUS OF BILGE CIRCLE  **SEA GRAVITY = 1,02500 **********************************	LENGTH DYFR ALL	167.730 H	·	
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1.0030	1.965	2.282	7.430	2.898	3.122	3.408	3.740	4. 235	3.798 4.582	4.241	4.482	5.14
1.2500	2.995	3.373 -	3.803	4.131	4.401	- 4.037	5.185	. 3.724	4.149	5.131	5.434 ' 7.482 '	4.19
1.5000	4.154	4.585	5-069	5.436	5.739	4.224	6.608	7.203	7.680	6.847 " 8.480	9. 174	#.124 9.43
1.7500	-5.421-	5.070	-6.390	-6.778-	7:098	7.609		::31	4:135-		70:441	11:26
2.0000	4.757	7.215	7.779	8.119	8.440	4.935	9.354	9.970	10.470	11-275	11.490	12-396
2.9330	- 4.455	9.853	14.301	10.439	10. 919	11.366 "	11.712 **	12.231 "	12.433	13.196 ***	-13.543 -	13.77
3.0000	11.404	12.095	12.411	12.645	12.637	13.130	13.370	13.699	13,912	14.117	14.185	14.20
3.5000 "	13.100	"13.341"	13.593	13.741 "	" 13.854 -	14.016	14.117	14.199	14.200	14.200	14.200	- 14. 2u
4.0000	13.143	13.515	13.721	13.863	13.968	14.109	14.182	14.200	14.200	14.200	14.200	14.200
3.6655	13:343	13.515	13.721	13.843	13.965	-14.109	-14-1 02	-14.200	-14.200-	-14.200-	-14.200-	14.20
4.0500	13.343	13.515	13.721	13.843	13.968	14.160	14.102	14.200	14.200	14.200	14-200	14-200
4.3446	13.343	13.515	13.721 "	13.463	13.946	14.109	14.102	14.200	14.200	- 14. 200-	- 14.2JO -	14.20
7.0000	13.226	13.424	13.633	13.760	13.890	14.045	14.137	14. 200	14 - 200	14.200	14.200	14.20
7.5000	12.333	12.400 "	12.896	13.112	13.283	13.542	- 13.727 -	13.966 "	14.094	14.191	14.200	14 . 200
0.0000	10.432	10.995	11.400	11.703	11.949	12.348	12.435	13.044	13.358	13.706	13.670	13.92
8.2575	4.501	· • • • • • •	10.140	10.696	10.472	11.410	11.740	-12.276-	-12.445	-13.119-	-13.363-	-13.44
8.5000	0.204	8.647	9.132	7.495	4.742	10.270	10.655	11.237	11.464	12.236	12.545	12.67
8.7533 "	4.771	· 7.:35	" 7.740	0.115	* 8.433 ****		9.335	9.954 ""	10.417	11.035	11.302	11.54
.0000	9.250	5.715	4-224	6.407	4.920	7.428	7.834	4.458	0.716	9.534	7.881	10.07
4.1756	4.478	4.935	5.440 -	5.615	4.124	6.628	7.027	7.637 ""	8.083	8.670	9.014	9.21
9.1506	3.711	4.155	4.640	5.010	5.313	5.601	4.106	6.775	7.200	7.757	8.080	0.270
9.3740	1.447	3.345	3.846	-4.220	4.486	4.954		- 5.075	6.272	6.707	7.005	7.26
4.5000	7.244	2.430	3.041	3.347	3.657	4.091	4.434	4.940	5.361	5.760	6.032	4.164
9.4256	1.570	1.410	2.291	2.549	2.82+	3.215	3.518	3.975	4.242	4.700	4.425	5.044
1.7504	0.974	1.265	1.506	1.820	2.020	2.337	2.590	2.975	3.248	3.588	3.762	3.844
9.8750	0.540	6.797	1.070	1.248	1.426	1.467	1.052	2.128	2.319	2.564	2.642	2.704
0.0000		0.375	0.704	0.910	1.070	1.315	1.465	1.708	1.848	2.004	2.047	1.90
0.0135			3.170	7.655	0.874	1:184	1:356-	1.560	1.726	1.020	1.830	1.710
0.127u 0.1905		··				0.835	1.151	1.420 1.030	1.286	1.464	1.618	1.410
					i .'.					1:370:		0.435



1084 784 884 784 681 581 481 381 281 181 C. 181 281 381 481 581 681 781 881 981 1081

U

			BON	JFAN	* SQU.	ARE HETFR.	BOTH SIDE	(EXT) +	·	
ORDINATE NO	31	32	33	34	3.5	36	37	<sub>38</sub>		40 -
DISTANCE FROM AP	~137.6: H	162.525H	"167.456H"	172. 375H	177.3004	179.762H	182.225H	184.6478	187.1508	189.612H
DRAFT EXT (H)	•••		***			• ,	••		•	
0.175	1.82	1.69	1.53	. 1.33	1.09	0.96	0.03			
U. 15 U	3.06	3.60	3.27	2.05	2.36	2. 48	0.83	0.69	0.55	0.41
0.275	7.34	4.06	6.25	5.46	4.56		1.80	1.50	1.21	0.92
.41:	14.00	10.20	9.31	8 • 19	6.05	4 - 05	3.51	2.96	2.40	1.84
9.525	14.49	13.60	12.43	10.97	9.21	6.10	5.31	4.50	3.67	2.85
0.775	71.67	70 .53	18.85	16.72 ····	14.12	8.23	7.19	6.11	5.01	3.92
1.025	79.29	17.62	25.43	22.65	19.24	12.67	11.12	9.51	7.86	6.22
1.525	44.53	42.17	34.02	34.96	24.63	17.31	15.26	13.10	10.95	0.69
7.025	60.05	57.08	53.05	47.79	41.19		24.01	20.76	17.41	14.04
2.525	75.71	72.19	67.34	60.94	52.79	. 37.38 .	33.27	28.91	24.30	19.81
3.625	91.56	07.56	61.96	74.46	64.78	40.05	42.90	37.41	31.67	25.86
3.525	107.49	103.00	96.79	08.23	77.05	59.10	52.90	. 46 • 25	39.28	32.19
4.025	123,49	118.72	111.79	162.21	09.53	70.43 81.97	63-16	55.33	47.10	38.70
4.525	139.53	124.45	1 26.61	116.35	102.25	- 33.76	73,63	64 • 67	55.10	_ 45,37
5.025	155.60	150.24	142.13	130.61	115.01	105.56	114.28	74.06	63.25	52.14
f.:·25	171.73	166.12	157.47	145.01	127.95	117.57	95.07	. 83.63	71.49	59.0U
h.025	167.03	182 00	172.82	159.43	140.95	129.64	105.99	93.33	79.85	65.94
6.525	261.94	197.90	108.20	173.90	154.00	141.76	116.97	113.08	08.25	72.89
7.025	2:0.04	213.80	203.61	100.41	_167.09	153.92	120.01	112.89	96.69	79.87
7.525	236.15	229.71	219.03	202.92	180.20	146.11	_139.69	_1:2.74	_105.17	_ 86.86
f.:25	251.25	245.62	234.45	217.44	193.37	170.32	150.21	132.62	113.67	93.86
0.525	268.36	261.53	249.07	231.96	216.44	193.52	161.34	142.52	122.19	100.85
9,025	284.46	277.44	245.29	246 - 48	219.56	202.73	172.40	152.43	130.71	107.04
9.525	310.57	243.35	200.71	241.01	232.67 "	214. 93	103.62	1(2.35	139.24	114.82
10.025	316.67	309.26	296.13	275.53	245.79	217. 13	194.76	172.26	147.76	121.70
11.025	348.80	341.00	326.97	304.57	272.02	251.53	<u> </u>	_162 • 18	156.28	_128.74
12. 25	381.09	372.90	357.81	333.61	298.25	275.91	278.16	201.99	173.32	142.62
12.025	413.36	464.72	316.65	362.65	324.40	367.33	250.42	221.81	190.33	156.47
14.025	445.44		419.42	391.63	350.66	324.69	272.70 294.94	241.62	207.33	170.34
15.025	477.65	460.29	450.26	420.67	376.91	349.16		261.42	224.37	104.31
16.025	509.85	500.11	401.10	449.71	403.22	373.73	317.33	281.41	241.68	198.67
17.025	542.06		511.93	478.75	429.61	398.46	362.67	_3 01 -65	259.36	213.57
16.125	574.27	563.74	542.77	5:7.79	456.13	423.40	305.80	322.26	277.58	279.21
19.025	597.23	504.50	563.36	527.6R	475.04	441.62	463.18	343.37	296.50	245.79
20.025	595.23		563.30	527.68	475.04	441.62	403-18	359.75	311.69	259.63
21.025	545.23	584.58 1	563.30	527.68	475.04	441.62	403.10	359.75	311.69	259.63
22.025	595.23		563.30	527.68	475.04	441.62	443.18	359.75	311.69	259.63
							743.17	359.75	311.69	259.63
PP.DK.S.L. HETGHT	16.326	18.335	10.360	18.404	18.464	18.501	10.542	10 647	10 / 04	
NASLNON	563.97	573.61	553.11	518.81	467.84	435.38		18.587	10.634	18.685
PP.DK.C.L. HEIGHT	18.058	18.858	18.858	18.858	18.858	10.050	397.94 18.858	355.52	308.45	257.34
BONJEAN	¨595.23``¨		563.30	527.68	475.04	441.62	403.18	18.858 359.75	18.058	18.858
						1 14.6 07.	743410	337413	311.69	259.63

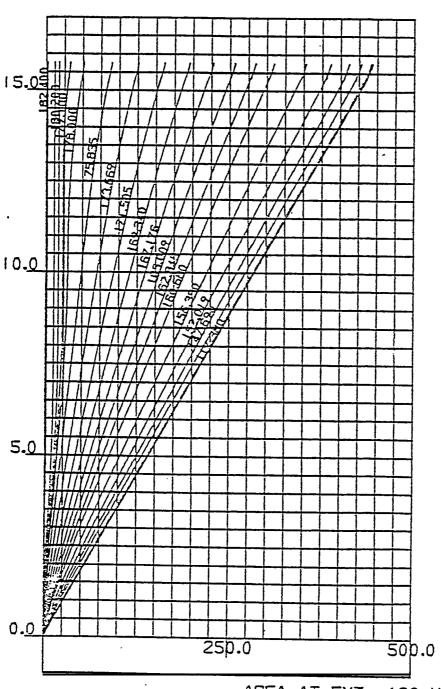
DRDINATE NO	31	32	33	34	3.5	36 .	37	38	39	40	
DISTANCE FROM AP	157.630H	162 -5 254	167.4504	172.375H	177.300H	179.762M	102.225M	164.607M	187.15CH	146-915H	
DRAFT MLD (M)		······································	<u></u>		·					<del></del>	·
0.050	76.76R	24.952	22.607	19.703	16.255	14.357	12.373_	10.333	8.275_	6.244	
0.125	27.393	25.627	23.335	20.472	17.043	15.143	13.140	11.006	9.001	6.944	
0.250 _	20,106	26 • 3 93	24 - 1 50	21.345	. 17.945	16.045	14.040	11.957	9.840	7.737	
0.375		~~26.91°	24.795	*** 22.624 ***	18.651	16.755	14.742	12.643	10.499	8.369	
0.500	Z9•140	27 • 4 98	25.342	. 22.605	19.255		15.344	13.230	11.063	8.701	
0.750	29.953	70.370	26.279	73.604	70.294	18.405	16.378	14.243	12.038	9.626	
1.000	30.653	29.125	27.093	24 . 46 9	21.190	19.304	17.271	15 .116	12.082	10.632	
1.500	31.876	30.444	28.511	25.974	22.747	20.066	18.817	16.630	14.343	12.028	
2.:00	32.979	31.622	29.760	27.302	24.115	22.234	20.173	17.956	15.630	. 13.266	
2.500	34.028	32.723	20.931	20.572	25.365	23.484	21.416	19-170	16.612	14-411	
3.000	35.050	33.782	32.037	_, 29.671	.26.526	24-654	22 -570	20 - 312	17-930	15.561	
3.500	36.061	34.816	33 -1 04	30.770	27.652	25.770	23.679	21.407	19.006	16.557	
4.000	37.046	25.826	34 . 1 . 46		20.724	26.847	24.750		20.053	17.509	
4,500	30.069	36.047	35.171	12.075	29.700	27.897	25.797	23.508	21.084	10.607	
5.033	29.770	37.854	36.176	33.099	30.810	28.929	26.827	24 - 533	22.102 _	19.617	
5.500	40.070	74.05	37.194	34.912	31.627	24.947	27.845	25.549	23.112	20-621	
6.000	41.070	39.859	30.198	35.918	32 • 836	30.957	28.855	26 • 5 5 8 _	24-118	_ 21.623	
h. 50f	43.070	40 - 859	39.199	36.921	33.841	31.962	79.861	27.562	75.121	22.674	
7.000	43.070	41.859	40.200_	37.972_	34.843	32.96%		20.565	26.123	23 • 624	<del></del>
7.500	44.010	42.159	41.200	30.922	35.843	33.965	31.865	29.566	27.124	24.624	
c.600	45.070	43 . 8 59	42 - 2 (10	39.922	36.843	34 . 965	32 • 865	30 .5 67	20.124 .	25.624	
0.500	46.070	44 . 0 59	43+200	40.972	37.043	35.965	33.865	31.567	29.124	26.624	
9.110(	47.070	45 . 8 59	44.200	41 • 927	38.843	36.965	34 - 8 45	32 • 567	30.124 _	27.624	
9.500	49.070	46.859	45.200	42.922	39.843	37. 465	35.865	33.567	31.174	28 • 625 29 • 625	
10.000	49.070	47.059	46.200	43.922	40.843	38.965	36.865 30.865	34.567	32.124		
11)	51.070	49.859	40.200	45.922	42.843	40.965 42.965	45.865	38.567	34.124 36.124	33.625	
17.600	>3.070	5i •059	50.2()	47.922	_ 44.843 .	44.965	42.865	···· 40 • 5 67	38.125	35.627	
13.000	55.070	53.859	52.200 54.200	49.922 51.972	46.843 48.843	46.966	44.866	42.569	40.130	37.642	
14.000	57.070 59.070	55 •859 57 •859	56.200	53.922	50.843	40. 967 ···	··- 46.870	···· 44 • 5 78 ···	42.152	37.642	
	61.070	57.859 59.859	58.200	55.922	52.845	50. 971	48.880	46.602	44.204	41.792	
16.000	63.27)	61.059	60.200	57: 922	54.844	52.979	30.901	48.650	46.298	43.961	
10.000	65.070	63.059	62.200	59.922	56.853	54.995	52.938	55.728	48.444	46.208	
19.000	- 67.677"-	66 • 479	64 . 8 70	62 . 680	39.740	57.972	- 56.035	53.919	51.925	49.978	****
20.000	67.672	66.479	64.870	62.600	59.740	57.972	56.035	53.989	51.925	49.978	
21.000	67.672	66:479	64.070	62.600	59.740	57. 972	56.035	53.989	- 51.925	49.978	
27.000	67.672	66.479	64.870	62.600	59.740	57.972	56.035	53.989	51.925	49.978	
DK.S.L. HFIGHT	18,361	18.310	10.335	18.379.	18.439	18.476	10.517	18.562	18.609	18.660	•
GIRTH	** 67.672 ·	66 .479	64.870	1.62.600	59.740	··· 57.972 ·	56.035	53.989	51.925	49.978	

## ボンジャン(作画)

# BONJEAN CURVES

(FORE BODY)

DRAFT IN M (EXT)



AREA AT EXT. (SQ.M)

排 水 量(MAP APPEN)

SONN	AR DOOH 1	•	COMPARTHEN	Y NO	1						
			•	• -		•	• `•	• • • • • • •			
W.L.	AUTHE		K.G.HT.	C.L.HT.	LCB HT.	W.P.A.	LCF HT.	CL.F HT.	1.4TAV.1	1. (LONG)	<del></del>
-2.900	. 0.20	<b></b>	····· -1.	-0.	27.	. 7.	'762.	··· ··· ··· ·· ·· ·· · · · · · · · · ·	···· · j.	137470.	
2.100	1.11	0.	-3.	0.	358.	11.	1523.	-0.			
-2.600 000.5=	3.71 5.45	0.	-10, -13,	-0. 	527. 778.	14. 16.	. 2321. 2633.	-0. 0.	17.		
-2.400	7.43	V:		-0	1057	···· ·	3044	-0.			
	12.10	Ö.	-30.	-0.	1723.	26.	3679.	-0.	42.	-514691.	
-5.400	17.50	Ų.	-42.	-0.	2494.	24.	4008.	~o.			
-1 •000 	27.68 30.38	0. 0.	-53. -65.	-0. 0.	3362. 430 <i>0</i> .	32.	4572. 4869.	. ~U.	61.		
	37.70	0.	70	-0.	4005	36.	5074.	0	71		
-1.419	37.54		-73.	-0.	5319.	37.	5200.	-0.	73.	734341.	
	45.07	. 0∙	-03.	-5.	6377.	38.	5361.	~U. ~O.			
-1.000 000.0	52.54 60.79	٥. ٥.	-94. -101.	-0. -0.	7472. 8592.	40. 40.	5565. 5614.	~u.			
-0.600	64.76	0.	-106	-0.	7711.	40.	5566.		71.	782359.	
	72.70	0	=109		10764	39,	5505.	-0.			
-1,4((	76.54 04.14	٥.	-110.	-0-	10011.	39.	5425.	-0.		762357. 725967.	
0.0	91.23	. 0. U.	113. -113.	0. -3.	11 n72. 12968.	. 37.	5166. 4781.	-0.			
0.050	91.23	o.	-115.	-0.	12008.	0.	0.	ů.		V.	
0.100	21.53	٥.	-113.	-0.	15000-		0.	g.	0.	0.	
<u>0.150</u> .	9],23 9],23	°°	≓113•. -113•	-0.	12960 <sub>=</sub>	0,	D.	0. 0.			
0.773		6.	-113.	-0.		0.		0.			
0.100	91.23	ů.	-113.		17568.	0.	0.	0.	0.	0.	
0.750	91.73		~!!!	0.	12867.					υ. 0.	
1.000	91.73 81.23	· 0.	-113. =113	-0. -0.	1286 <b>0.</b>	0.	0,	0.	· 0:	0.	
1.1:6	91,73		-113.	-6.	12864	0.	0.	0.			
1.750	91.13	٥.	-113.	0.	12060.	. 0.		u.	0.	0.	
2.000 2.500	91.73	ý.	-11).	-0.	17866.	0.	. 0.	Ģ.	ģ.	0. 0.	
3.000	91.73	0. 0.	-113. -113.	~0.	12865.			ů.	0.	0.	
3 . 500.	91.23	0			12869	0,_	ρ			ρ.	
4.366	21.23	¢.	-115.	-0.	12060.	. 0.		' 0.	0.	0.	
4.010	91.23 91.23	9• .	-113.	·-·· -0.	12868 17868.	0.			0.	0.	
1.000	91,23	n.	115.	-0.	12060.	. 0.	. 0.	ů.	J.		
5.150	91.23	n-	-113.	-0.	15944*	0.	٠.	0.	٥.	·	,
35 <u>0</u> .		0•:	113		12468. 12666.	0.	0.	·· 0•	0:	O•	
7.000	y13	٥.	-113. -113.	-0.	12868.	. 0.		U.	. 0.	0.	
8.000	. 91.23	u.	-113.	~0.	17868.	Ď.	ŏ.	o.	0.	0.	
7.221	91.23	٥.	-111.	-0.	12868.	0.	. 0.	Ů.	.· v.	Ų.	· '
#.300 9.000	91.73	0. 0.	-113. -113.	-0. -4.	12869.	0.	.0.	0.	0. 0.	, 0. 0.	•
9.590		٠ ن: ٠	::::::::::::::::::::::::::::::::::::		12060.	. 4, 0,		0.	0.	0.	·
10.090	913	0.	-113.	-J.	linan.	٠.	0.	U.	٥.	o.	
11.000	91.73	٥.	-111.	-0.	3286A.	٥.	0.	٥.	0.	0.	
12.000	91.23 91.23	n.	-111. -113.	-0. -0.	17868.	0.	. 0.	٠. ٠ ٥.	. 0.	u. Ga	

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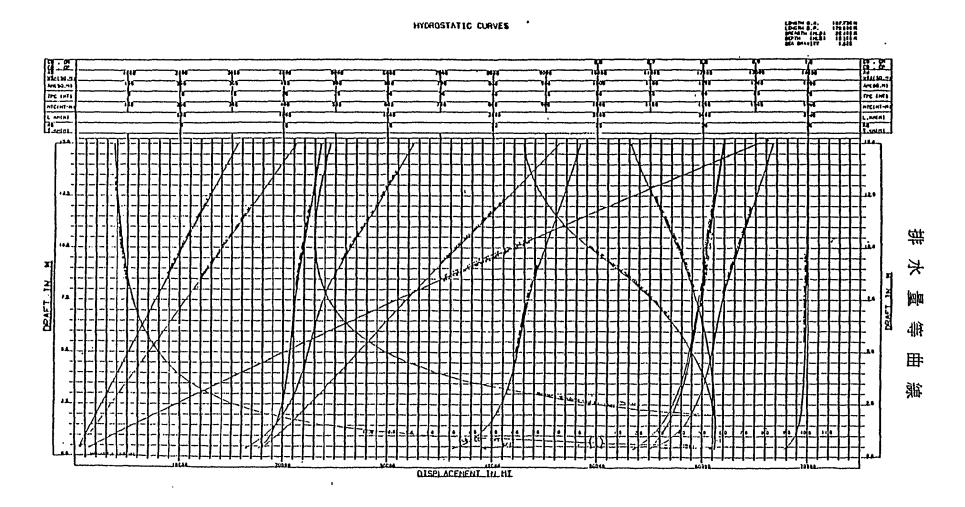
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		DAYA_	_€ื่อง ำเล้ō:	<u> </u>	CURVES	<u> + HETA</u>	1C SYSTE	н •							
ORAFT (FXY)	D15P1'	01 (°1	APPEN	D.COAR	HIC	TPC	κа	LCB	LCF	Y KH	L KH	1 C B	3 C F		
141	(HI)	(41)		(HT)	[H1-4]	(AT)	<u>(H)</u>	(4)	141	(4)	(4)	(H)	(n)		
0.073	456.78	150.72 400.31	54.70		330,52	31.95	0.025	-5.53[-	-5.524		27069.039		-0.000		
0.393	1310.21	1230.00	50.42	27.17 71.81	305,15	34.04	0.127	-5.551 -5.564	-5.502 -5.502	143.217	5073.617 4034.061	-0.000	-0.000		
0.523	1741,19 2654.14	1614.70 2543.33	61.60 64.00		312,40 327.45	35.49 36.52	0.344	-5.591-	5.601_ 5.634_	75.941		-0.000 -0.000	-0.000		
1.623		3512.60 5491.60		77.40	339.30'_ 359.14	37.24_ 30.25	0.513	-5.603	-5.624 -5.653	40.066	1643.057	-0.000 -0.000	-0.000		
3.0	7405.47 11356.11 15383.93	7312.17 11:17.29 15294.64	71.25	30.12	373.43 393.85	34.40	1.029	-5.62 <u>2</u> -5.576	-5.593		402.776	-0.000 -0.000	-0.000 -0.000 -0.000	<del></del>	
4.023 5.023 4.023	7056	23491.25		31.10 31.31 31.47_	407.7 <u>6</u> 418.34 427.72		2.059 2.572 3.046	-5.476 -5.327 -5.130	5.010_ _4.507 _3.86#	17.372 14.847 13.404	314.017	-0.00.0- -0.00.0-	-0.000 -0.000 -0.000	 4-4.	
	~~27745.91`	27545.87 31944.67	101.03	31.63 31.78		41.75. 42.24	3.599-	-4.002 -4.582	-3.001	12.489		-0.000 -0.000	-0.000 -0.000	##	1
-4.0: j 10.023	14.51	40471.83	117.07	31.94	473.57	43.57	4.631 5.151	-4.771	-0.403 0.373	11.647	231.181	-0.000	-0.000	· 2	
11.023		44315.47	124.78	32.25 32.40	521.58 545.45	44.30	5.674	-3.332 -2.855	2.333	11.514	704.643	-0.000 -0.000	-0.000	—	
17.021 14.023	58451.40	53013.00 50413.44	134.24	32.54 32.48	567.87 541.17	46.66	7.734	-2.390 -1.952	3,349	11.705	182-129	-0.000 -0.000	-0.000	<b>J</b> enju	Ì
15.0:3	- Kr. E CA	47017.44	147.73	31.74	813.05	46.90	7.402	-1.552	3,543	12.267	175.741	-0.000	-0.000	<b>b</b> mba	
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1,73	2012	533.57 533.00	43.56	432.49	-6.36 -0.34	-7.10 -7.04	16.96		
1.40	19184.	533.00	40.60	624.36	-0.33		16.72		*****
5.40	7972u.	\$24.06	49.62	625.72	-0.31	-4.92	16.54		·
^.	30254.	534.00	40.64	627.08	-0.28	-6.05	14.38		
7:10	J0727.	-333:00-	45.64	621.43	FF.24	-6:19	16.25		<del></del>
4.20 4.30	31224.	333.00	40.68	621.17 631.10	-A.23 -8.20		14.00		
4.4#	32344,	137.00	40.72	A32.42	-8.16	-6.66 -6.59	15.41		
A. 50	12937.	134.00	40.74	433.74	~9.15	-6.52	15.64		
<del></del>	-33469:	377.00	40.76	-435:94-	-6.12 1.15	- <u>h</u> -h-1.5	-12-21-		
4,50	34544.	137.00	40.79	637.61	-8.07;	-6.3A -6.31			•
4.40	· sanec ·	339.00	40.81	430.41		-6.23			· : :
7.00	75622.		·	440.25					•

RAFT	DISPT	TPC		A H	A W	K n	L KH	C B	C P	C H	СН	
FXT)	~~ (MLD) ``			••••		•	•				•	THE REPORT OF THE PROPERTY AND ADDRESS OF THE PROPERTY ADDRESS OF THE PROPERTY ADD
(4)	(51)	(41)	(H.02)	150.41	[H, DZ)		(H)					····
7.60	9297.	_ 50.12	5256.3	62.0	4009.4	1.62	11,30.3	0.7245	0.7424	0.7730	6.9752	:
2.50	11921.	~ 50.41 °	~~5473.8 '~	70 -1	4957.2	1.27	9 26 -8	0.7346	0.7494	0.7115	0.9802	** * * * * ** ** * * * * * * * * * * *
3.00	1437å <b>.</b>	51.37_	5685.9	94.2	5017.0_	1 .53 _	702.6	U.7432	0.7556	0.7901	6.9836	
3.50	16957.	51.84	5 994.2	110.3	~5u57.3	1.79	678.2	0.7505	0.7612	0.7972	0.4859	
4.00	19559.	52.23	6100.1	1 46 . 4	_5095.5_	2.05	5 97 • 0_	0.7568_	0.7662	U.8U33	0.9677	
4.50	22172.	52.57	6304.5	142.5	5120.6	2.30	5 36 . 9	0.7623	0.7707	0.0005	0.9091	•
1.00	74815•	. 52.07.	6507.9	158 • 6	5157.6	7.56		0.7671				
5.50	27459.	53.13	6710.9	174.7	5183.7	7.87		0.7715				
۸.00	30124.	_ 53.3A	6913.B	190 • 8	5207.6	3.07		0.7755				
5.50	32803.		7116.7	206.9	523v.u	3.33	382.4	0.7791	ŭ.7051	0.8745	0.9925	
[-]	35467.	53.83	732,	223.0	5251.4	3.59	357.7				0.9930	
7-50	PA 18/1.	54.05	7:42.7	239 • 1	. 5273•2	3 .04	3 36 . 5				6.9935	
0.00	40794.	64.70	7743.7	255 .2	5796.9	4.10	310.6				0.4939	
1.50	43615	54.54	7940.9	271.3	5322.T	4.36	303.4				0.9942	
00 . 50	46350.	54.86	8154.6	207.4	5351.0	4 -62	240.5				0. 9946	
9.50	47102.	55.19	<u></u>	303.5		<u>4 •90_</u>	2.79 • 7				0.9946	
0.00 0.50	51169. 54654.	55.52	8509.4	319-6	5416.6	5.13	269.9				0.9951	
1.60		55.86	. #.EOAB .	. 335.7	5449.6	5.39	561.5				6.9953	
1.50	60774.		9019.4	351.8	5483.7 5517.1	5.65 5.91	753.4	0.8051				
		JO • J J	,,,,,,,		331741 .		, 4 70 • 7,	_0.8078	0.0113	.0.0641	U. 445 I	
2.00	63110.	56.90	7451.N	384.0	5551.2	6.18	240.0	0.0105	0.8139	0.8751	U.9959	
2.50	65963.	57.25	9667.7	400.1	5585.0	6.44	234.1	0.0132	0.8164	0.0004	0.9961	
3.90	68834• <u></u>	_ 57.59 _	9003.3	. 416 -2	5610.8	6.70	220.7				0.9962	
3.50	71727.	57.94	10097.5	432.3	5657.6	6.96		0.8106				
4. w	74628•	. 58 • 29	_10311.4	. 448 • 4	5686.5	. 7.23		0.8213				
4.50	7755 0.	58.62	10526.2	464.5	5718.9	7.49		0.8240				
5.51 5.50	n)490.	58.95	13741.1	40.6	<u> </u>	7.75	210.4	0.8267				
	03445.	59.20	10954.0	496.7	5703.7	B.02	206.6	0.0293	V. 8319	0.9118	0.9968	
	86417.	59 • 62 <u></u>	11 168.9	512.8	_5616.+ <u>4</u>	0.25			0.8345	0.9169	.0.9969	
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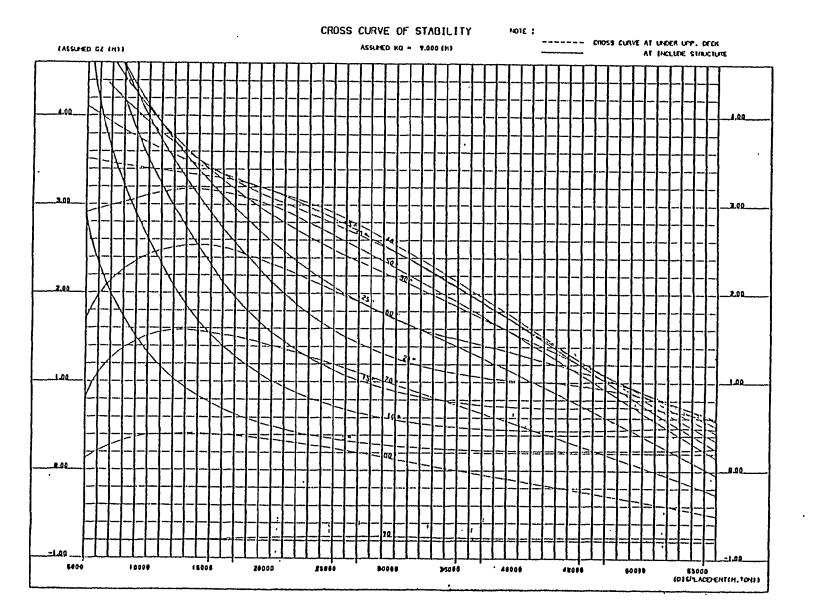
			UND'R UP	PPTA DECK					·	
ANGLE C	.0 DLG.	AYGLE3	.Q.DCG	ANGLE _ 10	).0 DEG	MGLE 1	.0 DEG	ANGLE 20	.0 DEG	
312E1	<u> </u>		6.7 (H)	015PT (T)	G • Z		- G 12 (H)	D1557	(H)	······
5740.90	0.0	1417.73	13.70 !	932.27	15.431	790.97	15.365	737.02	14.787	
11499.97	- n. n	5670.91	6. 150	3779.00	12.381	3163.86	13.224	2948.07	13.175	
17244.96	0.5	.7219.37	3.697	P395.43	7.331	7118.70	11.383	6633.16	11.562	<del> </del>
2 > 9 0 9 . • 5		18975.46	Z. 398	14916.30	6.241	12655.49	8.941	11792.26	9.949	
21744.96	0.0	25731.57	1.717	22646.48	3.904	10774.71	6.800	18425,43	0.337	
34490.96	0.0	32407.60	1.349	34444.47	2.912	28 3 37 . 13	4.788	26532.65	6.724	<del></del>
49.96	0.0	39243.75	1.129	30222.23	2,336	37160.54	3.658	36064.23	5.148	
45777.96	2.3	45799.66	6.972	45999.95	2.004	45999.95	3-057	45999.67	4.170	
51740.93	(·. C	57755.98	0.007	53777.72	1.811	54831.28	2.730	55935.47	3.684	
\$7490.00	0.0	57512.04	0.655	41555.37	1.703	63662.69	2.562	65610.37	3.374	
6 1249.94	0.6	86268.12	0.827	69333.12	1.653	72 340. 37	2.454	74361.62	3.010	
1 40003-64	4.3	73024.25	0.017	77078.00	1.639	80187.00	2.236	82119.50	2.635	<del></del>
741-9.95	0.7	79710.25	941.0	84264.74	1.528	87074.62	1.943	86584.17	2.222	. —
80499.87	3.0	86473.19	C.815	90558.44	1.300	92605.75	1.577	93574.81	1.814	
86747.81	Ů. Ú	92453.62	0.687	95463.44	0.983	96555.12	1.214	97091.44	1.465	<del></del>
98649.75	0.0	98649.79	0.325	98649.73	0.648	98649.75	0.964	90649.75	1.277	
		I_N	.E.L.O. W.	A N. G LE						
DISPT,		D1SP.T		. , DISPT.,		DIS PT_		DI SP.T		·
91789.	41	A1224	94	15111.	.75	A7350.	Δ2.	58515.	מר	
				0_H_FROH_A	•	<u></u>	<del>'</del>	····		INFLOW ANGLE 14.03 D
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INCLINANGLE .	\$ U.UA7	10 0.174	15 0.259	20 0.342	25 0.423	30 0.500	35 0.574	~				
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	-1:333-		3.398	3.985 3.860	4.122	4.003	3.928					*** ****** ** *
11000.	1 : 172	. 2 . 3 30	3,252	3.740	1.910	3.921	3.011			•		
11300. 12000.	1.034	2.093	3.121 2.496	3.625	3.622	3.844	3.757					•
12500.								<del>,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,</del>				******
13000.	0.929	1:959-		3.404 3.298	3.642 3.558	1.703 1.637	3.453					
13500-	0.884	1.761	2.636	1.194	3.474	3,572	3.557					
14000. 1450-i.	0.639	1.696	2.530 2.472	~ \$.00ñ~ 3.007	<u>"3.3</u> 93" 3.317	3.448	3.5104					
				.,,,,,,,		.,,,,,,	3,707	· · · · · · · · · · · · · · · · · · ·	······································			
15000.	<u>0.74</u> 2-	-1:533-	<del>2-221</del> -	2.90 <u>0</u> -	-}:?;}-	3,300-	-3:424	*	·			
14000.	0.695	1.401	2.133	2.733	3.097	3.275	3.341				·	
16400.	0.663	1.343	3.042	2.645	3.027	3.250	3.302					•
17 100 .	0.633	1.256	1.961	2.567	2.957	3.166	3.263					•
		-1-533-	} :40 }	2 - 400	3:093	3.114.	3.230	•	•			•
14000.	0.402	1.134	1.003	2.413	2.824	3.014	3.196	•				
19000.	0.537	1.042	-1:671-	2.766	7.700	- 2.972-	3.132			• • • • •		
19370.	0.114	1.050	1.613	2.195	2.641	2.936	3.102	<del></del>			<del></del>	
30000	0.446	1.000	1.554	2.124	2.542	2.002	3.074					
20500.	0.480	0.972 " 0.918	1.453	1.998	2.469	2.834	3.017	•				•
71500.	:	- 6.967 -	- 1:363 -	[ .970	-2.415	2.796	2.989	•		,		
>20n0.	0.431	0-074	1.359	1-474	2.361	2.715	2.961			· · · · · · · · · · · · · · · · · · ·		<del></del>
27500.	0.416	0.844	1.316	1.823	2.310	2.476	2.933					
23000.	0.402		1.273	~ i . 7 & j~~	* 2.261 "	2.636	2.405		•			
<u>73400.</u>	0.3/19 0.377	0.773	1.234	. 1.717 1.660	2.212	2.547 2.561	2.075	•	•			
24500.	በልየል	0.750	1.163	1.424	2.120	2.524	2.013					
24	0.244	0.744			2.074	> 400						
73500.	0.354 .	0.724 0.706	1.131	1.544	2.075	- 2.490 ·	2.702 ··· 2.751 ···		••		* *	***-
\$6700+	0.336	0.490	1.072	1,508	1.790	2.424	2.719					
24500.	0.328	0.472	1.046	1.473	1.931	2.347 2.361	2.656					
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271202	0.311	- 0.437 -	_ 0.997 0.474	1.376	1.871	2.310	2.625 2.543					
\$4500.	0.196	0.605	0.952	1.344	1.802	2.244	2.561					
zvnoo.	0.242	0.594.	~ d. 934 ~	1.318	1.746	2.237	2.524		·• · · · ·	• •	,	
29500.	0-542	0.514	0.915	1.292	1.734	2.204	2.495		<del></del>		<del></del>	
39499.	0.779	0.572	0.897	1.247	1.704	2.174	3.441			•		
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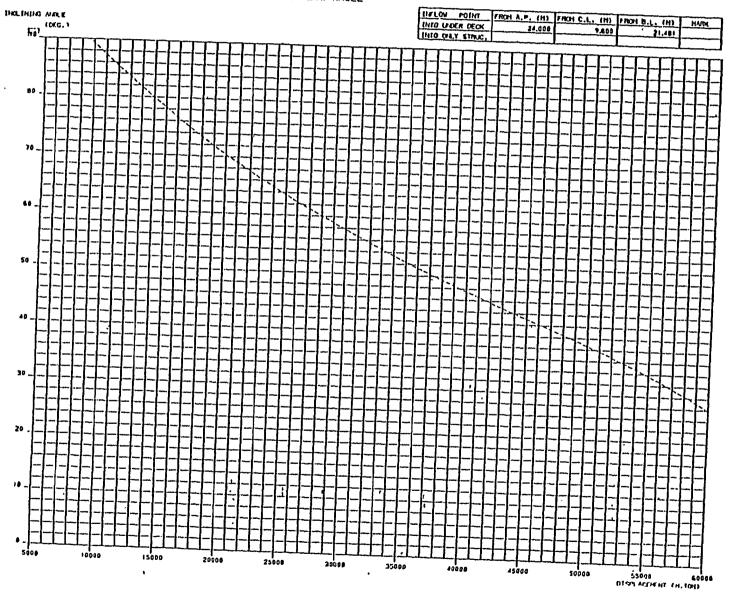
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		UND. UPP.	DECK			
INCL IN ANGLE	n	5		15	30	
	66177.50	66 170 - 45	66176.37	64078.11	64557.16	
INCL IN ANGLE	25	30	35	40	45	
D1SP • 1 (H1)	61569.74	57738.2,1	53199.02	47940.11	42194.07	
INCLIN.ANGLE	44	55.	60 .	45	70	
DISP!T , [HT]	00,46445	31999.05	27714 <u>.1.7</u>	23839.45	£43.40°4.20	
INCL IN. ANGLE	75	10	05	90		
OTSP-T (HT)	17007.75	14007.29	11228.92	8624.49		***************************************
					** FOR REFERENCE **	
INFL	OH ANGLE POINT	Y <u>*</u> 9,	.000 H FROM A.P	•	INFLOW ANGLE 43.89 DEG AT 015P*( 43465,42 MT	
•		2 = 21.	481 M FROM B.L	•		
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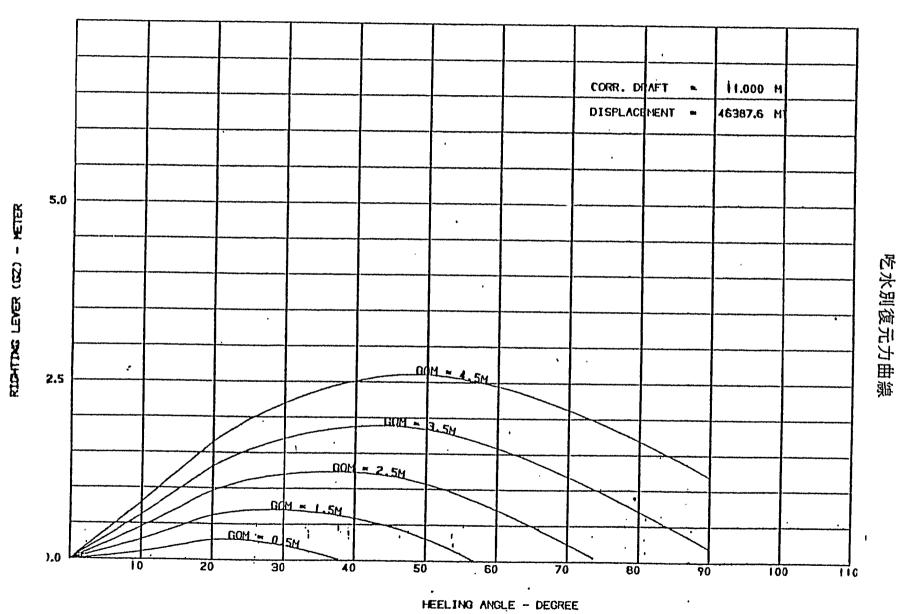
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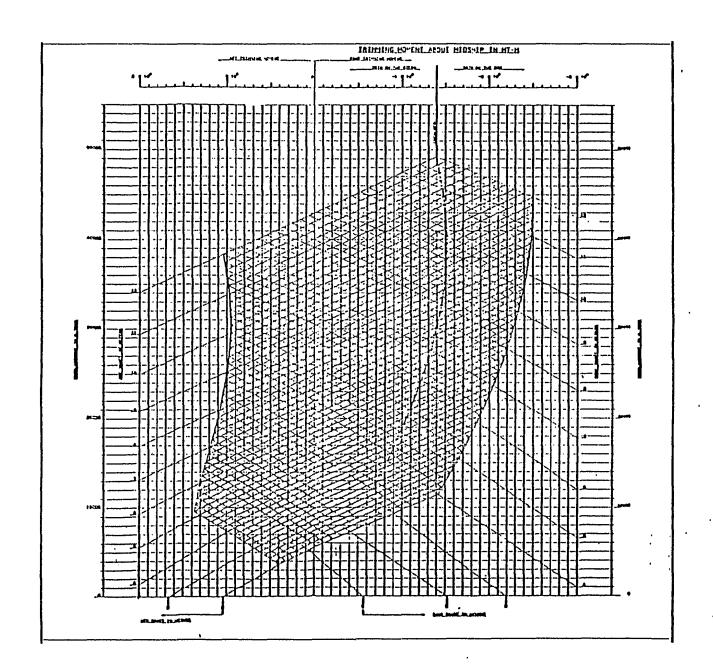
海水流入角

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NO. 1	DRAFT	10.000 H	TT DI ZU	ACEMENT .	195689.	7 HT				<b></b>	· · · · · · · · · · · · · · · · · · ·	<b></b>
INCLI	NING ANGLE"(I	DEG. ) 10	20 -	30	40	50 ····	60	70	80 ···	····· 90 ·····	山上。	r.两卷12
	8.000	GZ 1.473	G2 2.913	GZ 2.547	1.020	-1.669	-5.051	-8.665	-12.216	-15.432		
	10.000	1.620	3.597	3.547	2,305		-3.319	-6.786	-10.246	13.435	e tage of delet they a tage of pages and the desirence	
	12.000	2.167	4.281	4.547	3.591	1.395	-1.587	-4.906	-8.276	-11.438		
	14.000	2.514	4.965	5,547	4 - 6 76		0.145	3.027_	-6.307	-9.441		
	16.000	2.862	5 . 6 49	6,547	6.162_	.4.459_	1.877	-1.147	4.337			*
	18.000	3.209	4.333	7.547	7.448	5.991	3.609	0.732	-2.368	-5.443		
•.	20.000	3.556	7.017	8.547	_ 6.733_	7.523	5.341	2.611	-0.398	-3.445		,
76 1)	22.000	3.904	7.701	9.547	10.019	9.055	7.073	4.491	1.572	-1.444		
-1c	24.000	4,251	4.305	10.547	11.304	10.587	8.005	6.370	3.541	0.558		
46	26.000	4.598	9.049	11.547	12.590	12.110	10.537	8.249	5.511	2.561		
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HEAN DRAFT	-0.50	0.0	0.50	1.00	1.30	2.00	2.50	3.00					
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3,00	<u> </u>		-602	-1192 -	!???	-2340	-2900	-3440		·			
3.50	413	0	-601	-1193	-1172	-2143	-5306	~3459					
4.10	<u> </u>	<u> </u>	-54A	-1107	-1767	-2341	-5x61 -5.461	-3460 -3452				<del></del>	
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4.50	5 112		-575	-1144	liói_	-22.5	-2019	-3366					
7.00	174	0	-769	-1133	-1 497	-7746	-5 348	-3341		·			<del></del>
7.50	464	J	-564	-1172	-1876	-1218	-2773	-1111					
4.00	114.3		-550	-1111		-2204	-2743	-1270					****************
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9.00	549		-540	-1074	-1609	-2139	-366 <u>8</u> -	-3143					
10.00	544	<u> </u>	-437	-1044	-1564	-7137 -2077	-5271 -5754	-3144		<del></del>			
_10.50 _19.50 _	530	å		-1028	-1537	-2444	-2551	-3454					
-11.65 -	471	<u>-</u>		-iaiž	i s i ž	-2011	-2501	-3004		<del></del>			
11.50	911	0	-449	174	-1445	-1975	-2464	-2949					
17.00	560		-440	-975		=1436							
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- jzījō	424	<u>_</u>	<u>-441</u>	-476_			-3164	- 25 67					
15.00	441	0	-427 -413	-04A -041	-1265 -1224	-1661 -1671		-2414 -2414	,				
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"j4.00"	;i.i.	0	-269 -271	-571	-04 A	-[[5]	-1395	-1656		•			
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10.50	453	ő	-/41	-473	-701	-928	-1150	-1366					
71.00	: 36	Ü	-3:5	-443	-L55	-064	=1070	-1271					
21:30 -		<u>-</u>	- 209		- 40 8	-801	-490	-1174					
27.110	205 191	0	-175 -176	-366	-141	-71H	-91(-	-1675 -107					
33,50			=:	249_		-676		-101					<del></del>
73.00	177	<u> </u>	-163	-319	-469	-616	-760	-400					
71.50	144	<u>.</u>	-152	-205	-432	-567	-769	-1123			*******		
74.00-	<u>125</u>		=				-644	-751		······································	_,	·	
24.50 25.00	137	0	-112	-5-0	-340	-4117	-:48	-639					
75.13	[5]	<u>5</u>	-114	-220	-319	-412		-5A2		<del></del>			<del></del> ,-
:4.00	110	٥	-103	-179	-284	-377	-453	-577					
24.50	100	<u> </u>	-45	-[43-	-;65	-338	-411	-476					·
.27:00_		<u> </u>	-117	-146	-240	-300		-433		<del></del>		<u> </u>	·
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21)

<del></del>				FORE D	RAFY CO	RRECTIO	N TABLE	FOR DR	AFT HAR	K (UNI	T IN C.	H.)	····			PA	CE = 1
FT							FORE D	RAFT (H		•	· ·		1. 4.34				• •-
AFT				<del>.</del>													
3.00	4.00	5.00	6.00	7.00	00.0	9.00	10.00	11.00	12.00	13.00	14.00	15.00	16.00	17.00	10.00	19.00	20.00
.00 -0.7 5.00 -0.5	, 0,3			0.7	(1.9				*****	*****	*****	*****	****	*****	****	*****	*****
.00 -0.7	-0.7	1 2			<u>0:</u> 7					*****	*****	*****	*****	*****	****	*****	*****
.00 -0.9	~0.7				0.5					*****	*****	*****	*****	*****	****	*****	*****
.00 -1.2	-0.3		-0.5	-0.2	====		0.5			. * * * * * *	******	*****		*****	*****	44444	4 4 4 4 4 4
.00 -1.4	-1.2	-0.9	-0.7	-0.5	-0.2				0.7	0.9	1.7			*****	-		
. UJ -1.6	-1:4		-0.9				4.0			0.7	0.3	1.2					44444
<u> </u>	1.4		-1.2	-0.9	-0.7	-0.5	-0.2			0.5	0.1	<u></u>	1.2		44444	*****	*****
.00 -2.1	-1.11			-1.2	-0.9	-0.7	-u.5	-0.2		0.2	Ü.3	3.7		1,2	*****	*****	*****
<u>.00 -2.1</u>	-2.1			-1.4	-1.2		-0.7		-0.2	0.0	0.4	0.5	0.7	0.9	1.2	*****	*****
•00 -2•5	-2 -3	-2.1	-1.6		-1.4	-1.2	-0.9		-0.5	-0.2	0.1	0.2	0.5	U.7	0.9	1.2	04000
· )( +3 · 4 · 4	~, c	-7.3 -2.5	<u>-2.1</u>	-1 . B	-1.6	-1.4	-1 -2		-0.7	-0.5	-0.3	0.0	0.2	0.5	0.7	0.4	
.07 *****		-2.8	-2.3 -2.5	-2.1 2.3	-1.8 -2.1	-1 . 6 -1 . 8	-1.4	-1.4		-0.7	-0.3	u.2	U.U	u. 2	0.5	: 0.7	.0.
.00 *****					-2.3		-1 :6		-1.2	<u>-0.9.</u>	-0.1	-0.5 -0.7	-0.2	0.0	<u>u • ż</u>	0.5	0.7
.Ou +++: +				-2.8	-2.5		-2.1	~1.0	-1.6	-1.4	-0.4 -1.2	~0.9	-0.5	-0.2 -0.5	0.4	0.2	
00 444334					-2.8		-2:3		-1 -8 -	-1.6	<del>-1:</del>	-1.2	-0.9	-0.7	-0.2 -0.5	-0.2	<u> 0.2</u>
*****	*****	*****	*****	4++++			~2.5		-2.1	-1.8	-1 -1	-1.4	-1.2	-0.9	-0.7	-0.5	-0.2
		<del> </del>							<del></del> -					-0.7	-001	-043	-012
.00 +++++				-2.7	-2.4	-2.3	-2.0	-1.8	-1.6	-1.3	-1.1	-0.4	-0.7	-0.4	-0.2	4.0	6.2
.00 +++++					-2.7	-2.4	-2.2	-2.0	-1.8	-1.6	-1.7	-1.1	4:0.9	-0.7	-0.4	-0.2	<u> </u>
• <u>00 * 65 / 14</u>						-2.7	-2.4	~2.2	-2.0	-1.A	-1.4	-1.3	-1.1	-0.9	-0.7	-0.4	-0.2
.00 *****						*****	-2.7	-2.4	-2.2	-2.0	-1 ·B	-1.6	-1.3	-1.1	-0.9	-0.1	-0.4
·7/ +++++									-2.4	-2.2	-2.1	-1.8	-1.6	-1.3	-1.1	-0.9	-0.7
.00 *****	414444	*****	1-144	*****	*****	*****	*****	*****	-2.7	-2.4 -2.1	~ko k	-2. ú	-1.0	-1.6	-1.3	-1.1	-4.9
00 ***//*	7				*****					*****	-2.5	- <u>2.2</u>	-2.U -2.2	-1.8	-1.6	-1.3	-1.1
00 *****	*****	*****	*****	*****	****	*****	*****	* 40 + 44	****	*****	****	-2.7	-2.4	-2.0 -2.2	-1.8 -2.0	~1.6 ~1.6	-1.3
· <del>- · · · · · · · · · · · · · · · · · ·</del>														-242	-2.0	-1.0	-1.6
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			**	FOR OND.	DATA 🤚	•	••	••••	• • • •		•	•	
_	DAAFT DA	TÀ	<del> </del>	<del></del>									
_			(0-1		-21	(D-3)		11. (0-9					
	DRA		(H) 6.		500	0.0	0.0						
	rus	. IFROM A	P) (H) 0.0	, L/B	.000	0.0	0.1	)	'				
	DIST.	DRAFT	SEC.AREA	ÄŘĚA	GIRTH			DIST.	DRAFT "	SEC. AREA	AREA	GIRTH	<del></del>
	(FROH AP)		(NOTH S.)	DITAN	LENG	rH		IFROH AP I		IBOTH S.1	RATIO	LENGTH	
	(H)	[H]	(\$0,H)	(SA/AH)	(H)			[H]	TAI	· (30.4)	TRAVARI	TAT	
	4.131	6.570	0.0	0.0	0.0		31	167.176	7-318	1 25 -345	0 .50973	28.829	
	4.331	6.373	1.652	0.00672	14.11		32	169.340	9.354	106.643	0.43368	26.965	
	4.530	6.276	2.591	0.01054	14.13		33	171.505	9.391	46 . 738	0-35273	25. 03 9	
	5.513	6.593	7.273	0.02958	14.2		34	173.669	9.427	65 .728	0.26729	23.145	
	6.495	6.609	12.052	0.04901	14.4		35	175.835	9.464	45.608	0.18547	21.606	
	8.660	6.646	22.923	0.09322	15.2		36	176.859	9.401	37.132	0.15344	22.304	
_	12.990	- 6.687 6.719	34.383 .	0.13982	16.60		37 38	177.883 178.000	9-498	30.485	_ 0 •1 2397 . 0 •12077 .	21.626 21.458	
	15.155	6.755	46.387 58.625	0.10064	20.11		30 39	178.000	9.500	29.697 28.690	0.12077	21.204	
	[7.3[9	6.792	70.929	0.28844	21.8		~ 40 ···	179.100	9.519	22.753	0.09253	10.938	
	21.650	6.865	95.247	0.28733	25.29		41	180.200	7.537	17.245	0.07013	16. 191	
	25.981	6.938	110.306	0.48143	28.43		42	161.300	9.356	10.878	0.04424	12.640	
	37.310	7.011	139,675	0.56802			43	182.400	9.574	2.121	0.00063	5.399	
	34.841	7.044	158.500	0.64456	33.6		44	182.515	9.576	1.207	0.00441	4.430	
	43.300	7.230	187.569	0.76277	37.5		45	162.630	9.578	0.0	0.0	0.0	
	51.960	7.376	204.539	0.03178	40.40	)2							
	60.620	7.522	217.211	0.86298	42.17								
	69.279	7.660	216.573	0.88072	42.6	15				,			
_	<u>69093</u>	<u>0.000</u>	726.012	0.91911	43.31								
	106.800	8.300	234.532	0.95376	43.9						1		
	117.380	0.479	239.596	. 0.97435	44 . 21					*******			
	126.040	8.624	243.604	0.99065	44.43								
	134.700	8.770	245.904	1.00000	43.79			<u> </u>				····	
	143.359	8.916	240.737	0.97899	42-15								
	147.690		$-\frac{237.170}{218.406}$	0.88818	40.87			<del></del>				····	
	156.350	9.135	199.039	0.80942	34.77								
_	-160.686-	- 9.208-	173.735	0.70651	33.97						····		
	167.845	9.245	158.920	0.64627	32.37						•		
	165.009	9.201	142.770	0.38059	30.6			<del>-,</del>	*************	······································			
			2.20.00								•		
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DRAFT DAT		(H)(D~	500	D-21 9.500 8.000	0.0	0.0 0.0 0.0	(n-5) 0.0 0.0				
	(HT)	01SPT(HLD) (HT) 33072,	APPEN (HT) 132+	T P C (HT) 43,00_					H T C (HT-H) 463-47	7 C B (H)	† č F (H)
	(HT) "	(H. P2)	(20*H)	(150.4)							
	32.59 C B	6524.9	<u>204.5</u>	_41 95 +6	<u>4 • 16</u>						
	-	0.8862			ii						
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\$50.40	FR.HAME	FR.NO	FR.SRACE	DIST. (AP)	OEST. (MID)	
1	FH	-0.000	0.0	-4.800	71-400	
	-FG	-7:000	-0.600	-4.200	70.800	
<u></u>	- <u>FE</u>	-4.000 -5.000	0.600 	-3.600	90.200	
	FD	-4.000	0.600	-2.400	89.000	
<u>`</u>	-FC	=3.000		=1.000	65.400	
Ť	FB	-5.000	0.600	-1.200	87.500	•
<u>-</u>	- FA		~~ 0.670	0.656	*** 87.200 ****	
•	Få	<b>v.0</b>	0.600	0.0	66.600	•
10	FI	1.000	0.600	0.600		
11	FZ	3.000	0.600	1.200	85.400	
13-	F Z=1/2	\$ 200.	0.600	1:800	84.800	
13	F3	3.000	0.600	. 2.400	84.200	
14	-F4	4:000-	0:800	3.000	83.600	
17	F5-172	5. vv0	0.600	3.600	83.000	
17	F6	5.500 6.700	0.600	4.200 4.800	#1.700	
·	- <del></del>	7.000	0.600	- ··· 5.400 ···	··· 61 -200	
17	FR	8.000	0.600	6.000	NO-600	
Žớ ·	F8-172	6.550	- 3.670	··· 006.8	***** 80.000 ***	
ži	F9	7.000	0.400	7.100	79.400	,
	F 10	10.000	-0.800		78.600	
23	FIL	11.000	0.800	0.800	77.800	
	-F12	17.000	0.800	9.600	77.000	
25	F13	13.600	0.800	10.400	76.200	
<u> </u>	k 14	14.000	~ 0.800	~ 11.200	75.400	
27	F15	15. ໜ	<b>4.</b> 600	12-000	74.600	•
	?F16	16.000	0.800	12.800	73.800	
29	F17	17.000	0-800	13.600	73.000	·
30	.k18	18:000	0.600	14.400	72.200	
31	F19	17.:00	0.000	15.200	71-400	
31	F 21	20.000	0.600	16.800	70.600 69.800	•
34-	- # 22	21.000		17.600	69.000	
15	F23	23.000	0.800	18.400	68.200	
	- F24	24 :000 .	0.800	19.200 -	67.400 ·	айманандар ву на населения разправода у процествение на труго от отности в обществения и в серойний от населений на насел
37	P29	25.000	0.800	20.000	66.600	
38	·· ¥ 26	26.000	0.800	20.800	65.8CD	
39	F27	27. 000	0.800	21.600	63.000	
40	- 1 S A	20.000	0.400	22.400	64.200	<u></u>
43	PZ4	27.000	0.800	23.200	63.400	
47	. kas		0.000	24.000	62.600	
43	F31		0.800	24.000	. ' 61.860	
	F32	77.000	3:803	75.600	61.000	
45	F 33.	33.000	0.100	26.400	. 60.200	
46	- F74 F34	34.000°	0.800	27.200 20.000	57.400 57.600	
	-F35	35.000	-0:00	24.000	57,800	ومينسا البيان يتقد الأشياط البيان والمستوري المراوي والمراوية والمنافعة والمراوعة والمراوعة والمنافعة والم
49	F37	37.000	0.000	27.400	57.460	•
7.7	- F 39	39:000	9: 303	30.400	54.200	

### FR DIST TABLE

Frame Ho.	4.000 H W.L.	7.000 M	0.000 H W.L.	9,000 M	10-000 H		12-000 H		14.000 M	13-000 M	14-000 M	17.000
		W.L.		W.L.	W.L.	Vol.	HeL:	H.L.	W.L.	H-L.	<u> </u>	<u></u>
FH						1-084	2.394	3.454	4.330	9.033	3.444	5.444
76					•	1.469	2.716	3.773	4.45)	3.313	4.012	4.012
re					0.095	1.731	1.024	4.087	4.971	1.711	4.355	4.355
rę					0-442	2.053	3,337	4.311	5.246	4.048 "	- 6.473	4.473
0					750.0	2.374	3.447	4.708	5.376	6.367	7.024	7.024
C					1:110		3.954	5:014	3 -40 3	4-480	7.344	7-346
•					1.525	3.009	4.243	5.316	4.211	6.765	7.459	7.459
A					1.849		4.545	3.414	4.509	Y. 184	7.942	7.742
'0 ' <b>1</b>				0.445	2.210	3.442	4.864	5.710	4.401	7.575	4.255	0-255
1				0.819	2.547	3.454	5-141	4-177	7.609	7.041	W-541	0.541
2-172	·			<u> </u>	<del></del>	4.363-	\$ .455 \$ .746		7:273-	!:!:!_		8.020
• • • • • • • • • • • • • • • • • • • •				1.924	3.532			L.745			9-091	7:01
4		•••	0.458	2,286	- 3.051	4.472	- 4.035 .	7.315	7.925 -	0.745	- 4.355 4.412	9.355 9.412
·			0.834	2.448	4.167	5.444	4.310	7.502	0,471	7. 200	7.043	7.863
9-1/2	•	0.221	1.254	3.002	4.441	5.757	···· 4.971 ··	7.044		- 1.447 -	10.107	10.107
4	0.348	0.413	1.451	3.351	4.794	4.045	7-139	6-100	8.450	7.488	10.344	10.344
7	388.0	K:003	7.060	- 3.415-	5.108	4:374-	7.402	-4.353 -	190	4.421	-10:574-	10.579
•	4.764	1.34+	2.495	• 4.034	3.415	4.408	7.440	8.574	9.424	10.146	16.403	[8.03]
1=177	1.304	1.771 **	2-044	4.368	5.717	4.443	··· 7.914 ···	0.434 -		-14.346-	11.017	- Îl.uly
•	1.424	2.149	3.278	4.477	6.015	7.154	8.144	9.04 B	9,073	10.540	11.220	11-226
10	2.056	2.444	3.779 ```	" 3.177 -	4.403	" 7,508 <del>"</del>	- 0.459	9.372	10.150	10,033 ***	- 11.493	·· 11.4+3
11	2.415	3,135	4.214	5.344	4.740	7.652	0.005	1.444	10.430	11.115	11.746	11.746
17	7:413	7:417	4:19	3.743	7.147	4:146	7:113	9, 44 5	10:441			<u> </u>
13		4.387 _	_ 5.151	4.345	7.304	4.310	7-411 _	10.217	10.942	11.404	12-209	11-201
15		" 4.146 <sup>-</sup>	3.597 "	4.797 "	7.849		7.418	10.400	`` 11.145 `**		12.415	12-415
ii.——	4.101 4.664	4.993 5.430		7.137 7.507	- 4.513	7.130	_ 9.976 _	10.734	11.410 11.43 -	. 12.044	12.404	11.404
17	5.)22	5.651	4.047	7.564	6.035	**************************************	10.503	11.213	11.063	12.246 -	12.763 12.945	12.703
i j j		-6:269-	7.239	6:219		—•;;•• <u>;</u> -	10.753	-11.436-	i z . 676-	—i:::ii—	-15:675-	-13.0+3
19	5.875	6.672	7.414	0.359	9.449	10.260	18.772	11.452	12.242	12.777	13.232	13.232
10	4.151	7.065	7.941	. 1.117 -	- 4.740	10.510	11.222	11.454	11.440	12.733	13.350	13.35
11	6.422	7.447	4.333	4.205	10.011	10.767	11.441	12.030	12.402	13.076	13.472	13.472
11	7.0.3	7.010	A.673	- 1.511 -	- 10-291 .	" 11.005 "	11-450		12.752	13.205	13.572	13.572
113	7. 114	0.170	4.0(2	7.875	10.331	11.233	11.647	12.404	12-4+4	13.323	13.662	13.461
14	7:743	0. 32 5	4:320	TO:OAE	10:400	11:450	-ii:03#-	-12.558	-13:311-	~(3.433~	-13.744-	13.744
15	8.100	1.042	7.674	10.35#	11.034	11,437	12-217	12.72#	13.157	13.554	13.019	13.611
15	1.440	· 9.387	- 9. ¥20 -		-11.265 "	·· 11.854 -	15.302	12.841	" 13.275 "	13.427	13.004	13.044
117	0.700	_ +.501	10.203	10.469	11.402	12.040	12.543	12.778	13.304	13.711	13.544	13.744
111	9.114	~ 4. NOA ~	10.473	~ 11. los ~	11-646	" 12-214 <b>"</b>	12.440	. 13-113	13.479	13.74 L	13.944	11.794
116	<u>*-429</u>	17.094	10.714	11.332	11.002	12.300	12.427	13.226	11.540	13-044	_ 14-037_	14.031
10	9.753		_13.484	71:94	72.346	17.536	12.960	13.33 <i>X</i>	<u> </u>	13.902	-14.075	14.075
21		9 4 4 9	_11.222	_11.727	1 2 • 24 2		— 13° δ63	12.420	12.724		j y • j n g	la_lui
37	10.317	16.467	11.457	11.954	12.413	12,020	13-199	13.528	13.793	14.001	14.135	14-135
33		_{\!*!?}	112401	_13:141	12.570	12.724	13.304	13.404		14.034	<u></u>	
74		11.391	11.074	12.317	12.710	13.079	13.401	73.27	15.702	14.076	14.157	14.167
35	!!-!04	_!!-!!4	_12.073	12.405	_12.878_	_!!!!#_	_13.492_	_13.747	_13.949	_ 14.027_	_14-17#	_ 14-174
37	11.700		12.240	-12.644 12.795		-13.331-	-13.577 13.655	~~13.010'~ 13.048	13. 691	14.122	-14.168 '-	14.100
14		-11:31	-{1:::::	-{{{1}}	-{{::}}}	13.402	-13:937-	5.426	-14.023	12:128	14:175	14 - 1 15 14 - 200

1.617				<del></del>	(H)	. (M)	(H)	182. HI	(IRV.)	(LONG.)		
				**************************************								
K +1717 J	0.0 89.70	0.0 P0.51	1.017	0.0 -10.600	-10.600 -10.600	0.0	0.0 0.0	481.3 409.0	14/05.	25169. 25593.	0	
3.000		596.99	2.418	-iö.300	-10.600	0.0	0.0	531.0	19861.		- <u>č</u>	
3.343	783.66	791.30	2.596	-10.600	-10.600	ö.č	0.0	565.7	22442.	32347.	ĭ	
4.000	1166.47	1162.97	2.950	-10.600	-10.600	0.0	6.0	547.5	26076.	33464.	Ü	
4.070	1702.88	1697 -77	3.419	-10.600	-10.600	0.0	0.0	642.1	31656.	37492.	1	
4.897	1715.73	1710.50	5.430	-10.600	-10.200	0.0	0.0	662.5	327:0.	40832.	3	
_5,009	1703.07	1703.51	3.472	10.504_	<u>-10.300</u>	0.0	0.0	667.5	33501.	41143.	0	
5.000	2479.21	2471.17	4.053	-10.477	-10.200	0.0	0.0	713.1	40353.	43979.	0	
7.000	3,15.17	3205.52	4.014	-10.413	-10.700	<u>0.0</u>	<u>0.p</u>	750.8	49209.	46709.		
7.300	3444.85	3434.51	4.703	-10.399	-10.200	0.0	0.0	772.5	51421	47635.	1	
9.000	4754.05	3973.63 4743.76	5.327	$\frac{-10.372}{-10.344}$	-10.200 -10.200	0.0	0.0	712.5 112.5	51921.	4 16 35 .	<del>- 8</del>	<del></del> .
10.000	5530.54	5513.74	6.275	-10.324	-10.200	0.0	0.0	772.5	51721.	47635.	Ö	,
10.300	5742.211	5744.97	6.379	<del>10.319</del> -	~12.200	0.0	0.0	772.5	- śióżi	4 7635.	<u>i</u>	
11.000	6.40.04	6:61.29	6.744	-10.309	-10.200	0.0	0.0	707.0	39000.	43594	Õ.	'
1:.000	6440.35	6919.52	7.200	-10.299	-10.200	0.0	0.0	613.5	£600y.	37622.	0	
13.000	7507.04	7484.55	7.599	-10.29l	-10.200	0.0	0.0	520.0	15836.	32054.	0	
13.973	7973.04	7747.11	7.442	-10.206	-10.200	0.0	0.0	420.0	8834.	26364.	1	
14.271	P074.97	8072,-56	0.037	10.291	10.600	0.0	0.0	307.5	6957	37505.	!	
14.703	8314.52	1204.55	n. 209	-10.299	-10.900	0.0	0.0	332.6	4401	19310.	1	_1
<u> 15-000                                 </u>	H 144.30	03:4.23	0.236	-10.300 -10.300	-10.600	0.0	0.0	332.6	4401.	19319.	Y	t
15.695	65.00.63	8554.07	0.4i8 n.451	-10.300	-10.600 -10.200	0.0	0.0 0.0	`221•B	2934	5725.	•	
15.783 16.700	#805.79 #655.11	0581.14 7629.12	B.497	-10.308	-10.200	0.0	0.0	221.0	2434.	5725.	0	
17.000	8876.07	6650-51	0.692	-10.305	-10.200	0.0	0.0	221.0	2934.	5725.	Õ	Д
17. 370	1947.04	3920.46	0.759	-10.305	-10.200	0.0	0.0	221.8	2934.	5725.	1	
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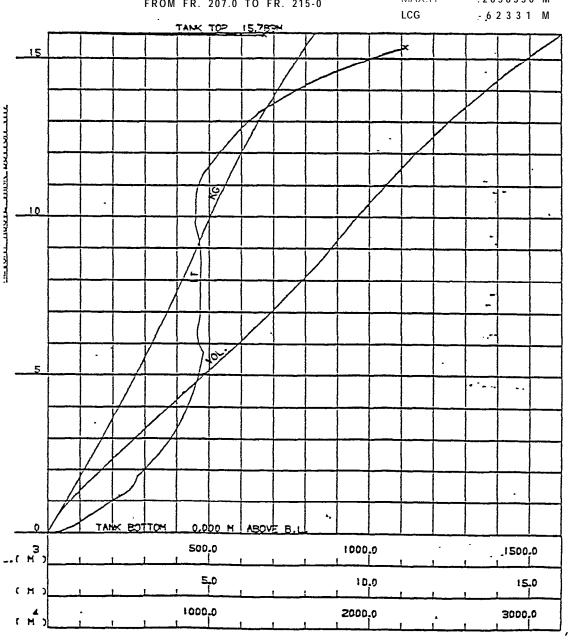
NO	TANKZIDED NAME PL	AC E	AFT. BIID.	FWD. BHD.	VOLUMF	L C G	K G	T.INERTIA	T C G	
			FRND DIFF.	FHIO DIFF.	( CUR. H)	(H)	(H)	(H4)	(H)	
	FRESH WATER TANK (P)		11		106.92	<u> </u>	16.995	150.		
	FRECH HATTR TANK (S)	4	y	17	176.30	07.879	17.004	183. 32.		
	DRINK WATER TANK (P)				27.54	90.407 -31.300	17.035 2.156	5488		
	NO.3 F.O.T. (P/S)	2	147	174	903.86	31.903	2.106	4616.		
	40.5 F. Q. T. (P(S)	_ <u>-</u> 2	<u> </u>	93	<u> </u>	68.968		240.		
	NO.7 F.C.T. (P)	6	26	40	104.90	72.038	1.075	249		
	NO. 6 F. U. T. (P/S)		<u></u>	<del></del>		5 2 . 90 0		22.		
	NO.1 T.S.T. [P/S]	2	202	227	346.45	-73.535	16.972	621.		
	NO.: 1.5.1. (P/S)	<del>-</del>	- <u>iřš</u>	202	592.65	-54.641	16.607	1;19.		
	NO. 3 T. S. T. (P/S)	2	121	175	1191.93	-22.300	16.595	2431.		<b></b>
. z	NO.4 1.5.1. (F/5)	- <u>-</u> -	67	12.1	1191.00	20.699	16.595	24310		
	NO.5 T.S.T. [P/S]	Ž	49	67	581.34	53.150	16.654	1232.	rigg plant com a company or merce about the section of	
	47.1 9.5. t. (PZS)	<u> </u>	701		565.21	-74. 28 Y	1.802	3173.		
15	NO.2 W.N.T. (P/S)	2	174	201	872.114	-53.659	2.165	5189.		
11	N7.4 W.P.T. (P)	3	93	147	1067.69	0.100	2.156	10974.	•	•
17_	NO.4 H. 1. T. [5]	_4_	93	147	1007.69	0.100	2.156	16974.		• • • •
	NG.6 W. W. T. (P/S)	Z.	40	06	547.63	53.463	2.782	1100 <b>.</b> 5594 <b>.</b>		
	<u> </u>	_ļ_	_??4,,	. 232	2963.91	-90.252	9.506 12.740	2254		
	A.P. T.	ij	-5 -0.308	9	270.53 7067.02	94.055 -74.991	10.613	36162.		
	NULZ CARGO HOLD	~-ţ	20i 17:	227	97237	-54.775	10.157	59970.		
	NO.3 CARGO HOLD	- 1	140	175	9 861 . 69	-33.273	10.124	66695.		
	NO.4 CAPGO HOLD	-:-	-171	- 148 ······	9441.45	-11.637		60795	gar i gigan a' sanagallar di juga una di julius yan kanan mengalaban kentan pada i menjalah di selahan di sela	•
	NO.5 CARGO HOLD	i	94	121	9 967 . 70	9.900	10.141	60095.		
	NO.6 CARGO HOLD	∽ i	67	94	9841.78	31.770	10.172	60095.		• . •
	NO.7 CARGO HOLD	i	40	67	9 106 . 09	52.760	10.569	56087.		
	DILCE TANK		9	3 0	26.46	87.681	1.101	6.		
31	SEP. HILGE HIL TANK	4	10	26	17.42	79.733	1.266	9.		
	F. O. OVER FLOW TANK	i	36	40	30.53	65.639	01990	25.		
32	LUT. OIL SUMP TANK	1	21	20	24.31	_ 78.050 _	1.695		نجه جدوره و موملت فيستح ليميني بيان يا و يا يا يا يا يا يا يا يا يا يا يا يا يا	
34	CHOLING WATER TANK	1	, , , , , , , , , , , , , , , , , , ,	9	24.04	92.111	6.443	21.		
	NO.3 C.H. IIN DECKI	_1_	1411	175	11331.96	-33.100	10.765	60095.		
	NO.3 C.H. (CHECK)	į	140	175	8 354. LG	-33-100	9.600	14400.		
47	FOR W.L. CHECK	1	100	110	2519.79	12-100	11.294	5333.		

#### TANK CAPACITY CURVES

FORE PEAK TANK

FROM FR. 207.0 TO FR. 215-0

MAX. Va. -1596.000 M  $^{3}$ M A X . K G 8-260 M MAX.IT .2658330 M′



UNIT SIGN ( 0) 0   HETRIC 1   FEET  CAL_SIGN ( 0) 0   INNAGE 1   ULLAGE  ITH DATA NO. ( 3)  INTERTAL DEPTH ( 1.0000)  O/P UNIT	(P') SIBE DAYA CIENCE INTERVAL
TYPH DATA NO.   3    INTERTAL DEPTH   1.0000)   O/P UNIT	(P) SIBE DAYA CIENCE INTERVAL X Y Z
TNTERTAL DEPTH   1.0000)	(P) SIBE DAYA CIENCE INTERVAL X Y Z
1   CUP. 4   1.00000   1.00000   1.00000   1.00000   1.00000   1.00000   1.00000   1.00000   1.00000   1.00000   1.00000   1.00000   1.000000   1.000000   1.0000000   1.00000000   1.0000000000	(P) SIBE DAYA CIENCE INTERVAL X Y Z
1   CUP.4   1.00000   5   1   H.70N   0.25000   5   1   H.70N   0.25000   6.28781   1   U.S77L   6.28781   6.28781	(P) SIBE DAYA CIENCE INTERVAL X Y Z
## TANK NAME DATA **  ** TANK NAME DATA **  ** SMINDING TUBE DATA **    CHENSE INTERVAL    CO	(P) SIBE DAYA CIENCE INTERVAL X Y Z
** TANK NAME DATA **  NO.: FUEL DIL YANK (P75)  ** SOUNDING TUDE DATA **  IST SIDE DATA CHEWSE INTERVAL  X Y Z 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	(P) SIBE DAYA CIENCE INTERVAL X Y Z
** TANK NAME DATA **  NO.: FUEL DIL YANK (P75)  ** SOUNDING TUDE DATA **  IST SIDE DATA CHEWSE INTERVAL  X Y Z 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	(P) SIBE DAYA CIGNOE INTERVAL X Y Z
** SOUNDING TUBE DATA **    SOUNDING TUBE DATA	(P) SIBE DAYA CIGNOE INTERVAL X Y Z
** SOUNDING TURE DATA	X Y Z
X	X Y Z
X Y Z 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	X Y Z
0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	
0.0 0.0 0.0 0.0 0.0	1.456 '4.400 0.014 0.0 0.0 1.450 4.400 15.759 0.0 0.0 0.0 0.0 0.0 0.0 0.0
	0.0 0.0 0.0 0.0
v.v 0.0	0.0 0.0 0.0 0.0
0.0 0.0 0.0	v.ù 6.0 U.U
0.0 0.0	
0.0 0.0 0.0 0.0 0.0 0.0	0.0 0.0 0.0
∪.∪ ∪.∪ 0.0 ∩.ე ე.∩ ∩.∪	0.0 0.0 C.U 0.0 0.0
0.0 C.0 C.0 2.0 0.0 0.0	0.0 0.0 0.0
nan nan han hai i '	0.0 0.0 0.0
1.01 1.00 1.00 1.00 1.00 1.00 1.00 1.00	0.0 0.0 0.0

タンク容韻歌

(4)	OUN.H	4.1011	U-5-85E		CUR. H	H. 10H	U.3.66C	<del></del>		TARK HOLT	TET	
0.0	4.1	3.9	27.6	0.50	158.1	100					.,,	
7.01	7,0	4.6		V. 51	161.7			•		4.00	3.17	
0.07	7.9	7.4		0.58	144.5			•		2.71	3.17	
7,03				0.53	167.6	159.1				2.94 2.94	3.10 3.10	
0.04	11.5	15.0		0.44	170.5		1074.4				- <del>3</del> :{ -	
2.76	19.9 11.9	17.9 20.7	114.2	0.55	174-0					2.78	3. 18	
0.07	24.0	23.5	195.8	0.56 0.57	177.2					2.78	J. 1 B	
0.01	27,8	24.4	174.7	0.58	100.4 100.6			•		3.00	3. 19	
0.07	10.0	77.7	197.6	0.37	106.0	177.4				3.00 3.01	3.20	
.17	33.8	37.1	212.6	1.60	19.0	180.5		***************************************				
0.11	34.4	31.0	131 . 6	0.61	193.2	143.5				3.01	3.20	
0.12	30.4	37.9	250.6	0.05	174.4	106.5				3.01	3.23	
7.13	42.9 45.9	47.7	267.7	0.67	177.6	109.6	1255.2			3.03	3.20 3.21	
7.15	47,0	<u>+</u> }•\$				]"[ •4				3.04	3.21	
7.14	1 . 1	40.4	127.1	0.66	7 05 . T	195.7	1.795.5			3. øś	-3.31	
U.17	57.7	5 . 3	346.3	0.47	217.4	.01.0				3-05	3.71	
7.19	55.1	55.7		0.40	215.6	204.1				3.06 3.06	3.21 3.21	
0.19	61.7	79.1	314.4	0.69	214.0	207.9	1176,2	•		3.06	3.21	
7.10	54.7	41,0	404.2	0.70	777.0		137674		<u>`</u>	3.07	<u> </u>	
1.21	67.3	44.0	427,4		225.7	214.0	1416.5			3.00	3.21	
1.21	70.4 77.5	86. Y 87. 4	442.9 487.3	7-72	250.4	:17.0				3.65	3.71	
0.74	74.6	77.4	401.7	0.73 0.74	231.6 234.0	770.J 223.1				3.00	3.21	
2+29	77.7	75.7	501.7	0.75	239.0	2276.1	1477.1 1477.3			3.09	3. 21	
7.5	77.6	1:.5	570.7	0.14	241.1	7 9 9 5	1517.5	<del></del>	~~~~~		-3: {{}	<del></del>
1.27	77.9 77.0	71.A	14	0.77	744.9	217.1				3-10	3.22	
0.27	77.1	67.5	559.A 579.2	U.7A 0.79	747.7 250.9	235.3				3.10	3. 22	
	•		.,,,,,	0.,,	2.111.4	234.4	1578.2		•	3.10	3.22	•
0.30	91.7	97.4	179.0			741.4	1120.5			1.12	3.22	
1.11	1"1.4	44.4	639.0	0.11	78: .6	747.6	1610.7			7. i ž	3. 2 ž	
v. 11	104.6	99, 3	657.6	0.73	263.9	:50.4	1639.N. 1659.3			3-12	3.55	
7.14	107.7	102.3	677.3	0.74	267.0	253.1	1679.5			3.13 3.13	3.27 3.22	
7,31	110.8	101.3	677.0	0.85	270.2	246.7	1679.0	, ,		3.13	3. 23	
-H. 17	177	111.2	716.6			<del></del>	!???•!			3.13	3.23	
P . 34	1'0.7	114.7	756.2	0.89	1279.9	163.9	1740.4 1740.7			3.14	- 3. 2 <b>3</b>	
7.30	121.4	117.7	775.9	4.09	203.2	769.0	1701.0	•		3.14 . 3.14	3.23	
0.47	174.5	120.2	793.7	0.90	286.4	272.1	1001.3				•	re e
7-41	177,4			0.71	207.6	777.	1071.6	•		3.14	3. Z3	•
0.43	177.5	17/10.7	115.4	1.47		771.7	1041.7	<del></del>	<del></del>	<sup>3</sup> •14 3•16	_3· 2 3.····	
7.44	130.0	1. v. t 177. 2	975.0	9.91	294.1	241.3	1062.2			3.16	3.23	
0.41	147.3	174.5	712.U 894.9	0.94	197.3 107.5	284.7	1982.5	•		J. 16 .	.3.23	
0.46	144.4	137.2	914.6	0,96	377	1 100.5	1902.6		•	3.16	3. 23	
P+47	1400	[있당-	774 -7	0,77	.0110	291.1	1743.4	t'	. '	3. 16 3. 16	3. 23	•
0.44	151.8	144.7	074.3	, 64	117.2	277.5	7945.7		······································		-3. ? 3	
	* >	- 17/17	974.4	9.44	315.4	. 44. 4	1474.0			3, 17	3.73	

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	TA	HYHEEL CH	ECK TANK											
		• •	•				•	•						
	INDICATION OF	CALCULAT	ION AND	PRINT OUT	<del></del>				······································		<del></del>			
	PA INT FO			9 (USING U	NDER DAT	A)						¥		
	•	IN DATA	-1.		1.000	1.500	<b>3</b> .000	** * /	P 86 - PM- 1	*******				
	HE	EL DATA	-2.	0JU -1.500 500 1.000	-1.000	-0.500	. 0.0		<del></del>		. ,	-		
٠.	DR A	AFF DATA	1.	00 1.500 100 4.000	2.000	2.500	3.000 2.200			•				•
			۵.	000 7.644 000 12.644	8.500	9.000	10.000	~~~~			• •	•		
	AHENDHE	IT VALUE	(	)	1001	CATION OF	5CALE		•		•			•
	INDICATI	ON OF TUBE	E	,	SOUN	DING								
	TANK ARE	ANGE		l	P/S	DR C					•			
•	EXCHANGE	AATE	` u.s						•	•		,		•
		•••	•••	USE DATA	****								•	
	X-C(	100	Y-C(	000	2~CO	70 <b>P</b>						•		
		1.040	0^0	4.000	0.0	0 . U					<del> </del>			
	J. J J. J	1.000 1.000	0.0 0.0	4.000 4.000	0.0 0.0	3.000 13.000				•				
	0.0	0.0	0.0	0.0	0.0	0.0							•	
	0.0	0.0	0.0	0.0	0.0	0.0						•		
	0.0 	0.0 V.V	0.0 k.i	0.0 	0.0 V.Q	0.0 								
	0.0	0.0	0.0	0.0	0.0	0.0	·····			.,				
	0.0	0-0	0.0	0.0	0.0	ů. v								
	0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0								
• •	0.0	0.0	0.0	0.0	0.0	0.0								
·		! <u></u>		4.4	0	_9•Ý								
	y. y	<b>0.</b> 0	0.0	0.0	0.0	0.0			•					
••	0.0 0.0	0.0 0.0	0.0 0.0	0.0 0.0	0.0	0.0			•	•	ži v	***		٠.
	0.0	0.0	0.0	0.0	0.0	0.0						,		
	0.0	0.0	0.0	0.0	0.0	0.0			•				•	
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SOUND.	DY BOH		**	(H)	AY	STERN_											
(H)	-1.00	0.0	1.00	1.50	2.00	2.50	3.00	3,50	4.00				······································	<del></del>			
	•						~	_	_								
1.000	2.	0.	-2.	~3• ~3•	-4.	-6 a	-7•	_ ~!•	<b>-9.</b>						•		
2.000	2• . 2•	U.	···· ~2•.	~3.	<del>_4</del> 。		7•. -7•	<b>-1</b>	74 -9.								
2.500	2.	Q.	-2.	-3.	-4	-6.		-0.									
3.000	2.	0.	-2.	-3.	-4.	-6.	-7.		-9.					<del></del>			
3.500	ž.		-2.	-3.	-4	-6.	-7.	~8.	-9								
4.000	ž.		-2.	-3.	-4.	-6.	-7	~0•	-9							•• • • • • •	
5.500	1.		-2.	-3.	-4.	-6.	-7.	-8.		******							
5.000	-1 -	U.	-z.	-4.	-5.	-6.	-7.	-8.	-ý•	******	•		•				
5.500	-50	-50		-50	-50	-50		-50				•					
ba000		·~100 <b>.</b>	-100.	-100.	100.	_~100.	-100.	100.	100								
7.000	-200		-200.	-200 -	-200 •	-200.	-200 •	~200•	-200 •								
0.000		-300.	-3 vú.		-306.	<b>-3</b> 00.	. −30 <b>.</b>	-3 Ü U	-300								
9.000	-400.		-400.	~400.	-400.	-400.	~4UO•	-400.	-40u.				•				
10.000_		200 a_	500	500	500_ح_		200	500a_	ــه ۱۷۹ جـــ								
11.000 12.000_	~600•	-600.	-600+	-600•	-600.	-600.	-600 •	-600.	-600 • -700 •								
																	•
	COURT	CTION	TTITE	DUF TO	THE FI	711117 7					· · · · · · ·					·····	
	CORRE	CTION	TAULE	DUE TO	HEEL	TUNIT I	CH)				,		,,				
		· <del>······</del>	116	DUE TO		TUNIT		·				· · · · · · · · · · · · · · · · · · ·		····			
_soung.		P(	IBT.	EL (DEG	•1		STARE				· · · · · · · · · · · · · · · · · · ·	· · · · · · · · · · · · · · · · · · ·					
SOUND		P(	116	EL (DEG		0.50		·	2.00					***************************************			
(H)	-2.00	-1.50	IBT	-0.50	0.0	0.50	\$ TARE	1.50	2.00					***************************************			
(H) 1.000	-2.00	P( -1.50	IBT	-0.50	0.0	0,50	\$TARE 1.00	1.50	2.00		<u></u>		-	***************************************			
(H) 1.000 1.500	-2.00 14. 14.	-1.50 10.	7.	-0.50	0.0 0.0 0.	0.50 -3.	\$TAR6 1.00	1.50 -10.	2.00 '-14. -14.		<u></u>		-	***************************************			
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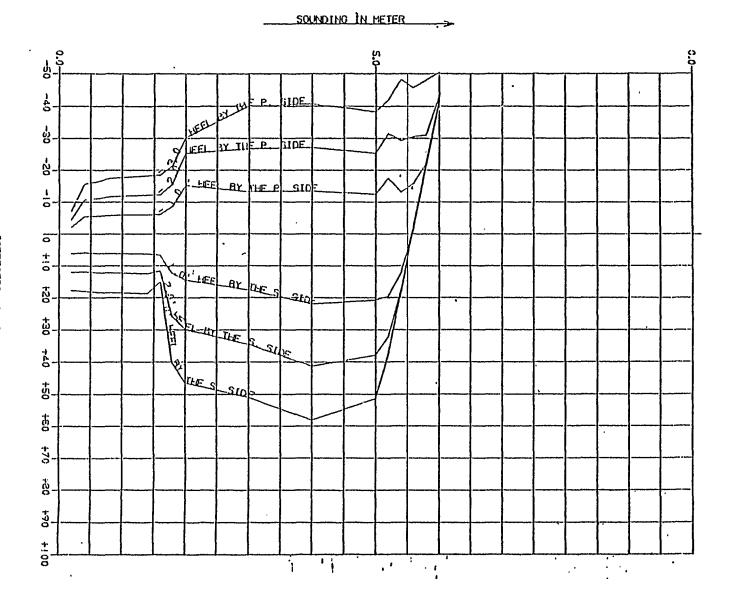
		IEEL UII	ECK TAN	· · · · · · · · · · · · · · · · · · ·	· · · · · · · · · · · · · · · · · · ·			· · · · · · · · · · · · · · · · · · ·			<del></del>				<del></del>	· · · · · · · · · · · · · · · · · · ·
	CORRE	CTION	TABLE	DUE TO	HIRT C	(UNIT	CH)						, <u></u>		···	···-
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3.000	2.			-3.	-4 <u>-</u>	-6.	-7.		-9.					<del></del>		<del></del>
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b_000	-100 -	-100.	~100.	~100.	-100.	~100.	<b>-1</b> nn -	÷100.	-100 -					A		
7.000	-200		~200~	-200 -	-200 -	-200-	-200 <u>-</u>	~700~	-2 no -							
U.000	-300.	-300.	-3 vů.	-360.	-304.	-300.	-306.	-360.	-300.							•
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11.000	~600∙	-600.	-600•	-600.	-600.	-600.	-600.	-600.	-600.							
12m0QQ_	790	_=7.90 •_	700		700,+_	<u>-700,</u>	<u>-700.</u>	<u>-700,</u>	-700 ·							
						•										
·	CORRE	CTION	TAULE	DUE TO	HEEL	TUNIT I	CH)				,					
	CORRE	CTION				TUNIT I	CH)				· ;-····			***************************************		
SOUND	CORRE	· <del>····································</del>	116	DUE TO		TUNIT I				••••	· · · · · · · · · · · · · · · · · · ·			**************************************	, again an Aorta. Philippin a ( a. )	
soung,	CORRE -2.00	P	III ORT			. (UNIT I	STARE		2.00		. <del></del>					
(H)	-2.00	-1.50	18T	-0.50	0.0	0.50	\$ TARE 1.00	1.50	2.00					**************************************		
(H) 1.000	-2.00 14.	-1.50 10.	IBT	-0.50	0.0	0.50	\$ TARE 1.00	1.50	2.00							
(H) 1.000 1.500	-2.00 14. 14.	10.	181 -1.00	-0.50	0.0 0.0	0.50 -3.	\$TARE 1.00	1.50 -10.	2.00 '-14. -14.							
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# タンク TRIM/HEEL 修正表

NO.2 F. o. T. (P/S) NO.4 F. o. T. (P/S) NO.1 W. 6. T. (P/S)

SOUNDING

## CORRECTION FOR HEEL



CORRECTION IN C.M.

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GRAIN RU	KE THEO	· X · 1 264		<del>.</del>			<del></del>	·	<del></del>	<del></del>	<del></del> -
. SIGN OF	CALC. TRAN	.) 1	TONEO	cxic.	TTCACC. ANI	D SECTAR	EX PRINT	ZICALC.			
	- IVERY		<del></del>		······	<del></del>			<del></del>		
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·	TATCH AND HATC	H CUVIR 1	£4								
• IN PUT DATA T	(II) (III) (III) (III) (III) (III) (III)	HATCH	COAMING			IAYCH CO'	VER				
•	(II) (III) (III) (III) (III) (III) (III)	HATCH	COAMING LENGTH HE I	GHT DIS	TANCE	HE LGH	T CINTER				
HATCH TY	PE DREAD JAY (H)	HATCH OTH FOR (	COAMING LENGTH HE AY (	GHT DIS	TANCE:   C.L. : SI (M)   (	HE LGH	CINTER (H)				
HATCH TY	PE NREAD AFT (M) 14.400	HATCH OTH FORT (M) 14.400	COAHING	GHT D19	TANCE, (M) ( 0.0 0	HE LGH   DE   M     0.704	CINTER (H) 0.704	- AWE	υ		
HATCH TY	PE NREAD	HATCH OTH FOR ( (M)	COAHING	GHT D19	TANCE, (M) ( 0.0 0	HE EGH DE M)	CINTER (M)	- ANKL	<i>u</i>		
HATCH TYI	PE READ AFT (H) 14.400 14.400 0.600	FOR ( (M) 14.400 14.400 0.000	COAHING	GHT D19	TANCE, (M) ( 0.0 0	HE LGH   DE   M     0.704	CINTER (H) 0.704	- ANNO	0		
HATCH TYPO	PE NREAD  AFT  (M)  14.400  14.400  0:600	HATCH  PORT  (M)  14.400  14.400  0.000	CDAHING	GHT D19	TANCE, (M) ( 0.0 0	HE LGH   DE   M     0.704	CINTER (H) 0.704		0		
HATCH TYPO	PE READ AFT (H) 14.400 14.400 0.600	HATCH  OTH  FOR!  (M)  14.400  0.000	CDAMING	GHT D19	TANCE, (M) ( 0.0 0	HE LGH   DE   M     0.704	CINTER (H) 0.704	- AWG	9		
HATCH TYPO	PE NREAD  AFT  (M)  14.400  14.400  0:600	HATCH  OTH  FOR!  (M)  14.400  0.000	CDAHING	GHT D19	TANCE, (M) ( 0.0 0	HE LGH   DE   M     0.704	CINTER (H) 0.704				

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TAIAT TAIAT	NOTT# 123 T SEC	TIONAL" AREA"	*****	(5	(C. # + 2)	M. Mit · L c	AL ISWT	RIT		
TA MCTTOR	FR. 154.(33 DI	FFERENCE 0.0				te- area al vacu vil a b		HELD ELLINGBERNER	•	
FVEN H.L.	SEC.AREA	CL 97	KG HT	14 H.071	C.AREA RP	CL HI	KG HI	ווננג אונג	)	1
41jyr 11-	150.4-7-151	5:H-H 1"[50.	4 4H ).	TIEEL ANGLE 25	-0"DEG		H AH I	"AANVE AL ATTI	**********	
(H.)		- EVIN H.L.		(TRAN.) (	VERT.	HELL H.L		IN I	MAL UL OF	1.77
1.095	0.3	0.0	0.0	0.0	0.0		V. 0	-3:K01	_ <del></del>	'  <b>'</b> -
	6.5	0.0	0.0	0.0	0.0 0	0.0	0.0	-3.601	1	1
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	0.0	0:0	0.G-		0.0 i · · ·	0.0		-3.661	. 1	1
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	14.65	• • • •	67.6	-100.6	17.1 6	-1:0.6	06.8	0.491	1	l
٠ ۲. ۲۶۶۶	1 22.34	o.j	144.6	-349.7	40.9 6	-349.7	- 107.5	2.435		
۸.00,	44.47	0.0	233.4	-444.4	67.5 0	-474.4	30 D. n	3.599	1	l
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7.000	75.10	0.0	437.3	-747.7	173.6 0	-142.7	561.0	5.530	1	l
1:440 0	4 41.44		7554	-641.7	147:7-0	-841.7	702: 3	8.394		1
· 7.(3 ·	3-124.16	6:3	. 1.79.1.	-919.5	170.2 0	-919.5	P44.2	7.414	L AWAY.	
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	- 144:33		1101:4-		20110 -		- 1151-5 - 1367. 6-	P-346		l
10.000	17: 40		1250.6	-1037.2	200.4 0	-1039.2	1466.9	10.160	j :	١
11.	776.87		1598.7		iñ#:00	-706:0	795.5	11.486	-	
17.00: 4	6 214.93		1467.6	-467.3	170.4 0	-947.3	2145.4	12.034	1	ſ
12.003	72/2.54		2766.4	-793.2	-151.3 ·· 6	-793.2	2437.7	14.664	·	
13.000	264.93		4,000	-754.4	143.1 0	-750.5	2511.6	14.308	1	l
14.000	\$ 2110.71		2770.3	-584: 2	49.4	-51.4.2	~\$044.7	15.660		
15.000	375.03		3151.6	-344.9	55.7 ù	-346.9	3207.3	17.555		1
186835	1-1:1:4:		356577	114:3	15.4	-111:	-322121-	19:505		
7 17.000	. <u>( 357 . 73</u>		3469.5	0.0	0.0 . 1	0.0	3665.5	23.562		l
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0.728	··· o.:o ··· `		Ú. J		.5 '0.0			.0.0	}	
OK. TOTAL	17445	SFC.	0.0	VERT. SEC.	0.0	1 3.0	44.		""	1
	Hr (	. 11.		HLEC HIS		<del></del>	44,43	<u> </u>		
SEC. TOTAL	TRAYS	5.6	0.6	VEAT-SEC.	0.0		That () 21/1	A A SEIN COYRL		
	HCE(	. 47:		HEEF HAS	· - ··-	san-tenzill	, ·	47.44 Sty	- [ ]	
***********	PAT 414 150 2 600	FF4F1.PP				ASA DIVATES		LOIN/REAT	ر	
SELETON AT 1	F47.148.2333 DIF	restact 0.4	40 M	ursigne of se		,	• • • •		•	
(V N P.C.	-Strakter	CLAT	76" HT	1,20.4,44	C. AREA , KP 1	· CE HY	KC-HT	-,,,-,-,,-,,-,-,-,-,-,-,-,-,-,-,-,-,-,-,		
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		- EVCY W.L.			VERY	# 150.H +H ) (50.		WINDLE UT WITH		
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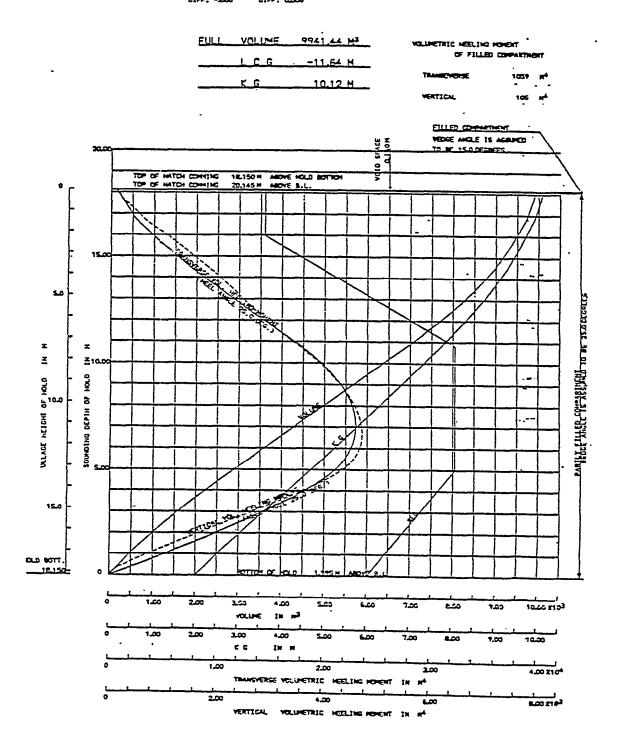
23		FR. 148.00									
	(UNDER UPPER DECK)	DIFF. V.	ያ0ጥ	F	J						
	COMPARTHENT	A1	FŢ	TT TON FO	nr F	NO.	VOID DEPTH	VOLUHF VOLUHF	YOLUHE	TRANS.	VERT.
	AFT PART OF HATCH	FH. 148.050	0.0	FR. 154.000	DIFF.	1	0.728	27.01	(COB.H)	[H++4]	H4-47
	HAYCH PART	154.000	0.0	169.000	0.0	1	0.150	00- TOTAL 147-57	27.8	152.0	-17
	FORE PART OF HATCH		<del></del>	175.000	<del></del>	1		50.30	147.6	546.5	-74
······						<del></del>	2	UB. TOTAL	56.3	214.9	-31
<del></del>					<del></del> -	<del></del>	<u> </u>	ECK TOTAL	225.7	973.4	-124
	<del></del>		<del></del>				GR	AND TOTAL	225.7	973.4	-124
			······			<del></del>	<del></del>				
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<u> </u>	DETAIL OF VOLUMEYA	ac Heef and	- ROHENT	TAT FYCCE	о сонракти	ent . (Jæec	TNG ANGI	.F 15.0 DE	toin' mi	West wo	
NO.	DETAIL OF VOLUMETR COMPARTHENT NAME		ריטודרט	AT FYEEE	O COHPARTHI VOLUHI: (M. M.)	ENT THE EL	KG	100	TRAN. VOL.	VERTAVOL. HEELAH	ř <del> </del>
	COMPARTHENT NAME NO.3 CARGO HOLD (F)	PU AFT 1.148.330	511104 F)	OR E 5. 23%)	VOLUHI:	enttæeï	KG (H)		TRAN. VOL.		6)
NO. 23	COMPARTHENT NAME NO.3 CARGO HOLD (F)	AFT	511104 F)	OR E 5. 23%)	(cnu 'W ) AOEAHE	(H) · CCG ENT THEED	KG (H)	ICG (H)	TRAN. VOL. HEEL.HT. (H++4)	HEEL . HI	6)
	COMPARTHENT NAME  NO.3 CAAGO HOLD  O  AFT PART OF HATCH IFF	AFT  1.148.330 1FF - 0.600	FR.17 - FR.17 - DIFF.	DR E 5. 73%) 0.000	(cnu 'W ) AOEAHE	(H) · CCG ENT THEED	KG (H)	ICG (H)	TRAN. VOL. HEEL.HT. (H++4)	HEEL . HI	2
	COMPARTHENT NAME  NO.3 CARGO HOLD  O  AFT PART OF HATCH  O  HATCH PART  (F)	AFT 1.148.333 1FF. 0.600 (.140.000 1FF. 0.0	- FR.17 - DIFF. - FR.18 - FR.18	0RE 5. 2341 0.000 4.000) 0.0	(cnu 'W ) AOEAHE	(H) · CCG (H) ·	KG (H)	1CG (H)	TRAN. VOL. HEEL. HT. (H*+4) 973.4	HEEC.AT (M++4 124.2	2
	COMPARTMENT NAME  NO.3 CARGO HOLD  TET PART OF HATCH  O  HATCH PART  O  FORE PART OF HATCH  (F)	AFT  1.148.370  1FF. 0.600  1.148.000  1.148.000  1.154.000  1.154.000	- FR.17 - FR.17 - FR.18 - FR.18 - FR.18 - TR.18 - TR.18	OR E 5. 73%) 0.000 4.000) 0.0 9.000) 0.0 5.000)	(cnu 'W ) AOEAHE	(H) · CCG (H) ·	KG (H)	1CG (H)	TRAN. VOL. HUEL. HT. (H++4) 973.4	124.2	2
	COMPARTMENT NAME  NO.3 CARGO HOLD  TET PART OF HATCH  O  HATCH PART  O  FORE PART OF HATCH  (F)	AFT  1.148.370  1FF. 0.600  1.148.000  1.154.000	- FR.17 - FR.17 - FR.18 - FR.18 - FR.16 - DIFF.	OR E 5. 73%) 0.000 4.000) 0.0 9.000) 0.0 5.000)	(cnu 'W ) AOEAHE	(H) · CCG (H) ·	KG (H)	1CG (H)	TRAN. VOL. HEEL.HT. (H++4) 973.4 152.0 546.5	124.2	2
	COMPARTMENT NAME  NO.3 CARGO HOLD  TET PART OF HATCH  O  HATCH PART  O  FORE PART OF HATCH  (F)	AFT  1.148.370  1FF. 0.600  1.148.000  1.148.000  1.154.000  1.154.000	- FR.17 - FR.17 - FR.18 - FR.18 - FR.18 - TR.18 - TR.18	OR E 5. 73%) 0.000 4.000) 0.0 9.000) 0.0 5.000)	(cnu 'W ) AOEAHE	(H) · CCG (H) ·	KG (H)	1CG (H)	TRAN. VOL. HEEL.HT. (H++4) 973.4 152.0 546.5	124.2	2
	COMPARTMENT NAME  NO.3 CARGO HOLD  TET PART OF HATCH  O  HATCH PART  O  FORE PART OF HATCH  (F)	AFT  1.148.370  1FF. 0.600  1.148.000  1.148.000  1.154.000  1.154.000	- FR.17 - FR.17 - FR.18 - FR.18 - FR.18 - TR.18 - TR.18	OR E 5. 73%) 0.000 4.000) 0.0 9.000) 0.0 5.000)	(cnu 'W ) AOEAHE	(H) · CCG (H) ·	KG (H)	1CG (H)	TRAN. VOL. HEEL.HT. (H++4) 973.4 152.0 546.5	124.2	2
	COMPARTMENT NAME  NO.3 CARGO HOLD  TET PART OF HATCH  O  HATCH PART  O  FORE PART OF HATCH  (F)	AFT  1.148.370  1FF. 0.600  1.148.000  1.148.000  1.154.000  1.154.000	- FR.17 - FR.17 - FR.18 - FR.18 - FR.18 - TR.18 - TR.18	OR E 5. 73%) 0.000 4.000) 0.0 9.000) 0.0 5.000)	(cnu 'W ) AOEAHE	(H) · CCG (H) ·	KG (H)	1CG (H)	TRAN. VOL. HEEL.HT. (H++4) 973.4 152.0 546.5	124.2	f . ( ) 2
	COMPARTMENT NAME  NO.3 CARGO HOLD  THE PART OF HATCH  OF THE PART OF THE PART OF HATCH  OF THE PART OF THE	AFT  1.148.370  1FF. 0.600  1.148.000  1.148.000  1.154.000  1.154.000	- FR.17 - FR.17 - FR.18 - FR.18 - FR.18 - TR.18 - TR.18	OR E 5. 73%) 0.000 4.000) 0.0 9.000) 0.0 5.000)	(cnu 'W ) AOEAHE	(H) · CCG (H) ·	KG (H)	1CG (H)	1 RAN. VOL. HEEL.HT. (H*+4) 973.4 152.0 546.5 276.9	124.2	2

				DIFF. O.		TRAH.VUL.	VERT.VOL.		· · · · · · · · · · · · · · · · · · ·		
H.L.	DEPTH	NET V.	K G	LCG	TCG	HEEL . HT.	HEEL.HT.	KP	•		•
ANOVE AL	Annyt Ho.	- (CUBTH T	(H)	N 1		125.0 DEGI	T25.0 DEG )			<del></del>	
<u>[M.]</u>	<u> </u>					(H ++4)	[H ++4]				
1.975	5.0	. 0.0	395	0.0	0.0	0.0	0.0	7.		<del></del>	
13.077	0.503	235.43	2.516	-33.641	0.0	2616.9	274.2	00			
	1.005	472.39	2.659	-33.041	0.0	\$697.9	103.7	U			
3.240	1 .245 1 .5 05	307.40	2.72.6	-33.841	0.0	62.29.2	9.0.6	1			
4.000	2.005	974.46	3.051	~33.795		7536.7	1168.6	00			
4.483	2.475	1266.41	3.311	-33.741 -33.712	0.0	9903.9	1661.7	. 0	·		
5.000	3.005	1573.23	3.589	-33.601	0.0	12617.0	2137.4	1			
5.500	3.505	16:5.71	3.051	-33.531	0.5	14369.4 16319.8	2650.5	0			~~~
6.000	4.005	2202.33	4.119	-33.476	0.0	10319.8	3134.3 3590.9	0			
	4.505	25 33 . 00	4.379	-33.432	<del>8:3</del>	15705.0	4001-1	<del>8</del>	····	····	
7.000	5.005	2072.48	4.684	-33.398	0.0	20981.0	4327.7	0			
7.57.	5.5.5	3216.36	4.95	-33.370	ŏ:ŏ	21911-0-	4549.6	<del></del> 8	<del></del>		
n.365	5.005	3349.20	5. 22 B	~33.348	0.0	22409.4	4661.3	č			
8.530	6.565	3454.18	5.494	-33.330	0.0	22007.9	4700.0	<del>6</del>		<del></del>	
9.000	7.005	4248.06	5.750	-33.315	0.0	22928.9	4696.7	ŏ			
5.505	7.505	4541:48	410:8	-33.302	0.0	22854.4	4650-1	<del>ö</del>			<del></del>
10.000	8.005	4935.86	6.279	-33.291	0.0	22592.2	4565.8	Õ			
33	9.075	5623.64	6.795	-33.273	0.0	21513.9	4294.4	<u> </u>		<del></del>	~~~~
12.000	10.005	6311.43	7 . 30 (	-33.25 A	0.0	19710.9	3904.7	0	•		
13.000	10.005	6001.66	7.716	-33.249	0.0	17726.9	3512.5	0	***************************************		····
13.000	11.005	6992.19	7.770_	-33,247	0.0	17157.8	3403.8	0			
14.065	12.005	7622.52	0.207	-33. 238	0.0	14230.2	2097.6				
15.000	13.005	0194.50	0.719	-33.232	0.0	11414.1	2473.8	00			
16.000	14.003	16.00.78	9.7.95	-33.228	0.9	8663.9	2043.6	0			
17.000	15.005	9072.59	9.259	-33.226	0.0	7351.6	1014.8	<u>l</u>			
17.659	15.664	9300.76	9.416	-37.235	0.0	6040.4	1560.7	0			
17.844	15.950	9340-25	9.607	-33.245 -33.254	0.0	4541.2	1206.9	!	·		
18.775	16.005	9405.61	9.697	-33.254	0.0	3930.1	1055.5	į			
10.745	18.750	7613.79		-33.275	0.0	3023.5 2271.8	1025.n 614.9	<del>0</del>	<del></del>		<del> </del>
18.433	16.030	9635.66	9.905	-33.273	0.0	2122.3		1			
19.000	17.005	9864.43	6.632		0.0	1919.2	517.0	<del>-å</del>			
20.000	18.005	9836.71	10.099	-33.273	0.0	738.3	110.1	ŏ			
20.145	18.150	9861.69	10.124	-33.273	0.0	614.7	84.4	<del></del>		<del></del>	
						~.,,,	0.74.4	•			
		_	<del></del>			<del></del>				<del></del>	
			ENGN AS	DA729							
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,	•		* ************************************	1	<del></del>		_HōLt		<u>"- 7</u>	<del></del>	
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COMPARTHENT	VOLUME	353	KG	TRAN. VOL.	VERT.VOL.	
40.	(CUB.H )	(H)	(H)	(H ++4)	(H ++4)	•
21 NO.1 CARGO HOLD	7067.02	-74.991	10.613	1021.8	100.4	
22 NO.2 CARGO HOLD	9923.09	-54.775	10.157	1131.2	112.4	
23 NO.3 CARGO HOLD	9861.69	-33.272	10.124	973.4	124.2	
24 NO.4 CARGO HOLD	9941.45	-11.637	10.121	1 65 9 . 4	104.6	
15 NO.5 CARGO HOLD	9967.70	9,988	10.141	1132.2	112.6	
CANDA CARGO HOLD	9841.78	31.770	10.172	1 132.2	112.5	
T NO.7 CARGO HOLD	9106.09	52.760	10.569	1178.4	118.1	<del></del>
11 NO.3 T.S.T. (P/S)	1191.93	-22,300	16.595	4266.6	1793.2	
(5.5105)				4143.3	1792.9	
(P.STDE)	<del></del>			123.3	0.3	
12 NO.4 T.S.T. (P/S)	1191.88	20,899	16.595	378.3	46.4	
(S.SIDE)		······································	·	252.1	33.6	
(P.SIDE)	~~			126.1	12.8	** <u> </u>
O NO.3 C.H.(IN DECK)	11331.96	-33.100	10.765	0.0	0.0	
1 NO.3 C.H. (CHECK)	8354.00	-33,100	9.680	0.0	0.0	***************************************
····				<del> </del>		
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### GRAIN HEELING MT.

NO.4 CARGO HOLD (FR.121.000 - FR.148.000)



			PERATINAL IN		000. → HAXINUH	D1 SPT=	** 50000. • IN	YERVAL DISPI-	500.	
TANK NO.	SG	HL ( 1 )	INERTIA(1)	HL [ 2 ]	INERTÍA (2)	WL(3)	INERTIA(3)	HL(4) INE	ERTIA(4)	•
6	1.025	0.0	Q.	0.0	0.	'a.o	0.	0.0	٠ ۵.	
<del></del>	1.025	0.0	0.	0.0	<u> </u>	0.0	<u> </u>	4.4	0.	
9	1.025	0.0	٥.	0.0	0.	0.0	0.	0.0	0.	
١č	1.025	0.0	0. 0.	0.0 0.0	0.	0.0 0.0	0. 0.	0.0 0.0	0.	٠.
ii	1.025	0.0	ő.	0.0	0.	6.0	Ű.	0.0	0.	
13	1.025	ŭ.ŭ	٥.	0.0	Ů.	0.0	0.	0.0	. 0	
14	1.025	0.0	0.	0.0	Ď.	0.0	ů.	0.0	0.	
15	1.025	0.0	. 0.	0.0	6.	0.0	Ŏ.	ō.ō	ŏ.	
16	1.025	0 • 0	0.	0.0	, o•	0.0	. 0.	0.0	0.	
17	1.575	11.0	0.	0.0	6.	.0.0	D.	0.0	0.	• • • • •
1A · 19	1.025	0.0	0.	0.0	<b>0.</b>	0.0	. 0.	J.6	0.	
20	0.950 0.950	0.0	U. 0.	0.0 0.0	ý.	0.0	, <b>0</b> •	0.0	0.	
<u>2</u> 1	- <del>0.955</del>	<u>0:0</u> -	ă:	0.6	<u> </u>	<u>0.0</u>	<u>0.</u>	<u>0.0</u>	<u>0.</u>	
2 2	0.900	0.0	0.	0.0	0.	0.0	0.	0.0	0. <i>F</i>	
23	1.000	0.0	ű.	0.5	ě.	0.0	G.	0.0	0.	
6 7 23 0 19 20 21 22 0 0 0 0 0 0	0 0 0 0 0 0 0 0	10 11 13 0 0 0 0 0 0 0 4 6 0 0 0 0 0 0 0 0 0	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0		FRESH WATE FUEL DILIC- FUEL DILIA- OF THE DILIA-	R OT L. )	.,	:		
			I SPLACEHENT DA							~~~~.
	* D	ISPT NO.=	. 4 HINIHUH	DISPI=	2 GOUD. + HAXI	HUM 015P T=	50000. +	INTERVAL DISP	Y= \$00.	
,	· · · · · · · · · · · · · · · · · · ·	20000. 25000. 30000.	25500. 26 30500. 31	000 . 20	1500. 22000 5500. 27000 1500. 32000	· 27500 · 32500	20000. 33000.		24000. 24500. 29000. 29500. 34500. 34500.	***
		40000.			5500. 37600 1500. 42000				39000	

TANK\$	BALLA	ST WATER					•
	NO.1 T. S. 1	TNO.2 T. S. 1	INO.3 T. S. TI	IO.A +. 4. 3			A C A S S S S S S S S S S S S S S S S S
. 14210	. (2/5)	. 17/51	. (7/5)	IP/S)	(P/S)		
<u>(HT)</u>	3.G1.025		erije ma j ve jej kare				<u>.</u>
	3.01.029	3.61.025	"\$.G1.025"	2.0.=1.053	3.61.025		
20000.	0.032	0.040	0.040	0.040	0.042		
20500.	0.011	0.039	0.039	0.039	0.041		
21000.	0. 033	t. u36	0.03#	0.038	0.040		
_31255-	<u> </u>	<u> </u>	0.037	J. 017	0.039		
22000.	0.019	0.036	0.034	0.034	0.031	<del></del>	
22500. 23000.	0.02A 0.02Y	0.035	0.033	0.035	0.037		
23103	6.527	0.034	0.031	0.035	0.037		• • • •
24 300.	0.036	u. 034	0.034	0.034	0 • 0 3%		
24100	0.074	0.033	0.011	0.033	0. 333		* 4 deter e des space vans and i
		<u>v. v. v. 4</u>	<u> </u>	0.033	<u> </u>		
25000.	0.025	0.032	0.032	0.032	0.034		
23500.	0.025	0.031	0.011	0.031	0.033		
260(· <b>U</b> •	0.374	0.031	0.031	0.031	0.033		, , , , , , , , , , , , , , , , , , , ,
24500.	0.014	0.030	0.010	0.033	0.032		
	<u> </u>	0.029		0.030	0.031		
27500.	0.021	0.029	8:8:3	0.029		·	
30000.	0.023	0.024	0.078	0.02	04030		
37100.	0.022	0.020	0.010	0.024	0.030		· , ·
79500.	9.372	V. 327	0.027	0.011	0.029		
243004	0.021	0.027	0.017	0.627	0.020		
30000	0.031	0.016	0.027	-0.011	0.020	·	•
30500.	0.021	0.024	0.074	0.024			
31000.	0.020	0.024	0.024	0.024	0.02# 0.027		•
31560.	6.420	0.025	0.075	0.02	0.027		•
37000.	0.0:0	0.025	0.025	0.029	0.024	•	
32500.	0.019	J. 424	0.025	0.025	0.014		,
33000.	0.014	0.024.	6.624	0.024	. 0.024	<del></del>	
33500.	0.019	0.024	0.014	0.024	0.025		
34000.	0.019	0.053	0.053	0.023	0.025	•	•
34500.	0.314	0.023	0.023	0.423	0.074	•	
35000.	0.018	U. UZ 3					
35300	<u> </u>	8.022	<u> </u>	0.023	0.024		
36 000	0.011	0.022	0.022	0.02	0.024		
34500.	0.017	0.022	0.022	0.022	0.021		•
370 //.	6.517	151.0	0.672	0.672	0.023		•
37500.	0.017	4. 02 1	4.031	0.021	0.072		
31000.	0.017	0.021	0.021	0.021	0.072	•	,
16 500 .	0.016	0:01	0.021	i.ozi	6:022	<del></del>	
37000.	0.016	0.020	0.020	01.020	0.022		
39505;	0.316	7.753	0.525	o.dza	10.021		
•							

TTT CARD INF	UT DATA ***
TITLE	TRIH TABLE - DRAFT CORRECTION TABLE BY LOADING UNIT WEIGHT
WETCHT	`100.00 (HT)
HIN. DRAFT	2.000 (H)
HAX. DRAFT	1?.000 (H)
INTERVALS	0.500 (H)
TINU	0 ,
TANK NO.	6 7 8 9 10 11 13 14 15 16 17 18 0 0 0 0 19 19 27 21 22 23 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
,	
ADDED DATA	TANK NO FANK NAME H.G(H)  ( 0.0 0 0.0
	0 0.0 0 4.0 0 0.0
	التوالية المنظلية ال
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	en de la companya de la companya de la companya de la companya de la companya de la companya de la companya de La companya de la companya de la companya de la companya de la companya de la companya de la companya de la co
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	TRIH TABLE
	DRAFT (H)
TANK NAME	2.0 2.5 3.0 3.5 4.0 4.5
HO.1 T. S. T. (P/S)	FND 9.1 8.9 0.7 8.6 8.5 8.5 AFT -5.0 +4.8 -4.7 -4.6 -4.5 -4.4
NO.2 T. S. T. (P/S)	FHD 6.2 6.1 6.0 5.9 5.9 5.9
NO.3 T. S. T. (P/S)	AFY -1.8 -1.7 -1.6 -1.6 -1.5 -1.5 FHO 3.0 3.0 3.0 3.0 2.9 3.0 AFY 1.9 1.8 1.8 1.0 1.8 1.7
NO.4 T. S. T. (P/S)	FHD -0.2 -0.1 -0.1 -0.0 -0.0 0.0 AFT 5.5 5.4 5.3 5.2 5.1 5.0
NO.5 T. S. T. (P/S)	FHO -3.5 -3.3 -3.2 -3.1 -3.0 -3.0 AFT 9.2 9.0 8.8 8.6 9.4 8.3
NO.1 W. H. T. (P/S)	FHD 9.2 9.0 8.8 8.7 8.6 4.6 AFT -5.1 -4.9 -4.0 -4.7 -4.6 -4.5
NO.7 W. A. T. (P/S)	FHD 3.0 2.9 2.9 2.9 2.9 2.9 AFT 1.9 1.9 1.8 1.8
NO.5 H. B. T. (P)	FUD -3.4 -3.2 -3.1 -3.0 -2.9 -2.9 AFT 9.1 8.9 8.7 8.5 8.3 8.2
NO.5 W. B. T. (S)	AFT 9.1 8.9 8.7 8.5 8.3 8.2  FHD -3.2 -3.0 -2.9 -2.8 -2.7 -2.7  AFT 8.9 8.6 8.4 8.3 8.1 8.0
FORE PEAK TANK	FMD 11.6 11.3 11.1 11.0 10.0 10.6 AFT -7.9 -7.6 -7.4 -7.2 -7.1 -7.0
AFT PEAK TANK	FHD -0.3 -8.0 -7.6 -7.5 -7.4 AFT 14.7 14.3 14.0 13.7 13.4 13.2
NO.3 CARGO HOLD [HII]	FND 3.0 3.0 3.0 3.0 3.0 3.0 AFT 1.0 1.8 1.8 1.8 1.7
REMARK : TABLE O FOR EAC	CHANGE IN DRAFT IN CENTIHETERS

TANK/IDLD CAPACITY SUMMARY TABLE			12 ++	102581 **	DAT	E	11.16.	
+ CALCULATION UNIT SIGN = 0	o <u>oo_</u>	<u> </u>	0 0					
			<del></del>	·····				
TANK / HOLD APPLICATION DAT	A ++++		······································		•			
KIND DE CARGO 1 TYPE OF TABLE	O CARGO TITLE	FRESH WATER TAN	к			<del></del>		
TANKZHOLD NO. SUH1 23 00	000.	0000	000	0_0_0	00	0		
TANK\HOLD NO ZUHZ O O O	0 0 0 0	<u> </u>	0 0 0	0 0 0	00	0		
TANK/HOLD NO. SUH3 0 0								
CAPCLIYILL CUB.H CAPACITYI2						,		
CONVET (1) 1.00000 2 CONVET (			-		•			
	•				14.4.4.41.8			\$
		** ** ****** *** * **********			** ************************************		~~ ~~ ~~~	
rational substitute to the even beautiful and the first to the first termination was			randa ayrını merenden	1 v::n -r-v===		=======		
					•			
+++ FRES	H-WÂTER TANK	¥ <b>••</b>		S.G. = 1.	000			
		·				- ·· · · ·	• • • •	
	CAPACIT	A	NEIGHT	S.G. = 1.				
TANK NAME FR.NO	CAPACIT CUB+H	CUB.FT	NEIGHT B FULL LHTI	CENTER OF	GRAVITY			•
	CAPACIT	A	NEIGHT	CENTER OF	GRAVITY			
TANK NAME FR.NO	CAPACIT CUB+H	CUB.FT	NEIGHT B FULL LHTI	CENTER OF	GRAVITY K G			
TANK NAME FR.NO.	CAPACII CUB-H 385,39	CUB.FT	NEIGHT B FULL LHTI	CENTER OF	GRAVITY K G			
TANK NAME FR.NO	CUB.H 385.39	CUB.FT 13609.8	HEIGHT DEFULL (HY) 385.	CENTER OF	GRAVITY K G			
TANK NAME FR.NO.  FRESH W. T. (P/S) 9 - 17  ++	CAPACIT CUB-H 385,39 DATA ++	CUB.FT 100	NEIGHT DEFULL (HT) 385.	CENTER OF	GRAVITY K G	0 0		
TANK NAME FR.NO.  FRESH W. T. (P/S) 9 - 17  ++	CAPACIT  CUB.H  385.39  DATA ****  ABLE O CARGO TIT  O O O O	CUB.FT 100	WEIGHT DEFULL (HT) 385.	CENTER OF LCG 78,35	GRAVITY K G			
TANK NAME FR.NO.  FRESH W. T. (P/S) 9 - 17  +++ TANK/HOLD APPLICATION  KIND OF CARGO 2 TYPE OF TANK/HOLD NO. SUH1 19 20	CAPACIT  CUB.H  385.39  DATA ***  ABLE O CARGO TIT  O O O O	CUB.FT 100  CUB.FT 13609.8  LE FUEL OIL T/	HEIGHT 0 \$ FULL (HY) 385.	CENTER OF LCG 78,35	12.99 0 0	0 0		
TANK NAME FR.NO.  FRESH W. T. (P/S) 9 - 17  ++++ TANK/HOLD APPLICATION  KIND OF CARGO 2 TYPE OF TANK/HOLD NO. SUM1 19 20  TANK/HOLD NO. SUM2 0 0  TANK/HOLD NO. SUM3 0 0	CAPACIT  CUB.H  385.39  DATA ****  ABLE O CARGO TIT  O O O O O O O	CUB.FT 100  CUB.FT 13609.B  LE FUEL OIL Y  O O O O  O O O O	NEIGHT 0 8 FULL (HT) 385.	CENTER OF LCG  78.35	12.99 0 0	0 0		
TANK NAME FR.NO.  FRESH W. T. (P/S) 9 - 17  +++ TANK/HOLD APPLICATION  KIND OF CARGO 2 TYPE OF TANK/HOLD NO. SUM1 19 20  TANK/HOLD NO. SUM2 0 0	CAPACIT  CUB.H  385.39  DATA ****  ABLE O CARGO TIT  O O O O  O O O O  TY(2) BARREL WE	CUB.FT  13609.8  LE FUEL DIL TA  0 0 0 0  0 0 0 0  1GHT NAME DE CARG	NEIGHT 0 % FULL 14Y) 385.	CENTER OF LCG  78.35  0 0 0 0 0 0 0 0 0	0 0 0 0 0 0	0 0 0 0		

### タンク等サマリーテーブル 〔標準 FORM〕

		FRESH 1	WATER 1	TANK	5.G. <del>*</del>	1.000
TANK NAME	CD NO	CAPA	CITY	<b>AEIGHL</b>	CENTER OF	GRAVITY
TANK NAME	FR.NO.	C.B.X		100 x FULL	LCG	KG
			C.B.FT	(HT)	(H)	(H)
FRESH W. T. (P/S)	9- 17	385.37	13609.2	325	78.35	12.09

		FUEL (	ANT LIC	١K	s.c	.0.>50
TANK MANG	FD 15	CAPAC	CITY	WEIGHT	CENTER OF	GRAVITY
TANK NAME	FR.NO.	CLS.4		96 I FULL	LCG	KG
			BARREL	CMTS	(H)	(14)
NO.2 F. O. T. (P/S)	139- 173	1526.55	9601.7	1392	-36.65	1,15
NO.4 F. O. T. (P/S)	71- 105	1547.53	9733.7	1411	17.58	1,44
TOTAL		3074.08	19335.4	. 2 <del>0</del> 03		

		DIESEL	OIL TA	NK	s.c.	0.900
		CAPA	CITY	AETOHL	CENTER DE	CSAVITY
TANK NAME	FR.NO.			96 × FULL	LCG	KG
		CUB.H	BARREL	(ዘባን	(H)	-(4)
F. O. T. (A-OIL) (P)	15~ 39	97.43	812.8	84	65.03	0.97
F. 0. T.(A-OIL) (S)	17- 39	96,44	606.6	83	64.72	0.~
TOTAL		193.87	12194	157		

	h	ATER B	ALLAST	TANK -	\$.G	1.025
TANK NAME	50 NO	CAPA	CITY	AETOHL	כבאויבה: סד	GRAYITY
TANK NAME	FR.NO.			100 x FULL	LCG	K G
		CUB.H	C.E.FT	· (HT)	(H)	(#)
NO.1 T. S. T. (P/S)	174- 207	733.28	25875.6	752	-61.55	1410
NO.2 T. S. T. (P/S)	140- 174	993.68	35091,4	1019	-37.33	13.59
NO.3 T. S. T. (P/S)	106- 140	999-24	35298,4	1025	-10.17	13.27
FORE PEAK TANK 207- 21		1575.00	56362.2	1636	-62.83	8.28.
AFT PEAK TANK	-6- 8	283.46	10010.2	29!	85.96	10.26
SUB TOTAL	10531.22	371907.0	10797			
NO.3 CARGO HOLD (VB)	9447,63	333640,0	7999	-10.42	8_71	
SUB TOTAL		9447.63	333640.0	7584		
GRAND TOTA	\L_	1997 <u>8.8</u> 5	705547.0	20481	<del></del>	

### タンク等サマリーテーブル 〔変 則 FORM〕

FUEL OIL TANK									
T 5 1 1 6 1 6 1 7 7	55	CAPAC	CITY	CENTER OF GRAVITY					
TANK NAME	FR.NO.			LCG	Ķ G				
		CUB.M	BARREL	(M)	(H)				
NO.2 F. O. T. (P/S)	139- 173	. 1526.55	9601.7	-36.65	1.45				
NO.4 F. O. T. (P/S) 71- 105		1547.53	9733.7	17.58	(-,44-				
TOTAL		3074.08	19335.4						

WATER BALLAST TANK								
TANK NAME	בם אס	CAPACITY	CENTER OF GRAVITY					
TANK NAME	FR.NO.	A - 4	LCG	КĠ				
		CUB.H	(H)	(H)				
NO.1 T. S. T. (P/S)	174- 209	733	-61.55	14.10				
NO.2 T. S. T. (P/S)	140- 174	994	-37.33	13.89				
NO.3 T. S. T. (P/S)	106 140	1000	-10.19	13.87				
NO.4 T. S. T. (P/S)	72- 106	1000	17.01	13.89				
NO.5 T. S. T. (P/S)	36- 72	1045	44.26	13.91				
NO.1 W. B. T. (P/S)	173- 209	1801	-62.48	1.31				
SUB TOTAL		5853						
NO.3 W. B. T. (P/S)	105- 139	1551	-9:50	1,44				
NO.5 W. B. T. (P)	36- 71	567	43.97	. 2,12				
NO.5 W. B. T. (S)	36- 71	581	12.24	1.82				
FORE PEAK TANK	207- 215	1596	-82.83	8,25				
AFT PEAK TANK	<del>-6-</del> 8	283	<b>85.</b> 96	10.86				
SUB TOTAL		4578						
NO.3 CARGO HOLD (WB)	105 140	9448	-10.42	8.71				
SUB TOTAL		9448						
GRAND TOTA	L	19979						

	BSJA10 TAIM_CALCIU.ATIVM	
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	CONDITI	UM 40.	. * 4				***	CONOII	TON CAL	CULATION S	TART +++		
	*#_CHEC	K-MUTI	E_0E0	ARO_INEU	<b>Ι_</b> ΩΑΤΑ±	<b>4</b>	······································			Per b 10 1 6 daile ( P = m - P 40 m 40 m		emman sofe en	
<del></del>	TRCONO	4FUL	L LOAD	CONDI	TON A	T UKC ARE	ι.		DAENTA	BUNK.	<del></del>	•	
· · · · · · · · · · · · · · · · · · ·	TREDGE	11099	0	0	<del></del>	<del></del>	<del></del>			4. ····································			
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FRESH WATER	# 700 HT
FUTL NIL	'= 911 HT
OIFSFL OIL	180 MY
WATER HALLAST .	w o Nt
CARGO OIL	# - 448107 HT
CONSTANTS	* 540 HT
DEADHEIGHT	# 450496 HT
LIGHT WEIGHT	m 62679 HF
DISPLACEMENT	• 513175 H7
CURRESPOYDING DRAFT	# 24.72 H
T P G	# 225-23 HT
i.c.g	a ~19.30 H
LCA	= ~15,3ù M
чтс	# 5734-5 NT~ H
TAIH	, , , , , , , , , , , , , , , , , , ,
L C.F	-4-17 H
DRAFT AT FIR.	= 24.72 H
DRAFT AT A.P.	- 24.72 H
DRAFT (MLAN)	■ 24.72 H
PROPFLER INVERSION	• 232 <b>t</b>
T K4	27.77 H
K G	■ 16.34 H
·G GO	n 1-11 H
, Gn M	10:32 N
MOMENT EN HEEL 1 DEG.	# 10632 H # 92448 HT- H

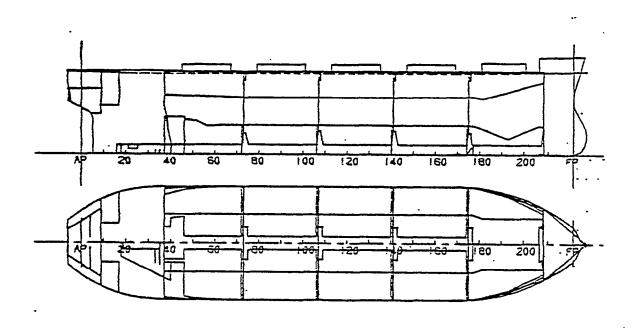
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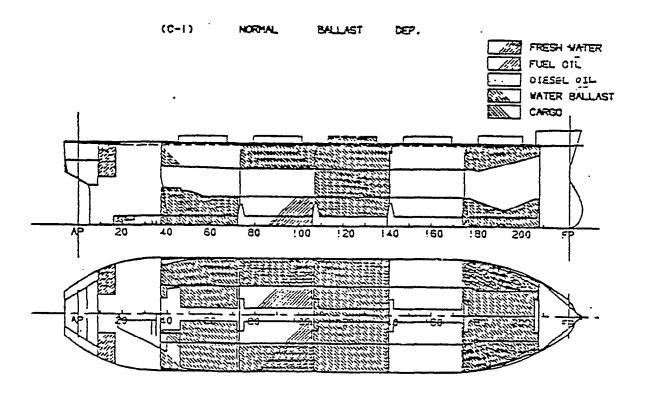
FULL LOAD C		OHEHAY, PUNK(	(2/2)
8 T E H \$	E WETGILF L C & HOHENT		G GO
LIGHT HEIGHT		. 17.721110670 14.44 4010	
FRESH WATER T. IP	1 50 139 179,44 22124	70,00 3646 0	0.0 2.4 0.0
_DIST. WATER T IS IFRESH WAYER TOTAL	) 98 190 (65.82 31506 ) 1 750) 1 1307301	28.26 _ 5369 _ U	0.0 )
FORE FUEL OIL TAYS  AFT FUEL O. SETT. I.	$\frac{5}{5}$ $\frac{0}{53}$ $\frac{0}{91}$ $\frac{137.63}{155.95}$ $\frac{0}{142070}$	25. 75 23094 0	0.0
<u> </u>	79 150 162,02 24303 1 150 ( 24303)	27.074040 0	0,0
FORF PFAK TANK	0 0-172,62 0	0.0 0 0	U.J 0.0
NO.6 W. M. T. CO RESERVE W.H.T. CO RESERVE W.M.T. CP/ AFT PEAK TANK	1 0 0 121.75 0	0.0 0 0	0.0 0.0 0.0
10.1 G. O. T. 10	( ))	1 5110	0.0
NO.2 C. O. T. (C NO.4 C. O. T. (C NO.3 C. O. T. (C	) , 94	16.06 395262 0 16.06 197077 0 16.06 591217 0	0.06 0.03 0.10
NO.7 C. O. T. 16 NO.1 C. O. T. 16 NO.1 C. O. T. 167 NO.2 C. D. T. 187	1 46 23615 146.64 2471366 51 20 21172 -150.41 -3104460	16.57 341350 0 17.14 363311 0	0.16 0.06 0.02 0.03
NO.3 C. O. T. 12/ -NO.4 C. D. T. 12/ -NO.4 C. D. T. 12/	\$}	15.67 312132 0 15.74 302255 0	0.06 0.06 0.14
10.0 C. O. T. (P/ NO.7 C. O. T. (P/ NO.6 C. O. T. (P/	<u>51_90</u>	15.74 302755 0 15.74 102270 0 14.74 342255 0	Q, uh Q, nb Q, nb
N2.9 C. N. Y. 1P/ N2.10 C. N. Y. 1P/ N0.11 C. N. Y. 1P/ N0.12 C. 2. T. 1P/	\$1 _ 93?41?7 <u>72,4917444</u> \$1 _ 982370654,251266051	11.04 301429 U	០ <sub>០</sub> ០
NO.13 C. O. T. (P/		17.46 349014 0 19.27 295563 0	0.06 0.04
	413175 =15.70 =7049544	•	1.11
	NOTE 1 - SIGN SHOWS FORWARD FR	OM MEDSHEP.	

CO - 11645	OVDITIONS	COMPITION	SIR UNLOAD. CONDITION AT UKC ARR.	POITIGNO	CONDITION		PORT CONDITS DW AT MINA A L AHENADI	TAIL SIRFT AFLOAT CONDITION	DOCKSHE
		UMENAY BUNG.	ONEWAY DUNK.	CHEHAY BUNK.	OMERAY BUNK.		NEW FACILITY		•••••
FRESH WATER	` '47 · "	747	785	74)	780		780	0	780
FUEL TIL	HT	411	*11	V1 1	<b>V11</b>	0	. 101		797
Offset of	HT	150	110	150	150		150		150
WATER PALLAST	HT	<u> </u>	4191R	30601	17000	0	149300	9861	40147
CARGO OIL	. 41	225367	222720	143073	105034	··· ··· · · · · · · · · · · · · · · ·	0	· • · · ·	·
CONSTANTS	41	148	948	140	544		348	54.8	546
DEADWERNIT	18 .	227776	771077	376463	126423	0	151575	10409	42422
FIGHT ASSOUT	HT	62679 "	62679	62677	: 62679	· · · · · · · · · · · · · · · · · · ·	42679	42679	42479
DISPLACEMENT	hf	290455	333706	430742		· · o · · · ·	214254	73008	105101
COTTESPONDING DT.	ÁFŤ " H""-	14.53	15.76	21.34	9.68	0.0	10.09	3.93	5.55
T * C	41	211.58	214.30	229.74	205.49	0.4	207.43	1 95.54	199.53
L C G	н	-9.38	-14.95	-13.41	-4.84	0.0	-10.75	. ~13.91	~23.14
<b>ι</b> σ • · · · · ·	4	~20.05	-17.22	~16,76	-21,75	0.6	-21.34	-23.73	-23.14
H T C '	" - IP	4715.5	4945.4	#n.2	4470.0		4546.3	3973.3	4153.7
TREM	и .	6.49	2.87	2.47	7.15	0.0	4-99	i . 1 1	0.0
LCF		-14.03	-12.55	-4.99	-14.86	U.0	-18.62	-72.38	-21.44
DRAFT AT F.F.	p	11.55	15.22	20.19	A448	0.0	9.65	3.14	 5.55 "
DRAFT AT A.P.	M	18.04	18,09	22.46	13.63	··· , ·· oin ··	13.64	4.95	5.55
DRAFT CHEARS .	ч	14.79	16.65	21.43	10.05	, 0.0	11.14	4.04	5.53
PROPELLER 144FRS	tun x	167	168	210	117	u	330	29	
THE	M	37.20	36,34	· 20.19	41.54	0.0	30.21	89.14	65,43
× G	4	10.26	16.74	16.54	15,56	· · o	14.49	16 . 5 4	13.37
G 60 ;	n '	1.07	0.45	0.48	1.11	0.0	0.21.	0.32	1.37
GD 4	٦	14.87	12.00	10-65		0.0	21.51	72.32	. 50.49
MOYENT TO HEFT 1	DEG. HT- H	75411	74179	81104	871141	in in	10465	. 92289	73019

# ト リ ム 計 算(積付状態作画)

GENERAL ARRANGEMENT SCALE: 1/1186





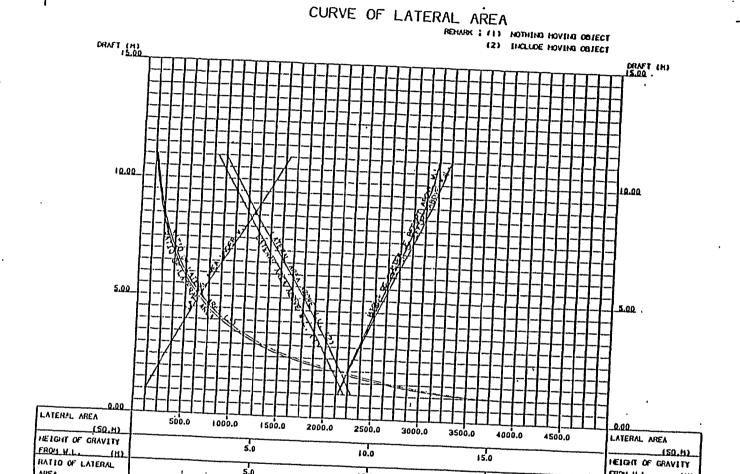
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DRÅFT	UNDER WAT	ER LINE	ABOVE WAT	ER LINE K G	AREA RATIO	ABOVE WAT	ER LINE	AREA RATEO
THI "	isoini —	—(н) ·—-	(80.H)	înî	<del></del>	(H. Ø2)	— (n) —	
1.000	194.30	0.50	1612.14	10.17	16.500	3612.14	10.17	18.390
2.000	390.23	1.00	7416.21	10.66	8.754	3416.21	10.66	8.754
3.000	586.64	1.50	3217.60	11.16	5.486	3219.60	11.10	3.486
4.000	713.16	2.01	3055.68	11.66	3.857	3022.60	11.66	3.057
5.000	980.14		2825.71	12.16	2.001	2825.71	12.16	2.651
6.000	1177.51	<u> </u>	2624.94	12.66	2.233	2628.94	1 <u>Z 1 6 6</u>	1:233
7.000	1373.78	3.51	2432.67	13.15	1.771	2432.67	13.15	1.771
ñ.000	1569.50		2236.95	13.65	1.125	2236.95	13.65	1.425
9.000	1765. 30	4 • 50	7041.14"" 1844.26	14.14	1.156	2041.14	14.14	1.156 0.940
_10.000 _11.000	1962. 19 2160. 35	<u>5.00</u> 5.51		14.64 15.14	0.762	1646.10	14.64 15.14	0.762
12.000	2359.73	5.51 6.01	1446.72	15.64	0.613	1446.72	13.64	0.613
13.000	2560.22	6.52	1246.23	18.14	0.487	1246.23	16.14	0.407
14.000	2761.47	7.03	1044.98	16.65	0.378	1044.98	16.65	0.378
15.000	2963.27		043.17	17:17-	0.285	843.17	i i . i i	0.285
16.000	3165.77	8.05	640.68	17.69	0.202	640.68	17.69	0.202
17.000	3365.54	8.56	437.50	16.25	0.130	437.50	10.25	0.130
18.000	3572,79	9.07	233.65	18.90	0.065	233.65	16.90	0.065
18.300	3634.08	9.22	172.37	19.17	0.047	172.37	19.17	0.047
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HEIGHT OF GRAVITY

RATTO OF LATERAL

FROIL V.L.



10.0

15.0

5.0

LATEA

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COND. NO.	<u> </u>	32	33	34	35	36	
	THEAVY DALL.	NORHAL BALL.		HOHOG EN EO US	GRAIN LOAD.	GRAIN LDAD.	
	JAPAN DEP."	JAPAN DEP	ALTERNATE HUASCO DEP.	NOFOLK DEP.	S.F.=4U VANCOUVER	_S.F.=55 ~¥ANCLUVER	the state of the second state of the second
	DAFAH DEFE	NALVA DEL	HONSON DIFF	MUPOCK DOTA	DEPARTURE	DEPARTURE	
	(1:0.=1)	(1.C.=3)	(T.C.=9)	11.0.=111	(Y.C.=21)	(1.C.=29)	
DISPLACEMENTANT		32 578 • 6	70965.0	70965.0	70917.0	56137.0	,
DRAFT (CORR) (M)	7,743	6.434	13.330	13.330	13.330	10.736	
DRAFT AT AP (P.)	B. 460	n.480	13.340	13.330	13.330	10.730	
DRAFT AT FP (M)	7.080	4.640	13.340	13.330	13.330	1 u. 730	
DRAFT (MEAN) (M)	1.70%	6.563	13.340	13.330	13.330	10. 730	
TPIH (M)	1.400	2.146	0.6	0.0	1.0	6.6	
KG (H)	7.670	10.760	10.200	10.190	10.240	10.420	
GM (M)	4.740	5.540	2.970	2.900	7.43U	2.760	
GGO (M)	6.770	0.600	0.260	U.260	0.120	0.220	*** ****** *** **
· GA4 (M)	3.970	4.740	2.710	2.720	2.810	2.540	
OG (M)	1.727	3.026	-3.130	-3.140	-3.090	-0.316	*
G7 HAX. (M)	3.155	3.120	1.247	1.353	1.413	2-138	I Manusana, Manusan (A.)
THETA MAX. (DEG)	42.265	42.034	30.013	30.621	40.000	40.757	
THE TA RANGE (DEG)	83.822	81.070	76.150	76.329	77.974	78.621	* Decimal property of the second seco
D.S. (M-RAD)	2.637	2.684	1.711	r.n7	1.170	1.609	
D.S. (MT-M)	164:01.3	87444.1	78817.7	79283.9	83567.4	90313.6	
D.S./01SP (M)	2.637	2.484	1.111	1.117	1.176	1.609	
FLOOD ANGLE TOUG !			,	f#-4H / n m4			
LATAHAREA (MZ)	2286.9	2543.5	1178.4	1170.4	1180.1	1699.3	
LAT.AREA RATTO	1.420	1.951	0.444	0.444	0.445	0. 78 i	<del></del>
ROLL-PERIOD IS	12.120	11.365	13.527	13.502	13.205	14.290	
REO. COSEL. N	U. 02364	0.62612	0.01985	0.01983	0.01991	0.02214	
MOLL.ANGLE (DEG)	16.090	19.930	14.814	14.838	15.056	14.616	
RESULT CI	11.995	7,965	11.644	11.644	11.665	10.313	,
ct oro							
- GZ REQ. (M) G7 44X./GZ RED			· · · · · · · · · · · · · · · · · · ·				
	1 400	1 430	1.794	1 204		1 250	
THE TA MAX /30.	1.409	1.428		1.294	1.333	1.359	
RESULT C2	0.033	0.046	0.023	0.022	0.022	0.026	
FRIGODARD AT FP	11.709	14.149	5.449	15.459	5.459	8.059	; "
FREEDCARD AT MID	16.520	11.740	4.960	4.976	4.976	7.570	
FRECHNARD AT AP	10.177	10.177		5.327	5.327	, 7.927	
RES.DISP (MT)	63098.1	70158.1	31771.1	31771.1	31819.1	46599.1	
MES.DISP/A.DISP	1.592	2.154	0.448	0.448	0.449	0. 630	

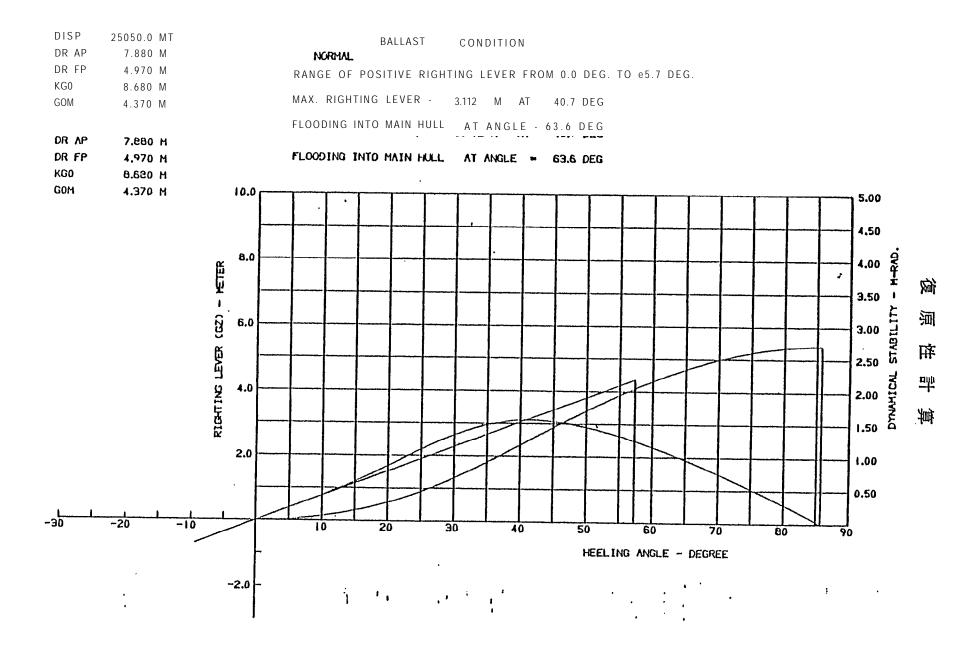
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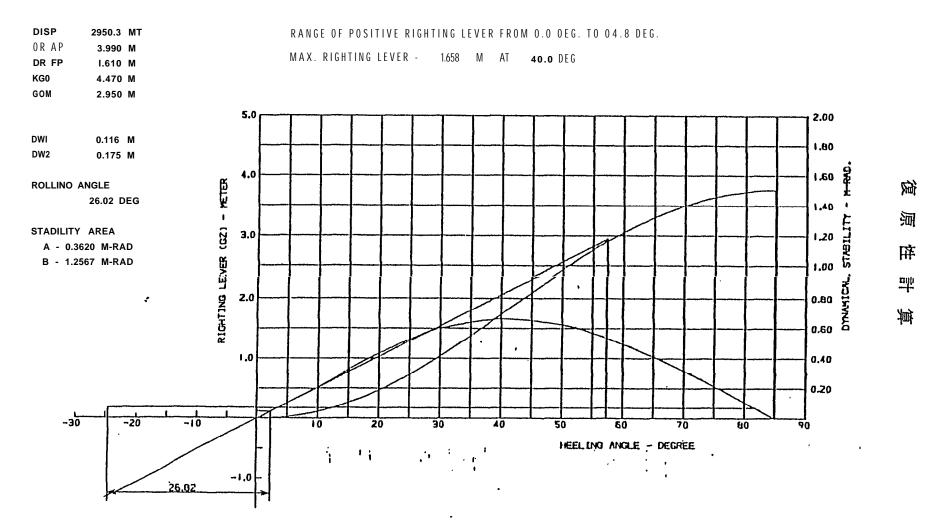
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AND E DHOS	MAL PALL.	MAGAL	DEP .				<del></del>
	S FOR STABILITY						
DISPLACEM	IENT	HT DR	AFT AT AP	4.300 H	and the second s	described appears to the garden at the	
CO30,80,780	0 KG (KGO) 11.220 0 GN (GJM) 24.120	н		··	***************************************	,	
** RIGHTING LE	EVER (GZ) ČÜRVE IN	иот туния		* * ** **			
MAY. RIGHT	POSITIVĖ RIGHTING 11 NG LEVER 5.026 FLOCOING INTO HATI	H AT 20	,63 DEG.	% .60 DEG.			
· -	C AREA FOR DYNAMI		•				
INTERNATE	AREA-2 2.795	M-RAD. FROM O M-RAD. FROM O M-RAD. FROM 30	•0 DEG 10 3	0.00 DEG.	•	•*	
HEFLING /	MGLE AT _ 42,42,1						
	PLINDA MÍTH MIND YNI					RITE **	
RIGHTING RIGHTING	LEVER AT STEADY H	IND DUL = 0.244	H (COEFF. KI	<u>* 0.0000</u>	LATERAL MI	NOAGE AREA = 3	367.08 SQ.M
ROLL ING	ANGLE = 38.82	DEG.		- 1.50003	CEACH OF R	IND JUNEAL -	70000 N
ROLL ING	ANGLE = 38 - 8 2 .1	QEG		• • • • • • • • • • • • • • • • • • • •	AOLLING PE	8100 T = 7.6	SEC. K. 18.701 H
STAPILITY STABILITY AREA RATI	ANGLE * 38,82 I  / AREA(A) 2.026 / AREA(B) 3.474 O C1* B/A * 1.187  VER AND DYNAMICAL	M-RAD. FROM-30 M-RAD. FROM O	.27 DEG. 10	0.81 DEG. 2.89 DEG.	AOLLING PE	8100 T = 7.6	SEC. K. 18.701 H
ROLLING  STAPILITY  STABILITY  AREA RATI  ++ HIGHTING LF  INCLIN - R  ANGLE	ANGLE = 38,82 ( ANGLE = 38,82 ( ANGLE = 38,82 ( ANGLE	M-RAD. FROM-30 M-RAD. FROM O STABILITY CURVE	27 DEG. TO .81 DEG. TO 7 S .27 DEG. TO 7	O.81 DEG. 2.69 DEG. DYNAMICAL STAULLITY	AOLLING PE	8100 T = 7.6	SEC. K. 18.701 H
ROLLING  STAPILITY  STABILITY  AREA RATI  ++ HYGHYING LE  INCLIN. R  ANGLE  LDEGI	ANGLE = 38,82 ( AREA(A) 2.026 ( AREA(A) 3.474 ( O C1 8/A = 1.187 (VER AND DYNAMICAL (IGHTING HIGHTING LEVER (M) 0.0	M-RAD. FROM-30 M-RAD. FROM O STABLLITY CURVE LEVER (GZ) IN (	27 DEG. 10 .81 DEG. 10 7 S H) DYNAHICAL STAP IL ITY 10.0 (M-RAD)	DYNAHICAL STAULLITY (HT-H)	VEDACTION VEDACTION	8100 T = 7.6	SEC. K. 18.701 H
ROLLING  STAPILITY  STABILITY  AREA RATI  ONLY HIGHTING LE  INCLINA R  ANGLE	ANGLE = 38,82 ( AREA(A) 2=926 AREA(B) 3-474 (O C1= B/A = 1-187 EVER AND DYNAMICAL LIGHTING HIGHTING LEVER (M) 0.0 4-32 : I	M-RAD. FROM-30 M-RAD. FROM O STABILITY CURVE LEVER (GZ) IN (	27 DEG. TO .81 DEG. TO 7  S  H) DYNAHI CAL STABILITY 10.0 (M-RAD) ————————————————————————————————————	O.81 DEG. 2.69 DEG. DYNAMICAL STAULLITY	VEDACTION VEDACTION	8100 T = 7.6	SEC. K. 18.701 H
ROLLING  STABILITY  STABILITY  AREA RATI  ++ HIGHTING LF  INCLIN. R  ANGLE  (DEG)  0.0  5.1) 10.00 12.00	ANGLE = 38,82 ( AREA(A) 2=926 AREA(A) 3-474 O C1= B/A = 1-187 EVER AND DYNAMICAL RIGHTING HIGHTING LEVER (M) 0.0 2-323 1 3-905 1 4-345 1	M-RAD. FROM-30 M-RAD. FROM O STABLLITY CURVE LEVER (GZ) IN (	27 DEG. TO .81 DEG. TO 7  S  H) DYNAHICAL SYABILITY 10.0 (H-RAD)	0.81 DEG. 2.69 DEG. DYNAHICAL STADILITY (HT-H) 0.0 1107.7 4051.3 5588.0	VEDACTION VEDACTION	8100 T = 7.6	SEC. K. 18.701 H
## HYGHTING LF    STAPILITY     STAPILITY     STAPILITY     AREA RATI     OF HYGHTING LF     INCLIN. R     ANGLE     LOEGI     O.O     5.1     10.00     12.00     15.00     20.00	ANGLE = 38,82 ( AREA(A) 2 2026 AREA(B) 3.474 O C1 = B/A = 1.187  VER AND DYNAMICAL LIGHTING HIGHTING LEVER (M) 0.0 2.32 :	M-RAD. FROM-30 M-RAD. FROM O STABILITY CURVE LEVER (GZ) IN (	.27 DEG. TO .81 DEG. TO .7  S  M) DYNAMICAL SYA9 IL ITY 10.0 (M-RAD)	DYNAMICAL STAULLITY (HT-H) 0.0 1107.7 4051.3 5588.8 8147.2 12731.1	VEDACTION VEDACTION	8100 T = 7.6	SEC. K. 18.701 H
## ## ## ## ## ## ## ## ## ## ## ## ##	ANGLE = 38,82 ( AREA(A) 2=926 AREA(B) 3-474 O C1= B/A = 1-187 EVER AND DYNAMICAL RIGHTING RIGHTING LEVER (M) 0.0 2-323 I 3-905 I 4-345 I 4-775 I 5-023 I 4-951 I	M-RAD. FROM-30 M-RAD. FROM O STABILITY CURVE LEVER (GZ) IN (	27 DEG. TO .81 DEG. TO .81 DEG. TO 7  S  H) DYNAHI CAL STABILITY 10.0 (M-RAD)	0.81 DEG. 2.69 DEG. DYNAMICAL STABILITY (MT-M) 0.0 1107.7 4051.3 5588.8 8147.2 12731.1 17385.2 21079.7	VEDACTION VEDACTION	8100 T = 7.6	SEC. K. 18.701 H
ROLLING  STAPILITY  STABILITY  AREA RATI  ++ HIGHTING LF  INCLIN. R  ANGLE (DEG)  0.0 5.1) 10.00 12.00 12.00 15.00 25.00 30.00 30.00 35.00	ANGLE = 38,82 ( AREA(A) 2-926 AREA(A) 3-474 O C1= B/A = 1-187 EVER AND DYNAMICAL RIGHTING RIGHTING LEVER (M) 0-0 2-323 1 3-905 1 3-905 1 4-345 1 4-775 1 5-023 1 4-951 1 4-664 1 3-772 1	M-RAD. FROM-30 M-RAD. FROM O STABILITY CURVE LEVER (GZ) IN (	27 DEG. TO .81 DEG. TO .81 DEG. TO .81 DEG. TO .81 DYNAHICAL .81 SYA9 IL ITY .82 IL ITY .83 IL ITY .84 IL ITY .85 IL ITY	0.81 DEG. 2.69 DEG. DYNAHICAL STADILITY (HT-H) 0.0 1107.7 4051.3 5588.0 8147.2 12731.1 17385.2 21079.7 26055.6 29804.1	VEDÁCTION VEDÁCTION	8100 T = 7.6	SEC. K. 18.701 H
ROLLING  STABILITY  STABILITY  AREA RATI  ++ HIGHTING LF  INCLIN. R  ANGLE (DEG)  0.0 5.1) 10.00 12.00 12.00 15.00 25.00 30.00 35.00	ANGLE = 38,82 ( AREA(A) 2-926 AREA(B) 3-474 O C1= B/A = 1-187  VER AND DYNAMICAL HIGHTING HIGHTING LEVER (M) 0-0 2-327 1 3-905 1 4-345 1 4-775 1 5-023 1 4-951 1 4-275 1 3-772 1 3-713 1	M-RAD. FROM-30 M-RAD. FROM O STABILITY CURVE LEVER (GZ) IN (	27 DEG. TO .81 DEG. TO .81 DEG. TO 7  S  H) DYNAMI CAL STABILITY 10.0 (H-RAD)	0.81 DEG. 2.69 DEG.  DYNAMICAL STABILITY (MT-M)  0.0 1107.7 4051.3 5588.8 8147.2 12731.1 17385.2 21079.7 24055.6	VEDÁCTION VEDÁCTION	RIOD T = 7.6: COEFF N = 0.0: GAMHA = 3.1: DELTA = 0.0:	SEC. K. 18.701 H



## CHECK STABILITY

SYOHI JYOTAI



	. <b>£</b>		GRA 1	N STABILI	TY CALCULATION	•
\$ 1GN	FOR UNIT		. 0		OIHETRIC	1:FEET
SIGN	FOR CZP COL. OF DISP		Ú		Qt ++	1: **.* 2: **.**
\$ IGN	FOR PRINT OUT OF TAB	LE				,
	GRAIN STABILITY SUMM	ARYs	1		SINDTHING	TIPRINT DUT
	ALLOWABLE HEELING HT	. 1	2		UINOTHING	1:PRINT DUT(CHANGE THE H. ANGLE) 2:PRINT GUT(CHAGE THE G
	HEELING MT. OF EACH	3F 1	1		DINOTHING	1:PRINT OUT
SIGN	FOR RULE 14	CO I	264			
S I G N	FOR CALCULATION METH	เกเว			······································	
	STABILITY CALCULATIO	N L	. 3	,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		
- · · · · · · · · · · · · · · · · · · ·	HEELING AT.	1	0		OLLOADFD LTVEL	
	GZ CHOSS CURVE	1	٥	<del></del>	OIFREE	1 LUNDER UPP.DK 21 INCLUDE SUP.STRUC.
	, , , , , , , , , , , , , , , , , , , ,					
	,	•				

	vol.		- 510	HAGE FACTO	R TOURSFYZET		~		-		
CARGO IDLD	HEFLING	40.0000					<del></del>		<del></del>		
	(H ++4)	~ ~ ~ ~ ~	TR'AI	15 V F A S E THE	ecting hohent	( HT=H-)-				<del></del>	
NO. 1 CARGO HOLD	1022	917	1194	073	<del></del>	····		<del></del>	<del></del>	<del></del>	<del></del>
NO.2 CARGO HOLD	1131	1015	996	286		<del></del>					
NO, 3 CARGO HOLD	973	075	ns 2	832				<del></del>	···	.,,	
NO.4 CARGO INICO	1054	950	927	905		<del></del>	<del></del>	***************************************		····	<del></del>
NOTE CANCOLLOCO	1132	1016	491	987		<del></del>	<del></del>				· · · · · · · · · · · · · · · · · · ·
YOTAL	5317	4771	4.654	4543			**************************************	,		····	
**************************************	· · · · · · · · · · · · · · · · · · ·	······································	············	<del> </del>	,	· · · · · · · · · · · · · · · · · · ·					
TABLEOF	VENTICAL S	INFTING RO			CED COXDED T		IGLE 15.0	pro. r			
	VOL.	MIFTING RO	- STOR	IAGE FACTO	LED COANED TY		IGLE 15.0	ore.			
YABLE OF	VOL. HEFUNG		- SYOF	IAGE FACTO		•	IGLE IS.O	pre-1	et e		
CARGO HOLD :	VOL. HEFUING HOHENT	40.000	- SYOF	IAGE FACTO	C (CUB.FT/LY)	•	IGLE 15.0	pre- 1			
CARGO HOLD .	VOL. HEFUNG HOMENT (H ++4)	40.606	- STOR	AGE FACTO	C (CUB.FT/LY)	•	IGLE 15.0	pre-			
CARGO HOLD :  YU. 1 CARGO HOLD  YO. 2 CARGO HOLD	VOL. HEFCING HOMENT (H ++4)	40.0J00 70	- STOR	IAGE FACTO IZAGOGO TCAL SHIF	C (CUB.FT/LY)	•	IGLE 15.0	pre- 1			
CARGO HOLD :  NO.1 CARGO HOLD  NO.2 CARGO HOLD  NO.2 CARGO HOLD	VOL. HEFUNG HOHENT (H**+4)	70 CJCO 70 COJCO 70 C	- SYON *1:0000 4 - VERT	TCAL SHIF	C (CUB.FT/LY)	•	IGLE 15.0	pre- 1			
CARGO HOLD - NO.1 CARGO HOLD NO.2 CARGO HOLD NO.2 CARGO HOLD	VOL. HEFUNG HOMENT (H"++4) 100	70 101	- SYON 61:0000 4 - VERT - OU 90 109	76 106	R (CUB.FT/LY)	•	IGLE 15.0	pre- 1			
CARGO HOLD - NO.1 CARGO HOLD NO.2 CARGO HOLD NO.2 CARGO HOLD	VOL. HEFUNG HOHENT (H**+4) 100 112	70 70 101 111	- SYON	18GE FACTO 12.6060 1CAL SHIF 08 98 108	R (CUB.FT/LY)	•	IGLE 15.0	pre-			
CARGO HOLD : NO.1 CARGO HOLD NO.2 CARGO HOLD NO.2 CARGO HOLD NO.4 CARGO HOLD NO.5 CARGO HOLD	VOL. HEFUNG HOHENT (H*+4)  100  112  124  105	70 50 101 111 94	- SYON (1.0000 4	16GE FACTO 16Z-60GO 16ZL SHIF 08 96 106	R (CUB.FT/LY)	•	IGLE 15.0			·	

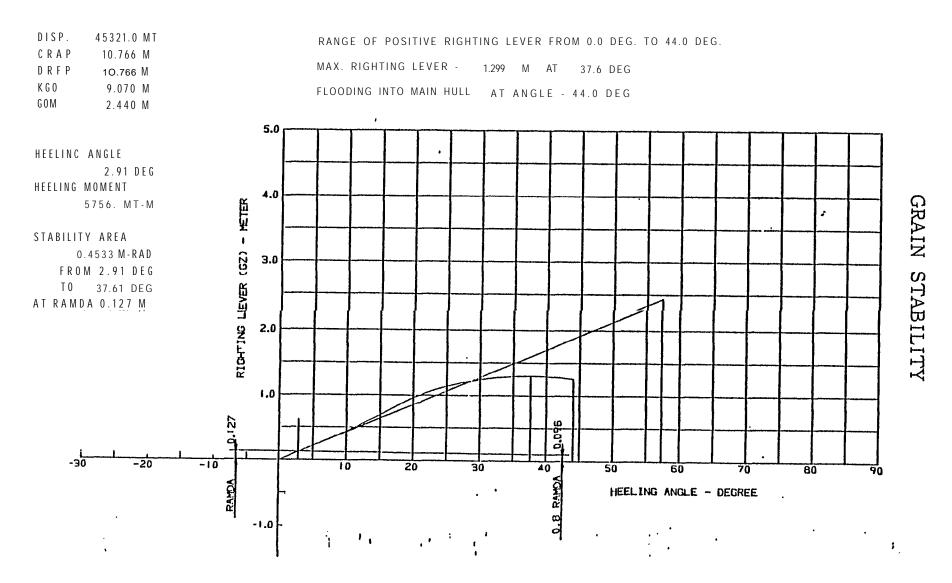
			TABLE	OF ALLO	HADLF HE	FLING H	OHINT					**			
015PT (411)	MIN.GOH	HTN. _(HT-4 )				0.40	. 0.70 0 H (H	0.00	-	1.00	1.10		CHE RESID. S.AREA (H-BAD)	HEEL.	EFFEC.
55000	0.30	0.65	6660	7577	8693	9910	1.11.74	1 23.43	13559.	1477.6_	15.997	,	0.145	12.0	35.0
<u> </u>	0,30	6326	6326	. 7565	8803	10042	11280	1 25 19	13758	. 14996	16 235		0.140	12.0	35.0
57200	<u>0,70</u>	6399	6399	. 7660	8921	101 P1	11442	1,2703	13964	15224	16485		0.135	12.0	35.0
5/1000	0,30	<u> </u>	<u> </u>	7.77.3	9.056	10139	11622_	1,290 <u>5</u>	14107_	1547 <u>u</u> _	16,753		0.124	12.0	35.0
59000	<u>(* 33</u>	. 6577	6577	7882	9187	10492	1 1797	1 31 02	14407	15712	17017		6.123	12.0	35.6
60000	0,30	6662	. 6662	7489	9316	10643	11970	1 32 97	14624	15951	1727 <u>a</u>		0.113	12.0	34.6
61000	0,30	6745	6745	8094	9443	_10793	1.41.42_	1.34.91	_14840_	16189	17539	<del></del>	0.100	12.0	33.1
62000	0.70	6437-	6832	8204	9575	10746	12316	13689	15060	16432	17863		0,085	12.0	32.4
63000	0.34	7397 ,	6074.	, 626B,	9661	11059	. 12448	1 30 4 2	15235	16679,	_18022_		0.075	12.0	31.4
<u> </u>	0.47	9385	4.901_	#397	9.11.2	11.220	1 2644.	14059	15475	16890	405 811		0.075	12.0	30.9
65202	i <u>a.</u> o	11395	. 6991	8429	. 9886 .	11364	1,2742	14179	15617	17055	•		0.075	12.0	30.4
60644	0.73_	13473.	7178	863 ji	10098	11558	1 30 17	14477	15937	17397	18857		U.U75	12.0	30.1
66360	6,70	14361	7735	8765	10176	11647		1 45 89	16060	17531	19002		0.075	12.0	36.1
67000	0.87	15781	7287	8768	10 250	11732	13214	1 46 96	16178	17663	19142		0.075	12.0	29.1
67500	0.24	16741	7350	8843	10 336	11029	1 3322		•	17801			0.075	12.0	29.6
60000	1,(3	10276	7406	8910	10414	11918	13422	14926	16430	17934	19438	····	0.475	12.0	29.4
60500	1.11	19722	7460	8975	10490	12005	1 35 20	15035	16551	16066	19581	rimeira de antiga esta esta esta esta esta esta esta est	0.075	12.0	29.1
MARK C	0M # COR	RECTED GM	HITH GO		<del>,</del>	***************************************		····							
					1	<del></del>						1	<del></del>	- <u>'-</u>	
					······································				•			<del>, , , , , , , , , , , , , , , , , , , </del>	<del>,</del>	<del></del>	<del></del>
······································						<del></del>	<del></del>	· · · · · · · · · · · · · · · · · · ·	<del></del>	<del></del>		***			
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<del></del>					<del></del>		<del></del>	<del></del> -		<del></del>	<del></del>		······································	<del></del>	<del></del>

PARTIC	LLAR OF STABILITY	· · · · · · · · · · · · · · · · · · ·	,	
	THIS SHIP	CRITERIA INCO	*** CHECK LIST ***	
DISPLACEMENT CORRECTED GM	4.736 HT 4.027 LT 2.510 H 8.235 FT	0.3 N	A 1 AREA BETWEEN 0.0 DEG.	AND 3 U. W DEG.
RESIDUAL STABILITY AREA	0.475 H-RAD. 1.502 FT-	-RAD. 0.075 H-RAD.	Q-369 M-RAD.	1.211 FT-440.
ACTUAL HEELING HOMENT	3756 MT-H 13386 LT-	·FT '	A & AREA SETTEN OF DES	AND 4 yes DEG.
MAX. RICHTING LEVER AT ANG ANGLE OF ELOCITIE	LE 37.6 DEG.		0.608 H-RAD.	1.995 FT-AAD.
RANGE OF PESIDUAL AREA	37.6 050. uilig M		C & AREA BETWEEN 30.0 DEG.	A1 AND 40.0 DEG.
THEREN DATABLE HEELING MOMENT	26283 MI-M 8+868 LT-	-F[	0.239 M-RAD.	U.784 FT-RAD.
INCLIN. RIGHTING	STATICAL RIGHTING LEV	ÆR (4.)		
	1.0 1.5 2.0 2.5 3.0	3.5 4.0 4.5 5.0		<del></del>
5.05 0.221 1 *				
10.00 0.453 [				t a managaguar y an an a a a a ga mainn a babay namag a denomer mag apara
15.12 J.7.8 I	•			
20.00 U.992 I		**************************************		
	*	rem to place y remove to their dy all the treatment page at remain sample grips, review grips, and specific treatment		
30,00 1,313:1	*			
	1			
40.20 1.309 [		•		
44.52 1.3+> [ _44.52 0.0 [+	IN FLOW FOIN	Τ		
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		SUMMARY T	ABLE OF GRAIN	STABILITY CALC	ULAT ION		
CONDITION	ıs	GRAIN LOAD (S.E.= 50)	GRAIN LOAD (S.F.= 52)	GRAIN LOAD	GRAIN LOAD		,
(ITEH)	(UNITS)	VANCOUVER DEP.	JAPAN ARR.	YANGOUYER DEP.	JA PAN A RR •		
S.G. OF SEA WATER	(HT/CUB.H )	1.0250	1.0250	1.0250	1.0250		
STOWAGE FACTOR	(CUB.FT/LT)	50.0000	50.0000	55.0000	55.0000		
DISPLACEMENT	( HT )	59759	, 59355	56137	55813		
CORRESPONDING DRAFT	( H )	11.38	11.31	10.73	10.68		
LCG	( H )	-6.41	-6.45	-6.65	-6.67	<u> </u>	<del></del>
L C 8	( H )	-6.06	-6.09	-6.33	-6.35	*	
HTC	(HT-H )	734.6	732.7	717.8	716.3		
TRIM	( H )	-0.28	-0.29	-0.25	-0.25		<u> </u>
L C E	t H )	-1.49	-1.58	-2.23	-2.30		······
DRAFT AT FP	( M )	11.52	11.45	10.86	10.80	·····	·····
AT AP	( H )	. 11.24	11.16	10.61	10.55	<del></del>	<del></del>
AT HID.	( H )	11.36	11.30	10.73	10.67		······································
T_KH	( H )	13.65	13.06	13.12	13.13		<del> </del>
K G	( H )	10.33	10.44	10.36	10.49		
GGO	( M )	0.15	0.17	0.22	0.23		
GOH	( H )	2.54	2.45	2.54	2.41		
HETLING MT. TO 1 DE	G. (MT-H )	2650.0	2539.0	2490.0	2349.0		<del></del>
ACTUAL HEELING HT.	( K-TM)	5474	5474	4977	4977		
ALLOWABLE HEELING N	T. (4T-H )	36031	34617	33 94 9	32163		•
HEELING ANGLE	(DEG.)	2.0	2.1	2.0	2.1		<del></del>
RESIDUAL STABILITY	IH -RAD.)	0.689	0.670	0.726	0.698	•	
ANGLE DE FLODDING	(DEG.),	49.8	53.1	52 .2	52.4		
HAX RIGHTING LEVER	L_H1_	2,00	1.96	2.1.6			

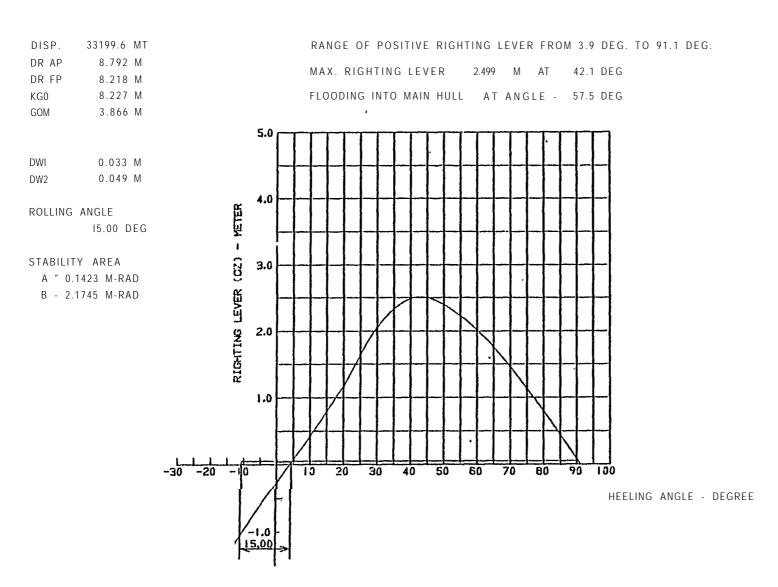
	OF EACH CO		SSUMED KG = 11.000TH 1			· · · · · · · · · · · · · · · · · · ·
CONDITION	(HT)	STOWAGE FACTOR ASSUM	GZ VALUES (H ) .			
	•	TCUR.FY /LT) 12 DEG.	10 DEG. 12 DEG. 40 DEG.			
RAIN LUAD	<del></del>			-4,	<del></del>	
(S.F.= 40).	70765	40.0000-0:479-	0.798 0.6UB 1.397			
VANCOUVIR DEP.		A		•		
	<u> </u>		0:000 0:000 0:000			
			. <del>- 1 - 1 - 2 - 2 - 2 - 2 - 2 - 2 - 2 - 2 </del>	•		
				***	B//ED	ATUD A MERLUIC ANCIES
<del></del>	<del></del>				(X IP	COUP A MEELING ANGLE E.
TABLE OF	TATULE TATE	Sverse hefung h	SHEAT OF EACH CONDITION	- (HEEE') NC ANGLE 'Y	5.0 DEG.1	
		TS 1 MT-M ) GRAIN COAD	CA ATH LUAR	GRAIN LUAD	URT LOAD.	
	VOL.	15.F.= 401	(5.F. + 4)		(ALTERNATE)	
Cree 7 400	HEELING	VANCOLNER	VANCOUVER	JA PAN	JAPAN ARR.	
	HOHENY (H ++4)	nep.	DEP.	ARR.		
0.1 tage into .	1022	7415	0 873	673	4707	
D.2 CARGO HOLD	1131	1015	0 1022 0 966	1085	+ 2830	
O. 2 CARGO INILD	913	1131	0 1131	1131	1557	
ID.4 CARGO HOLD	1059	973	0 905	, 973 905	\$73 0	
	·	16.59	3 Y059	1059		
D & CARCO MOAD	1132	1016	0 967	1132	1011	
		1014	0 . 967 0 1132	967 1132	<u> </u>	AS HOLD X
ID-5 CARGO HOLD	1132	3172	U 1132		15 10 ⊀	- EPR HEELING MIT.
	1176	1057	0 1007	1007		
ID.6 CARGO HOLD				1007 1178 6517	4 956 4- 9105	. FIZ VOL. HEELING NT.
O. A CARGO HOLD O. 7 CARGO HOLD TOTAL REPARK	1178 7627	1057 1178 9342	0 1007 0 1176 0 6517	1178	958-4-	
O. A CARGO HOLD O. 7 CARGO HOLD TOTAL REPARK :	1178 7627 ARYLY FILLED	1057 1178 9342 СОИРЛЯТИРИТ (ИРЕ	0 1007 0 1176 0 6517	1178 6517	♦ 958 <del>/</del> 9105	
O. A CARGO HOLD O. 7 CARGO HOLD TOTAL REMARK :	1178 7627 ARYLY FILLED	1057 1178 9342 СОИРЛЯТИРИТ (ИРЕ	0 1007 0 1176 0 6517	1178 6517	♦ 958 <del>/</del> 9105	

(B-1) HOMO. FULL D-10.763 S.V. DEP.

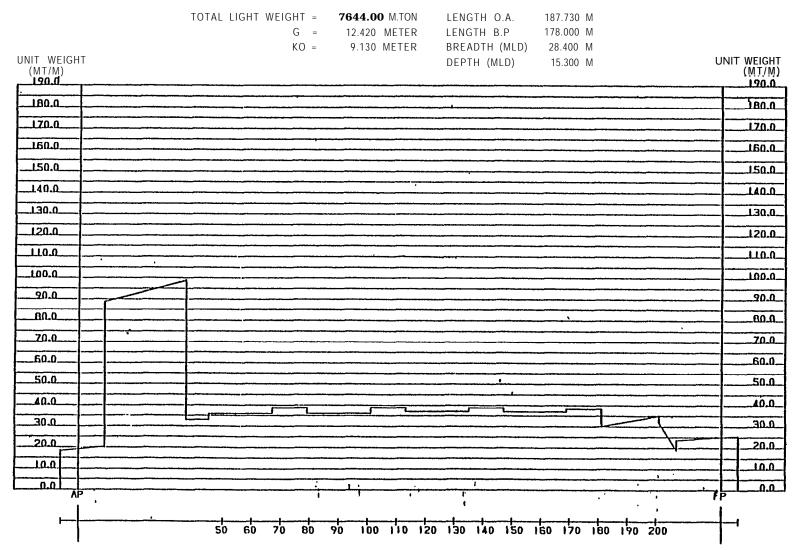


ÄELAT	IONS BETH	EEN WATER	LINE ANI	D DECK .	SIDE LINE .	IN FINAL S	TAGE .	
		DISTANC	:E	PRAFT	HEIGHT FRO	H_ DRAFT	HEIGHT FRO	н
- N	O STATION	FROY AP	,	AT CL	WE TO DK O	IL AT SL	WL TO DK S	L
	1 -0.258	(!!) -8 • 25		_ (M) 21 • 651		(H) 21.65	(H) 1 6.252	
	2 0.0	0.0		- 4		11111		
·	3 (.506	16.63		21.706	6.197	21.70		t todiada e i traria manamanataratar i ma i i i i i i i i i i i i i i i i i
	4 1.000	?2.00			6.160	21.74		
• • •	5 1.500	41.00		21.779	6.124	21.77		E DE NER WITHER REGERMANDEN PART OF THE RESERVE A TRANSPORT OF THE PROPERTY AND THE PROPERT
	6 2.000	44 .00		21.416	6.087	21.81		
	7 2.500	80.00		21.852	6.051	21.85		
	0 3.000			11.889		21.08		
	9 3.56	112.0		21.925	5.978	21.92		to the state of the common manual district the control of the cont
. 10	0 45500	128.00	0 2	21.962	5.941	21.96	2 . 5.63B	DE TOLOGISTUS DE DOS DE EMPLOYETES D. DE TOLOGOURS (MODES SINGE ) 193 3/2/2 D. 1520 5 10000 de Quant
1	1 4.500	144.00	α :	21.908	5.905	21.99	5.002	
1	25. 000	160.00		42.035_	5.860	22.03		
1		176.00		22.071	5.832	22.07		
. 1				22.100.	5.795	22. 10		THE RESERVE STREET STRE
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19		272.00	0 7	22.290	5.613	22. 29		
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5		3(4.6)		22.363	7.060	22.36		•
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	ANGLE AT	MATCH DEC	K 8066 (	nthrtn		TER LINE	16.010 DEG.	•
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RELAT	IONS STHE	EN WATER.	LINE AND	INFLO	POINT IN	FINAL STA	GE .	
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	BILITY			<u> </u>	***************************************
DISPLACEMENT	35600. HT	DRAF	TATAP 1	0 .809 M	
CORRECTED KG (KGO	1 10.240 M 1 10.560 H	DKAF	T AT FP 1	p-80A_H	
• TRIGHTING LEVER CZY	CURVE INFORMA	ION	***************************************	<del></del>	
RANGE OF POSITIVE MAX.RIGHTING LEVE ANGLE OF FLOODING	R. 2.876 H	AT 40.6	1 DEG. 10 9 7 DEG.	8.91 DEG.	
+ STABILITY STUDY AT	FINAL STAGE AFT	ER FLOODING			CHECK MITE
RIGHTING LEVER AT FIGHTING LEVER AT ROLLING ANGLE =	T COST WIND T				LATERAL WINDAGE AREA = 1056.74 SQ.M LEVER OF WIND NOMENT = 10.000 M
STABILITY AREA(A STABILITY AREA(I AREA RATIO C3= D/ AREA RATIO C3= D/	2.360 M=RAL A = 0.0		4 DEG. TO 1 5 DEG. TO 9	6.95 DEG. N.79 DEG.	
INCLIN. RIGHTING	RIGHTING LEVE	(GZ) IN (H)		DYNAMICAL	
ANGLE LEVER	0.0		4.0 [M-RAD]	CHI-H)	
(DEG) ≠ (M)					
6.0' -2.443		1	0.0	0.0	
0.0 -2.443 5.00 -1.781	<u>i</u>		0.0	c.o	
0.0 -2.443 5.00 -1.781 . 10.00 -1.079 15.00 -0.316	i		0.0	0.0 0.0	
0.0 -2.443 5.00 -1.781 10.00 -1.079 15.00 -0.316 20.00 7.537	. 1		0.0 0.0 0.0 0.0 0.0141	0.0 0.0 0.0 500.3	
0.0 -2.443 5.00 -1.781 10.00 -1.079 15.00 -0.316 20.00 7.537 25.00 1.497			0.0 0.0 0.0 0.0 0.0141 0.1631	0.0 0.0 0.0 500.3	
0.0 -2.443 5.00 -1.781 10.00 -1.079 15.00 -0.316 20.00 7.537			0.0 0.0 0.0 0.0 0.0141	0.0 0.0 0.0 500.3	
0.0 -2.443 5.00 -1.781 10.00 -1.679 15.00 -0.316 25.50 5.537 25.50 1.497 30.60 2.776 35.00 2.775 40.00 2.074		•	0.0 0.0 0.0 0.0 0.0141 0.1031 0.2695 0.4899 0.7354	0.0 0.0 500.3 3671.6 9595.4 17441.5 26178.6	
0.0 -2.443 5.00 -1.781 10.00 -1.781 15.00 -0.316 20.00 7.537 25.00 1.497 30.00 2.776 35.00 2.775 40.00 2.037			0.0 0.0 0.0 0.0 0.0141 0.1031 0.2695 0.4899 0.7354	0.0 0.0 500.3 3871.6 9595.4 17441.5 26178.8	
0.0° -2.443 5.00 -1.781° 10.00 -1.079 15.00 -0.316 20.00 7.537 25.00 1.497 30.00 2.776 35.00 2.725 40.00 2.837 50.00 2.677		•	0.0 0.0 0.0 0.0141 0.1031 0.2695 0.4899 0.7354 0.9853 1.2264	0.0 0.0 500.3 3671.6 9595.4 17441.5 26176.6 35075.2 43660.4	
0.0 -2.443 5.00 -1.781 10.00 -1.781 15.00 -0.316 20.00 7.537 25.00 1.497 30.00 2.775 40.00 2.074		•	0.0 0.0 0.0 0.0 0.0141 0.1031 0.2695 0.4899 0.7354	0.0 0.0 0.0 500.3 3671.6 9595.4 17441.5 26178.6 35075.2 43660.4	
0.0 -2.443 5.00 -1.781 10.00 -1.079 15.00 -0.316 20.00 7.537 25.00 1.497 30.00 2.776 35.00 2.775 40.00 2.074 45.00 2.074 55.00 2.677 50.00 2.130		•	0.0 0.0 0.0 0.0 0.0141 0.1031 0.2695 0.4899 0.7354 0.9853 1.2264	0.0 0.0 0.0 500.3 3671.6 9595.4 17441.5 26178.8 35075.2 43660.4	



# SNO. ZZZZ LIGHT WEIGHT DISTRIBUTION CURVE



L/W 分格

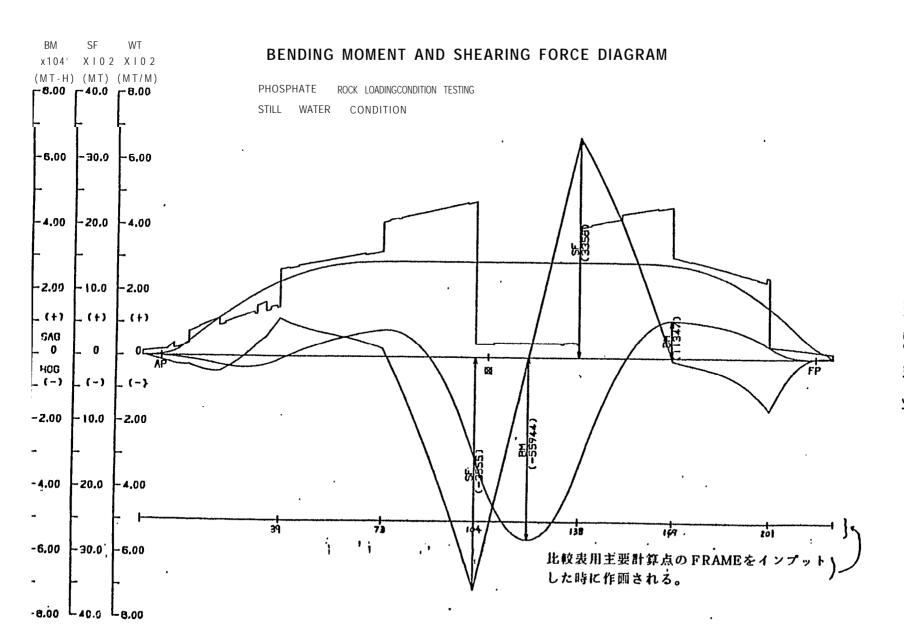
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COND 1 FULL STILL WATER CO	DADILIAN	DR AJ_P.G. PEP.						
MAYF LEI	HGTH = 0.3 M DH=[GHT = 26]	1181611 = 0.0 M	CREST = 0.0  TAP = 20.5004	H FROH AP				
0151	PLACFYENT = 204	TRO. YT DRAFT A	T EP . 19.8944					
\$9FI • 32 YAM WWO	CIFIC 6 • 1.02:	50 <u>at 144.300h. Fadh</u>	AP . +PEANS UPOD	NUW 68				
UPP HAX SF =		AT 209.4004 F10H						
\$AG_4AX_54		AT 212,4014 FROM						
HOG HAX SH =		AT 177.479H FROM		ING BH				
	DISTANCE DRAFT	TOTAL WEIGHT	BURYANCY ", SF	nH .		FLECTION		
NO ESTO DIFE	EROH VP (III)	HEFT) (ALOHI)	(\$9.H) (HY)	(717-11)	[SF/H]	100/01	TOTAL	
	<u></u>			·	-0.001	0.006	0.005	
2 -10 0.0 3 -9 0.00L	-7.500 23.515 -4.749 20.515				~0.0() -0.00l	0.005	0.004	
4 -0 0,350	-4.400 70.513	54.317 56.317	12.355 -41	. 1-1.7.	-0.001	0.005	0.004	***************************************
5 -7-0,001 6 -5-0,357	-5.251 20.510 -4.571 27.5 A		27.75A -125	-106.	-0.001	0.004	6.563	
7 -5-0.000					-0.000	0.003 0.663	1.002	
A -3 0.375	-1.875 20.504	A7.479 A7.479	44.930 -206	-746.	~0.000	0.061	0.001	
	<u>-1.400 20.503</u> 0.0 20.500				-0.0CG	0.001	0.001	The second secon
11_2-0.250.	1.250 20.400	A6.438 46.438	AA.717 -226	-1454.	0.000	-0.001	-0.001	
12 2 0, 1	1.5'1 27.498	67.44 66.746 67.477 67.472	67.693 -225 77.714272		0.000	-0.001 -:.7:2	-0.001	
14 4 0.700	3.200 20.444	ልባልበፋኝ ልለልበፋኝ	70.334 -213	-1 ms.	0.000	-0.032	-0.002	
15 2 0 250 16 6 0 0	**DEC 50.493		83.423 <u></u> -203 		0.20	~0.003 ~0.003	-0.0(1	
16 6 0.0 17 6 0.200	4.500 20.492 4.600 20.491				0.000	-0.003 -0.003	-0.003	
10 9-0.350		140.757, 149.367	106.204 -277	-2535.	0.000	-0.005	-0.004	
19 10 0 0 20 11-0.250	1.101 20.405	_112.271112.271 117.497 113.497			0.000	-0.065 -0.005	-0.005	
2114_0.0	<u> 10.562 20.400</u>	120.931 _ 96.302	145.275122	-1273.	_ 0.60%	-0.0GA	-0.007	<u> </u>
72 16 0.0 23 17 3.3		107.078 102.078 124.925 145.911			0.061	-0.00°C-	~0.0ca	
24 17 0.0:0	12.000 29.476	141.754 146.254	154.005	-3:37.	6.161		-7.049	·
25 20 0 10		181. 282 203.574		3G7A	0.000		-0.011	
0.100 15 45 	17.640 20.467	210,760 210,360 615,122 <u></u> 221,236	254.936 254 264.240 355	2860. -2317.	0.000	-0.017		
	19.200 20.464	232.014 232.014	311.764 407	-1 651.	0.000	-0.014	-0.014	
20 24-0.400		235.541 235.541 237.389 237.309			0.000	-7.014		
31_26_0.490	20.550 20.561	2524.762_ 242.763	337.194 634	- 762	0.000	-0.015	-0.015	
97 79-0.000 33 70 0.0		257.130 252.130 264.119 264.119			-0.0(0		A10.C-	
		274.726 274.726	413.090 1238			-0.019	-0.019	
35 33-2-20	26.702 23.455	· 201.905 293.176	430,120 1407	<u> </u>	-0.001	-0.070	-0.020	
		231.417 361.417				-0.020 -0.021	-0.021	
24 35 0.450	28.953 20.445	330.204 330.204	464.544 1738	A634.	-0.001	-0.021	-6.073	*
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41 27 0.400	20.7(0 27.442	336.617 336.617	478.152 1879. 490.914 2071		-0.002	-0.022	-0.023	
42 39-0,100	37.010 20.440	357.927 357.977	510.271 2236	. 34004.	-0.002	-0.024	-0.026	
45 42 0.400	_33.0 <u>00</u> ZQ.430_ 35.200_20.434	<u> </u>	524.984 2401. 556.895 2745.			-0.074		
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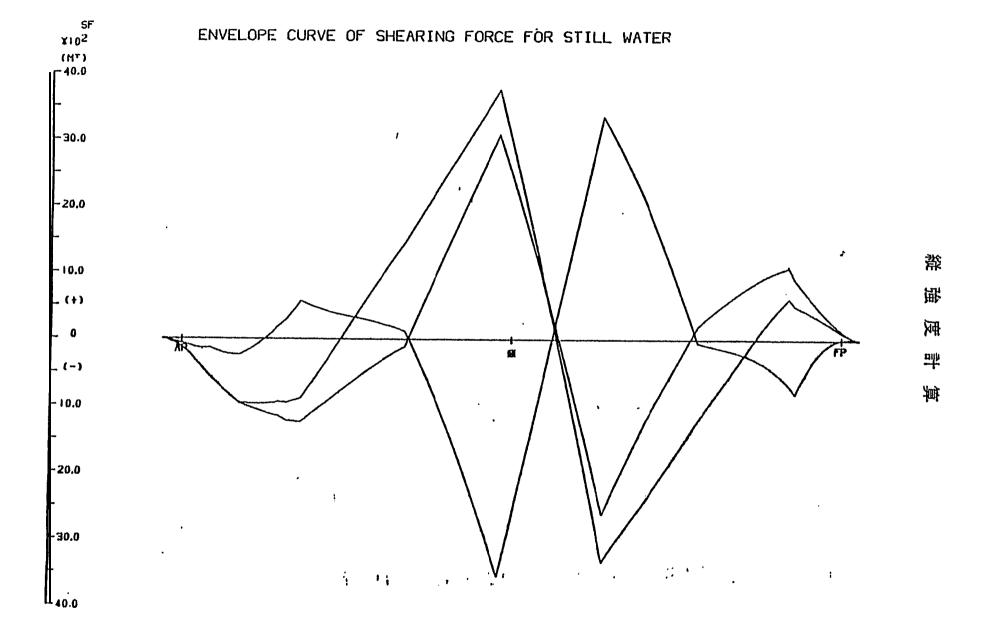
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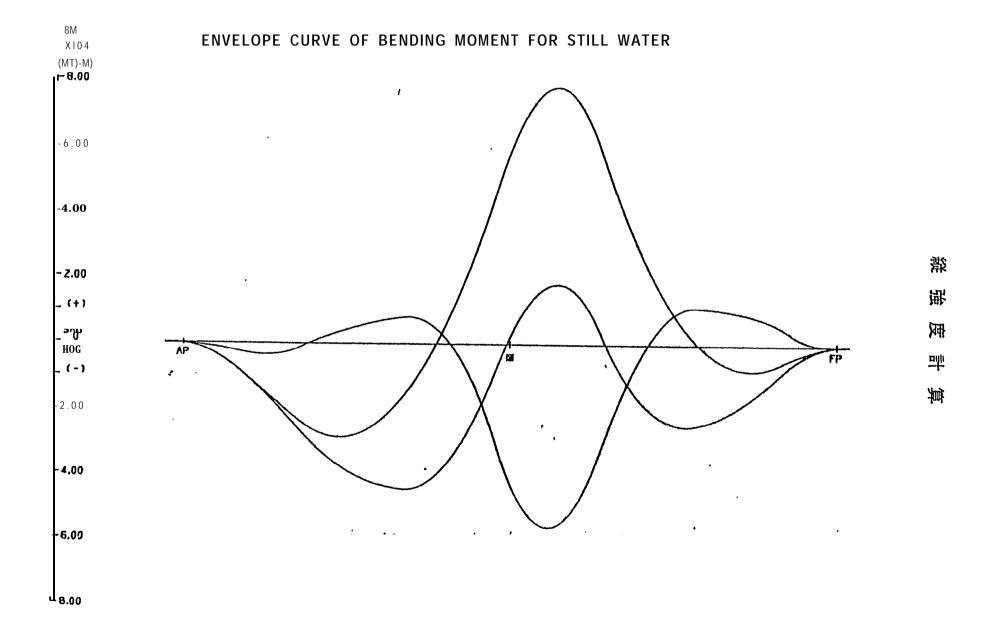
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		[H-IN]		<u> [HI-Y].</u>	<u>[#</u> ]) .	THI-HI	
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4 -9 0.350 -6.400	-61,	-47.	3.	;:	-A.		
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34 4 0.200 3.200 15 5 0.2504.000_	-213.	-1005.	144.	545.	-463	-1402	
16 6 0.0 4.500	203 -194.	2057. -2151.	172•- 107•	474.	575.		
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10 9-0.750 6.400	-207.	-7:35.	253.	316	-727	-3694	
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76 71 0.100 16.000	254.	-2400	767,-	337^+ 4429.	-1457. -1783.	-13724. -15446.	
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28 25-0.300 19.200 29_21.0.275_19.225	402.	-1651.	1000	8667.	-2213.	-21831.	
30 24-0.430 23.000	154	-1786. -1237.	1:51 1674.	7202•_	72P6. 7325.		
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37 74-0-00 73.700 000.45 0.0 06 65	797.		1 765.	12065.	-2444	-79110.	
74 72-1.2.1 23.600	1230	1045 3642.	1437 1509.	]449] •_	2915. -3163.	34113.	
75 74-0-400 76-760 76 74-0-400 77-800	1407	5 195	1757.		3136.	-30075. -42440.	
74 34-0.400 27.200 37 34 0.400 28.000	1481.	5010.	1754.	39500.	-3416.	-44237.	
38 35 0.450 29.9*0	1599 1738.	7696 8634.	1 A3A 1 941 .	21017.	3544.	47621.	
_79_,76-0.000 _ 29,403	1403		1091. ;	23697-	-3677.	-50461	
47 36 7:457 29.857	1477.	10260.	2041.	246031	-3646.	-52142. -53656.1	file and the second second second second second second second second second second second second second second
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CAL. POINT DISTANCE	SF (HT)	SF LATA	SF (HT)	SF	SF					<del></del>	
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64 37 Wast 29.685	-4057.	711	695.	730.				<del></del>			
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1+2 107 6-0 05-600 160-125-3-0 12-200	29770	-75.	-1458.	2980.	3302.	• •			***********		·
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12111. 121121. CONJUNCS. COLUMNS. CAL. PURIL MINES PARCE PURIL MIN	6H 4 nl - H) - 32 27 s - 740 47 s	##NI **  ##N	BM BM 4NI-M1 -3230 4.400	10 8H (HI-H) -2939,	11 6M 1H7-H1 22927a 2784a	•					
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COND.113.  CAL. PJINI JISTANCE  AA PORT FROM AP  AA PORT FROM AP  AA PORT FROM AB  AA PORT	6M 4 nl - Ml - 476: 0 740 47 62144 62144 6223 6023 -	BH (MI-M)  -32,77  139-57  196-52  19972  11-1005	BH BH -3230 6,600 746x -13926 -24083 -14784	10 8H (HI-H) -2932, -47147, -62265, -40788,	11 						
COND.HO.  COND.HO.  COND.HO.  CAL. PJINT JISTANCE  NO EMNO DIFF F309 AP  A	5M 4 m = m) - 32 v 7 s - 476 · 0 s - 700 v 7 s - 6100 v 9 s	BH (MI-M)  -32,77  139-57  196-52  19972  11-1005	BH BH -3230 6,600 746x -13926 -24083 -14784	10 8H (HI-H) -2932, -47147, -62265, -40788,	11 BH -4HT-H1 -22927a -2786a -4446a -5221a	•					



**辩 強 庚 計 算** 





### R5J530 DATA FOR SIMPLE CAL. OF LONGITUDINAL STRENGTH DSLA 1 O UNIT OF DUT PUT **HETRIC SYSTEM** DSLA 2 1.02500 DSLA 3 3.00 27.00 1.00 -3.00 3.00 DSLA 4 8 4 14 15 16 17 18 19 21 22 23 24 25 26 27 28 29 30 31 32 DSLB 1 33 34 9 10 0 0 0 0 0 0 0 0 0 0 0 0 0 0 DSLB 2 0 0 0 0 0 0 0 0 0 0 0 0 0 0 DSLB 3 DSLB 4 0 0 0 0 0 0 0 ٥ DSLC 1 0 0 0 0 0 0 0 0 0 0 0 DSLC 2 0 0 0 0 0 14100. 415690. -415690. DSLD 1 61 0.0 13080. 577540. -577540. DSLD 2 66 0.0 DSLD 3 70 0.0 14490. 739380. -739380. 901230. -901230. DSLD 4 74 0.0 13280. 940900. -940900. DSLD 5 78 0.0 13280. 13280. 980560. -980560. DSLD 6 82 0.0 13280. 1000390.-1000390. DSLD 7 86 0.0 13280. 1000390.-1000390. DSLD 8 90 0.0 13280. 1000390.-1000390. DSLD 9 94 0.0 13280. 1000390.-1000390. 98 0.0 DSLDIO 13900. 953160. -953160. 14490. 763680. -763680. DSLD11 102 0.0 DSLD12 106 0.0 DSLD13 110 0.0 13900. 574200. -574200. DSLD14 114 0.0 13280. 384720. -384720. DSLD15 119 0.0 11580. 164580. -164580. DSLD16 0.0 ٥. 0. ٥. 0 0.0 DSLD17 0. ٥. 0. 0. 0. DSLD18 0.0 ٥. DSLD19 0.0 0. 0. 0. DSLD20 0 0.0 ٥. ٥. 0. 0 0.0 DSLD21 0. 0. 0. 0.0 DSLD22 0. ٥. 0. 0.0 **DSL023** ۵. 0. 0. 0 0.0 DSLD 24 ٥. 0. 0. DSLD25 0 0.0 0. 0. 0. ; DSLD26 0 0.0 0. 0. ٥. DSLD27 0 0.0 0. ٥. 0. DSLD28 0.0 0. ٥. ũ. 0 0.0 ٥. 0.

0.

0.

DSLD29

DSLD30

0 0.0

\*\* PRINT OUT OF APPLICATION DATA \*\*

••	重量分	配係数									
. COEFFICINT	DF WEIGHT	DISTRIBUT	ION FOR SHE	AR ING FOR	CE ++		······································		<del></del> <u></u>	-	
2PAY 2PAT	FRAME NO. (_ 57.000)	FRAME NO.	FRAME NO. _L.63.0001_	FRAME NO.	FRAHE NO.	FRAME NO. [_81.000)	FRAME NO. . 87.000 ]_	FRAM: NO. [ 93.000]_	FRAME NO. (	FRAME NO.	
PPAT PETAV MIRO.	0.00	0.00	0.30	0.00	0.0q	0.00	0.00	0.00	0.00	0.00	
NO. 1 FRESH WATER T.	0.00	. 2.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	
_NT.2_FRESH_WATER_T OIST. WITER T. 1P/S1	0.0?	0.30	0.03	:-		0.;co	0.00	0.00	6·66	0.00	<del></del>
40.1 FUEL OIL T. 151	0.00	0.00	0.01	0.00	0.00	0.03	0.00	0.00	0.00		
43.2 FIEL MIL THIP!	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.03	
_ 47.112 F.O.S.T	0.00	0.07	0.00	0.33	0.00	0.00	0.30	0.00	0.00	0.00	
TESTE THE TANK	7.07 1.00	0.00 L.GG	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	
47. 1 5.9.7. (C)	1.00	1:00	1.03	<u>1.0;</u>	1.00	1.00	1.22	1.10	1.00	1.00	
40. 1 2.7.7. 10/51 .	. 1.07	_ i.00 _	i.oo _	1.00	1.00	1.00	1.00	1.00	0.45 0.37	0.00	
47. 2 C.O.T. (C)	1.00	1.00	1.00	1.00	1.00	1.00	1.00	0.00	0.00	- 0.00	
_42. 1 C.D.F. (*/S)		1.00	1 .00	1.00	1.00	1.00	1.00	0.00	0.00	0.00	
_ M3. 3 4.9.1.     (C) YC1_C.1.1.     (P/S)	1.02	1.02	1.00	1.00	1.00	0.49	0.00	0.00	0.00	0.00	
17. 4 C.O.T. (C)	1.00	1.00	1.00	l.07 0-50	0.00	0.00		?.^;}	0.00	0.00	
_ 47. 4 C.O.T[P/S]	1.02	1.00	1.00	0.51 _	0.00	0.00	0.00	0.00	0.00 0.00	0.00 0.00	
47. 5 C.O.T. [C]	1.00	0.69	0.00	0.00	0.00	0.00	0.00	0.00	0.00	0.00	
<u> </u>	1 • 00	1 • 20	0.00	0.00	0.00	0.00	0.00,	0.00	0.00	0.00	
SLOP TANK (P/S) AFT PENK TANK	1.00	0 <u>.00</u>	0.03	0.07	0.00	0.00	0.00	0.00	0.00	0.00	
						0.00	0_00	0.00	0.00		
Le n	M. 第出用	<u>LEVE</u>	<u>K</u>								
,											
,						<del></del>					
		ENT 10 DE	777 V 77								
•• LEVES -CO BI		ENT IN HE	TER UNIT +	·							
** LEALS & Cd B	ENDE NO HOH	FRAME NO.	FRAME NO.	FRAME NO.	FRAME NO.	FRAME NO.	FRAME NO.	FRANC NO.	ERANE NO.	En ang Up	
** LEALS & Cd B	ENDE NO HOH	FRAME NO.	FRAME NO.	FRAME NO.	. FRAME NO.	FRAME NO. .1_61.000]_	FRAME NO.	FRAME NO.	FRAME NO.	FRAME NO.	
ON LEVER FOR BI	ENDING MON FRAME NO. 1.57.0001	FRAME NO. 1.59.0001	FR AME 40. [ 63.000]	FRAME NO. 1.64.0001	L_L_75.030 <u>)</u> _		.(_87.000)_		1_99.000)_	FRAME NO. (104-000)_(	
BPAP POTENTIAL MILES	ENDING MON FRAME NO. 1.57.0001	FRAME NO. 1.59.0001	FRAME NO. _[_ 63.000]_ 0.00_	FRAME NO.	L_L_75.030)_ 0.00	0 , 00	.(_ 87.000)_ 0.00	0.0001_ 0.00	1_99.000)_ 0.00	0.00	
•• LEVER FOR BI	ENDING MON FRAME NO. 1.57.0001	FRAME NO. 1.59.0001	FRAME NO. _1_ 43.000)_ 8.00_ 0.00	FRAME NO. _1. 69.000] 0.00	0.00 0.00 0.00	0.00, 0.00	.(_87.030)_ 0.00	93.0001_ 0.00	4_99•000)_ 0.00	0.00 0.00	
BPAP POTENTIAL MILES	ENDING MON FRAME NO. 1.57.0001	FRAME NO. 1.59.0001	FRAME NO. _[_ 63.000]_ 0.00_	FRAME NO. 1. 69.0001 	0.00 0.00 0.00	0.00 0.00, 0.00,	.(_ 87.030)_ 0.00 0.00	0.00 0.00 0.00 0.00	1_79.000)_ 0.00 0.00	1.04.000)_( 0.00 0.00 0.00	
TAIN HAME  DATIN WATER THANK HILL FRESH WATER T. HILL STEEL WATER T. DIST. WATER T. [P/5] HILL FREE TO IT.	ENDING MOM FRAME NO. 1.57.0003 	FRAME NO. (.59.900)	FRAME NO. _[_ 43.000]_ 0.00 0.00 0.00	FRAME NO. _1.69.0001 0.00 0.00 0.00	0.00 0.00 0.00 0.00	0.00 0.00, 0.00, 0.00	.(_87.030)_ 0.00	93.0001_ 0.00	4_99•000)_ 0.00	1.04.000]_[ 0.00 0.00 0.00	
## LEVER FOR BI  TAIN HAME  DRING HATER TAME  17.1 FRESH HATER T.  17.2 FRESH HATER T.  17.3 FREE DIL T. (875)  17.4 FREE DIL T. (8)  17.6 FREE DIL T. (8)	ENDING MON FRAME NO. 1.57.0003 	PRAME NO. 4.59.001	FRAME NO. 1. 43.000) 0.00 0.00 1.72 0.00 0.00	FRAME NO. 1. 69.0001 0.00 0.00 0.00 0.00 0.00 0.00	0.00 0.00 0.00 0.00 0.00	0.00 0.00 0.00 0.00 0.00 0.00		0.00 0.00 0.00 	4_79.000)_ 0.00 0.00 0.00	1.04.000)_( 0.00 0.00 0.00	
TAIN HAME  DRIN HATER TANK  10.1 FRESH HATER T.  10.2 SRESH HATER T.  DIST. WATER T. (P/S)  HILL RIFE OIL T.(F)  NO.1 EVEL OIL T.(F)  NO.1 EVEL OIL T.(F)  NO.1 EVEL OIL T.(F)	ENDENG MON FRAME NO. 1.57.0001 	FRAME NO. 4.59.200]	FRAME NO. [ 43.000] 0.00 (-71 0.00 0.00 0.00 0.00	FRAME NO. 1.64.0001 0.00 2.27 0.00 0.00 0.00 0.00	0.00 0.00 0.00 0.00 0.00 0.00		0.00 0.00 		1.77.000)_ 0.00	1104-000]_[ 0.00 0.00 0.00 0.31 0.06	
TAIN HAME  DRING HATER TANK HILL RESIGNATION DISTON WATER TO (P/S) HILL RUSE OIL TOIL HILL ROSE OIL TOIL HILL ROSE OIL TANK HILL ROSE OIL TANK	ENDING MOM FRAME NO. 1.57.0003 	FRAME NO. 4.59.3001	FRAME NO. 1. 43.0009 0.00 (.21 0.00 0.00 0.00 0.00 0.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00, 0.00, 0.00 0.00 0.00 0.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0		0.00 0.00 0.00 0.00 0.00 0.00 0.00	0.00 0.00 0.00 0.31 0.00 0.00 0.00	
TAIN MAME  DRIN MATER TANK MOLE FRESH MATER T. MOLE STESH MATER T. DIST. MATER T. (P/S) MOLE FIEL OIL T.(P) NOLE F.O.S.T.	ENDENG MON FRAME NO. 1.57.0001 	FRAME NO. 4.59.200]	FRAME NO. 1. 63.0003 0.00 0.70 1.73 0.00 0.00 0.00 0.00 0.00 0.00 22°.65	FRAME NO. 1. 69.000] 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 19.05	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.31 0.06 0.00 0.00	
TAIN NAME  DRING HATER TANK 10.1 FRESH HATER T. 10.2 FRESH HATER T. 10.3 FRESH OIL T.(5) 10.3 FRESH OIL T.(6) 10.3 FRESH OIL T.(6) 10.1 FRESH OIL T.(6) 10.1 FRESH OIL T.(6) 10.1 C.O.T. (C) 10.1 C.O.T. (C)	ENDING MON FRAME NO. 1.57.0001 0.00 0.00 0.00 0.00 0.00 0.00 2.4.45 212.73	FRAME NO. 4.59.0001 	FRAME NO. 1. 43.0009 0.00 (.21 0.00 0.00 0.00 0.00 0.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00, 0.00, 0.00 0.00 0.00 0.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 11.05	0.00	
TANK MAME  DRINK MATER TANK MILL RESIGNATER T. MILL RESIGNATER T. MILL RESIGNATER T. MILL RESIGNATER T. MILL RESE OIL T. (8) MILL RESE OIL T. (8) MILL RESE OIL TANK MILL RESPONSE OIL	ENDING MOM FRAME NO. 1.57.0003	FRAME NO. 1.59.2001 	FRAME NO. 1. 63.0003 0.00 0.00 0.00 0.00 0.00 0.00 0.00 222.65 167.95 183.70	FRAME NO. 1.69.0001 0.00 0.00 0.00 0.00 0.00 0.00 0.00 19.85 150.15	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0		0.00 0.00 0.00 0.00 0.00 0.00 0.00 27.45 60.75 59.50	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.31 0.06 0.00 0.00	
TAIN MAME  DRIM WATER TAME  10.1 FRESH WATER T.  10.2 SRESH WATER T.  10.1 FRESH WATER T.  10.1 FRESH WATER T.  10.1 FRESH WATER T.  10.1 FRESH WATER T.  10.1 FRESH WATER T.  10.1 C.O.T. (C)  10.1 C.O.T. (C)  10.2 C.O.T. (C)  10.2 C.O.T. (C)	ENDING MOM FRAME NO. 1.57.0001 0.00 0.00 0.00 0.00 0.00 0.00 2.54.45 217.52 217.52 174.40	FRAME NO. 4.59.0001 	FRAME NO.  1. 43.0001  0.00  0.00  0.00  0.00  0.00  0.00  1.72  171.75  143.10  143.10	FRAME NO.  1. 69.0001	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 159.25 124.35 179.10 79.50	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 15.90 15.90	0.00	1.77.000) 	1104-000] [	.1
TAIN NAME  TAIN NAME  DRING HATER TANK 10.1 FRESH HATER T. 10.2 FRESH HATER T. 10.3 FREE DIL T. (5) 10.3 FREE DIL T. (6) 10.1 FREE DIL T. (6) 10.1 FREE DIL T. (6) 10.1 C.O.T. (6) 10.1 C.O.T. (6) 10.2 C.O.T. (6) 10.2 C.O.T. (6) 10.3 C.O.T. (6) 10.3 C.O.T. (6) 10.4 C.O.T. (6) 10.5 C.O.T. (6) 10.5 C.O.T. (6) 10.6 C.O.T. (6) 10.7 C.O.T. (6) 10.7 C.O.T. (6) 10.7 C.O.T. (6) 10.7 C.O.T. (6) 10.7 C.O.T. (6) 10.7 C.O.T. (6) 10.7 C.O.T. (6) 10.7 C.O.T. (6) 10.7 C.O.T. (6) 10.7 C.O.T. (6)	ENDING MON FRAME NO. 1.57.0001 0.00 0.00 0.00 0.00 0.00 0.00 254.45 217.57 217.57 174.90 124.80	FRAME NO. 4.59.0001 	FRAME NO.  [. 43.000]	FRAME NO.  1. 69.0001  0.000  0.000  0.000  0.000  0.000  1.000  1.56.15  1.170.90  1.11.30  63.20	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 1.50 15.90 0.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 12.45 12.44	0.00	.1
TAIN NAME  DRIN WATER TANK  10.1 FRESH WATER T.  10.2 FRESH WATER T.  10.5 FRESH WATER T.  10.1 FRESH WATER T.  10.2 FRESH WATER T.  10.1 FRESH WATER T.  10.1 FRESH WATER T.  10.1 C.O.T. (C)  10.1 C.O.T. (C)  10.2 F.O.T. (C)  10.3 C.O.T. (C)  10.4 C.O.T. (C)  10.5 C.O.T. (C)  10.6 C.O.T. (C)  10.7 C.O.T. (C)  10.8 C.O.T. (C)  10.8 C.O.T. (C)  10.8 C.O.T. (C)  10.8 C.O.T. (C)  10.8 C.O.T. (C)	ENDING MOM FRAME NO. 1.57.0001 0.00 0.00 0.00 0.00 0.00 0.00 2.54.45 217.52 217.52 174.40	FRAME NO. 4.59.0001 	FRAME NO.  1. 43.0001  0.00  0.00  0.00  0.00  0.00  0.00  1.72  171.75  143.10  143.10	FRAME NO.  1. 69.0001	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 127.25 97.54 97.53 47.70 47.70 47.70	0.00 0.00 0.00 0.00 0.00 0.00 0.00 25.45 60.75 59.50 15.90		0.00	0.00	
TAIN NAME  DRIN WATER TANK  10.1 FRESH WATER T.  10.2 FRESH WATER T.  10.3 FRESH WATER T.  10.1 FRESH MATER T.  10.2 FRESH MATER T.  10.2 FRESH MATER T.  10.3 FRESH MATER T.  10.3 FRESH MATER T.  10.4 FRESH MATER T.  10.5 FRESH MATER T.  10.7 FRESH MATER T.  10.7 FRESH MATER T.  10.8 FRESH MATER T.  10.9 FRESH MATER T.  10.9 FRESH MATER T.  10.1 FRESH MATER T.  10.1 FRESH MATER T.  10.1 FRESH MATER T.  10.1 FRESH MATER T.  10.1 FRESH MATER T.  10.1 FRESH MATER T.  10.1 FRESH MATER T.  10.1 FRESH MATER T.  10.1 FRESH MATER T.  10.1 FRESH MATER T.  10.1 FRESH MATER T.  10.1 FRESH MATER T.  10.1 FRESH MATER T.  10.1 FRESH MATER T.  10.2 FRESH MATER T.  10.3 FRESH MATER T.  10.4 FRESH MATER T.  10.5 FRESH MATER T.  10.5 FRESH MATER T.  10.5 FRESH MATER T.  10.7 FRESH MATER T.  10	ENDENG MON F944E NO. 1.57.0001 0.00 0.00 0.00 0.00 0.00 0.00 2.12.73 217.73 174.40	FRAME NO 57.001.  D.D	FRAME NO.  1. 43.000)  0.00  0.00  0.00  0.00  0.00  0.00  17.27  17.75  143.10  95.40  31.74	FRAME NO.  1. 69.000]  0.00  0.00  0.00  0.00  0.00  0.00  17.05  156.15  170.00  111.30  112.30  63.20  63.60  11.90	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 127.25 97.55 97.55 47.70 47.70 15.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 27.45 60.75 59.30 13.90 0.00		1.77.000) 	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00	,1
TAIN NAME  TAIN NAME  DRING HATER TAME 10.1 FRESH HATER T. 10.15 FRESH H	ENDING MOM FRAME NO. 1.57.0001 0.00 0.00 0.00 0.00 0.00 2.7.7.7 217.72 174.90 124.80 127.72 63.54 43.97	FRAME NO. 4.59.0001 000	FRAME NO.  [ 43.000]  0.00  0.00  0.00  0.00  0.00  197.95  143.10  95.40  31.74  32.17	FRAME NO.  1. 69.0001  0.000  0.000  0.000  0.000  0.000  1900  156.15  111.30  63.20  63.60  12.68  17.88  17.99	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 159.75 124.35 177.10 79.50 77.50 31.40 31.61	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 127.25 97.55	0.00		0.00	0.00	
TA'N NAME  DRIN HATER TANK 17.1 FRESH HATER T. 17.2 SRESH HATER T. 17.2 SRESH HATER T. 17.1 FUEL OIL T. (P/S) 17.1 FUEL OIL T. (P/S) 17.1 C.O.T. (C) 17.1 C.O.T. (C) 17.2 C.O.T. (P/S) 17.3 C.O.T. (P/S) 17.3 C.O.T. (P/S) 17.4 C.O.T. (C) 17.4 C.O.T. (C)	ENDENG MON F944E NO. 1.57.0001 0.00 0.00 0.00 0.00 0.00 0.00 2.12.73 217.73 174.40	FRAME NO 57.001.  D.D	FRAME NO.  1. 43.000)  0.00  0.00  0.00  0.00  0.00  0.00  17.27  17.75  143.10  95.40  31.74	FRAME NO.  1. 69.000]  0.00  0.00  0.00  0.00  0.00  0.00  17.05  156.15  170.00  111.30  112.30  63.20  63.60  11.90	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.0	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 127.25 92.55 90.30 47.70 135.00 125.97 0.00 0.00	0.00	0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00 0.00	1.77.000) 0.00 0.00 0.00 0.00 0.00 0.00 0.00 31.05 12.75 12.44 0.00 0.00 0.00	0.00 0.00 0.00 0.00 0.33 0.06 0.00	

## \*\* LONGITUDINAL STRENGTH DAYA, NOT INCLUDED IN LIGHT WEIGHT \*\*

### 3.00 METER BASE DRAFT S.G.= 1.02500

### UNIT: SHEARING FORCE IN METRIC TON/1000 BENDING MOMENT IN METRIC TON-METER/1000

### SHEARING FORCE BENDING HOHENT BASE DRAFT TRIH BASE DRAFT TRIH LOCATION VALUE CORRECT. CORRECT. VALUF CORRECT. CORRECT. (BH) (SF) (CD) (CT) (CD) (CT) FR. (1119.000) 1.20 0.52 -0.49 9.07 4.29 -4.05 , -28.18 30.24 FR. (114.000) 4.51 1.76 -1.61 73.59 FR. (110.000) 8.30 3.10 -2.76 203.32 79.58 -72.67 FR. (106.000) 12.44 -3.89 414.52 157.28 ~140.57 4.52 16.66 FR.(102.000) 5,94 -4.95 711.28 263.99 ~230.86 FR.( 98.000) 1094.23 399.72 -341.97 20.88 7.37 -5.93 FR. ( 94.000) 25.11 1563.36 564.48 -472.25 8.79 -6.83 FR.( 90.000) 29.33 10.21 -7.65 2118.69 758.27 -620.06 FR.( 86.000) 33.56 -8.39 2760.20 981.09 -703.76 11.63 FR.1 82.0001 37.78 3487.87 1232.93 -961.71 13.06 -9.05 FR.( 78.000) 4300.79 1513.75 -1152.24 41.69 14.47 -9.62 FR.1 74.000) 45.66 -10.09 5194.54 1822.90 -1353.47 15.82 FR.( 70.000) 48.91 17.03 -10.44 6160.22 2158.26 -1563.13 FR.( 66.000) 2516.13 -1778.78 51.49 18.02 -10.68 7185.45

-10.81

8255.75

2891.70 -1998.15

FR. ( 61.000)

53.32 18.76

COND 1 FULL LOAD CONDITION AT P.G. DEP.

ROUND BANKER

AFT DRAFT
BASE DRAFT
TRIM
25.04 METER UNIT1. SHEARING FORCE IN METRIC TON
BENDING MOMENT IN METRIC TON-METER
0.01 METER
S.G. OF S.M. 1.02500

LOCATION	SHEARING FORCE	BENDING MOMENT
FR.(I19.000)	-6990.	53630.
FR.(I14.000)	-1720.	153990.
FR.(II0.000)	<b>1560</b> .	154800.
FR.(106.000)	4650.	91260.
FR.(102.000)	-4150.	86040.
FR.(98.000)	-1040.	139060.
FR.(94.000)	2130.	127890.
FR.(90.000)	5370.	51160.
FR.(86.000)	8530.	-90570.
FR.(82.000)	-190.	-175610.
FR.(78.000)	-8070.	-82960.
FR.(74.000)	-5540.	64280.
FR.(70.000)	-1860.	140490.
FR.(66.000)	2430.	133970.
FR.(61.000)	6290.	44470.

** LIGHT MFIGHT (LW) = 3000, MT, MIDSHIP G. (XC) = 2.500 H. (LM) * (XG) = 7500, MT-H, (LM) * (XG
#IDSHIP G. (XG) = 2.500 H.  (LH) * (XG) = 7500, MT-H.   CAL.P.NO. IST INTEG. 2ND INTEG.  0 120. 326. 1 261. 1304. 2 761. 1304. 3 522. 5217. 4 522. 5217. 5 522. 5217.  *** BONJEAN DATA FOR SEAMATE-4, 6 ***  #INIHUM DRAFT 1.00 METER DRAFT INTERVAL 4.00 HETER NUMBER OF DRAFT 5  HEAN DRAFT.
120   326   1304   13
1   261   1364   1304   1304   1304   1304   1304   1304   1305
## BONJEAN DATA FOR SEAMATE-4,8 **  ## BONJEAN DATA FOR SEAMATE-4,8 **  ### BONJEAN DATA FOR SEAMATE-4,8 **  ### BONJEAN DATA FOR SEAMATE-4,8 **  #### BONJEAN DATA FOR SEAMATE-4,8 **  ##################################
5 522. 5217.  ** BONJEAN DATA FOR SEAMATE-4.8 *4 MINIMUM DRAFT
MINIMUM DRAFT 1.00 METER DRAFT INTERVAL 4.00 HETER NUMBER OF DRAFT 5  HEAN DRAFT
MINIMUM DRAFT 1.00 METER DRAFT INTERVAL 4.00 HETER NUMBER OF DRAFT 5  HEAN DRAFT
MINIMUM DRAFT 1.00 METER DRAFT INTERVAL 4.00 HETER NUMBER DE DRAFT 5  HEAN DRAFT
80NJ. NO. 5.00 9.00 13.00 17.00
0 40.00 700.00 360.00 520.00 680.00 1 40.00 200.00 360.00 520.00 680.00 2 40.00 200.0° 360.00 520.00 680.00
OLFF - SIMPSON - TRAPEZQID)
MEAN DRAFT '
1.00 5.00 9.00 17.00 PREP.NO.OF
CAL-P-NO- PONJEAN DIFFERENCE IN CUB. H.
1 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0
2 0 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0, 0
5 0 0. 0. 0. 0. 0.

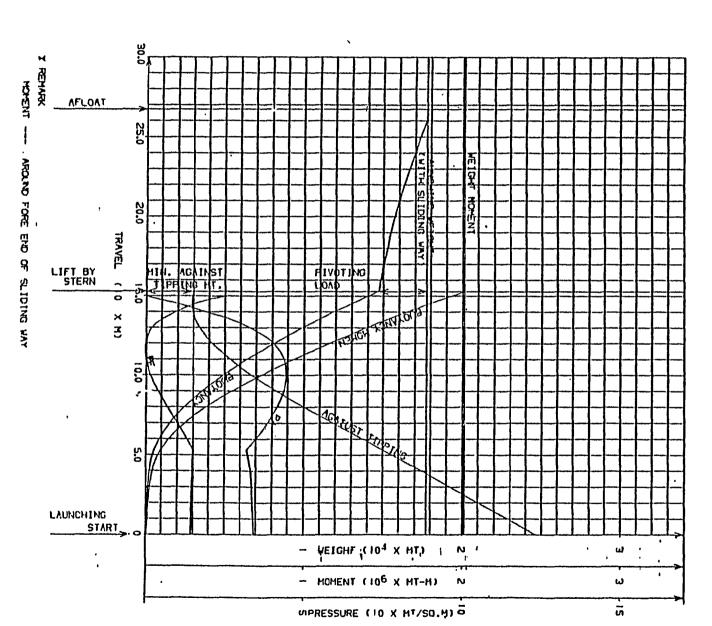
TANK	.NO						EH	PTY.	HEDDL	.E	FULL		HANHUH	
arui			TANK	AME		MtD.G. (H)		1 (KG) 3) (M)	(V)L) (H++3)	( KG ) ( M)	(VOL) (H++3)	(KG) (H)	T.INERTIA (M++4)	
.10	1,		PFAK T			52.500	0.0	0.0	2000.0	5.000_	4300.0	_10.760_	26667	
11	2		H- N-		(PGS)	-45.000			. 600.0	1.000	1500.0	2.500	5625.	
12 13	3		И. В Н. Р.					0.0	1200.0	1 •000 _ 1 •000	1500.0. 3000.0	2•500 2•500	5625 11250.	<del></del>
14	3		W . A .		(PGS)	0.000		0.0	1200.0	1.000	3000.6	2.500	11250.	
15	6		FAK TA		ICI	55.000			4000.0	5.000	8324.0		53333.	
. 1	7		HATER			24.265				13.357_	1700.0		1667	
5	6		F.O.			20.000			1200.0	1.000		2.500	11250.	
- <u>:</u> ?	9 10		L DIL. Cargo:			35.000 -45.000		<u>0.0</u>	200.0_ 2000.0	1.000 8.500	6720.0	8.500	208. 53333.	
22	11		CARGO			-35.000		5.000	2800.0	8.500	6960.0		53333.	
23	12		C AR GO			-20.000		5,000	5600.0	8.500	13800.0	النصبب بنيست بنيات	106667.	<del></del>
24	1.3		C AR GO_			0,0	_0.0	5,000	5600.0	8.500	13666.6	13.633	106667.	
	14	NO E	CARGO	unt n	1.01	20.000	0.0	5.000	5600.0	8.500	13800.0	13.708	106667.	
25	** TA	NK TAR		SE A	1ATE-4	,8 **								
	** TA	NK TAR	LE FOR	SE AI	1ATE-4	,8 ** HID+G+	AFT	FOR	E					
TANK DLD) (	** TA	NK TAR	LE FOR	SE AI	1ATE-4	,8 ++ HID.G. (M)	AFT_ FRNO	DIFF FRN	E	FLAT				
TANK DLD) ( 10	** TA NO NEW)	FORE	TANK (	SEAT	1ATE-4	#1D.G. (M) -52.500	AFTFRNO	F DR F FRN 0.0 105	E0 DIFF	FLAT				
TANK DLD) (	** TA NO NFW)	FORE (NO.1 NO.2 )	TANK (	SEAT	(C)	#ID.G. (M) -52.500 -45.000	AFT_FRNO	PDIFF FRN 0.0 105 0.0 100	E 0.0 DIFF	FLAT				
TANK DLD) ( 10	** TA NO NFW)	FORE ( NO.1 1 NO.2 1 NO.3 1	TANK I	SEAF	(C) (C) (PGS) (PGS)	#1D.G. (M) -57.500 -45.000 -35.006 -20.000	AFT_FRNO	FOR DIFF FRN 0.0 105 0.0 100 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	E	FLAT				
TANK DLD) (	** TA NO NFW)	FORE ( NO.1 ) NO.2 ( NO.4 ) NO.4 ( AET P)	TANK (	SEAN	(C) PES) PES) PES)	#ID.G. (M) -52.500 -45.000	AFT	PDIFF FRN 0.0 105 0.0 100	U.0 0.0 0.0 0.0 0.0	FLÀT				
TANK DLD) (	** TA NO NFW) 1 2 3 4 5 6 7	FORE NO.1 1 NO.4 1 NO.4 1 PF T PSH	TANK (	SEAT	1ATE-4  (C) PGS1 PGS1 PGS1 PGS1 (C) PGS1	#ID.G.  MI  -57.500 -45.000 -35.776 -20.000 5.000 34.265	AFT	DIFF FRN 0.0 105 0.0 100 0.0 92 0.0 60 0.0 60 0.0 20	E	FLAT				
TANK DLD) (	** TA NO NFW)  1 2 3 4 5 6 7	FORE   NO.1   NO.2   NO.3   NO.4   AFT PRINCE   NO.1   NO.4   NO.1   NO.	TANK ( FAK T.  I. I.  I. B.  I. B.  AK TAN  MATER	SEAT	(C) PGS) PGS) PGS) PGS) PGS) PGS)	#1D.G. (M) -57.500 -45.000 -35.76 -20.000 0.000 55.000 34.265	AFT	F.DR DIFF FRN 0.0 105 0.0 100 0.0 95 0.0 80 0.0 60 0.0 20 0.0 20 0.0 40	E	FLAT				
TANK DLD) (10 11 12 13 14 15 15 6	** TA NO NFW)  1 2 3 4 5 6 7	FORE   NO.1   NO.2   NO.3   NO.4   AFT   PH   FR   SH   NO.1   DIFSFI	TANK TANK TANK TANK TANK TANK TANK TANK	SEAT	1ATE-4  (C) (PGS) (PGS) (C) (C) (C) (PGS) (C) (PGS)	#ID.G. (M) -52.500 -45.000 -35.000 0.000 55.000 34.265 20.000 35.000	AFT	FUR DIFF FRN 0.0 105 0.0 100 7.6 92 0.0 60 0.0 0 0.0 0 0.0 20 0.0 20 0.0 20 0.0 20	E	FLAT				
TANK DLD) ( 10 11 12 13 14 15 15 6	** TA NO NFW)  1 2 3 4 5 6 7 8 9	FORE   NO.1   NO.1   NO.3   NO.4   NO.4   NO.1   NO	TANK  FAX T.  I. B.  I.	SE AI	(C) PES) PES) PES) PES) PES) PES)	#ID.G. (M) -52.500 -45.000 -35.000 0.000 55.000 34.265 20.000 35.000 -45.000	AFT_FRNO 100 90 60 -10 10 20 10	FOR DIFF FRN  0.0 105 0.0 100 0.0 90 0.0 90 0.0 00 0.0 20 0.0 20 0.0 20 0.0 100	E 0 D1FF  0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0	FLAT				
TANK DLD) (10 11 12 13 14 15 15 6	** TA NO NFW)  1 2 3 4 5 6 7	FORE   NO.1   NO.2   NO.3   NO.4   AFT   PH   FR   SH   NO.1   DIFSFI	TANK TANK TANK TANK TANK TANK TANK TANK	SE AF	(C) PGS) PGS) PGS) (C) PSS) PGS) PGS) (C)	#ID-G. [M]  -52-500  -45-000  -35-000  55-000  34-265  20-000  -55-000  -55-000  -55-000  -55-000	AFT_FRNO 100 90 80 40 -10 10 20 10 90 80	FUR DIFF FRN 0.0 105 0.0 100 7.6 92 0.0 60 0.0 0 0.0 0 0.0 20 0.0 20 0.0 20 0.0 20	E 0 DIFF  0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.0 0.	FLAT				
TANK DLD) (10 11 12 13 14 15 5 6 21 22 22 24	** TA NO NFW)  1 2 3 4 5 6 7 10 11 12 13	FORE   NO.3   NO.4   NO.3   NO.4   NO.1   NO.1   NO.2   NO.3   NO.4   NO	TANK I FAK T. I . I . I . I . I . I . I . I . I .	SEANAMF  ANK  I. (1)	(C) PES) PES) PES) PES) PES) PES) PES) PES	#ID.G. (M) -57.500 -45.000 -35.000 -36.000 34.265 20.000 -35.000 -45.000 -35.000 -25.000	AFT	FUR DIFF FRN 0.0 100 0.0 100 0.0 90 0.0 0 0.0 0 0.0 0 0.0 20 0.0 20 0.0 20 0.0 20 0.0 20 0.0 30 0.0 30 0.0 40 0.0 30 0.0 40 0.0 40 0.0 40 0.0 40 0.0 50 0.0 50 0.0 50 0.0 60	E	FLAT  0				
TANK DLD) (10 11 12 13 14 15 5 6 21 22 22 24	** TA NO NFW)  1 2 3 4 5 6 7 10 11 12 13	FORE ( NO.1 ) NO.3 ( NO.3 ) NO.4 ) NO.4 ) NO.4 ) NO.4 ) NO.4 ) NO.1 ( NO.2 ( NO.3 ) ( NO.3 )	TANK I FAK T. I . I . I . I . I . I . I . I . I .	SEANAMF  ANK  I. (1)	(C) PES) PES) PES) PES) PES) PES) PES) PES	#ID.G.  MID.G.  MID.G.  MID.G.	AFT	FUR DIFF FRN 0.0 100 0.0 100 0.0 90 0.0 0 0.0 0 0.0 0 0.0 20 0.0 20 0.0 20 0.0 20 0.0 20 0.0 30 0.0 30 0.0 40 0.0 30 0.0 40 0.0 40 0.0 40 0.0 40 0.0 50 0.0 50 0.0 50 0.0 60	E	FLAT  0				

LAUNCHING HEIGHT (HITHOUT	Priding AVAF = TOTITO	uo MT	
LCG FROM MIDSHIP	10.5	920 H	
KG	• 11.	150 H	
TYTINCHING HELGHT THILH PTI	DING WAY! 10579.4	o MT	
LCG FROM MIDSHIP	<u> </u>	586 M	and the second contract of the second contrac
KG	<b>a</b> 10.5	769 H	LAUNCHING PARTICULAR (1-2)
SLIDING WAY A		,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	WHEE MSF. 7-9
DECLIVITY RETHEEN AP	AND FP 7.1	167 H	na chamain ann a chuir ann a chuir ann an 1 a sa sa she guadhe a chuar a chuar ann an chuir ann an chuar ann an chuar ann ann an chuar ann ann an chuar ann ann an chuar ann ann an chuar ann an chuar ann an chuar ann an chuar ann ann an chuar ann an chuar ann an chuar ann an chuar ann an chuar ann ann ann an chuar ann an chuar ann an chuar ann ann an chuar ann ann ann ann ann ann ann ann ann a
DECLIVITY BETWEEN POP	PET * '5.8	921 H	(SUPUT 12 1811 - 9 15 x 24 2 8 5-981)
THE RATE OF INCREASE	OF VERTUAL HASS	= 6.20000	
MEAN PRESSURE PEFDRE LAUNCE INITIAL LAUNCHING FORCE AT			
PAG DRAG HT. TRAVEL			
	nter de y de t uras gr. 1 gant cross-francoundrad	GROUND	
1 200.00 10.0 2 100.00 50.0	949.0 0.330000 0.330000	0.036000	
TOTAL	v v ··· m 1 00 mmemo		

CYZE HOY	<u> </u>	66 'UYT' 1	1 TT 32 TO 1	123 - C. T. S. T. S. T. S. T. S. T. S. T. S. T. S. T. S. T. S. T. S. T. S. T. S. T. S. T. S. T. S. T. S. T. S.	WEIGHT IM		F[6:14]			TALL WAY	CHECK T		
	LEXUNCAL	NO NI. (	MITIMUT SE	IDING WAY)		00	11.300			. *** FUK	CHECK 4	••	
	IF IGHT				•		•	17	ALL CASE NO. TAKE				
	14- 10111	ne itne	4.53	TIPPING	LIFTING	** *			NA CASE NE. EMUJ.				
TRAVEL		TIM )	HUNYANCY	MOYENT	HOHENT!	PRESS.	IMT/SO.P			TRAVEL	LCn	T.KH	
14 1	AT AP	AT FP	]Ht}~	(HI~H_).			FORE			1H1		_14.1	
0.0	0.288	-7.966	16.4	2444059.	3255.	34.32	14.27			0.0	74.04	0.38	
5.0	1.242	-7.805	40.6	2374957.	7000.	34.70	14.33			5.0	72.07	0.24	
10.0	1.497	-7.643		2744463.	11503.	34.10	14.37	٠.		10.0	89.45	0.24	
20.0	<u>1.759</u> 2.014	-7.479 -7.312	76.0. 94.1.	_2195127 _2105899	14613	34 <u>.02</u> 35.94	4:4 -	·		<u>1</u> å • <u>þ</u>	87.54_	8 - 27	
25.0	2.215	-7.144	116.1	2016554	1 1712. 21418.	33.74	14.45			20.6 25.0	83.90	0.70	
33.0	2.538	-4.974	141.5	1927167.	76035	33.72	14.52			30.0	81.62	1.08	
35.0	2.003	-4.405	102.5	1037253.	32901.	33.55	14.58			35.0	79.70	2.96	
49. U 45.0	3.47¢	-6.678 -6.452	374 7	1746977. 1655715.	41831. 56672.	33.33	14.66			40.0 45.0	75.29	5.92	
30.0	-3.616	-6.274		1563772.	30162	- 32.44	4:42			30.0"	71 .56 - 67.78 -	2.59   9.44	
55.0	7.883	-6.074	627.1	1469753.	104554.	32.54	14.75				44.29	2 5. 47	
63.1	4.150	-5.013	864.1	1775577.	142304.	14.95	12.99			٨٨٠٨	61.54	33.71	•
45.0 70.3	4.713	-5.729 -5.543		1201100.	148719. 243946.	36.85	11.47		进水 計井 6% 果	45.0	59. 43 96.76	36.10	
75.0	4.971	-5.356	1957.2	1094241	307430.	40.26	0.37			75.0	54.81	34.90	
0.01	3.716	-5. 167	2443.3	1003515.	370403.	71.65	6.45			80.6	51.33-	-33.33	
15.0	5.500	-4.973	2771.9	914263.	450200	. 42.A3	5.33			05.0	50.97	33.55	
99.9 84.0	1.846 6.135	-4.702 -4.587	3541.6 4337.1	028957 <b>.</b> 747066.	544323. 635020.	43.77	3.91 2.53	•		90.0	49.17 47.50	31.47	
100.0	6.425	-4.190	4930.2	6 70295.	731170.	44.58	1.36			100.0	45.92	28.99	
105.0	6.717	-4.171	5667.7	_ 596210.	n3151n.	44.37	0.33			105.0	44.43_	27.67	
110.0	7.011	-3.790 -3.797	64 19.9	332103.	736404.	43.72	-0.53			110.6	~43.02~	26.47	
1 20.0	7.404	-3.502	7248.4 0097.1	472326. 419055.	1044317.	42.47	0.64 -0.56	** ** *** *		115.0 120.0	41.68 40.42	25.39 24.44	
125.0	7.904	-3.376	1977.7	374448.	1771741.	37.23	0.36		, , , , , , , , , , , , , , , , , , , ,	125.0	37.26	23.63	
130.0	0.205	-3.167	9901.0	330070.	1391850.	33.02	3.00	••	, ,, ,, ,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,	130.0	30.14	22.69	
140.0	ባ <u>፣</u> የሰካ	-7.757 -7.744	11875.0	310047. 293144.	1517097.	26.P6 19.07	5.1? .			135.0 140.0	. 37.20_	. 22.20	
145.4	9.177	-7,530	12935.4		1744430	N. 36	10.55			145.0	36.32 T 35.56	21.50	
\$50.0	9.472	-2.317	14046.0		1927400.	-6.36	26.05		•	150.0	34.07	20.53	
155.0	9.644	~2.095	15009.2	<u></u>						155.0	34.21	20.14	
160.0 165.0	9.740 9.573	-1.675 -1.653	15309.0 15722.3		LIFT BY		1 %			140.0	33.43 32.54	19.97	
170.0	2.770	-1.427	- 15011.4		LINI # J.	J/6KK			<del></del>	170:0-	-31:37-	-17.73	
174.0	9.101	-1.293	15139.8							175.0	30.62	17.45	
110.0	4.974	-9.975	15248.5							180.0	29.67	19.75	
185.0 190.0	# ,953 # ,725	-0.744	15359.7 15477.7					•		1 A5 . 0 19 U . B	28.72 27.75	19.63	
195.0	M.595	-4.240	15594.9							195.0	26.74	19.44	
200.0	M.4/17	-0.045	15721-9	•	<i>'</i>					200.0	25.70	19.32	
275.7	·#•329	0.473	11056.6							205.0	24.61	19.21	
71.0 215.0	#.143 #.054	0.412	1597A.3 16147.5		4 1 1					215'-0	2 2 2 40	19.00	
220.0	7.913	0.016	16303.8		- j - 1 j		• '	r	, •	220.0		18.04	•
774.0	7,769	1.16?	16464.5							225.0	_19.92	10.71	
235.0	7.475	1.659	15030.5							230.0	18.70	10.58	
244.4	7.126	1.9119	16974.3							235.0 240.0	17.46 16.21	10.45	••
745.0	7.175	2.161	17150.3							245.0	14.97	18.20	

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	HE 16H	OF T1	OHTINI Be	SLIDING WAY S OUT SLIDING WAY! 4.35 (M ) 20.00 (MI)	17744		11.								•	•
TAAVEL			SPEED (4/S)		·		<del> </del>						·	<del> </del>	·	
0.0	0.00	230.0	5.38									•••				
10.0	1.11		5.34 5.30			· ' ·	٠,					•				
	2.76		5.26		<del></del>	18 1/5	SPEE	<u> </u>			<del></del>			· 		
25.0	3.00	275.0	5.17			•			•.							· • • •
30.0 11.0			5.10 5.ul	•				•				•				
40.0 45.0		270.0					•	•			•			•	••	
	4.39		4.70								•					~ ~
60.0	4.78	310.0	4.59		• • •	• •		• • • •				- · · ·	• • • • • • • • • • • • • • • • • • • •	•• • •••	· ••· •	
170.0	4.94	315.0		•		•	·· ·		•				• • • •			
75.0	5.27	376.0								·					·	· · · · · · · · · · · · · · · · · · ·
99.0	5.43	335.0	3.97				,							٠.		
99.3	4.59	345.0	3.75										,			
100.0	5.70	350.0 355.0							•	•						
	5.75	760.0 765.0	7.41													
120.0	5. A1	370.0	3.17	•			•	•		, •	• •			,	•	·· •,
130.0	5.63	375.0	2.92								*			•		
135.0		391.0						<del></del>		***************************************		<del></del>	<del></del>			
145.0		395.C 400.0									•				,	· •
155.0	5.74	405.0	2.22							•			:			
145.0	5.79	410.0 415.0	1.98													
	5.49 1.47	420.0														••
	5.66	430.0	1.24													••
190.0	9.64	440.0	0.46													*
200.0	5.57	77 1.1	1140												·	
731.7	5.59			•							•				4	**
215.0	5.57					i										•
225.0	5.53											<del>,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,</del>				·····
230#0 235.0	5.49				<i>i</i>	1		,	,		٠.,		•			
240.0 '					i '	1		•	r			•	•	•		



CASE NO. (1-1-1)
LAUNCHING WT. (WITH SLIDING WAY)
(WITHOUT SLIDING WAY)
4.35 M

W.I. 17745.03 MT **174H0.00 MT** 

1.C.G. 11.09M 11.30M EZEZONS

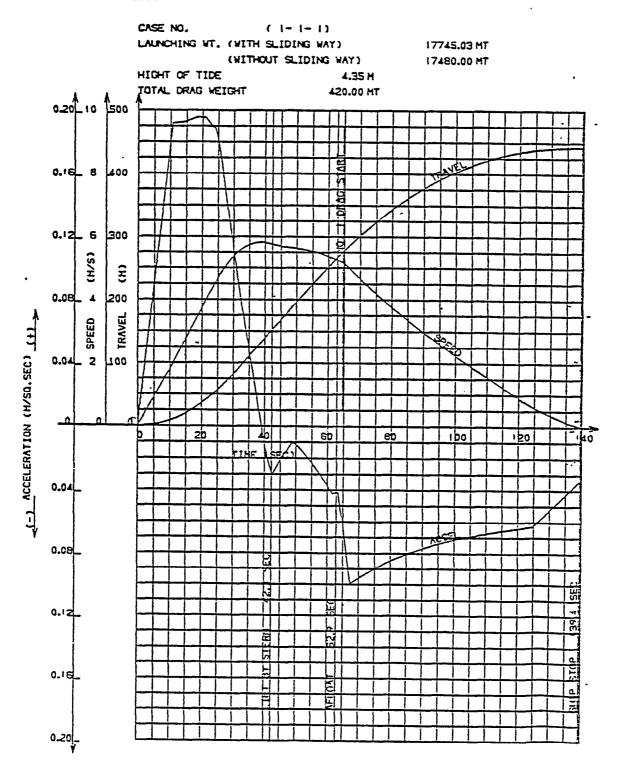
# LAUNCHING CURVES

# 進 水 計 算

## 進水計算

## LAUNCHING SPEED AND TRAVEL CURVES

### SN02323



, MARE OF L	BJECT 1	BON CHOC	. IUP						,		
		OBJECT3	297000-M 32.850 M	FROM	B.L.	EAE6021			FROM ATP.	· · · · · · · · · · · · · · · · · · ·	
<del></del>		<del></del>	U10 M	FROM	C.L.	<del></del>		.0.0H	FROM C.L.	<del>,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,</del>	
····	TRI	н вү	5 T E	R N	( - MEAN	45 : B '	Y 8 C	) W }	·		
EAN DRAFT	-2.0		0.0	1.0	2.0	3:0	4.0	5.0	0.8		
(H)	D-1-2-	TANCE	F-X-	O-H	F-U"R"E	END	(-UNT	T = METR	IC-)		
4.00	482	520	562	-608	663	725	191	802	983		-
6.00	448	403	523	568	619	677	746	826	921		
10.00	413° 378	446	445	5 26 4 8 5	574 530	630 582	694 643	770	7 99		
12.00	343	373	408	443	<del>486</del>	534	591	658	737		
14.33	309	336	367	402	441	487	540	602	676	•	
36-00	274	299	320	366	397	439	488	546	614	<del>,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,</del>	<del></del>
18.0¢ 20.00	239 204	263 226	269 250	<sup>319</sup>	352 308	39 L 34 4	437 385	490 434	593 491		·
22.00	170	189	211	236	264	296	333	378	430		
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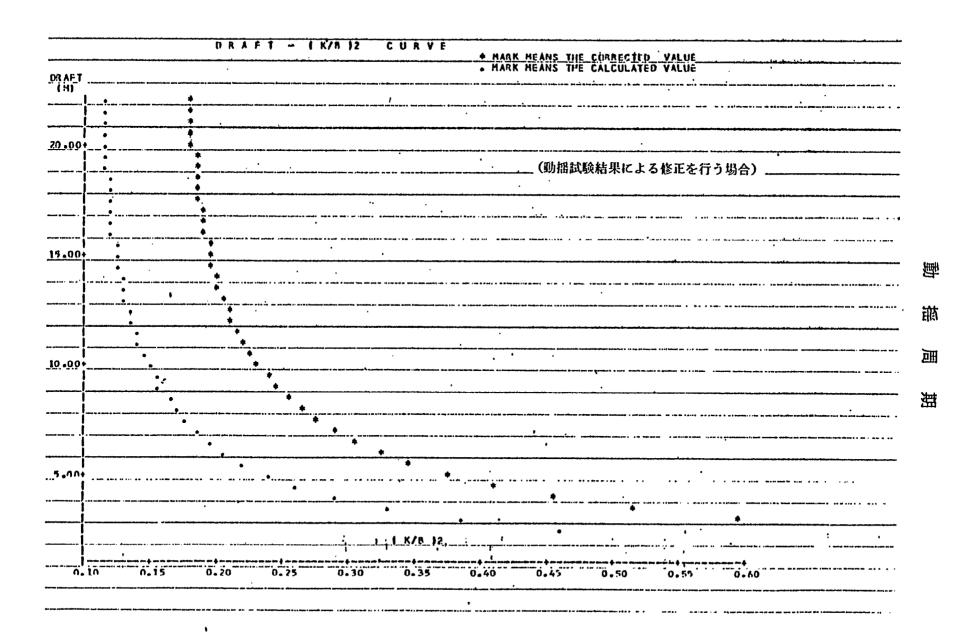
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RAFTIEXT				ROL	LING PER	IOD (SEC	:1	<del></del>		<del></del>					
(H)	6	7 .	8 ,	, 9	10	11		. 13	14	15	20 .	25	30	· 35	40
				CORRI	ECTED GH	≖ GOH (	1913								
2.50	26.43	19.42	14.07	11.75	9.52	7.86	6.61	5.63	4.85	4.21	2.38	1.52	1.06	0.78	0.5
<u> 270</u>		18,16_	13,90_	10.98	8.90_	7.35	6,10_	5,26	4,54	3.95	2,22	1 .4 2	0,99	0.73_	0.5
7.90	23.25	17.09	13.08	10.34	8.37	6.92	5.81	4.95	4.27	3.72	2.09	1.34	0.93	0.68	0.5
3.10	22.01	16.17	12.30	9.70	7.92	, 6.55	5.50	4.69	4.04	3.5?	1.98	1.27	0.88	0.65	0.5
3.30	20.93	15.38	11.77	9.30	7.54	6.23	5.23	4.46	3.84	3.35	1.00	1.21	0.84	0.62	0.4
3.50	20.00	14.69	11.25	8.89	7.20	5.95	5.00	4.26	3.67	3.20	1.80	1.15	0.80	0.59	0.4
3.70	19.16	14.09	10.79	8.52	6.90	5.71	4.79	4.08	3.52	3.07	1.73	1.10	9.77	0.56	0.4
3.90 4.10	_ !2•45 -	13.56	10.38	8.20	6.64	5.49	4-61	3 •93	3.39	2 .9 5	1,66.	1.06	0.74	0.54_	0.4
4.30	17.81 17.24	13.08 12.66	10.02	7.92	6.41	5 - 30	4.45	3.79	3.27	2.85	1.60	1.03	0.71	0.52	0.4
4.50	16.72	12.28	9.70 4.41	7.66 7.43	6.20	5.13	4.31	3.67	3.17	2.76	1.55	0.99	0.69	0.51	0.3
4.70	16.26	11.94	9.14	7.23	6.02 5.85	4.97	4.18	3.56	3.07	2.68	1.50	0.96	0.67	0.49	0.3
4.90	15.84	11.64	8.91	7.04	5.70	4.84 4.71	4.06 3.96	3.46	2.99 2.91	2.60	1.46 .	0.94	0.65	0.48	0.3
5.10	15.46	11.36	8.69	6.87	5.56	4.60	3.86	3.37 3.29	2.84	2.53 2.47	1.43	0.91	0.63	0.47	0.3
5.33	15.11	- ii .iō-	_ B.SU_	6.72	5.44	- <del>4.50</del> -	3.78-	3.22	2.78	-2:42-	1.36	0.89_	0.62_	0.45 0.44	0.3
5.50	14.79	10.87	8.32	6.57	5.33	4.40	3.70	3.15	2.72	2.37	1.33	0.85	0.59	0.43	0.3
5.70	14.50	10.65	8.16	6.45	5.22	4.31	3.63	3.09	2.66	2.32	1.31	0.84	0.58	0.43	0.3
5.90	14.23	10.46	A.01	6.33	5.12	4.24	3.56	3.03	2.61	2.28	1.28	0.82	0.57	U.42	0.3
6.10	13.99	10.28	7.87	6.22	5.04	4.16	3.50	2.98	2.57	2.24	1.26	0.81	0.56	0.41	0.3
6.30	13.76	10.11	7.74	6.12	4.95	4.09	3.44	2.93	2.53	2.20	1.24	0.79	0.55	0.40	0.3
6.50	13.55	9.96	7.67	6.02	4.88	4.03		2.89	2 44	-2.17	<u>1.22</u>	0.78	0.54	-0.40-	— ŏ.3
6.70	13.36	9.81	7.51	5.94	4.81	3.97	3.34	2.84	2.45	2.14	1.20	0.77	0.53	0.39	0.3
6.90	13.17	9.68	7.41	5.86	4.74	3.92	3.29	2.81	2.42	2.11	1.19	0.76	0.53	0.39	0.3
7.10	13.01	9.56	7.32	5.78	4.68	3.87	3.25	2.77	2.39	2.08	1.17	0.75	0.52	0.38	0.2
7.30	12.05	9.44	7.23	5.71	4.63	3.82	. 3.21	2.74	2.36	2.06	1.16	0.74	0.51	. 0.38	0.2
7.50	_ 12.70	9.33	7.15_	5 . 45	4.57_	3.78	3.18	2.71	2.33	2.03	1.14	0.73	0.51	0.37	0.2
7.70	12.57	9.23	7.07	5.59	4.52	3.74	3.14	2.68	2.31	2 .01	1.13	0.72	0.50	0.37	0.2
7.90	12.44	9.14	7.00	5.53	4.48	3.70	3.11	2.65	2.28	1.99	1.12	0.72	0.50	0.37	0.2
0.10	12.32	9.05	6.93	5.40	4.44	3.67	3.08	2.62	2.26	1.97	1.11	0.71	0.49	0.36	0.2
9.30	12.21	3.97	6.87	5.43	4.40	3.63	3.05	2.60	2.24	1.95	1.10	0.70	0.49	0.36	<b>U.</b> 2
8.50	12.11	8.89	6.01	5.38	4.36	3.60	3.03	2.58	2.22	1.94	1.09	0.70	0.48	0.36	0.2
7.70	_12.01	8 • B 2	6.76	_ 5•34_	4.32_	3.57	3 •00	2 • 5 6	2.21	1 .97	1.08	0.69	0.48	0.35	0.2
A . 90	11.92	8.76	6.74	5.30	4.29	3.55	2.98	2.54	2.19	1.91	1.07	n •69	0 -48	0.35	0.2
9.10 9.30	11.83 11.76	8.69	6.66	5.26	4.26	3.52	2.94	2.52	2.17	1.89	1.07	0.68	0.47	0.35	0.2
9.50	11.68	8.64 5.55	6.61	5.22	4.23	3.50	2.94	2.50	2.16	1.88	1.06	0.68.	0.47	0.35	0.2
9.70	11.61	8.53	6.57 6.53	5.19 5.16	4.21	3.48	2.92	2.49	2.15	1.07	1.05	0.67	0.47	0.34	0.2
9,90	11.55	8.48	6.50	5.13	4.18 4.16	3.46 3.44	2.90 2.89	2.47	2.13	1.86	1.05	0.67	0.46	0.34	0.2
10.10	-11.49	-::44-	6.46	-::::-	-7:12-	-3.42	2.87	2 •46	2 •1 2	1 -85	1.04	0.67	0.46	0.34	0.2
10.30	11.43	0.40	6.43	5.08	4.12	3.40	2.86	2.45	2.11	1.84	1.03	0.66	0 46	0.34	0.5
10.50	11.38	8.36	6.40	5.06	4.10	3.39	2.85	2.44 2.42	2.10 2.19	1.83	1.03	0.66	0.46	0.34	0.2
10.70	11.34	8.33	6.35	5.04	4.08	3.37	2.83	2.41	2.08		1.02 1.02	0.66	0.46	0.33	0.2
10.90	11.29	8.30	6.35	5.02	4.06	3.36	2.82	2.41	2.00	1.81	1.02	0.65	0.45 0.45	0.33	0.2
11.10	11.75	8.27	6.33	5.00	4.05	3.35	2.01	2.40	2.07	1.80	1.02	0.65	0.45	0.33 0.33	0.2
11.30	11.21	0.24	6.31	4.98	4.04-1-	3.34		7 2 39 ~~	5.06 ·	- 1.79	1.01	- 0.65-	0.45	0.33	0.2 0.2
11.50	11.16	0.21	6.29	4.97	4.42	3.33	2.79	2.38	2.05	1.79	1.01	0.64	4.45	0.33	0.2



			FLOODABL	.É LENGTH C	ALCULATION	•					
ADTIME OL	DISPLACENT	NT AT FULL LO	AD DRAFT	DESIGN DRAF	FT) 13.325	(H) •	69148.75 (1	13 )	····		
DRAFT F.	DRAFT A.	SHIP VOLUME	SHIP HID. B.	FLODDING VOLUME	FLOOD ING	CENTER POINT	FLOODANLE	PERHISSIBLELENGTH		, 14 detemperates , mp detem ,	-
(X)	(н)	(M3)	(H)	( CH)	(M)	(H)	(H)	·			
18.626	10.600	76655.19	111.307	. 6005.00	182.372	104.232	19.878	19.878			
. 18.647	11.455	79038.81	110.277			156.428		14.560			<del></del>
18.667	12.309	81463.44	109.182	9851.60	140.067	. 139.956	17.995	17.995			
18.688 16.709	- 13.164 14.019	83926.81 . 86425.00						21.429		••	
16.709 16.729	14.074	88934.69_	107.036	1 3820.85 1 5828.60	120.466		24.538	74.530			
10.731	15.728	91429.12	104.929	17824.15			28.650. 32.283	28	<del></del>		<del></del>
_ 18.627		93660.62 .		19604.39	000.00	100.900	34.204	34.204			
10.484	17.428	95742.04	102.749		100.326	100.467	37.189	37.169			
18.292		97474.50 .						38 . 8 6 6			
18.292	16.292	97474.50	101.776	2 2660.45	97.123	97.075	38.866	38 . 8 66			
_17.438	18 - 460	95887 <b>.</b> 31	101.041.	21390.70							*
16.583	18.564	93799.25	100.248	19720.25	90.615	91 .004	34.004	34.004			*
15 • 7 25		191569.31	<b>99.</b> 484 .	17936.30				32.439			<del> </del>
14.874 14.019	18.649 18.646	69232-37	784735	16066-75	61-702	61.746	29.237	29.237			• •
13.164	18.630		71,701 97,224	14133465 12170450				25 • 150			<del></del>
12.309	18.612	81905.00_	96.424	12170.50		67.600 57313.	23.000	23.000 19.122	• •		
11.455	18.595	79444.12	95.576	8236.15		40.884			<del></del>		<del></del>
_ 10 -600	18.578	76907.25	94.671.	6270.65	15.180	4.612	44-945	44.544			
_ 10 - 600	10.578		94.671,	6270.61	15 .1 80	4.812	44 . 5 65	44.565			
_ 10 • 600	18.578	76907.25	94.671,	6270.61	15.180	4.612	44.565	44.565			
_ 10 - 600	16.578	76907.25		6270.63	15.180	4.812	44.365	44.565	**** *** *****************************		
10.600	10.570	76907.25	94.671,	6270.63	15.160	4.812	44.565	44.565	**************************************	•	
_ 10 - 600	10.570	76907.25	94.671,	6270.63	15.160	4.812	44.365	44.565			
_ 10 - 600	18.578	76907.25	94.671,	6270.63	15.180	-4-812	44.365	44.565			
_ 10 . 600	16.576	76907.25	94.671,	6270.63	15.180	-4.812	44.365	44.565			
_ 10 . 600	16.576	76907,25	94.671,	6270.63	15.180	-4.812	44.365	44.565			
_ 10 . 600	16.576	76907,25	94.671,	6270.63	15.186	4.812	44.365	44.565			
_ 10 - 600	16.576	76907,25	94.671,	6270.63	15.180	4 + 8 12	44.365	44.565			
_ 10.600		76907,25	94.671,	6270.63	15.180	4.612	44.365	44.565			
_ 10 - 600	18.578	76907,25	94.671,	6270.63	15.180	4 + 8 12	44.365	44.565			
_ 10 . 600		76907,25	94.671,	6270.63	15.180	4.612	44.365	44.565			
_ 10.600		76907,25	94.671,	6270.63	15.180	4.612	44.365	44.565			
_ 10.600	16.576	76967,25	94.671,	6270.63	15.180	4.612	44.365	44.565			
_ 10.600		76907,25	94.671,	6270.63	15.180	4.612	44.365				
10.600		76907,25	94.671,	6270.63	15.180	4.612	44.365	44.565			
		76907,25	94.671,	6270.63	15.180	4.612	44.365	44.565			
		76907,25	94.671,	6270.63	15.180	4.612	44.365	44.565			
		76907,25	94.671,	6270.63	15.180	4.612	44.365	44.565			
		76907,25	94.671,	6270.63	15.180	4.612	44.365	44.565			
10.600		76967,25	94.671,	6270.63	15.180	4.612	44.365	44.565			

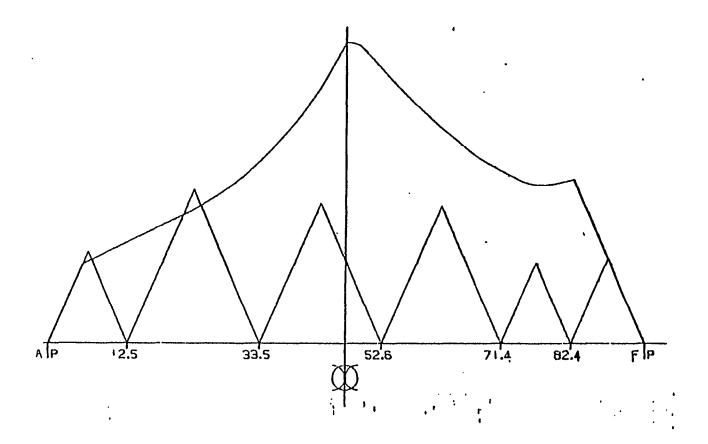
	FL000	PLE LENGT	H	PERMISS	IPLE LENGT	H
	015. AP (1')	L[4684 [4]	K.P.	DIS. AP	LFNGTH - IM)	K.P.
	147.010	0.0	•	147.000	- 0.0	PoP
	173.255	14,275	i	_173.250	14.375	_ 1 4 TAN LINGE FISS DABLE LENGTH ENATH
	173.101	14,366	<u>;</u>	173.163	14.766	
	172.856	14.329		172.006	14.729 .	· · · · · · · · · · · · · · · · · · ·
	172.467	14.777	Ö	172.402	14.277	
	171.454	14.190	. Ď .'.		14.190	<b>0</b>
	170.044	14.140	Ó	170.884	14.140	0
	749. <del>24</del> L	14,446_	O	_169.967	14.066	
	169.039	14,000	0 -	169.039	14.000	0
	119.486		_ 3	164.400	13.764	
	3 47 .5 75	13.710	Ō	167.536	13.716	0
	165.065	13,194	0	166.968	13.094	· · · · · · · · · · · · · · · · · · ·
	166.397	13.675	D	166.397	13.075	0
	1 65 . C . S	1 2 . 8 6 7	◘	_164.055	13-667	- 9
	1 45 . 070	13,452	n.	165.070	13.652	
	164.513 143.935	13.049	2	164.513	13.649	
	163.592	17.851	Ž	143.926	13.055	
	163.016	13.260		163.016	13.665	Y
	16:.674	13.678_	Ň	_162.674	13.878	0 A
	142.172	13.902		162.132	13.902	^
	161.115	12.057	ň.		13.757	Ă
	160.617	13.976	0	160-617	13.995	
	160.040	14.639		. 160.060	14.039	· ·
	159.732	14,000	_ D	159.732	14.068	• V
	159.168	14.121_	ō	_159.188	_14.121	_ 0
	150.060	14.156	D	158.869	14.156	0
	1:4.371	14.210	0	110.371	. 14.210 .	• • • • • • • • • • • • • • • • • • • •
	1 47 . 4 42	14,291	D	147.862	14.291	0
	157.546	14.337	O .		14.3?7 .	
	147.107 #	14.416	D	157.107	14.416	0
	156.672	14.502	O	_156.672	14 • 5 0 3	0
	156.425	14,560	Ō	156.428	14.560	
	129.456	17.975	. 0	137.956	17.995	•
	1 79 . 5 30	21.429	D	120.520	31.429	0
<b>3</b> 1	1:7.993	21.598	D .		21.590	. 9
	127.519	21.703	D	127.518	21.703	
	126.954 126.504	21.067_ 21.775	ÿ	1 26 - 9 54	21 - 867	— <u>V</u>
	125.0:6	22.142	. D	126.594	21.975	
	125.0:0	27.314		126.054	. 22.143 22.314	The second of th
	124.738	22.581	ň	124.738	22.581	
	1:4.410	27.605	Ď	174.410	22.691	
	123.722	22.073_	Ď	123.922	_ 22.073	
	123.446	21.153		123.446	23.053	0
	122.725	22.332	Ď.	122.735	23.332	ŏ
	1 12.465	73.517	Ď.	172.285	23.517	ŏ
	121.616	23.402	ō	121.616	23.802	
	1,70 - 949	24.090	Ŏ	120.969	24.090	, and the second second second second second second second second second second second second second second se
	1:0.004	24.530	ō	1 20 - 0 0 4	24.538	
	119.346	24,937	)	119.346	24.907	0
	118-004	25.246	ō	118-404	25.246	<b>•</b>

FLODABLE LENGTH CURVE

FLOODABLE LENGTH
---- PERMEABILITY LENGTH
(SCALE ICM = 4.33M)

SUBDIVISION LENGTH 94.0 M BREADTH 14.0 M DEPTH 7.6 M

CRITERION NUMBER 16 FACTOR OF SUBDIVISION 1.0 NUMBER OF CREW 325

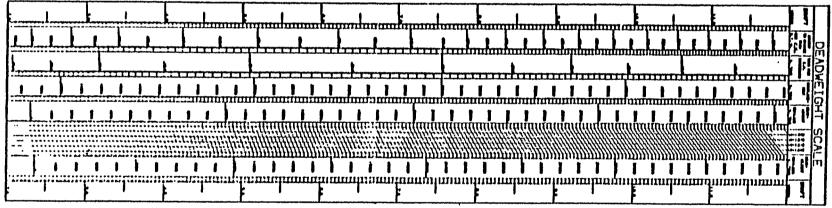


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	*** INPUT DATA FOR TURNING TEST ***		*** SPFED-DATA ***					
		0						
	DATE FEBRUARY 4 1977		HEASURED SPACE OF SPEED 40.0					
	PLACE KURE-1 SHIPYARD	.0	DISTANCE FROM C.L. TO MEAN MEASURED BUARD 47.3					
	COND. FULL	0	AND MATERIAL PLANTING AND ADDRESS OF THE PARTY OF THE PAR					
•	WEATHER BC	•	(HINUTE) (SECOND) (ELAPSED TIME)					
	\$EA COND. 5	Q	1 0 10.00 4.90 2 0 20.00 5.00					
	SFA DCPTH 1500.00	0	4 0 40.00 , 5.50					
	A. WEND DIRECTION 25.00	V	5 0 50.00 5.70 6 1 6.00					
	R. WIND VELCCITY 11.00	0	7 1 20.00 5.50 8 1 40.00 7.20					
	ENGINF PONFA M.C.R.		9 2 · · · · · · · · · · · · · · · ·					
	INITIAL SPEFD 16.20	0	11 3 0.0 10.70					
	Energial State	^	12 3 30.50 12.10 f					
	INITIAL COURSE 270.00	0	14 4 30.00 14.70					
····	INITIAL APH 83.00		15 - 3 - 0.0 - 15.70 - 16.60					
	DESIGNED NUDDER ANGLE 35.00	O_	17 6 0.0 17.40					
	DESTANCE NAMED MADEE 33.880	•	18 6 30.00 10.10 19 7 6.6 10.60					
	ELAPSFO TIME FROM THE ORDER TO DESIGNED AUDDER ANGLE	Ο.	20 7 30.00 19.20					
	25.20	0	22 8 30.00 20.00					
	MFA SURED RUDDER ANGLE 35.00	U	23 9 0.0 20.30					
	ULN 2045 NODDSK NUMBE 35 PAG	_	24 9 30.30 20.60 25 10 0.6 20.90					
	DISTANCE FROM PIVOTING POINT TO SHIPS GRAVITY	٥	26 10 30.00 21.20					
	100.00		27110.021.40					
	100.00	O	20 11 30.00 21.60 29 12 0.0 21.80					
			· 30 12 30.00 22.00					
	LINGTH DETWEEN PERPENDICULARS'	D	31 13 0.0 22.20					
			32 13 30.00 22.30					
	UNIT FOR OUT PUT OMCTRIC	D .						
	INDICATION OF DRAWING 1 DRAW							
		D	,					
	DAAHING SCALE	·						
	INDICATION OF TURNING O BOTH	٥	•					
	INDICATION OF RIGHT OR LEFT 1 PORT	r a						
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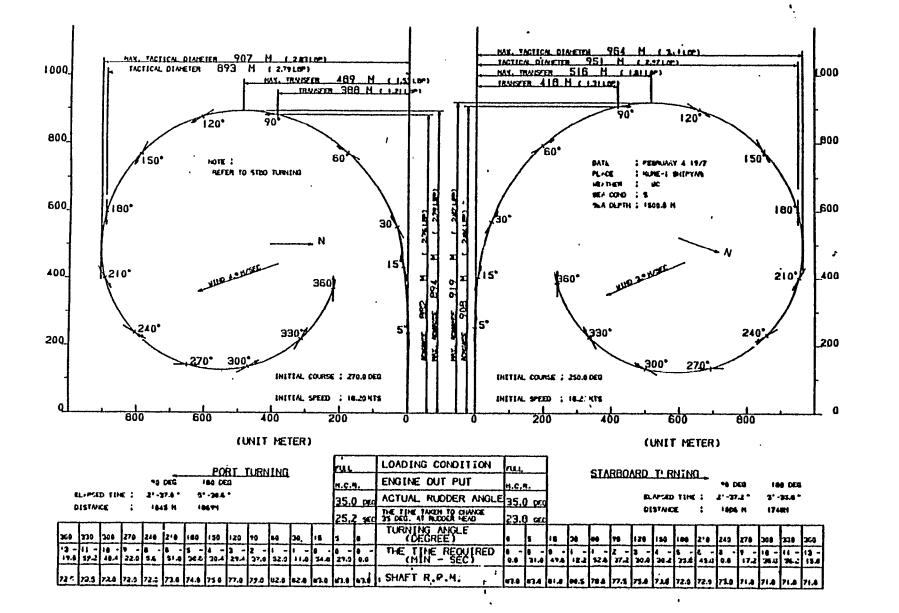
回力試験

樲

旋回力試験

······			HEASURED		MIGULAR -		- SHIP'S	TURNING .	
•	CN	TIME	SPEED	SPEED	AE FUCT IA	POSITION	POSITION	ANGLE	
	•	iseci o.o	(H/SEC)	(M/SEC)		(ADVANCE)	(TRANSFFR)	(DEG)	<del>-</del> - ,
	ż		N.160	4.16;	0.0	0.0	0.0	0.00	
	1	5-3	n-1n3	8.726	0.000923	40.9	0.4	0.16	•
	3	10.0	8.163	A.759,	0.032024	82.2	1.0	0.85	
	<del></del>	15.5	6.165	0.238	- 0.002873	123.5		1.67	· · · · · · · · · · · · · · · · · · ·
	?	20.0	0.000	8.174	0.6.3669	144.4	1.6	2.72	
	6	75.0	7.021	N-021	0.004221	205.1	1.4	3.73	
	<u>'</u>	30.0	7.547	7.766	0.004632	244 <b>.</b> T	0.0	3.76	
	8	35.0	7.466	7.628	0.004708	<b>283.2</b>	-0.7	4.40	
	9	40.0	7.273	7.514	0.005091	. 321.0	-3.2	1.46	
······		45. 3	7.147	7.424	0.005862 .	358 .4	6.0	1.74 -	
	11	50.6	7.018	7.373	0.007512	375.4	-7.3	11.87	
	12	55.0	6.845	7.386	0.011424	432.6	-12.3	15.17	
	13	60.0	6.667	7.372	0.014919	470.1	-16.1	14.44	••
	14	45.0	6.520	7.761	0.015680	506.7	-22.3	23.73	
	15	70.0	6.464	7.268	0.018265	543.5	-30.4	27.16	
	16	75.3	6.285	6.724	0.013523 -	577 .4	43.0	33.04	•
	17	MO.0	6.154	6.775	0.013140	607.2	-57.2	36.00	
	10	85.0	5.998	6.613	0.013008	637.4	-73.0	40.53	
	19	90.0	5.821	6.430	0.012876	667.9	-40.2	44.72	
	20	95.0	5.648	6.250	0.012744	474.4	-108.6		
	21	100.0	5.479	6.076	0.012612	717.2		47.67	****
	<del></del>	105.0	5 . 3 ? 3	5.923	O. 012480	742 . 1	-128.0	\$1.48	
	23	110.0	5.194	5.77n	0.012348	763.3	148.3	\$5.u4	· · · · · · · · · · · · · · · · · · ·
	24	115.0	5.063	5.646			-169.4	\$8.60	
	25				3.012216	762.7	-171.2 -	42.10	· · · · · · · · · · · · · · · · · · ·
		120.3	4.938	5.510	0.012084	800.4	-713.5	45.56	
	26 27	125.0	4.823	5.389	6.611957	816.5	-236.3	66.78	P
		130.0	4.712	3.771	0.011820	830.9	-259.5	72.37	•
· · · · · · · · · · · · · · · · · · ·	28	135.0	4 • 6 04	5 • 157	0.011688	843.7	i03.0	<b>15.</b> 72	
	29	140.0	4.500	5.046	0.011556	054.9	~306.6	79.03	
	30	145.0	4.399	4.739	6.611424	264.6	-336.3	82.30	
	31	150.0	4.301	4.835	0.011292	A72.0	-354.0 ·	85.53	
•	7 32	155.0	4.199	4.727	6.611160	874.5	-317.6	**.73	•
	33	160.0	4.101	4.639	0.011382	885.1	-401.0	91.99	
	34	165.0	4.0U5	4 .535	. 0.011192 -	2 . 9 . 2	-424.3	95.20	
	35	171.2	3.913	4.415	0.010604	. 891.9	-447.1	98.24	
	36	175.0	3.824	4.309	0.010254	893.2	-469.5	101.17	
	?7	100.0	3.730	4.213	0.010030	893.5 .	-491.4	104.05	
	38	145.0	3.657	4.121	0.009795	892.7	-512.7	104.85	
	39	190.0	3.581	4.035	0.009592	890.8	-533.6	104.60	•
	40	1 95.0	_ 3.509	3.959	0. 009516	006 .1	-554.0	ii k.jj	
•	41	200.0	3.439	3.084	0.359404	884.6	-573.8	115.02	
	42	205.0	3.371	3.011	0.009292	e 8 6 . 2	-593.1	117.68	•
	43	210.0	3.366	3.741	0.009766	875.1	-611.8	120.32	•
•	44	215.0	3.242	3.681	0.004289	664.3	-630.1	122.98	
	45	220.0	3.100	3.615	0.009167	867.6	-647.7	125.41	
	46	. 225.0	- 4 - 4	3.552	0.007086	655.7	-644.8		
•	47	230.0.	3.066	3.492				126.22 -	<del></del>
	48	235.0	3.003		0.608998	647.9	-681.2		• •
•	49	240.0		3-432	0-600055	839.6	-697.0	133.33	•
•	50		2.4 (3	3.3751	0.008713	. # 830 . 8	-717.2	. 135.63	
•	·51	745.0	2.917	3.322	0.008570	1 021 -5	-724.4	138.28	
	52	250.0	2.874	3.272	0.006430	611.7	-740.7	140.70	
<del></del>		255.0	2.832		0.008255	101 -4	<u>753.†</u>	143.06	· · · · · · · · · · · · · · · · · · ·
	53	760.0	2.793	3.174	0-068063	190.8	-746.6	145.37	
	54	265.0	2.756	3.125	0.007812 .	777.7	-774.5	147.41	
	55	270.0	2.721	3.086	0.007723	768.6	-789.9	149.82	

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CRASH STOP ASTERN TEST	
DATE FEBRUARY 20, 1977	
PLACE KURE-1 SHIP YARD	
COND. FULL LOADED CONDITION	mentalistratura para manufati indica de la composita della composita della composita della composita della com
WEATHER O	* ************************************
SEA CUND. 4	
SEA DEPTH 2000.00	
R. WIND DIRECTION 120.00	
R. WIND VELUCITY 7.00	
INITIAL SPEED 16.20	
INITIAL COURSE 250.00	
INITIAL RPM 84.00	
SETTLED SPEED -4.50	
SFTTLED RPM 25.00	
UNIT FOR OUT PUT O	METRIC
INDICATION OF DRAWING . 1	DRAW
DRAYING SCALE 4	A4 SIZE
INDICATION OF TEST U	
	CRASH STOP ASTERN TEST
NON CANADIAN DATA 1	CONSIDER
DATA NAME FOR ENGINE PART [HIN.]	) (SECOND) .
AHEAD VALVE SHUT Q ASTERN VALVE OPEN D	
ASTERN VALVE OPEN 0 - SHAFT STOP AND START 0 -	14.00 · ·
ASTERN RPM SETTLED 4 -	20.60 0.0

MEASURED	SPACE OF SPEED		40.0				
DISTANCE	FROM C.L. TO ME	AN HEASURED BOAI	RD 27.0				
LHIM	UEL LSECONOL	TFF962ED_11HET			TE1ISECONDL_	(HEAD. ANGLE	)
1(	10.00	<b>4.</b> 40		10	10.00	250.00	
2 (	30.00	5.60		2 0	204110	253.00	
	0.0	5.50		0	30.00	255,40	,
4 1	30.00	7.60	•	4 0		257.00	
5	0.0	B.50		0	50.00	264.00	* <del></del>
<u>6</u>	30.00	10.50		6 1	U.O	266.00	
	0.0	16.00		<u>1</u> <u>1</u>	30.00	170.00	ļ <del></del>
A	30.60	19.60		8 2	0.0	276.00	
10	30.60	23.50 20.50		10 3	30.00	203,50	,
10 4	0.0	32.00		11 3	30.00	289.00 295.50	
12 5	30.00	37.00				302.00	, <del></del>
13 6	0,0	47,00		13 4	30.00	307.00	
14		-45.00		14 5	0.0	304.50	·
15 7	0.0	-43.60		15 5	36.00	311.00	
16 7	30.00	-41.60		16 6		312.00	·
17 8	0.0			17 .6	30.00	312.30	· .,'
10 0		-30.60		16 7	0.0	312.00	
199	0.0	-35.50		19 7	30.00	311.90	·
20 9	30.00	-32.60		20 B	0.0	311.00	
21 16		-29,70		21 8	30.00	304.70	· · · · · · · · · · · · · · · · · · ·
22 10		-26.00		22 9		307.20	
2311				23 9	30.00	305.00	,
24 11		-20.00		24 10		303.00	•
25 12		10.0C		25 11		301.20	·
26 12		-15.70		26 12		290.26	,
27 13 28 13		-13.50 -10.20	<del></del>	27 13 28 14		207-00	, <del></del>
28 13 29 14		10.20 10.40		28 14 29 15	Û.Û 0.0	284.00 282.00	
30 14		-10.00		29 15 30 0	0.0	0.0	·
	20100	10100		30 0	0.0	0.0	
	<del></del>		**************************************	VIII. III. III. III. III. III. III. III			
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\*\*\* SPEED DATA \*\*\*

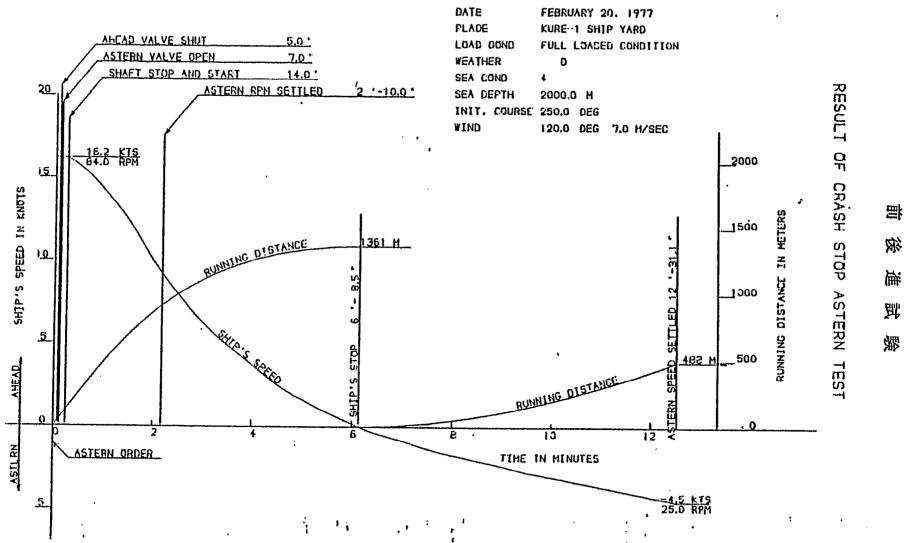
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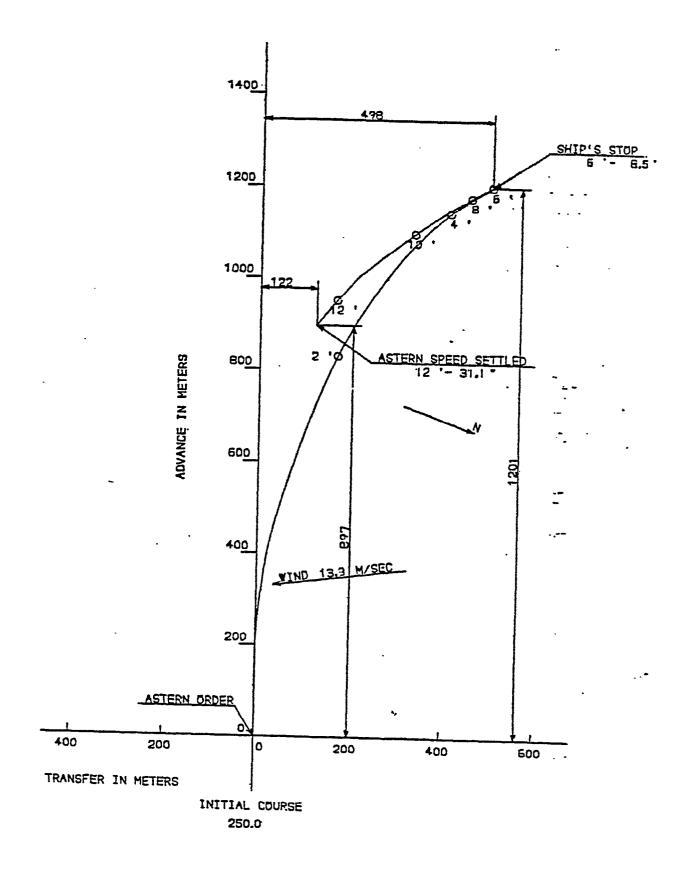
+++ HEAD. ANGLE DATA +++

	DATE PLACE LIBUS COND MEATHER SEA COND	FERRUARY 20. HURE-1 SHIP FULL LCADED D	YARD			••••	• • • • • • • • • • • • • • • • • • • •		
	SCA (IEPTH TNITE COURS	70c0.0 N E 25 147. PEG	7 . M/f16	<u>.</u>	•		·		
•	INITIAL SPECO		6.7 KTS			•	·		<del></del>
ì	ASTERN STEEL	ID SPEED -	4.5 KTS 5.0						***************************************
	SHIPES STOP		6*-15.6*					·· <del>····</del>	
1 1		ION OF CRASH	1. Pritzagora,	TF_ST ***	•			********	
	. HEASUNE	<u>о С</u> ілі.	ANGULAH	\$HIP*\$	SHIP*S				
W.	11 ME SILEN 15 CO		ACFOCILA	POSSTION (AUVANCE)	111204 1244AF	ON ER)			
1 2	0.0 8.413	F.46L	U.G	ί.0	0.0	)			
î	10.0 6.331	8.340	0.0005.4	84.3	0.1 -0.1				
<del></del>	13.4		-0,000063				<del></del>		
	25.0 f.07		-3.00654	166.6 217.3	-0.2 -0.3				
7	30.0 8.0()	7.556	-0.003410	746.3	-2.1			··	
Ţ	40.0 7.793		-0.003410	204.5	5.0 -10.2				
<u>}10</u>	50.0 7.5(F	7.501		360.8	15.å				
12	75.0 1.437	1.4.6	-,64.8 -3.514928	397.0 433.1	i.l 1.00				
13	65.0 7.473	7.116	~u.L06016 ~J.u02849	467.2	~38.0			<del></del>	
15	70.0 4.871	1.017	-0.00/844 -0.00/305	501.1 524.1	48.0 -58.k				
15	75.0 6.579	6,323	-0.00:00.	505.8	-08.4				
ii	85.6 5.947		-0.60154 -3.602154	594.0	-76.4				
19	90.0 5.714	5.644	~J.:0;543"	451.7	-47.0				
20 21	95.0 5.521 100.6 5.353			677.7 702.7	-107.6				·~~
21	105- 0	5.520	-0.403274	7:6.6	1 & 7.1.				
23 24	110-0 517	4.756	-3.003590 3.003588	749.6	-157.6 -147.0				
25	120.0 4.796	4.598	-0.003986	747.8	-157.0				
26 27	123.0 4.558	4+441	0.004338 _	013.0	-167.1				
	135.0	4.291	-0.664443	032-3 050-7	-177.4 -187.7				
29 30	140-0 0-151	4 1.29	mts - i.f. A.A. 2 f.s.	tra h					•
	145.0 3.406.	,_,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,,		<u>.,</u> 685.2, <u>,</u>	ZU8.4	···	<del></del>	<del></del>	
		<del> </del>		·-··	···································	•			
								1 3-1-27 Tru-	
					***************************************	**************************************			<del></del>

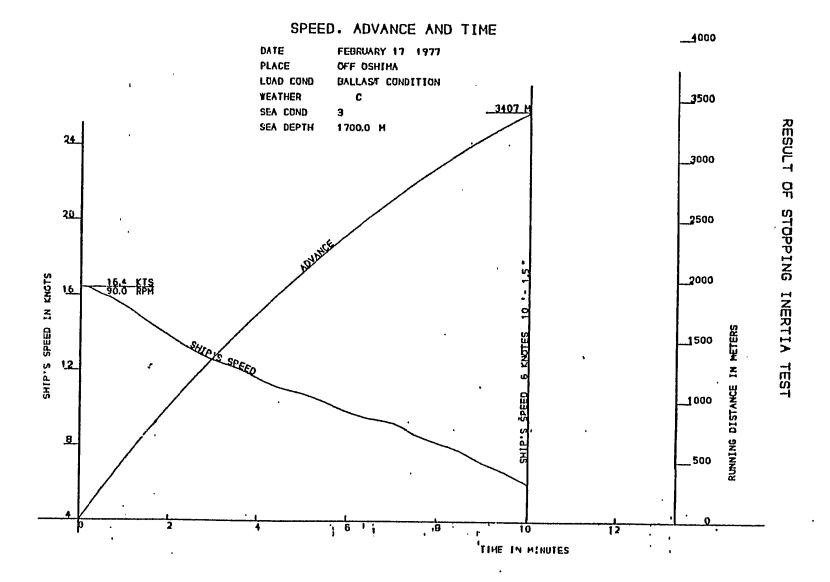




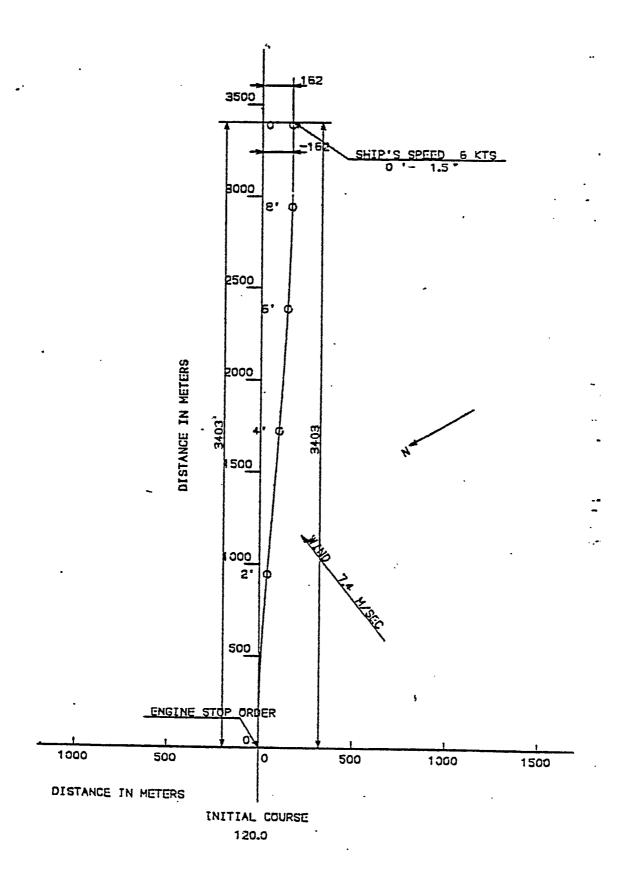
前後進試験 COURSE OF CRASH STOP ASTERN TEST



								•
		DATE		RUARY 17 F OSHIHA	1 977			
		LOAD	COUD BAL	LAST CONDI	T 10N			
		HEAT		. C 3	<del></del> -	<del></del>		
		SEA	OEPTH 170	0.0 M				
		INE1	ITAL COURSE	120 1.3 DEG 12	O DEG			
		•••	11.00		10 11/3/6	,		
·····		INII	TAL SPEED	10	. 4 KTS	<u> </u>		
			RPM		. 0			••
	•	5417	'S SPEED	4	. ZTN O.			
		ELAP	SFD TIHE		01- 2.3"		•	
							1	
		***	CALCULATION	1 OF 2101PP1	NG ENFRTIA TI	:51 ***	•	•
	410	TA	MEASURED	CORR.	ANGULAR	\$112 P S	SHIP'S	
	NO	_11 4E 15EC}	\$PEEO 	\$PEFD L_H_\$ <b>EC</b> }_	AEFOCILA	POSITION	POSITION LIRANSEERD	
	1	0.0	8.304	8.384	0.0	0.0	0.0	
	2	5.0	8.263	8.154	-0.004052	41.3	1-1	
	3	10.0 15.0	B.163 R.07B	0.079 r.01n	-0.003129 -0.002207	81.9 122.2	1.4 1.2	
	3	25.5	6.000	7.955	-0.001264	162.1	6.6	
		25.0 _	7.974	7.899 _	0.000926_	201.4		
	7	30.0	7.043	7.819	-0.000891	241.0	-1.1	
		35.0	7.689	7.666	-0.000851 -0.000823	274.7	-2.3 -3.5	•
	10	40.0 45.0	7.547 7.477	7.585 7.4 <b>6</b> 3	-0.000668	317.7 355.1	-3.5 -4.9	•
	iĭ	30.0	7.407	7.384	-0.000883	392.2	-6.4	
	i ž	- 55.0 -	7.319 <u></u>	7.314 _	0.000880 -	478.9	4.1	
	13	60.0	7.273	7.250	-0.000859	465.3	-9.9	
	14	45.0	7.215	7.194	-0.000764	501.3	-11.9	
	15 16	70.0 73.0	7.163 7.117	7.143 7.099	-0.000721 -0.000678	537.1 572.6	-14.0 -16.2	
	17	80.0	7.376	7.661	-0.000635	608.0	-18.5	
	ia	BS.O -	7.045	7.030 _	0.000344-	643.1		· · · · · · · · · · · · · · · · · · ·
	19	90.0	7.018	7.014	-0.000149	67A.1	-23.3	
	50	95.0	4.966	6.966	0.0	713.0	-25.7	
	21	300.0	6,927	4.920	3.0	747.6	-2 <b>4.1</b> -36.5	•
	22 23	105.3	6.678 6.841	6.878 6.841	0.0 0.0	767.0 816.2	-32.9	
	24	-115.0 -	A.BOA	6,804	0.0	650.3		
-	25	120.0	4.700	A.760	0.0	484.2	-37.7	
	26	125.0	6.761	4.761	0.0	917.9	-40.0	
	27	130.0	4.742	4.742	3.0	951.6	-42.4	
	28	135.0	6.723	6.723	0.0	205.2	-44.7	
	30	140.0 143.0	\$.707 \$.68*	4.707 6.489	0.0 0.0	1018.7 1052.1	-47.1 49.4	
<del></del>	30 31	150.0	 4.667	6.667	0.0	1005.4	-31.0	
	32	155.0	4.636	6.430	0.0	1118.4	-54.1	
	33	160.0 .	6.605	4.405	0.0	1151.4	-56.4	
			4.564	6.564	0.0	1104.5		



前進惰力試験 COURSE OF STOPPING INERTIA TEST



### \*\*\* INPUT DATA FOR ZIG-ZAG TEST \*\*\*

### \*\*\* RUDDER ANGLE DATA \*\*\*

DATE HARCH 9, 1977	(HINUTE)	(SECOND) (REDDER ANGLE)	
PLACE KURE - 1 OFFICE	1 0	0.0 0.0 5.80 15.00	
COND. BALLAST CONDITION	3 0	23.00 15.00 32.80 -15.00	
WEATHER 6	5 0	84.00 -15.00 93.40 15.00	
SEA COND. 3	7 O	141.60 15.00 152.40 -15.00	
SEA DEPTH 1500.00	9 0	200.00 -15.00	
R. WIND DIRECTION -90.00			
R. WIND VELOCITY 20.00			
INITIAL COURSE 220.00			
INITIAL SPEED 14.00	•		2
L B P 94.00			
INDICATION OF DATA O USING INPUT DATA AS IT IS	,		理
INDICATION OF DRAWING O HOT DRAW		•	Ph
DRAWING SCALE O AO SIZE-		•	
ANGLE VELOCITY DATA	•		歌
AHGL2 = 0.9833 ANGL4 = -1.2833 ANGL6 = 1.4267	•		

\*\*\* SPEED DATA \*\*\*

\*\*\* HEAD. ANGLE DATA \*\*\*

(	(HINUTE)	(SECOND)	(KNDT)	i	1 ;	,	, ,	(HIŅUTE)	(SECOND)	(HEAD. ANGLE)
1	ο '	<b>0.0</b>	25.80	•			1	<b>'</b>	0.0	. 0.0
2	2 '	0.0	24.45				ž	ŏ	41.50	30.00
3	4	0.0	23.50				3	Ŏ.	100.00	-28.00
4	6	0.0	23.00				4	Ò	158.30	31.60
5	7	10.00	22.85				5	٠Õ	0.0	0.0
6	22	0.0	18.00				<u> </u>	Δ	0.6	0.0

```
- RESULT OF ZIG-ZAG TEST -
```

DATE MARCH 9, 1977
PLACE KUAE - 1 OFFICE
LOAD COND BALLAST CONDITION
WEATHER B
SEA COND 3
SEA DEPTH 1500.0 M
INITIAL COURSE 220.0 DEG
INITIAL SPEED 14.0 KT\$

A. WIND -90.0 DEG 20.0 M/SEC

LENGTH BETWEEN PERPENDICULARS 94.00 M

### MANEUVERABILITY

K 4 (1/SEC) 6.14402 K 6 8 (1/SEC) 0.10086 KM (1/SEC) 0.12244 T4 (SEC) 31.933 T68 (SEC) 17.942 TM (SEC) 24.938 DELTA A (DEG) 0.89892 (VS BASE) K 1.5980 T 1.9107

操 縦 性 指 数 (連立方程式による)

即

2

鉄

## 記号の説明

### ABBREVIATION

DI SPT : displacement with appendige.

DISPT (HID) : Displacement without appendage.

 $\mbox{APPEN} \qquad \qquad \mbox{: Appendage displacement.}$ 

DIFF : Displacement difference.

D.CORR. : Increase of displacement for one CM/inch sagging,

in case of hogging this value means decrease of displacement.

TPC : Ton per one CM immersion.

TPI : Ton per one inch immersion.

LCG : Longitudinal center of gravity from midship.

LCB : Longitudinal center of buoyancy from midship.

L C F : Longitudinal Canter of floatation from midship.

K G : Vertical center of gravity above base line.

KB : Vertical center of buoyancy above base line.

T KM : Transverse metacenter above base line.

L KH : Longitudinal metacenter above base line.

G H : Height from center of gravity to transverse metacenter

CO M : Corrected height from center of gravity to transverse

metacenter by free water effects.

G GO : Increase of center of gravity by free water effects.

A W: Hater plane area.

CW : Water plane area coefficient.

AH : Midship sectional area.

C M : Midship sectional area coefficient.

W.S A : Wetted surface area.

M T C : Moment to change trim one CM.

MTI : Moment to change trim one inch.

I : Moment of inertia.

ICL : Moment of inertia around center line.

INA : Moment of inertia around neutral axis.

I / D : Propeller immersion ratio.

# APPENDIX G CADS - PIPING DESIGN SYSTEM

PART V. CADS Piping Design System

## CONTENTS

1.	CADS in Piping Design System	1
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	3. 2 Step-2 Layout	10
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### 1. <u>CADS in Piping Design System</u>

In piping design system, CADS is used for development of production engineering drawing for the purposes of:

- (1) Modification of piping layout made by module and automated piping design system.
- (2) Input of piping layout prepared by manually drafted sketch.
- (3) Design of new module, and
- (4) Modification of the module.

The data developed by CADS are transferred as a input data of pipe piece calculation program and material control program.

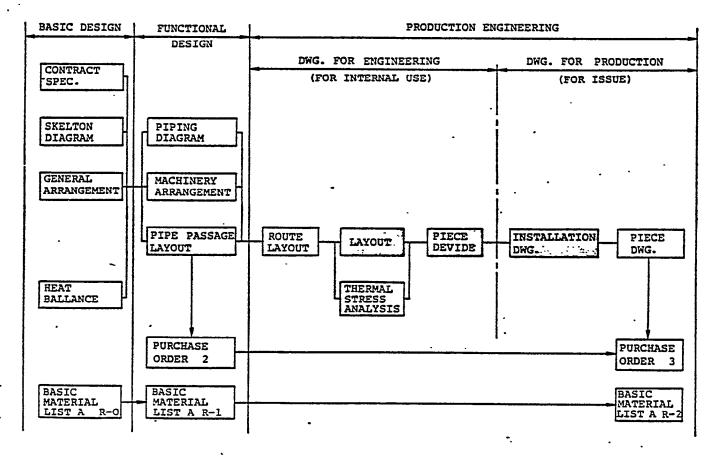


FIG 1.1 PROCESS OF PIPING DISIGN (BY DWG)

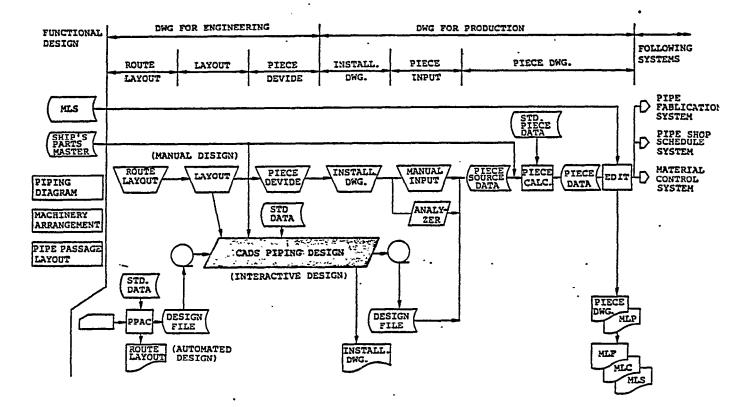


FIG 1.2 PIPING PRODUCTION ENGINEERING SYSTEM (ENG. RM)

- 2. Functions and Features of CADS Piping De sign System
  - 2.1 Functions of the System
    - (1) To use background drawings of hull structures and obstacles for piping, which are supplemented by transparent sheets, photographic techniques and additional hand-writing.

Among the background drawings, main hull structures (such as frames, longitudinals, decks) are registered and drawn by the system.

- (2) To obtain tripartite drawings of pipe planning with plan, side and section views.
- (3) To define pipe lines by indicating the start and end points and bent points.

Computer takes the following role by rough manually indicated data.

Running a pipe line in parallel with an axis..

Rounding pipe length

Setting bent piece standardized.

- (4) To check intersection among pipes and others (tanks. hull constructions. etc.). by means of optional sections.
- (5) To calculate the clearance between the designated pair of pipes.

- (6) To set pipe fittings on pipe line.

  Computer adjust the rough input data with Pen Analyzer to set it on pipe line and to round pipe piece length.
- (7) To check the pipe piece dimension based on the fabrication standard.
- (8) To design the shape of pipe support.
- (9) To set the same shaped pipe lines and pipe supports corresponding to indicated portion that is designed already..
- (10) To decide and draw leaders and characters such as pipe piece code, code of pipe supports, title of drawings.
- (11) To provide data for pipe piece fabrication system and material control system.

### 2.2 Features of the System

1) Large Panel:

Optional sections of tripartite with optional scale are easily drawn out on large panel so as to check them in one-glance.

2) Easy Operation:

The designer does not required special knowledge of computer. The designer can operate the system through a few day's training.

- 3) Minimized Input Data:
  - (i) Simpel input operation of command and symbol with key-sheet and of X, Y, Z value with defined coordinate, either indication is performed with Pen-Analyzer.
  - (ii) Command and symbol operation is mini-mized with "Default" and "Modal" function.
  - (iii) The computer supplements the attributive data to input data of a few key words by means of the master data based on standardization.
- Assured Quality of Drawings and Data to relevant System:

  When the drawing is finished, every relevant data, such as dimension, fabrication practice, material and installation control, are defined simultaneously.

  So that the data missing, that is occured frequently in case of manual input, is eliminated.

## 5). Easy Maintenance:

Data, which have possibility of changing in accordance with ability or restriction of facility and condition of purchasing such as fabrication practice and dimensional table of outfittings, are registered in master file.

Therefore, it is not necessary to change programs to cope with alternation of condition.

Revision of the master file is rather easy than revision of program, *furthermore*, conversational revision measure of master file is established.

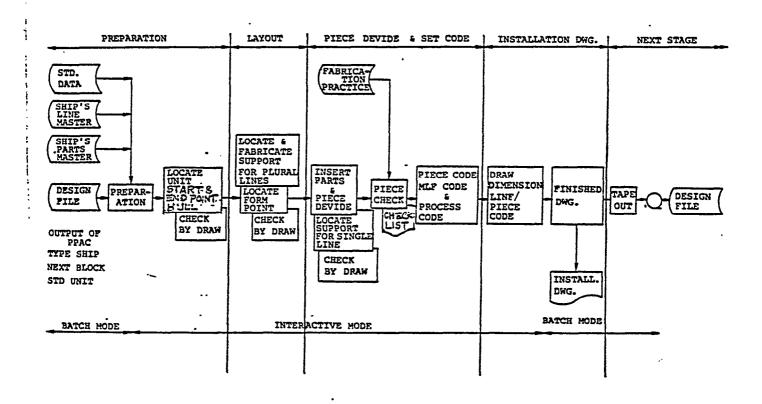


FIG 3.1 . OUTLINE PROCESS OF PIPING ENGINEERING WITH CADS

# 3. I Step-1 Preparation Locate Hull Unit & Start and End Point

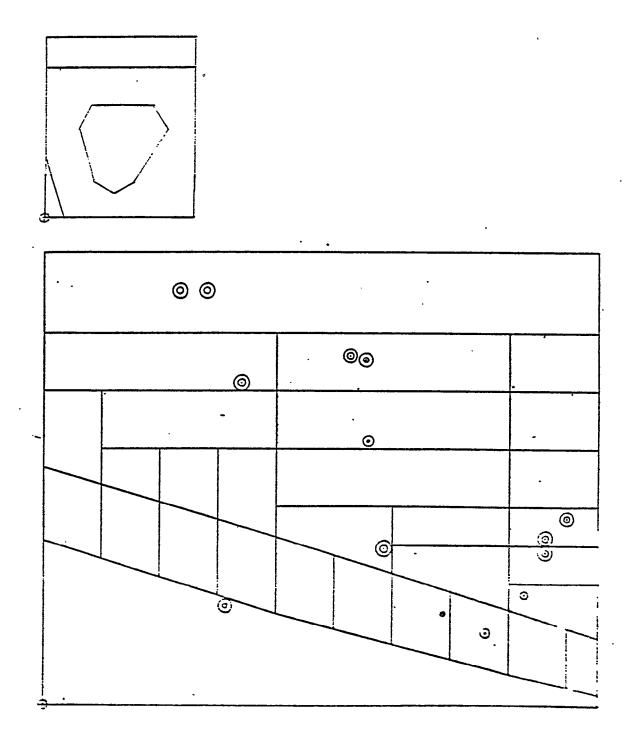


Fig. 3.2

3,2 Step-2 Layout

Locate Form

Locate & Fabricate Support

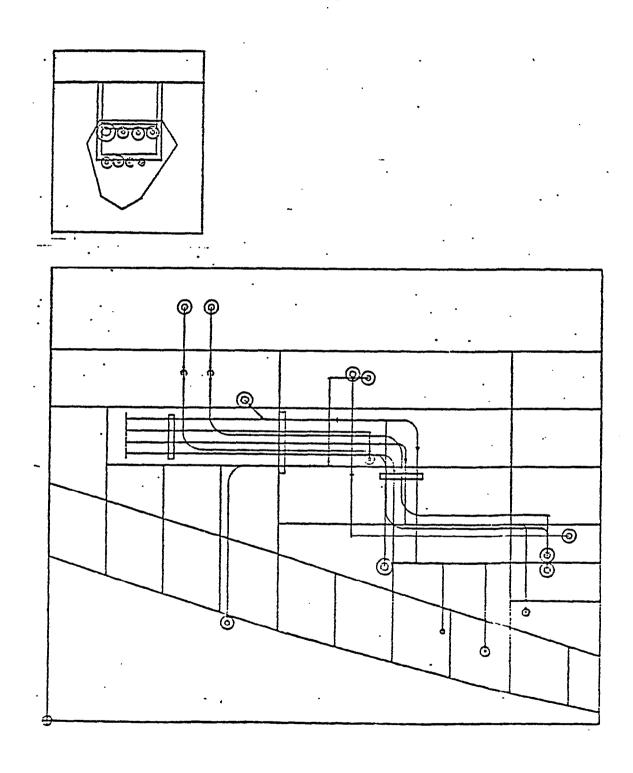


Fig. 3.3

## 3:3 Step-3 Piece Divide Insert Parts & Piece Devide

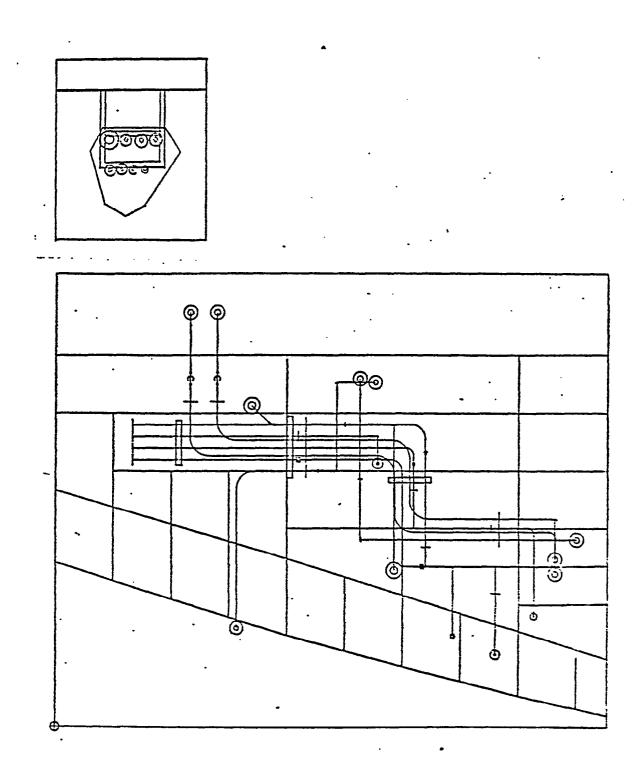


Fig. 3.4

## Table 3.1

\*\*PI ECE CHECK\*

#### \*\*AS03210

E042 LESS THAN BASE LENG. LENGTH= 50 BASE= 305 E042 LESS THAN BASE LENG. LENGTH= 200 BASE= 300 \*PIECE CHECK\*

#### \*\*AS03210

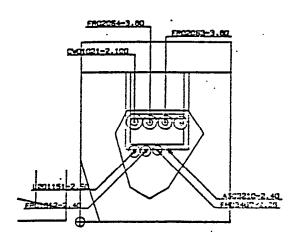
#### \*\*L001151

E042 LESS THAN BASE LENG. LENGTH= 100 BASE= 126 \*PI ECE CHECK\*

## \*\*FR01012

\*PI ECE CHECK\*

# 3.4 Step-4 Installation Drawing Draw Dimension Line / Piece Code



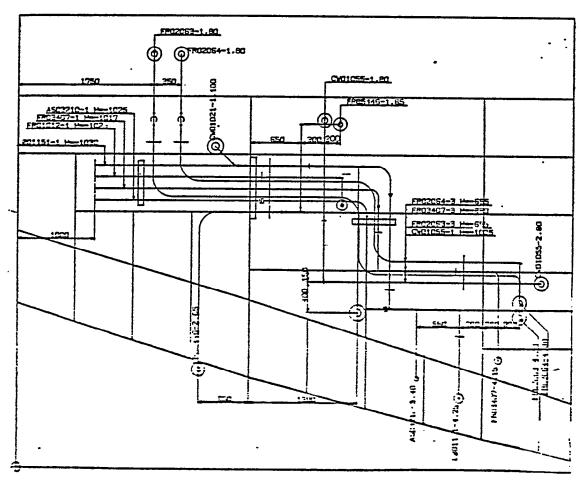
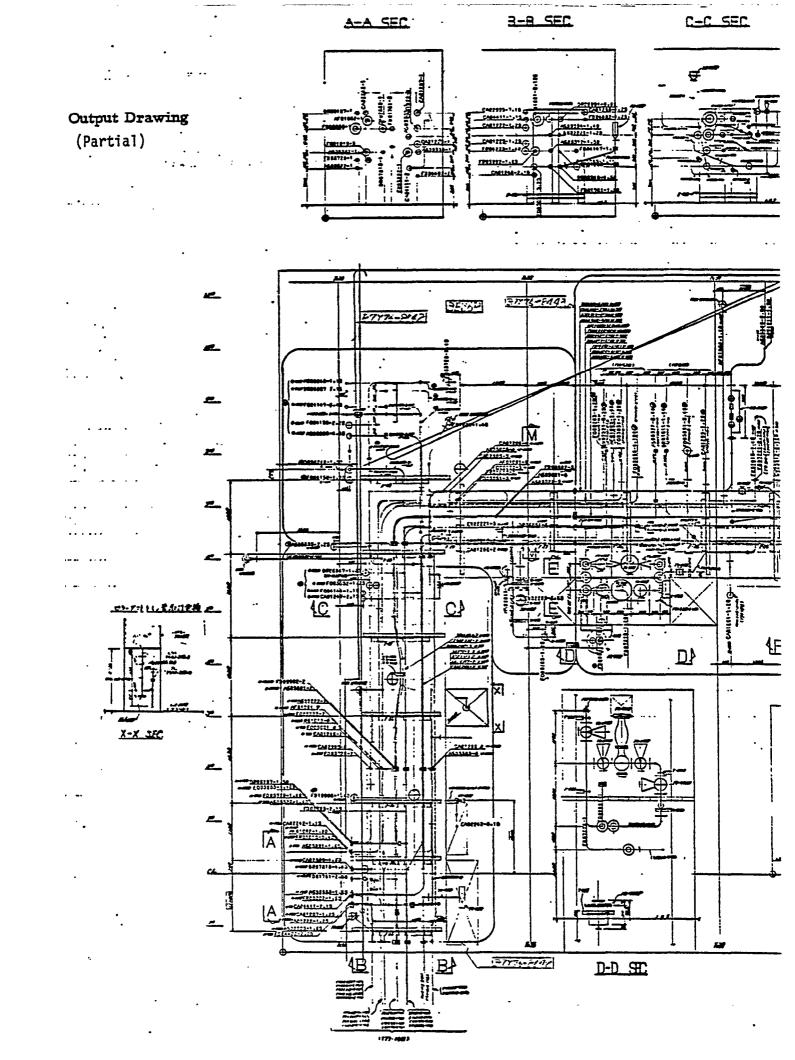


Fig. 3.5



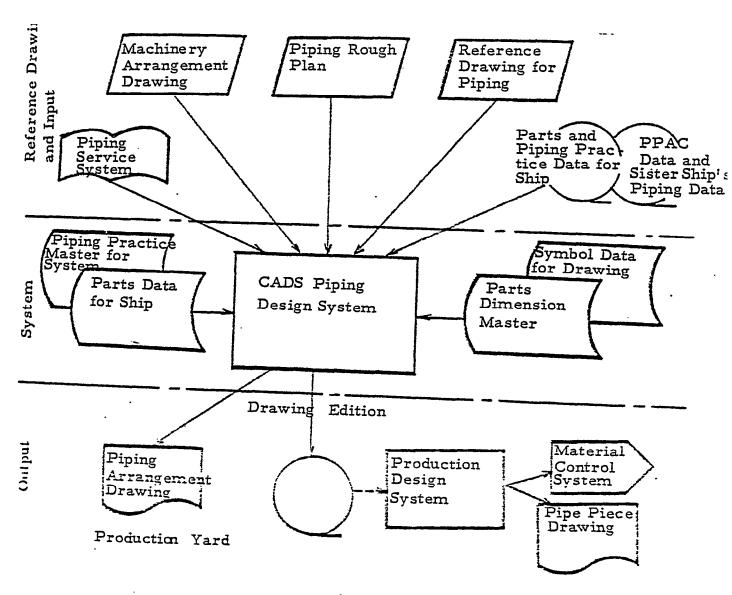


Fig. 5.1 CADS Operation

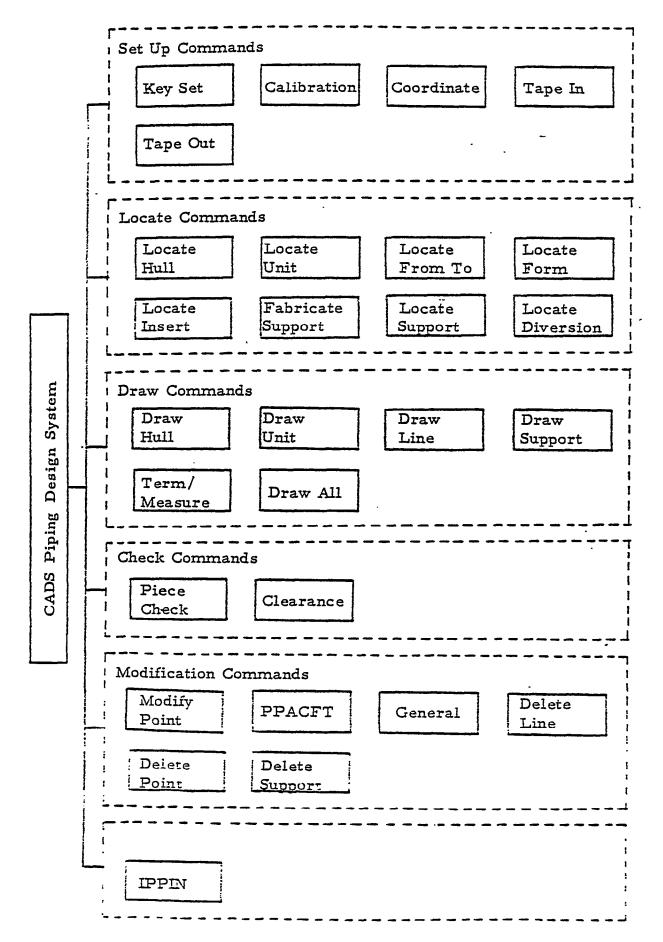


Fig. 5.2

## 5.2 Function of Commands

## 5.2.1 Set up Commands Start

- 1) To set initial values in the system.
- 2) To set initial values in data files.
- 3) To input particulars of piping arrangements such as Ship No., Drawing No., Zone, etc.

#### Coordinate

To define coordinates and their boundaries to be work out on AD table.

## Tape In

To read data from the magnet tape and set the data on the file.

Data in the file are pipe data and master data.

## Tape Out

To write data on the magnet tape for data transmission to host computer.

Data in the magnet tape are pipe data and master data.

## 5.2.2 Locate Commands

Locate Unit

- 1) To generate unit data to be stowed in unit standard file.
- 2) To locate unit data derived from unit standard file.
- 3) To alter unit data in deviated points from standard arrangement.

## Locate From-To

To input data of systems and to locate pipe endpoints in piping arrangement.

## Input data:

- (1) System names
- (2) Pipe end points by coordinates (X. Y. Z.)
- (3) Diameter
- (4) Pipe piece or fitting name
- (5) Others

#### Locate Form

To locate beding points on the line of the pipe end points through "Locate From-To" command.

#### Locate Insert

To set parts on the pipe line determined through "Form  ${\tt Command}$ "

#### Locate Diversion

To set the new pipe line on the designated position by using the piping data defined previously.

## Locate Support

- 1) To set pipe supports on the pipe line
- 2) To determine the position of U-bolt for pipe on the pipe supports.

## Fabricate Support

To design pipe support combining basic pattern of support.

Locate Hull

To define hull structure such as frame, longitudinal and deck as a background drawing.

#### 5.2.3 Draw Commands

Draw Hull

To make a hull structure drawing by data given by "Locate Hull" for checking.

## Draw Unit

- 1) To draw unit data derived from unit standard file.
- 2) To draw only the end points of the pipe line by data given by "Locate From-To" for checking.

#### Draw Line

To draw the pipe line by data given by "Locate Form" for checking.

To draw letters and leaders belonging to the pipe.

## **Draw Support**

To draw pipe support by data given by "Plural Support" for checking.

To draw letters and leaders belonging to the pipe support.

#### Term/Measure

To draw the following marks on the designated positions.

- Pipe Number
- Pipe Piece Number
- Pipe Fittings Number
- Pipe Support Number
- Pipe Size

To draw letters and leaders on the designated position.

To draw comments on the designated position.

Draw All

To make a final piping arrangement drawing by input data given in previous stage.

#### 5.2.4 Check Commands

Piece Check

To carry out piece dividing of the line and to check the length between two points in the unit of system.

Items to be checked are shown as follows.

- condition of piece fabrication
- condition of Pipe Piece Program
- direct welding
- length of short pipe, etc.

#### Clearance

To check the clearance between designated two pipes.

## 5.2.5 Modification Commands

**Modify Point** 

- (1) To modify entities of fittings.
- (2) To modify all of the coordinates or diameters of the pipe.

#### **PPACFT**

- (1) To edit pipe data prepared by PPAC.
- (2) To separate one pipe lines into plural pipe line

#### General

To delete and modify co-ordinates.

To delete hull structure, letters, leaders and comments.

Delete Line

to delete pipe data for each system or between designated two points on the piping route.

Delete Point

To delete a designated point of the pipe piece.

**Delete Support** 

To delete pipe support data in the unit of system or piece.

## 5.2.6 Fabrication and Outfitting Data

**IPPIN** 

To prepare data for pipe fabrication such as:

- Pipe piece Number
- MLF No. and/or MLC No.
- unit sign
- Loose sign, etc.

# APPENDIX H I HI REPORT ON COMPUTER-AIDED DESIGN SYSTEM

## K2123

TASK 2 ENGINEERING AND DESIGN

SUB-TASK 2.1 COMPUTER-AIDED DESIGN SYSTEMS

- 1. Examine and Study the SPADES System
- 2. IHI System
- 3. Comparison of the Capability between Levingston's SPADES System and IHI System

March, 1979

Prepared by: Masumi Hatake

IHI MARINE TECHNOLOGY, INC.

## 1.0. Examine and Study the SPADES System

The SPADES system of computer -- aided ship design has been examined and studied by IHI to determine if full utilization and benefit is being realized from the use of the system at the Levingston Shipbuilding Company. The study has been proceeded from November, 1978 through February, 1979 at the Engineering Office and the NC Department of Levingston. This report involves "Over View", "SPADES Modules" and "The Usage of the System at Levingston".

## 1.1. Over View

## 1) General

The SPADES system can be seen enough to cover almost over all from the design engineering to NC lofting and preparation for production in its function.  $_{\rm Its}$  capability shall be evaluated adequate to support many users which build various shapes and sizes of ships. In fact, it has been used by many shipyards in the U.S.A.

Moreover SPADES has enough space and more applicable field to be developed in the future. Newly developed programs, DEMO and SPAC, will make it possible to expand the users operation.

## 2) The Usage of the SPADES System at Levingston

The Levingston Shipbuilding Company is not using all modules of the SPADES system. Indispensable modules for minimum NC lofting are now in use. From the view. point of usage, SPADES system shall not fully display its worth at Levingston. The main reason shall be in lack of a large drafting machine. FAIRING and ship's hull calculation must be trusted to the Cali and Associates and others. And many difficulties and uneffective matters due to the lack of drafter can be found. If the drafting machine is provided, a useful module of SPADES, DEMO, shall be easily installed to this shipyard. In this concern, more detailed description shall be presented in 1.3.

# 3) Some Problems Pointed Out by IHI

(3-1) The output for making a material cutting list:

## 1.1. Over View (Continued)

The output through the system is good enough for full support of NC burning machine. On the other hand, the output for hand marking and hand cutting shall be rather poor. A material cutting list for flat bars, angles, slabs and face plates shall be obliged to be prepared by hand through the results of parts generation.

## (3-2) Stiffeners development on a web plate:

Concerning PARTS GENERATION, a web plate and the stiffeners on it, such as brackets, are separately defined. Most of stiffeners on a web have close relationship with it. If they are defined at a time, the output through PARTS GENERATION shall be automatically provided with the useful data such as:

Precisely marked starting and ending point of stiffeners in taking account of stiffeners plate thickness.

Marking the shifting direction of stiffeners plate thickness.

Drawing of stiffeners identification number on a web plate by drafting machine.

In this concern, more detailed study shall be continued later.

# (3-3) Installation of the shipyard's standard data:

Some of the designing standard of a shipyard can be easily installed to the system with a few input. The standard of cutout is a good example. However it

## 1.1 Over View (Continued)

does not seem to be *easy*, at this moment, to install the standard data in the field of production such as a bevel angle for welding, excess at joints and detailed end shape of stiffeners.

Taking a bevel angle for welding between stiffeners and a web plate, as an example, this angle shall be decided by the intersection angle between them. It shall be troublesome work to input the bevel angle by referring the manual of bevel angles as the key data of the intersection angle or referring working drawing. If the process can be treated by the system, it shall be more helpful for the users.

But this problem shall be resolved when the most of shipyards in the U.S.A. establish-their own standard. The urgent theme shall be to set up the scheme of the said standard in Levingston.

## (3-4) Problem concerning a curved shell unit:

The output for plate assembly from the PIN/JIG module might be inadequate. Dimensions for checking the shape of a curved shell unit, such as girth length at both end seam and end butt and diagonal length of a unit, are necessary as well as positioning data for the corner points of the unit and height dimension. at each pin/jig position.

However this problem has a close relationship with the fabrication method of a curved shell unit. Therefore it shall be prior theme to establish the fabrication method for a curved shell unit from gas cutting, bending, plate assembly and fitting frames and fitting web frames.

In this concern, more detailed description shall be presented in 1.2.

## 1.2. SPADES Modules

The following modules have been studied by IHI:

FAIRING
HULLCAL\*
HULLOAD
PARTGEN
NESTING
PLATDV
ROLL SET TEMPLATE\*
FRAME BENDING
PIN/JIG\*
DEMO\*
SPAC\*

Study has been extended to the modules which have not been installed to Levingston. (\* Marked Module)
Because it seemed to be necessary to evaluate the SPADES system correctly.

Any comments from Levingston shall be most appreciated.

## **FAIRING**

The Levingston Shipyard indirectly uses the FAIRING Program, because a large drafting machine to draw the faired frame lines is not provided. The Cali & Associates, Inc. calculates for Levingston and drawings as output and data to be stored on the SPADES Data Base are sent back to Levingston.

Lines fairing by computer has been one of the most basic matters for every shipyard in the world since a computer was first applied to Naval architecture. Because almost all of primary technology necessary for digitizing are involved in lines fairing. Smoothing, curve fitting, interpolation 'is a good example. Without these technology, NC (Numerical Control) and development of pieces of ship's hull, such as a curved shell plate, a curved frame and even an internal structure might not have been performed. Moreover from the view points of manhour saving, keeping high accuracy and scarce skilled loft man, lines fairing shall be a most essential theme.

It seems to be very hard to have a perfect fairing program for every kind of ship's hull. The ship's hull form is usually designed to fulfill the ship's purpose and ship's performance, so that the shape is varied very widely. Initial trim, skeg, bulb, notch, sonor dome and appendage are not easy to handle. Therefore, lines fairing program should be provided with the function of how to combine the said complex parts to the ordinary surface.

The FAIRING Program of SPADES system has been. observed to be well-designed. The principles of analysis and processing are very clear to understand for designers and loft men, and very reasonable. For example, curve fitting method of the program is very similar to

## FAIRING (Continued)

the one of fairing by hand. The other outstanding function is to recognize the characteristics of curvature and point; such as straight, knuckle, tangency, a large radius curve and a small radius curve.

The Fairing Program has been serving to not only Levingston but also many other shipyards in the U.S.A., which build many kinds of ships. From this fact, the program is to have adequate flexibility to the variation of ship's hull form.

## HULLCAL

The Levingston Shipbuilding Company does not use the HULLCAL program. SHCP (Ship Hull Characteristics Program) developed by the U.S. Navy is now in use at Therefore, it is not possible to report the usage of the HULLCAL program in practice.

However, a brief comment on the HULLCAL program can be presented through the reference of its user's manual. The following calculations can be performed by the HULLCAL program.

Curves of Form and Hydrostatics

- . Bonjean Curves of Stability
- . Floodable Length Curves
- . Tank Capacities and Sounding/Ullage Tables
  . Prim calculations
  . Damage Stability

- . Longitudinal Strength
- . End Launching Calculations

Tables of the results are printed out. Drawings by the N/C drafting machine of the results can also be generated as an option.

Levingston has a problem to be solved in launching calculation: Side launching is obliged at Levingston, however, there is no suitable calculation computer program.

In addition, it seems to be useful if information and data necessary for ship's operation and test trials before ship's delivery, such as, Turning Test and Crash Stop Astern/Ahead Test, are performed.

## **HULLOAD**

The HULLOAD program is a module to generate descriptions of hull structures and related design data by utilizing the hull definition and frame contours, generated and stored by the FAIRING program or obtained as data in other forms.

These descriptions are permanently stored on the SPADES Data Base, which can be retrieved by other SPADES modules.

The following structures and hull geometry can be generated and stored by the HULLOAD program.

Additional Frames and Canted Frames at an angle to the centerplane

- . Breasthooks
- . Decks of any shape, defined by sheer and camber contours
- . Deck Longitudinal and Seams
- . Shell Longitudinals and Sight Edges
- . Longitudinal Bulkheads
- Cut-Out Requirements
- . Bulkhead Stiffeners and Seams

The HULLOAD program is provided with sufficient, capability to generate hull structures and hull geometry necessary for succeeding design works. The high availability of the program can be supposed from the feature of generating CANTED FRAMES and BREASTHOOKS which requests an interpolating technique of high grade. Input data of the program is not so much and re-input due to design change is also easily performed.

## PARTGEN (Parts Generation)

PART GENERATION is a program primarily designed to allow the lofting of all flat plate parts needed in the hull of ship. A limited amount of developed parts such as cambered deck can also be generated.

The output of the program consists of one or more of the following:

- . Storage of the individual part or burning tape into data base
- . Nesting information for use with the nesting program
- . Paper tape for the drafting machine
- Calculation list for making a material cutting list by hand

PARTGEN program shall be considered good enough in its function which should satisfy the needs from NC lofting. The adequate function to define element surfaces of parts; such as points, straight lines, circles and curves; and to define contours, holes, cutouts and stiffener's location are provided. In addition, the function to retrieve the description of ship's hull previously loaded and the function to calculate geometrical matters used for defining parts and for making a material cutting list.

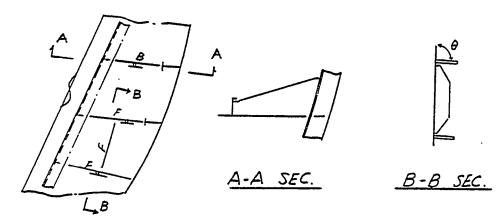
However, the following can be pointed out by IHI for future betterment.

# 1) <u>Definiation of Stiffeners on a Web:</u>

A web plate and the stiffeners on it such as brackets are separately defined by this program even if most of stiffeners on a web have close relationship with it. The stiffeners also correlate each other.

## PARTGEN (Parts Generation) (Continued)

This may be very clear by the following figures:



If they can be described at a time, following merits shall be expected.

- Input for parts generation of stiffeners can be saved.
- Output for all parts of the sub-unit can be generated at a time.
- Marking line for stiffeners can be precisely drawn in taking account of plate thickness.
- The shifting direction of stiffener's plate thickness can be drawn automatically.
  - Stiffener's identification number (piece mark) on a web plate can be automatically drawn by a drafter.

# 2) Output for a Material Cutting List

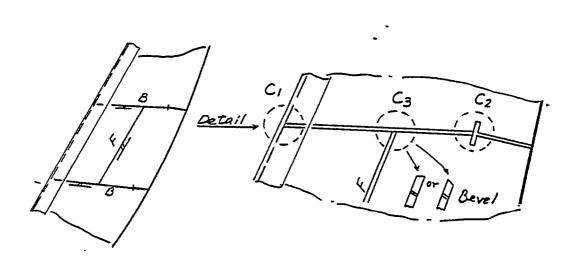
Necessary information and data are generated by PARTGEN and printed out to make a material cutting list for flat bars, T-bars and face plates. Comparing with the function for full support of NC burning machine, this output seems to be rather poor. If some features for automatically drawing a material cutting list are added, the output shall become a finished material cutting list.

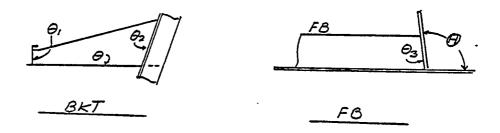
## PARTGEN (Parts Generation) (Continued)

## 3) <u>Installation Shipyard's Standard</u>

Some of the designing standard, such as shape of bracket, end cut of frame and cutout are easily installed by the system. (HULLOAD and PARTGEN) However, it does not seem to be easy at this moment to install the production standard, such as a bevel angle for welding and excess at joint.

Taking a bevel angle for welding at a bracket on a web, as an example, this angle shall be decided by the intersection angle with the correlating structures. Refer to the following figures.





## PARTGEN (Parts Generation) (Continued)

Bevel angles  $\theta$ ,  $\theta_1$ ,  $\theta_2$  and  $\theta_3$  shall be decided from  $\Theta$  and from the intersection angles at  $C_1$ ,  $C_2$  and  $C_3$  respectively. Though these intersection angles can be calculated and printed out by PARTGEN, the actual bevel angles  $(\theta, \theta_1, \theta_2, \theta_3)$  will have to be decided by user. In addition, it seems to be important to remind that the precise length and the precise location of the stiffeners can be finalized through this process.

If these standards or schemes are established and treated by the system, users can be released from this troublesome works.

## NESTING

The NESTING Program defines location (in the plate) and alteration of pieces, center punching, inner hole and outer contour. Finally the program generates a paper tape for a NC burning machine. This program is provided with almost all necessary functions, such as Piercing, Common cut, Automatic center punching sequence and Bridging, except Bevel cutting.

For bevel cutting, a programmer codes in NC tape to stop the machine at both start and end point of a surface with bevel. Then a machine operator sets the bevel angle of the cutting torch by hand. This work seems to be inconvenient for operators because they might have no information on the direction where the machine goes ahead. It shall be recommendable to have automatic cutting of bevel, however, it may be complicated to be provided with the function of automatic bevel cutting which can be of great help for saving time and precise cutting.

In this concern, more detailed description shall be presented in another report "Numerical Control Steel Fabrication".

## <u>PLATDV</u>

The PLATDV program is a module to develop **three** dimensional surface into flat plate, commonly used for shell plate development, for cold roll process.

The methods used in the program are triangulation and girth length. There can be seen some problems in accuracy at the surface with large curvature as described in the user's manual. Though it is not very clear that the said inaccuracy is caused by the developing method itself or by cold roll process, there may be problems in the both.

Concerning the developing method itself, the following shall be considered to be the cause of inaccuracy:

- 1) The girth length is calculated from the arc length of the circular arc to be defined with three space points. In case of "S-curved" surface and sparse arrangement of space points, the girth length of the circle shall differ from the *true* length.
- 2) Error shall be accumulated during repetition of development from the starting part to the final part of plate by the triangulation method. This tendency shall become conspicuous as the length of plate is longer and as the curvature is larger.
- Function to correct the error during cold roll process is not provided. Generally the plate is known to be stretched by cold roll. This margin should be considered.

As described at 'FAIRING', SPADES provides  $a \mod 1$  interpolation algorithm. If this algorithm is fully applied to approximate and interpolate the curved surface, the result shall be much amended.

## ROLL SET TEMPLATE (Manufacturing Aids)

For lack of a large drafting machine, the ROLL SET TEMPLATE program is not now in use at Levingston. So that the templates are obliged to be made manually by loft men.

A brief comment on the ROLL SET TEMPLATE program can be presented through hearing from Levingston's personnel who used to utilize the program.

A number of templates at frames including two extremes can be provided by the program. In case that the plate to be bent has a double curvature or a tight curvature lengthwise, other templates at longitudinal frames and/or water lines shall be prepared by hand referring output lists through the other programs of Manufacturing Aids.

It seems to be troublesome a little for loft men to judge the curvature of plate and to handle those programs. It may be more applicable if the ROLL SET TEMPLATE program handle the said conditions by itself.

## FRAME BENDING

The FRAME BENDING Program generates the information required for bending longitudinal and transverse frame and produce templates for the end cuts of the beams. The output can be in one or more of several different forms:

- Full size/reduced scale templates or drawings of the true curvature or the inverse curvature.
- Tabulated offsets of the inverse or true curvature at specified increments along the chord of the beam.
- . Full size templates for the end cuts of the beam.

Using these output, the following works can be performed:

- . Marking full size templates by hand.
- . Preparation for conjunction with a frame bending machine.
- . Checking after bending by a previously marked inverse curve.

Concerning this subject, it might be meaningful to note that bending performed by the FRAME BENDING is limited to two-dimensional curvature. On the other hand, IHI system allows to bend three-dimensional curvature (space curvature or twisted curvature). The difference might come from the one of bending practice and design practice between U.S. shipyard and Japanese shipyard.

## PINJIG

The PINJIG Program has never been used by Levingston, because assembly on pin/jig has not been applied. Through the study by the users manual, following problems could be found.

Positioning data for the corner points of the unit and height dimension at each pin/jig position shall be fulfilled with output through the PINJIG Program. However, more information shall be required for assembly such as:

- . Girth length at both end seams and end butt.
- Diagonal length of a unit to check the shape of a unit.
- Fitting angle of internal structures (such **as** web frames, girders and longitudinal/transverse frames) to shell plate.

This information may be calculated by the program "Manufacturing Aids". It shall be recommendable that the system is provided with the function to totalize the output from the PINJIG Program and the output of the said additional data which may be calculated by the modules of Manufacturing Aids.

## DEMO (Engineering Detailing Module)

1) The Levingston Shipbuilding Company has not yet installed 'DEMO', which was a newly developed module to utilize the time and effort spent during the detail design phase for numerical description of the ship structure. Therefore, it is not possible to report the usage of 'DEMO' in practice.

However, it seems to be necessary to report on the new module DEMO. Because DEMO is one of the representative modules to evaluate the functions of the SPADES system, especially to evaluate its-future development.

A study by the preliminary description of DEMO module has been done and a few comments are presented as below.

- 2) The characteristics and purpose of the DEMO modules are as follows:
  - a) <u>Visual checking</u> of the previous loaded data base increases and the data base becomes more comprehensive, verification of loaded data becomes more and more difficult. By checking the data loaded on the data base with drawings performed through this module, the possibility of errors downstream during part generation can be greatly reduced.

# b) An efficient tool for detailing:

All through members affecting other surfaces must be handled by 'HULLOAD'. On the other hand, local details will be-defined by 'DEMO'. Details

## DEMO (Engineering Detailing Module) (Continued)

#### are defined as follows:

. Stiffeners:

Symbolic Name Contour Definition Shape Code Number Orientation (near side or far side)

. Seams:

Symbolic Name Contour Definition Welding Detail (bevel and gap) Thickness on both sides

. Holes:

Symbolic Name Contour Definition Thickness, Width and Offset of Face Bar

. Brackets:

Symbolic Name
Contour Definition or Standard
Detail Identification
Thickness
Width and Thickness of Flange

. Inner Lines:

Contour Definition Width and Thickness of Face Bar

O) Saving manhour input for part programming:

Using the function of parts separation of part generation to the detailed designed structures by DEMO, manhour input for part programming will be greatly reduced.

Programming capabilities and language are designed as close to 'PARTGEN' as possible, and all 'PARTGEN' tools, such as Math, Contours, Symbolic Calls, Loops and Reps will be available.

## DEMO (Engineering Detailing Module) (Continued)

- d) As a result of design work performed by 'DEMO' associated with 'HULLOAD', structural drawings can be completed with the exception of lettering and dimensioning.
- 3) IHICS (Integrated Hull Information Control System) of IHI involves the close function of 'DEMO' and 'HULLOAD'. There can be found close similarity on the purpose and aim between the SPADES system and IHI system, even if detailed approaching methods are different from each other.
- 4) Before installation of 'DEMO' for Levingston, some of the preparation should be requested.
  - a) A large drafting machine should be necessary.

    One of the main purposes of 'DEMO' is on
    verifying drawings. In order to make 'DEMO' useful,
    a large drafting machine will have to be used.
  - b) Preparation on Levingston's design standard:

In order to use 'DEMO' in full worth, design standards will have to be reviewed though some of them have been established.

Design standards are as follows:

- . Symbolic Name of Stiffeners, Seams, Holes and Brackets.
- Contour (shape) of Stiffeners and Brackets.
- Welding
- End Connections of Stiffeners (lap, snipes, knuckles)
- Drawing standard

These standards should be arranged for easy application to 'DEMO'.

## DEMO (Engineering Detailing Module) (Continued)

5) Future scope of 'DEMO':

It seems to be meaningful to try thinking of the future scope of 'DEMO'. As mentioned in 'Over View' of this paper, the SPADES system has enough space to be developed in the future.

'DEMO' and 'SPAC' (Ship Production and Control Module) shall be nucleus for the future development of the SPADES System. From the viewpoint of computer technology, there can be found interesting matters around the said modules. Technology of cathode lay display is one of them.

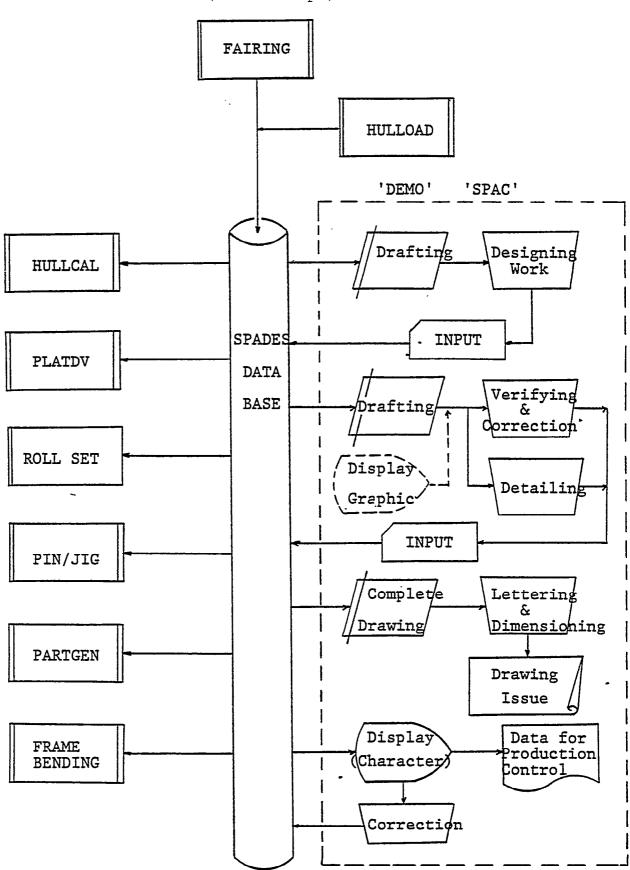
Quick verification of data on the Data Base, detailing by 'DEMO' and correction can be easily performed by a character display or a graphic display. After checking the data, drawings shall be completed by a large drafting machine. This process shall be of great help for reducing the turnaround time. A sample work flow on this matter is displayed as follows:

It does not seem difficult to attach these features to the SPADES System. However, the development will have to proceed in taking account of the cost to be required for users.

Though the future scope of 'DEMO' can be expanded to the said advanced computer technology, the recommended future scope at Levingston shall be to install 'DEMO'. associated with a large drafting machine and without the cathode lay tube technology.

DEMO (Engineering Detailing Module) (Continued)
 A Sample Work Flow with Advanced 'DEMO' and 'SPAC'

(Future Scope)



SPAC (Ship Production and Control Module)

The Levingston Shipbuilding Company has not yet installed 'SPAC' which was a newly developed module to utilize the data needed to generate the required reports for production. Therefore, it is not possible to report the usage of 'SPAC' in practice.

By preliminary description of 'SPAC', the following functions shall be provided to 'SPAC' as listed below.

- Unit weight of pieces and weight and centers of gravity of assemblies and sub-assemblies.
- Length and nesting within standard lengths of shapes of the various individual shaped pieces.
- Cross reference between assemblies due to the nesting into a plate of pieces belonging to different assemblies.
- Processing time for N/C burning tapes and flame planer sketches.
- Bulk material allocation for pieces produced through shearing or 'one-to-one' optical burning.
- Revision control is maintained by the system for all the issued reports generated, and many others.

IHI System involves such kinds of modules in use as:

Piece list issuing program for sub-assembly, assembly and erection.

of piece (including material dimension)
are involved.

#### SPAC (Ship Production and Control Module) (Continued)

Material control program and others.

Through IHI's experience, the function of 'SPAC' as a necessity is easily understood, especially the function of piece list issue required at Levingston. However, these matters shall request for user shipyards to make more clear their method for administration of their organization from engineering through production. Therefore, it might be prior matter to establish clearly their production process and material flow to install 'SPAC'.

#### 1.3. The Usage of the SPADES System at Levingston

1) The modules of the SPADES System to be used at Levingston are shown in the following table:

MODULE	USAGE	REMARKS		
HULLCAL	Unused	SHCP (Ship Hull Character- istics Program) developed by U.S. Navy is now in use.		
FAIRING	Used Indirectly	For lack of a drafting machine, Cali & Associates Inc. calculates for Levingston.		
HULLOAD	Used	Checking by a small plotter		
PARTGEN	Used	Ditto .		
NESTING	Used .	Ditto		
PLATDV	Used	Ditto .		
ROLL SET	Unused	Making templates by loftmen		
FRAME BENDING	Used Partially	Bending list only to be calculated		
PINJIG	Unused	Plate assembly on PINJIG is not performed.		
DEMO	Unused	Not yet installed		
SPAC	Unused	Not yet installed		

#### 1.3. The Usage of the SPADES System at Levingston (Continued)

As shown above, only five modules of the SPADES System are now in use and another one is partially used. As for HULLCAL module and module, concerning fabricating a curved shell unit, there can be seen clear problems as follows. For the others, a totalized study shall be required. In this concern, a detailed study and recommendation shall be presented later.

#### 1.3. The Usage of the SPADES System at Levingston (Continued)

- 2) For HULLCAL module, following problems have been pointed out by the user engineer at Levingston. Therefore SHCP (Ship Hull Characteristics Program) developed by U.S. Navy is now in use instead of 'HULLCAL' of SPADES.
  - (2-1) Problems pointed out at Levingston:
    - The program of Damage Stability Calculations cannot be sunk below the Margin Line in damage at 0 heel. The Margin Line cannot be made above the uppermost deck in the present version at Levingston.
    - Documentation does not always agree with the User's Manual.
  - (2-2) Additional problem at Levingston:
    - Side Launching is obliged at Levingston, however, there is no suitable launching calculation program neither in SPADES
  - (2-3) Information and data necessary for ship's operation and test trials before ship's delivery:
    - . It seems to be useful if the said information, such as Turning Test and Crash Stop Astern/Ahead Test are performed by
- 3) PLATDV, ROLL SET, FRAME BENDING, PINJIG:

These are the modules for fabricating a curved shell unit. PLATDV is only in full use and FRAME BENDING is partially used. The others are not in use. The usage of these modules seems to be unbalanced. Lack of a large drafting machine shall be the cause. However, it might be possible to make a template for bending a curved plate or a curved frame through the output of 'ROLL SET' or 'FRAME BENDING'.

#### PLATDV, ROLL SET, FRAME BENDING, PINJIG (Continued)

As far 'PINJIG', fabrication method for a curved shell unit is the primary theme. Levingston has not decided to adopt the assembling method of 'PINJIG'. One of the biggest differences in constructing technology between Levingston and IHI lies in this method. This theme shall be studied through the Technical Transfer Program.

4) Total View on the Problems Facing at Present and in Future

The urgent themes to be studied at Levingston are as follows:

- . Working drawings to be issued as fast as possible with a suitable advance for study on fabrication method by production people.
- . Easy and quick follow-up to design change caused by the Owner's opinion, claims from the classification society and convenience of production method.
- . Manpower saving both at the Engineering office and Lofting.
- . Error caused by mis-input to she SPADES modules to be minimized for the smooth operation of the system and for preventing the production process from confusion due to errors.
- . Consideration for scarcity of skilled loftmen.

The following are supposed to be concrete problems and difficulties to realize the said themes.

- 4) Total view on the problems facing at present and in future. (Continued)
  - (4-1) LINES FAIRING must be trusted in Cali & Associates, INC., so that TAT (Turn Around Time) of it gets longer and money is lost. FAIRING IS the most fundamental for quick start of every designing work. If the work can be performed by Levingston itself, the faired lines shall be brought more quickly in low cost.
  - (4-2) Visual checking is not so easy. Results of 'HULLOAD', 'PARTGEN' and 'NESTING' are obliged to be checked with small scaled drawings. This might cause some errors and inaccuracy.
  - (4-3) Duplicated works can be seen between the Engineering office and Mold Lofting.

Most of manhours are spent for making working drawings at the hull section of the Engineering office. The working drawing might be beautiful enough to express the ship's structure. The most detailed dimensions and all necessary information are involved in the drawings. Piece mark system, plate thickness shifting direction of stiffeners, cutouts, end cut of stiffeners and welding information are described. They are drawn by hand, as if drawn by a drafting machine.

On the other hand, input for 'HULLOAD' is simultaneously prepared at the Engineering office for the succeeding jobs. Then the input will have to be verified by drawing with a small plotter for only visual checking.

4) Total view on the problems facing at present and in future. (Continued)

If the input for 'HULLOAD' precedes and drawings are drawn by a drafting machine with a suitable scale., the drawings can be utilized as the base drawings from which more detailed design can be performed as well **as** checking drawing for input.

In addition, in N/C lofting at present, lettering and dimensioning will have to be obliged to be performed by hand referring the working drawings. This duplicated work is being performed in a small scaled drawing.

(4-4) Manual lofting in full scale still remains at pretty range.

In order to make a template for bending, a full scaled line is being drawn. This work can be performed by the computer system and by the sealed' lines. From the viewpoint of scarcity of skilled loftsmen, this kind of work might be replaced by the computer system as much as possible.

5) Some recommendations from IHI for the better usage of the SPADES system at Levingston:

Through IHI's study on the SPADES system used at Levingston, concrete problems described in the previous paragraph remain. What is needed at Levingston at this moment to solve the said problems shall be to make full use of the SPADES system. For each module of the SPADES system, IHI's recommendations have already been described in 1.2. The following are the additional recommendations from IHI.

- 5) Some recommendations from IHI for the better usage of the SPADES system at Levingston (Continued)
  - (5-1) To install a N/C Directed Drafting Machine for the Engineering and the Loft:

A small plotter has been provided for the loft. However, it cannot satisfy the need from Engineering and Lofting as aforementioned at 1.3-4. By utilizing  $a\,\mathrm{N/C}$  Drafting machine, these problems can be fairly proved.

The requested drafting machine shall have a  $70" \times 150"$  effective drafting area and its cost is estimated to be \$100,000-\$150,000. Installation of a N/C Drafting machine makes it easy to install 'DEMO', which shall contribute to the betterment described already, such as:

- . Fast issue of working drawing
- Easy and quick follow-up to design change
- . Exclusion of duplicated works
- Manpower saving
- (5-2) To confirm data flow from the Engineering office, including lofting through production:

In order to realize the said urgent theme, the actual data flow extended from Engineering Office through Production field, which is now getting clearshall be expected to be confirmed. This confirmation and selection of IHI's recommendation described in this paper (1.2, 1.3) are requested to be proceeded simultaneously.

#### 2.0. <u>IHI System</u>

The Computer-Aided Hull Design System of IHI covers almost over all of hull design field from the initial design, detailed design through production engineering.

IHI system consists of six (6) independent main systems, which support their particular functions. Some of them are designed to communicate with the other systems through their data base to keep a coincidence of common data between them and to save input. Main systems are as follows:

ZPLATE, ZVIBRA

Structural Analysis

SPECS

Ship's Hull Calculation
Operational Information Calculation

FAIRING

Lines Fairing

SHELL

Shell Plate Development
Template for Bending
Assembly Unit Marking on PIN/JIG
Supporting Jig Height Dimension

LODACS

Longitudinal/Transverse Frame Development Template for Bending

IHICS

Ship's Hull Description
Section Design
Parts Generation
NC Lofting (NC Drawing, NC Burning Machine)
Parts List Issue
Data Communication by Display Terminal

#### 2.0 IHI System (Continued)

Since many of documents concerning this subject have been already submitted to Levingston, only special characteristics of the said systems shall be presented here. The following are the documents to have been submitted.

(Through our Memo Ref. No. FPC-073 dated February 23, 1979)

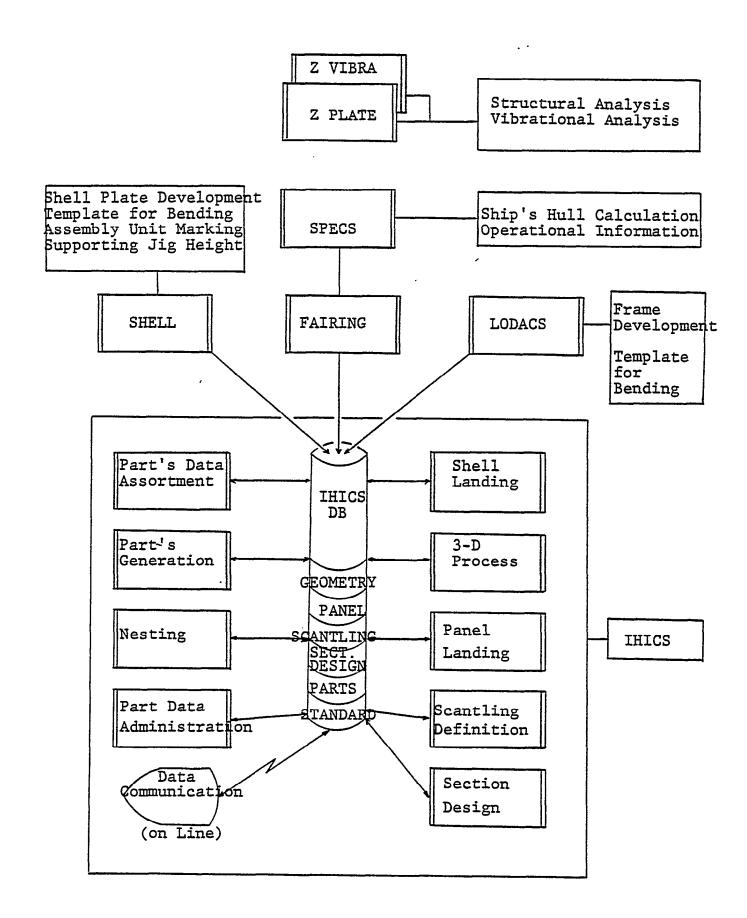
- 1. Brief Explanation of IHICS
- 2. IHICS Actual Input/Output Examples
- 3. Summary of IHI SHELL
- 4. LODACS Ship Frame Data Processing System
- 5. SPECS Ship's Preliminary and Exact Calculation System
- 6. SPECS Actual Output Example

(Through our Memo Ref. No, FPC-046 dated January 18, 1979)

- 7. The User's Manual 'Z-Plate' Output Sample. (Through our Memo Ref. No. FPC-029 dated December 20, 1978)
  - 8. General Purpose Program of Plane Stress Analysis by Finite Element Method and Its Application (ZPLATE)
  - 9. Matrix Method of Vibration Analysis of Framed Structure and Its Application

Relationship between the said systems and functions is presented in the following figure.

#### IHI Computer-aided Hull Design System



#### 2.1 Characteristics of Each System of IHI

Only special characteristics are described here. Other details are presented in the aforementioned documents. It is coherent in all systems that plenty of drawings can be generated through the systems and that designing know-how through IHI's long experience is concentrated in the systems.

#### 2.1 Characteristics of Each System of IHI (Continued)

#### ZPLATE, ZVI BRA

ZPLATE is a program for analysis of elastic plate structures under plane stress conditions, and especially a program for three dimensional structure constructed with plates or orthotropic plates will be distinctively useful for analysis of ship hull structure. There exists many good programs covering this field in the world. However, labor to make input data for the given problem is generally too large to use in practice.

ZPLATE solved this problem by adopting a substructure method with the function of automatic mesh generation and with periodic identical pattern. In addition, this program includes also a check device of input data and adjustment of calculated results by using a high speed plotter and a graphic display.

ZVIBPA has been prepared for analysis of vibrational responses in elastic range of three-dimensional framed structures under sinusoidal forced vibration. Because the effects of shear rigidity and rotatory inertia are taken into consideration, this method is particularly effective in the analysis of structures such as the ship hull structures where the effects of those factors cannot be ignored. For vibrational analysis method of framed structures, a method in general use obtains natural frequencies in free vibration by replacing the structure to be analyzed with a structure made up of a multiple mass and spring system. However, in framed structures where shearing deformation and rotatory inertia are taken into consideration, it is extremely difficult to obtain natural values by the use of the above method;

In ZVIBRA, an assumption of the framed structures to be a conglomerate of beams, which have infinite degree of freedom, has been taken and then their stiffeness matrices were obtained using such matrices, their vibrational responses were calculated, and as a result their vibrational modes and amplitudes were obtained.

## 2.1 <u>Characteristics of Each System of IHI (Continued)</u> SPECS

SPECS has been applied to detailed design in ship's hull preliminary calculation on Hull Form and its capacity. In addition, necessary information and data at ship's operation and at test trials before its delivery are also calculated. Almost all of the sub-programs of the system generate drawings and curves as well as tabular prints. The actual output examples can be referred to in "Actual Output Example of Ship's Preliminary and Exact Calculation System."

## 2.1 <u>Characteristics of Each System of IHI (Continued)</u> FAIRING

IHI FAIRING program performs Lines Fairing by adopting a newly developed interpolating method called 'N-Curve'. 'N-Curve', of which term stands on 'Natural Curve', can express straight line, true circle (arc) and a compound straight line with a knuckle point as well as 'natural curve'.

#### 2.1 Characteristics of Each System of IHI (Continued)

#### IHICS (Integrated Hull Information Control System)

IHICS is a series of program packages associated with the Data Base. IHICS serves for mainly describing ship's hull structure, section design, parts generation, NC lofting, NC burning machine and parts list issue. The special characteristics are as follows:

1) To be based on Data Base concept:

By excluding duplicate data, incoincidence of data is prevented and the space of the memory on Disk is much saved.

For easy maintenance of the systems and data, programs and data are independent from each other.

IMS (Information Management System) of IBM, Inc. has been adopted.

- 2) To generate sectional drawings at any location and at any phase of designing.
- 3) To generate a part program by the system itself (as well as manual coding).
- 4) To generate parts list for any stage of production (fabrication, sub-assembly, assembly and erection) at an early time.
- 5)- Parts generation with full information necessary for production (fitting angle, plate thickness shifting direction, bevel angle, piece mark and additional material).
- 6) Automatical reference of design standard and production standard:

Shape standard - cut out, scallop, bracket, stiffener end cut.

- . Standard how to select/apply standards.
- . Bevel and gap for welding.

#### IHICS (Integrated Hull Information Control System) (Cont.)

- 7) Register and revise of standard data.
- 8) Material cutting drawings (not lists) with full information necessary for marking.
- 9) Detailed working drawings for sub-assembly and assembly.
- 10) Consideration on design change (Refer to the attached paper).
- 11) Online capability -

Many online terminals are supported under the control of IMS/DB.DC.

12) Connection to Graphic Display.

Listed above are only a summary of the characteristics of IHICS. The more detailed description and actual output examples are presented in the submitted documents.

#### 2.1 Characteristics of Each System of IHI (Continued)

#### SHELL

The SHELL system constitutes an integrated and computerized data processing system which provides various highly accurated information pertaining to all the processes in production of a curved shell unit. The SHELL system covers shell plate development, making template for bending, unit marking on supporting jigs and jig height calculation. In addition, these calculations are performed under the common base which user planners can input the most suitable condition from the viewpoint of accuracy and workability in the production process. The standardization of production technologies and analysis of application know-how in the SHELL system has been established by mobilizing of the engineering power in IHI's five shipyards.

#### Special Characteristics of SHELL:

- 1) Shell system is a composite system for the geometrical calculation and data calculation and a data processing system relevant to the production of a curved shell unit.
- 2) Shell system displays a far higher level on accuracy as compared with the conventional system. The new ideas and method to be adopted for this matter are as follows:
  - . In order to make calculating algorithm simple as well as in order to keep the high accuracy of the output, the lines are drawn by the concurrence of points approximated by a certain supplementary straight lines.

The optional cut-plane method is fully adopted. This is a method to develop a curved shell plate in the view that the characteristics of its curvature are displayed best.

The desired plate is cut out of a larger expanded plan including the surrounding area of the plate.

In order to ensure the accuracy at plate bending time, the templates are set up at the right angle against the mean level of the curved plate.

(Refer to Fig. 1 - Fig. 3 in the 'SHELL Manual')

3) A remarkable improvement on workability and accuracy in the assembly stage can be expected. Because the various working

#### SHELL (Continued)

practice in the shop is taken into consideration from the first step of the system execution.

- . In order to keep the accuracy of angles between adjoining seam and butt at plate assembly, the intersection of datum planes in the supporting jig lines and shell plate are marked on each plate. In addition, the datum plane are orthogonal to the platform surface.
- Instructions can be given in connection with the position for setting plate, postion of stopper and the height of supporting jigs.
- The dimensions of unit, diagonal dimensions and the rate of curvature on seams and butts are calculated and displayed for checking at plate assembly time.
- The availability of automatic welding on the unit can be checked.
- Single panel assembly can be also available. (Refer to Fig. 4 - Fig. 5 in the 'SHELL Manual'),
- 4) Maintainability of data and system.
- 5) Easy recording of feed-back data.

More detailed description is presented in the submitted document.

#### 2.1 Characteristics of Each System of IHI (Continued)

#### LODACS (A Ship Frame Data Processing System)

The LODACS system covers frame development, frame bending and marking-cutting by hand or by NC burning machine. The LODACS system generates precise shape at both end of frame, inverse curve for bending, location of drain holes and template for bending. These data are drawn by a drafting machine, a plotter and dotto printer as well as printed out.

The LODACS supports three-dimensional bending as well as two-dimensional bending. The need of this function comes from fabricating method on bending.

More detailed description is presented in the submitted document.

#### ATTACHMENT

#### 1. CONSI DERATI ONS ON DESI GN CHANGE IN HULL PART

(IHICS: Integrated Hull Information Control System:) Ref. FPC-073 dated February 23, 1979

Some considerations on design change are involved in the system design strategy of IHICS system which was developed by IHI to perform hull design. The system has been utilizing full in worth.

Design change is considered to be caused mainly by the Owner's opinion and claims from the classification societies. It is very important to remember that designing work for detailing has not been allowed to keep until approval by the said agency is given to the prepared drawings. Therefore, the influences of design change have to be minimized on the designing procedure.

Another consideration to be taken in IHICS is flexibility to the design change due to the change of production method such as welding.

From the viewpoint of computer software, flexibility to this kind of change is fundamentally the same as design change occurring during the period of the actual ship's design.

Some actual examples of the subject to be taken in IHICS are as follows:

1) Relative expression on hull structure by defining language called 'LINE'. 'LINE' is designed to describe objective figures in relative expression as far as possible so as to minimize corrections caused by alteration of design.

This concept is coherent in all sub-systems. (Refer **to** "3) An example of 'LINE' description")

a) These descriptions will not have to be changed, 'even if the data on longitudinals such as offsets, material dimension and cutout are changed.

#### **ATTACHMENT**

These descriptions are executed by the system on both phase of designing and part generation.

b) Generation of physical data of every part can be postponed up to the execution of parts generation.

The shape of the cutout at every longitudinal frame shall be generated after all inputs are given to the system.

2) Independence of design standard data from the computer program.

Design standard data can be registered on the standard data base on the responsibility of designers. Registration and updating can be easily performed by the system.

The following are involved in the standard data base:

- . Shape standard: Cutout, Hole opening, Scallops, Bracket, End shape of stiffeners
- . Standard how to select/apply stands
- . Fabrication standard: Excess, bevels and gaps

#### 3. An example of 'LINE' descriptions

(An example for corner part of transv. section)

```
T
       F60
                                                         PAI
                                               UPP
    T1=F60----GEOMETRY DB REFERENCE
                                                DE
    PAl=PA-UD, ML--- PANEL DB REFERENCE
    P11=oUT(sL,UD,1) ---DESIGN DB REFERENCE
    Sl=PAl,Pll
    S11=PR-SL, L=2000, D
    LONG=SL, L40, L59
                       GEOMETRY,
                                   SCANTLING
    LONG=UD, L20, L21
                       DB REFERENCE
    Cl=TD-S11,T1,-S2
                                                                    P51
    Pl=CP-Cl
    P2=SLOT-PC2, UD-L20
    S5=PT-P2,PT-P1
    P3=INT(S5-C1),U
    P4=ON-C1, FROM-P3, GL=150, D
    P5=UD-L20,TOP
                            ----- COMPLEX SURFACE
    Al=P100,S11,C1,S2,
    S24=PT-P4, PT-P5, SCS=UD-L20, ECS=Al
    S21=SL-L57,
                     SCS=SL-L57, ECS=Al
    S25=UD-L21;
                     SCS=UD-L21; ECS=S23
C
    ----,Tl,MSL(SL,L40,L59),Tl,MSC(100),PAl,
                                                           CONTOURING
          MSL(UD, L21, L21), PA1, MBS(S5, UD, L20, P3, 21), C1
                                                           DEFINITION
Η
    MH(1,P51,150)
                                                           OPENING
                                                           DEFINITION
    S21=F,FIT=A,PD=D,MRK=U,TYP=FClSl,NAM=F15
X
    S22=B, FIT=A, PD=D, MRK=U, TYP=BClSlSl, NAM=B16
                                                     PARTS EXPANSION
                                                      (STIFFENER)
```

END OF PART PROGRAM

3.0 Comparison of the Capability Between Levingston's SPADES System and the IHI System

Comparison of the capability between Levingston's SPADES system and the IHI system is briefly presented in this paragraph, since deep study and detailed description has been reported in preceding paragraphs of this paper.

In addition, assessment of the interface problems and the adaptability of the IHI system software to the SPADES system is briefly tried for reference.

3.1 Comparison of the system component between the two (2) systems.

The following is a component figure of the both systems. The underlined is not used by Levingston.

	T				
APPLICATION	SPADES		IHI SYSTEM		
Structural Analysis	Unknown		Z-PLATE		
Vibrational Analysis	Unknown SPADES			Z-VIBRA	
Ship's Hull Calculation	HULLCAL* (Unused)		DB	SPECS	
Ship's Operationa Information	l Unknown		 	. SPECS	
Fairing	FAIRING		DB	FAIRLAND IHICS	
Ship's Hull Description	HULLOAD			Basic Data Creation Subsystem	
Section Design	<u>DEMO*</u> (Unused)	DB		Section Design Subsystem	
Part Generation	PARTGEN		DΒ	LINE System	
Nesting	NESTING			Nesting Program	
Part's Data Assortment	SPAC* (Unused)			Production Engineerin Subsystem	
SHELL Plate Developmen Rending Template					
Sending Template  Jig Height	Unused ROLL SET* PIN/JIG*		DB	SHELL	
Unit Marking	Manufacturing Aid			<u></u> _	
Frame					
Frame Development	FRAME BENDING (2-Dimensional)		ОВ	LODACS (3-Dimensional)	

3.1. Comparison of the System Component between the Two (2) Systems (Continued)

As shown in the aforementioned component figure of the SPADES system and IHI system, full configuration of the SPADES system covers much the same area as IHI system, except structural analysis and vibrational analysis through SPADES now in use at Levingston, is a part.

In conclusion, capability of SPADES is considered to be almost equal to IHI system. If the betterment for the problems described in 1.1-3) and 1.2 is taken into consideration, and if recommendations by IHI presented in 1.3-5) is studied, the SPADES system shall become full in worth.

## 3.2. Adaptability of the IHI System Software to the SPADES System

A short study on adaptability of the IHI system software to the SFADES system has been done for reference.

- 1) As far as the computer hardware configuration, there is no special problem for the subject, because the IHI system can be operated on IBM System/370 or IBM 3031 processor.
- 2) As far as the computer software configuration, there is no special problem for the subject except IHICS. IHICS requests to install the IMS of IBM Inc. (Information Management System)

#### 3) ZPLATE, ZVIBRA, SPECS

There is no problem to adapt to the SPADES system. These have (their own data file and do not request any interface.

#### 4) SHELL, LODACS

This system might request some interface for their data base unless a new data handler from the SPADES Data Base is attached to these systems.

#### 5) FAIRING

This system might request some post-processor form its data file to the SPADES Data Base unless the results are re-input to the SPADES Data Base from the offset book as the output.

#### 6) IHICS

It shall not be recommendable to adapt to the SPADES system, because almost all the modules were coded by IBM PL/1 based on IMS and too big to adapt.

However, algorithm of the system and design philosophy shall be considered to be convenient for reference.

#### Epilog:

This report **was** made by IHI through the investigation and the discussion with Levingston's personnel. IHI hopes that Levingston continues a study on the subject matter for the future betterment along the recommendations from IHI described in this paper.

# APPENDIX I I HI REPORT ON NUMERICAL CONTROL STEEL FABRICATION

K2123

TASK 2 <u>ENGINEERING AND DESIGN</u>

SUB-TASK 2.2 NUMERICAL CONTROL STEEL FABRICATION

March, 1979

- Prepared by: IHI MARINE TECHNOLOGY, INC.

M. HATAKE

#### NUMERICAL CONTROL STEEL FABRICATION

The N/C system now in use at Levingston for steel fabrication has been studied by IHI in order to quantify its utilization as compared to its potential capabilities. The study has been in process from November, 1979 through March, 1979, as a part of the extended computer-aided hull design system.

In this paper, a brief description on the usage of the N/C system at Levingston shall be presented since the more detailed description on the function of the system itself can be found in the attached paper. Then a brief study on the characteristics of the usage of the N/C system at Levingston shall be described comparing with the IHI system. Finally, some recommendations from IHI shall be presented,

#### Attached Papers:

- Specification summary for N/C 3000 and N/C 3300 SERIES FUME CUTTERS by C-R-O Engineering Co., Inc.
  - Detailed functions of the same machine as Levingston's are described.
- 2. IHI High-speed N/C Marking. System,-
  - This machine was developed by IHI to perform high-speed (about 60' per minute) line marking.
- 3. IHI Multi-torch N/C Cutting Machine.
  - This machine was developed by IHI to perform gas cutting by three torches simultaneously and to perform line marking by two torches simultaneously. This machine was provided with

### Attached Papers: (continued)

mini-computer in order to perform checking the interference of each torch, speed control, coordinate transformation and many others.

#### 1.0 The Usage of the N/C System at Levingston

C-R-O Double Model N/C 3000 flame cutter is now in use at Levingston. It is controlled by Kongsberg CNC 500 with the DNC (Direct Numerical Control) software. The machine is provided with almost all necessary functions to perform the N/C steel fabrication at Levingston such as:

Two sets of the master torch associated with two slave torches for each master torch.

Bevel torch for bevel cutting of (K, V, Y, X).

Center-punched marking equipment.

Automatic pierce rate control.

Automatic start-cut control.

Rotating triple-torch suspension.

Height sensing control.

Water spray system for preventing heat torsion.

Torch ignitor.

Others

The machine has been utilized to fabricate almost all hull pieces, which are cut out of raw plates, except square-shaped small pieces to be fabricated by shearing. After its installation at Levingston in April, 1975, this. machine has covered Levingston's needs with its capability since the work volume of steel fabrication at Levingston was not too extensive. (The past record was 500-metric tons per month.)

#### 1.0 The Usage of the N/C System at Levingston (continued)

However, since it is estimated that the work volume of Levingston's steel fabrication shall be increased to 1500-metric tons per month through the construction of the bulkers, the work load of the N/C machine is expected to become extremely heavy to fabricate the pieces aforementioned. Therefore, some considerations such as; leveling of the work load, rearrangement of supporting area of objective pieces and installation of an additional machine, shall be requested at this moment. In addition, an alternate method of N/C burning machine to compensate the work load in the time of "machine-down" and "repairing" shall be recommended to be studied. In respect to this concern, a little deeper consideration-shall be presented later.

## 2.0 Characteristics of the Usage of the N/C System at Levingston

In this chapter, a brief study on the characteristics of the usage of the N/C system at Levingston shall be described comparing the Levingston system with the IHI system.

#### 2.1 Objective Hull Pieces for the N/C Machine

A comparison of the subject title between Levingston and IHI-AIOI Shipyard which constructed the "Future-32" bulkers can be seen in Table 1. The Levingston N/C machine has been utilized to fabricate most of the hull pieces.

On the other hand, a quarter of all plates of a "Future-32" was fabricated by the N/C machine in IHI-AIOI Shipyard.

In regard to this concern, more detailed descriptions will follow as shown below:

1) Difference of the work volume between Levingston and IHI-AIOI Shipyard.

The past record of steel fabrication at Levingston equipped with one (1) N/C machine was 500-M.T. (metric tons) per month. IHI-AIOI Shipyard equipped with three (3) machines keeps 6000 - 8000 M.T. per month.

- 2) Difference of the philosophy on steel fabrication.
- (2-1) Levingston stands on the viewpoint that N/C burning is the best way from its high precision and from its capability of burning (2) to (3) plates at a time.
- (2-2) IHI stands on the following viewpoint:

  Though a N/C burning machine is best for high precision and it has a capability of burning (2) to (3) plates at a time, preparatory work for this such as, nesting operation by this system takes man-hours; nesting of twenty (20)

# 2.1 Objective Hull Pieces for the N/C Machine (continued)

pieces into a raw plate shall take 1.5 - 2 hours, for example: In addition, the layout for a N/C machine needs a Wide space in the shop, Therefore, the use of the N/C burning machine had best be limited to the pieces requiring high precision and the repetitively cut plate from the same N/C tape. In fact, IHI's N/C machines are being utilized for the main structures at cargo hold such as, web plates, floor plates and girders and for the curved shell plates.

# 2.3 Plate Assembly

For a flat plate unit such as main deck, both Levingston and IHI adopt the preceding plate-assembling method. However, the sequence of it is much different from each other.

At Levingston, marking on each plate is completed by a N/C machine before layout of plates and welding, On the contrary, at IHI each plate is neatly cut by a flame planer and welding of plates follows, Finally, marking on a unit (not plate) is performed before putting the internal structure such as sub-assembled webs on it.

3) Difference of Mold Lofting and Marking Equipment.

IHI has been performing "1/10 scaled mold lofting" except a few sull-scaled loftings which are scaled up from 1/10 output to full scale and converting from output prints by the computer system to steel tape templates for, unit working. A large and highly precise drafting-machine and an Electro Photo Marking System made it possible, "1/10 scaled mold lofting" - at IHI.

# 2.1 Objective Hull Pieces for the N/C Machine (continued)

An Electro Photo Marking System (called shortly (EPM) was developed more than 10-years ago by marriage of Electronics and Optics. It can work a raw plate in eight (8) minutes. It's principle is the same as the one used by photographers. In EPM System, a film is a raw plate on which powders (called photoner) charged static electricity are scattered, and the object is to be taken in photo and is a nested original pattern in 1/10 scale. The original pattern is expanded by a very precise lense into 1/1 scale. The precision of the printed pattern is good enough for the hull pieces.

On the other hand, neither a large drafting machine nor a precise marking equipment piece-is provided at Levingston. In addition, skilled loftsmen are scarce. From these stated reasons, the N/C burning machine is supposed to be very important and very vital at Levingston.

4) Re-arrangement of supporting area of objective pieces for the N/C machine.

The work load of the N/C machine at Levingston shall become more increased through the construction of the bulkers as described in the previous chapters. Moreover, a consideration for "machine-down" of the most vital machine shall be urgent; so that Levignston started thinking of shifting a flat plate (square shaped) from the N/C machine to the flame planer and/or hand cutting. It shall be recommended.

Table 1

# OBJECTIVE HULL PIECES FOR N/C MACHINE

Company Objective Pieces	LEVINGSTON	IHI-AIOI SHIPYARD (F-32)
WEB PLATE (HOLD)	·	
FLOOR PLATE (HOLD)		N/C BURNING
MAIN GIRDER		<u>117 PLT</u>
WEB PLATE (OTHERS) FLOOR PLATE (OTHERS) GIRDER (OTHERS) BKT	N/C BURNING	n/C DRAWING ERECTO MARKING HAND CUTTING 622 PLT*
CURVED SHELL		N/C BURNING 240 PLT
FLAT SHELL  FLAT DECK  FLAT WALL  FLAT BULKHEAD		Flame planer 606 plt
FLAT BAR  FACE PLATE  FLAT SHAPE	MATERIAL CUTTING LIST HAND MARKING	N/C DRAWING  ERECTO MARKING *Include Above  HAND CUTTING  MATERIAL CUTTING
CURVED SHAPE	HAND CUTTING	LIST P HAND MARKING 1900 HAND CUTTING
COLLAR PLATE, ETC.	SHEARING OR N/C BURNING OR HAND CUTTING	MAGNET TRACER OR PHOTO TRACER

I) Numbers of IHI's columns are the actual plate numbers of "Future-32" constructed at IHI AIOI Shipyard.

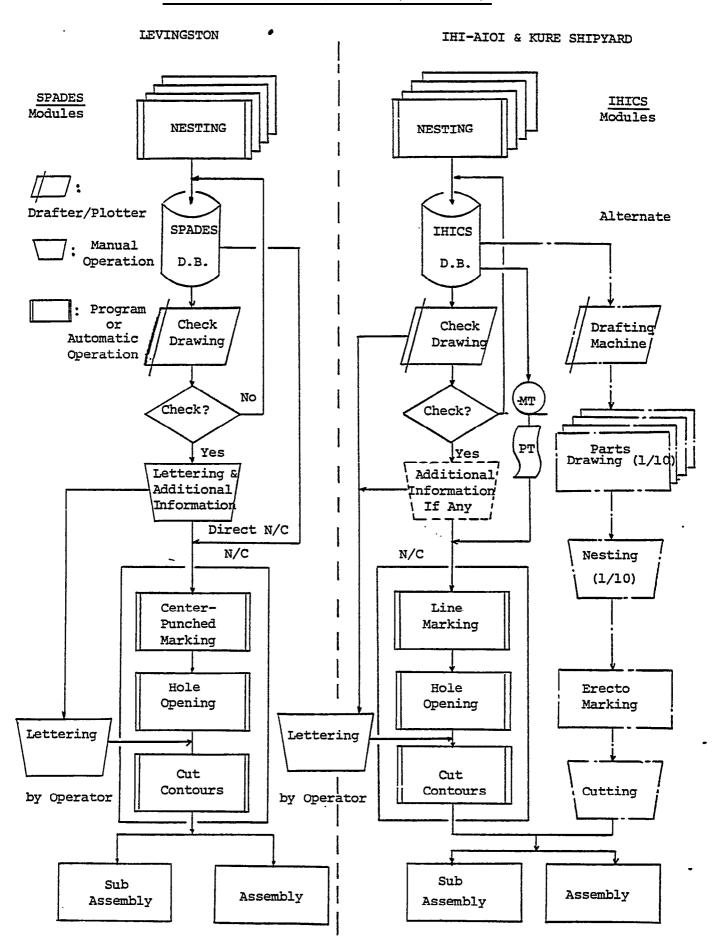
# 2.2 Process Flow of N/C Works (At Present)

A comparison of process flow of N/C works between Levingston and IHI-AIOI Shipyard can be seen in Figure 1. This figure displays a very simple flow from the computer system to the production through the N/C burning machine. The flow controled by SPADES and by IHICS (these are computer-aided hull design systems which have been described in another report "Computer-aided Hull Design System"), is almost the same except for the alternate method of IHI.

In this flow, a few differences between Levingston and IHI can be found. They are: the control method of N/C (direct N.C at Levingston and off-line control at IHI) and the marking method (center-punched-marking at Levingston and line-marking at IHI). These are described in the next chapter.

Drawings for checking purposes at Levingston shallbe improved if a large drafting machine is provided. In this respect, more detailed study has been completed in another report, "Computer-aided Hull Design System".

#### COMPARISON FLOW OF N/C WORKS (AT PRESENT)



# 2.3 The Function Of the N/C Burning Machine

A comparison of the function of the N/C burning machines between Levingston and IHI can be seen in Table 2.

Levingston has one C-R-O N/C 3000 System. The five (5) shippards of IHI have five kinds of N/C burning machines to suit their fabrication methods and a kind of N/C marking machine. IHI has eleven (11) N/C burning machines and three (3) N/C marking machines in total. In IHI-AI0I Shippard, three (3) N/C burning machines have been provided.

Only a few differences can be seen in the function of the N/C burning machine between Levingston and IHI. A brief explanation on this subject is described as follows.

# 1) Control Method of N/C

Levingston's control method of N/C is DNC (Direct Numerical Control), DNC is the system which the host computer controls directly the N/C machine, so that no gunched paper tape is necessary. And DNC system shall be more suitable for Levingston, because the host computer is at Ashland Co,, in Kentucky, very far away from Orange, Texas. Therefore it seems to be impractical to get N/C tapes from Kentucky.

IHI's control method is all off-line control. IHI also planned to adopt DNC system, however, it did not occur due to the cost of both initiation and running time. Since IHI has five (5) computer centers near the shipyards, paper tapes or magnetic tapes for N/C are easily provided.

# 2.3 The Function of the N/C Burning Machine (continued)

# 2) Bevel Cutting

At Levingston, bevel cutting is performed by the following sequence:

- Programmer defines a surface with bevel cut by the NESTING Program.
- N/C machine stops at both ends of the surface with bevel cut.
- N/C machine operator sets and reset the bevel degree and location.
- Machine restarts.

IHI system has several standardized bevel angles such as  $15^{\circ}$ ,  $25^{\circ}$ , and  $35^{\circ}$ . The selection of bevel angle and Ignition/Extinguish are automatically performed by the system without stopping the machine.

If the bevel angle can be standardized at Levingston (even not, the bevel angles other than standard are supposed to be quite a few), the automatic bevel cutting shall be performed by additional feature to the postprocessing program for the machine.

TABLE 2 COMPARISON TABLE OF N/C BURNING MACHINE BETWEEN LSCO & THI

-										
	7	_	COMPANY	LEYINGSTON			Ił	II		
ITEM FUNCTION AT		A11 ROUND	ALL ROUND	SIMPLE	MARKING	ALL ROUND	ALL ROUND	MULTI.TORCH		
INSTALLATION DATE		IOH DATE	April, '75	Oct., '71	Sept., '72	April, '75	1974	Aug., '71	Sept., '74	
MAKER Burning Machin		ming Machine	CRO N/C 3000	Koike,Japan	Tanaka,Japan	Tanaka,Japan	anaka Japan	Tanaka, Japar	Koike,Japan	
		Controller		Kongsberg CNC 500	Toshiba T-1500C	Toshiba T-1300G	IHI MK-240N	T-1500	T-1500	Ioshiba . T-40
			ffective	29'-7"	31'-2"	18'-2"	16'-3"	39'-0"	58'-6"	16'-9"
			Rail Span FFECTIVE	111'-0"	56'-3"	74'-8"	` 74'-8"	97'-6"	73'-2"	58'-6"
		_	leight		10 MT	2 NT	0.7 MT	8 NT	13 MT	1.5 MT *3
				0 • 120 pc7			<del> </del>	0.01	13 /11	1.5 81 -3
.		<b>"</b>	is Pressure Used	0 <sub>2</sub> : 120 PSI Hatural Gas: 15 PSI	0 <sub>2</sub> : 9kg/cm <sup>2</sup> LPG: 65kg/cm	•				
		┢	Cutting	2-50 IPM/15-250IPM	0.4-60 IPM	2-51 IPM		0.4-118 IPM		
		e	Marking	20.83'/Min.	39'/Min.	19.5'/Min.	58.5'/Min.	39'/Min.	-	
	뿢	SPEED	Rapid	20. 021 011	201 111	10 51 014	50 51 511			
	BURNING MACHINE	H	rking	20.83'/Min,	39'/Min. + 1/64"	19.5'/Min.	58.5'/Min. + 1.5/64"	39'/Min. + 1/64"		
	2		recision Is Used	Natural Gas	LPG		1.5/04	1704		
	2	-	Torch	6(2 Master, 4 Slave)	4	3	-	2		
	3		Station Bevel	I.K.Y.X.Y	I.K.V.X.Y	1		2	3 1.V.Y	3
					<del>                                     </del>		Tanada			
		×	Nozzle Tip	Oxweld Hade	IHI Made	Koike 106PD	Tanaka	Fanaka 3155A	Tanaka- Curtain	Tanaka- Curtain
CAPACITY		BLOCK	Height Sensing Rotating	Fluidie	Fluidic	Fluidic		Fluidic	Roller	Roller
APA		=	Torch	Equipped	-Equipped				Equipped	Equipped
٥		5	ignition Extinguish	Automatic	Automatic	Manual	Automatic	Automatic	Automatic	Hanual -
			Piercing	Automatic	Automatic					
			Marking	Center-punched	Plastic Burned	Zinc Burned				Plastic Surped
		Driving Motor		D.C. Servo	D.C. Servo					
			trol Axis	X.Y.0	X, Y1-2, 81-2	X.Y	X.Y	X.Y	X.Y.8	1.Y.0
	•		imum Dimen- n Inpur	1/64"	0.1/64"					-
		Maximum Dimen-		120'	1230'				123'	1230'
			e Format	ESSI	EIA-8U					
			erpolation	Linear and Circular	Linear & Cir.					
	ER	Ker	f Compen-	Dial Set	Dial Set			Dial Set		
	ONTROL	_	ersing	Direct on Path Program Rev.	Dir.on Path					
	8	Aux	i. Function	110gram Ners	Prom. Rev.	28	32	33	51	27
		Oth	ers		<del> </del>					
		┝								
		-			-					
	I		3/8" T 5/8"	<del></del>	51 IPH	24 IPH		30 IPH	28 IPM	24 IPM
		4	5/8" T 3/4"	Variable	47 IPH	20 IPM		26 IPH	26 IPH	22 IPH
g			3/4" T 7/8"	TOPIEDIE	31 IPM	20 IPH	/_	20 IPH	26 IPH	20 IPH
PAST RECORD		3	7/8° T		31 IPH	15 IPM	/	20 IPH	20 IPM	18 IPH
V			sigeration		Lutting Seq.	10 114		יייי איז טי		
ă	•		Torsion	Seq. Bridging	Wtr. Cooling					Cutting Seq. Bridging
		Troubles		j				1		1
		Up-	to-date							
AUHINISTRATION	. 1	À	Daily		Operator					
		MAM	Periodical		2 Times/Yr. Maker					
		Š	Repairing		Maker IHI Special:	<u></u>				
¥ 2 × 2		3	Number		2	2	2	4	4	1
		8	Experience		2-3 Years	3.5 Yrs.	1.5 Yrs.	1.5 Yrs.	3 Yrs.	1.5 Yrs.
	Max. Plate No. to be 4 PL(12'-0*50'-0")		PL(12'-0*50'-0")	2 PL(127*34)	1 PL(15'*75')	IPL(15'*75')	2PL(115*98')	PL(16'*46')	1PL(12**9')	
		DVAR		DNC :	Tokyo - 1 Chita - 1	Chita - 1	Chita - 1 Yokonama-2	Chita - 2 AIQI, - 2	Kure - 2	Kure - 2
	K		~	SAC .	A101 - 1		· Andresian-F			.

# 3. Some Recommendations on the Usage of the N/C Burning Machine from IHI

In this paragraph, some recommendations on the subject are presented. In addition, IHI's line marking method and its adaptability to the Levingston present system since it seems to be preferable from the viewpoint of workability by sub-assembly and assembly.

# 3.1 <u>Standardization of Bevel Angle for the Automatical</u> Bevel Cutting

Though this subject is concerned with not only the N/C system, standardization of bevel angle has the possibility for automatical cutting of bevel surface, Since automatical setting of optional bevel degree is very difficult, it is true that the angle has to be set by hand, However, if this standardization has been completed, pretty part of bevel surface can be cut auto-. matically by adding a feature for that to the postprocessing program which is connected to the output from the SPADES system.

Even if the said additional feature is not possible, necessary time for setting bevel angle shall be much saved by standardization of bevel angle,

# 3.2 <u>Re-Arrangement of Supporting Area 'of Objective Pieces</u> for the N/C Machine and a Consideration for "Machine-down"

Since the work load of N/C machine at Levingston is expected to be increased as described in Chapter 1 of this paper, re-arrangement of the subject shall be urgent.

3.2 Re-Arrangement of Supporting Area of Objective Pieces for the N/C Machine and a Consideration for "Machine-down" (continued)

From the viewpoint of the workmanship at Levingston it is recommendable to cut hull pieces by N/C machine as far as possible. However, in case of over work load at N/C machine, it is recommendable that a flat plate such as main deck plate is shifted from the N/C machine to the flame planer. In addition, the scaled mold lofting (1/10 or 1/12) is recommendable to be established with the aid of a large drafting machine; because making templates in full scale from the scaled drawing (1/10 or 1/12) is possible when "machine-down" of N/C continues for awhile. In this concern, more detailed description has been presented in another report "Computer-aided Hull Design System.

# 4.0 Line Marking Method of IHI System (For Reference)

Levingston's N/C system adopts center-punched marking system. This system can work (punch) points with several variations of intervals between them. From the viewpoint of workability at sub-assembly and assembly, this marking system shall not be so easy for production people to fit pieces on the marked (dotted) line. In fact, production people are seen marking a continuous straight line by stretching a thread. In this concern, it seems to be interesting to introduce IHI's line marking method for reference.

There exists two winds of line marking methods in IHI's system. One is to burn zinc powder by a working torch to a steel plate. Another-is to burn plastic powder. The clear continuous lines' (1/16" in boldness) can be marked with a speed of about 40' per minute. Comparing with the center-punched marking, it-might have a demerit to be faded away during the fabrication process. However, no particular consideration is requested at IHI. More detailed information on this marking system-is presented in the following table.

# LINE MARKING DEVICE IN IHI SYSTEM

	ZINC POWDER	PLASTIC POWDER
Price of One Device (Price in Japan)	400,000 Yen (\$2,000)	800,000 Yen (\$4,000)
Running Cost	Powder  3 Yen per 3'-3" of Marking  Nozzle Tip (per one) 25,000 Yen (\$125)	Powder  3 Yen per 3'-3" of Marking  Nozzle Tip (per one) 25,000 Yen (\$125)
Maintainability	Easy	· Very Complex

# Adaptability to Levingston's System

Some additional work shall be requested for adoption of the equipment to the present Levingston machine as follows:

- 1) Fuel gas, oxygen, Water and air must be supplied for the working torch.
- 2) Postprocessing program must have a signal of marking ON/OFF. This function is supposed to be satisfactory in the present program.

# EPILOG:

This report was made through the investigation and discussions with Levingston's personnel by IHI. IHI hopes that Levingston continues a study on the subject matter for the future betterment along with the recommendations described in this paper from IHI.

# APPENDIX J LSCO STUDY AND COMPARISON OF SPADES VS. IHI SYSTEM

# LEVINGSTON SHIPBUILDING COMPANY

# Technology Transfer Program

TASK 2 Engineering and Design

Sub Task 2.1 Computer-Aided Design Systems

STUDY AND COMPARISON OF SPADES VS. IHI SYSTEM

E. E. MAYER

4/25/79

#### INTRODUCTION

In an overall comparison of the SPADES system and the IHI system for computer-aided design, the SPADES system comes in second only due to the quanity of programs that compose the entire IHI N/C system.

Considering production aids, the SPADES system is equal to the IHI system, and because of ease in user coding, the SPADES N/C production oriented system is considered the better of the two.

# **Definitions**:

S.P.A.D.E.S. - Ship Production and Design Engineering Systems

I.H.I.C.S. - Integrated Hull Information Control Systems

# Appendices:

Appendix 1 - Spade System Ship Production & Control Module Cali & Associates Inc.

Appendix 2 - Spades System - Engineering Detailing Module Cali & Associates Inc.

# COMPARISON ANALYSIS

# Comparison Table

	APPLICATION	SPADES	IHICS
1)	Hull definition and data base creation	*Lines fairing  *Hulload	*Lines fairing  *Shell landing
		*Demo	*3-D process  *Section design sub-system
2)	Production L o f t i n g	*Parts generation  *Parts separation  *Plate development  *Nesting	Production Engineering subsystem consisting of:  *"Line" system for nesting w/manual and interactive graphics by "cards".  *Part generation
3)	Ships production and control programs	*SPAC (Ship Production and Control )	*Part data base administrative program  *Piece data assortment program
4)	Production and Lofting assistance programs	Mfg. aids module for:  *Frame bend offsets  *2 dimensional pin heights  *Girth length table  *Rol 1 set templates  *Mel d loft offsets	"LODACS" system for:  *3 dimensional frame bending (offset info only for twist-  *Mold loft offsets (from data creation sub-system)  *Part data base administrative program and shell program produce templates for shell plate bending and pin jig heights

•	APPLICATION	SPADES	IHICS
5)	Design engineering and hull calculation programs	Hull calls for:  *Bonjean curves  *Hydrostatics  *Longitudnal strength  *Tank capacity and sounding and ullage  *Trim calculations  *Damage stability  *End launching	*Bonjean Curves *Hydrostaties *Longitudinal strength *Tank capacities w/sounding and ullage *Trim calculations *Damage stability *Launching calculations *Grain heeling moment *Sea trial data

Programs that IHICS has that SPADES does not have are "z VIBRA" for vibrational analysis and "Z PLATE" for structural analysis. These, or similar, programs are available to LSCo from their sources.

# **Explanation Of Comparison Table**

(1) The Hull Definition modules compare equally since the end product or output comparisons bear close resemblance. Both FAIRING modules have excellent splines and produce very fair lines. However, the SPADES system utilizes more control line output for assistance in development of other modules. Also, skewed planes may be generated for cant framing and additional frames may be loaded for shell master butts.

The HULLOAD program is the SPADES module that defines shell landings of stiffeners, seams, decks, longitudinal bulkheads. HULLOAD also defines stiffeners and seams on decks. flats, and longitudinal bulkheads.

IHI generates this information with its I.H.I.C.S. shell and panel landing modules, which along with-the line module, comprise the data base. Without actual coding examples of the IHI system, a true detailed comparison is not possible. However, based on other available examples, coding the SPADES system in less difficult than coding the IHI system.

DEMO (Detail Engineering Module) and IHI's I.H.I.C.S. production engineering system (Integrated Hull Information Control System) both provide for the loading of structural details into the data base. The SPADES DEMO module (see enclosure) is a powerful tool which enhances the HULLOAD program and allows for structure definition loading in all planes including transverse and skewed. DEMO also places the emphasis of part definition within the engineering group and consequently allows part generation personnel to concentrate on parts separation.

The DEMO module can also provide drawings that are suitable for issue with hand finishing done by a drafter. The IHI system is capable of performing an equal task, but an accurate definition of how, when, and which module performs this function is not explicit in the description material IHI has provided.

(2) The PRODUCTION and LOFTING N/C modules, when compared based on available I.H.I.C.S. input information, shows the SPADES production module to have an easier input format and to be almost equal in output information. (See appendices No. 1 and No. 2)

Parts generation, plate development and other similar modules compare as to the assistance offered to production. However, the IHI group -utilizes more output information than the LSCo group needs at present.

The nesting modules offer no real comparison, as the SPADES system is totally interfaced with the data base and the IHI system uses a manual input of parts information to generate marking and burning tapes. Both systems have an interactive graphics capability in the nesting cycle.

The ships production and control modules are difficult to compare, as the IHI N/C system performs such tasks under different modules. However, the overall output far exceeds what Levingston or SPADES can generate. IHI adheres to the policy that Production Control and Planning system utilizes every means necessary to achieve and maintain total control. From the data submitted, it is concluded that data input is mainly performed manually using output information found in the data base.

IHI produces a cutting list, parts list, and piece drawing for every item on a ship whether processed by numerical control or manually .

The SPADES SPAC module (Ships Production and Control) has not been used nearly as extensively as the corresponding IHI modules. SPAC is a new module in use at this time only at National Steel and Shipbuilding Company and by Cali and Associates. The SPADES SPAC system utilizes the full data base for reporting and its limitations are set only by user option. This module provides a large growth potential.

In the area of Production and Lofting assistance, the IHI and SPADES systems are almost completely equal. Each offers the roll set templates, frame bending, pin jig heights, girth length tables, and mold loft offset talbes necessary in N/C shipbuilding.

The SPADES system module manufacturing aids interfaces the data base from which it pulls the output information requested, IHI use different sub-systems to gather required information and not all of them actually access the data base for data generation. In fact, the three-dimensional frame bending module, called LODACS, uses offset information from Fairland and other modules for input into the LODACS module which is a stand-alone module. The output information generated by LODACS for frame bending and end cuts is of exceptional quality.

5) The Design Engineering and Hull calculation programs of the IHI system is superior to the SPADES system in breadth and completeness, especially as applied by IHI. SPADES Hullcal does have the advantage of direct access to the data base, which simplifies coding. The IHI SPECS (Ships Preliminary and Exact Calculations System) requires manual preparation of input as it is a standalone module.

The SPADES output format is now out-dated and the Levingston Design Department has chosen to use SHGP (Ship Hull Characteristics Program). However, with a soon to be implemented Fortran subroutine, the Hullcal output should be more acceptable to regulatory bodies requirements and needs. IHI SPECS is a complete hull calculation system even including calculation of ship sea trial criteria.

6) IHI's structural analysis and vibrational analysis programs Z Plate and Z Vibra, are stand-alone modules similar to several programs used by Levingston but produced by other companies. The programs are not comparable to a SPADES module as SPADES is not intended to be involved with Desing Engineering analysis.

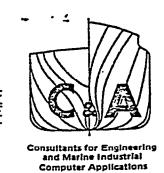
## CONCLUSIONS

In summary, the IHI N/C system and the SPADES N/C system are both very good. The IHI system is fully utilized to the best advantage of the IHI shipbuilding complex while the SPADES N/C system utilization at Levingston is limited by a lack of implementation and facilities.

The comparison of those N/C modules that actually duplicate the old style lofting and template making methods would actually give SPADES the advantage as the input coding of SPADES is much easier. Also, SPADES fully utilizes the data base set up with the Fairing and Hulload modules.

Implementation of the IHI system at Levingston is not necessary because the SPADES system currently utilized is more than adequate for present and future needs. All SPADES modules access the data base for ease in generating input data.

Overall, both systems serve the intended functions. IHI utilizes Computer-aided Design and N/C to their maximum capabilities which is one reason they are one of the world's foremost shipbuilders.



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#### APPENDIX 1

# SPADES SYSTEM

# SHIP PRODUCTION & CONTROL MODULE (Preliminary Description)

All information contained in this document is considered proprietary of Gali & Associates, Inc.





# INTRODUCTION

The use of extensive modular construction in shipbuilding, combined with the increased use of Numerical Control, has greatly improved in the last decade the efficiency of the industry.

In order to properly utilize these techniques, it was immediately apparent, however, that more and better planning was necessary.

The planning effort, per se, is neither too difficult nor too costly. The collection and updating of the data needed to generate the required reports is both difficult and costly, in order to obtain a reasonable degree of accuracy,

The The 'ship Production and Control (SPAC) Module' of the 'SPADES' System is designed to achieve in this area the following goals:

- 1. Reduce man-hours for data collection.
- 2. Improve the accuracy and timeliness of the reports.
- 3. Reduce ship construction costs by reducing errors and misinformation in the shops,

The 'SPACE' Module covers at the present only the hull construction. It is intended-t, in parallel with the development of modules to handle the design and production of other ships' systems, the 'SPAC' Module will be expanded accordingly.

# DESIGN CRITERIA OF THE 'SPAC' MODULE

Since the 'SPAC' Module properly falls in the category of management in formation systems, the basic criteria applicable to this type of system must be respected as follows:

1. The module must allow the collection of independent data at the origination source and make it immediately available to all interested shipyard functions.

As an example, for instance, assembly boundaries and schedule starts can be inputted directly to the system and the 'master erection schedule' report generated immediately after for dissemination.

2. All applicable data generated by other modules of 'SPADES' must be collected and used by the 'SPAC' Module without any user intervention.

This feature is the main justification for the development of the module, and the following is a partial list of examples:

.Allocation to the proper assembly and sub-assembly of all pieces generated through use of 'PARTGEN', 'PARTSEP' 'PLATDV' or 'MANF AID (frame bending).

\*Processing time for N/C burning tapes and flame planer sketches.

.Unit weight of individual pieces and weight and centers of gravity of assemblies and sub-assemblies.

\*Length and nesting within standard lengths of shapes of the various individual shape pieces.

- Cross reference between assemblies due to the nesting into a plate of pieces belonging to different assemblies.
- Bulk material allocation for pieces produced through shearing or 'one-to-one' optical burning.

3. Revsion control is maintained by the system for all the issued reports generated.

A summary report can also be generated, showing at any one point in time the current valid revisions of all the issued reports.

4. Any change of the independent data or other data used by the system must generate an exception report indicating which of the reports are affected by the change, so that the user can initiate the proper request.

For example, if planuing changes require different boundaries for any one assembly, the module must automatically update the allocation of all pieces effected by the change of boundaries, and give a report indicating which reports must be requested for re-issue.

- 5. Exception reports can be generated to indcate to the uaser at any point in time which pieces for any one particular drawinghavenot as yet been defined, or any material deficiencies.
- 6. The system must allow the introduction of data at levels other than the optimum, to override or enrich the data base, in order to be able to generate complete reports at anytime.

The Appendix contains the basic data flow for the module, a brief description of the input needed, and some examples of the reports generated by the system. The examples of the reports are simulated in this preliminary description, and they will be changed as the development of the module proceeds.

# **CONCLUSION**

The purpose of this preliminary description is to disseminate among all potential users the features and the capabilities of the module **as** they are conceived by Cali &Associates,. **Inc.** 

It is hoped that a review by the shipyards will provide us with the very much needed input and comments for incorporation during the development. Specific comments regarding the size and format of the data used by each shipyard as it is applicable to this module will be very helpful in avoiding future incompatibilities and/or restrictions.

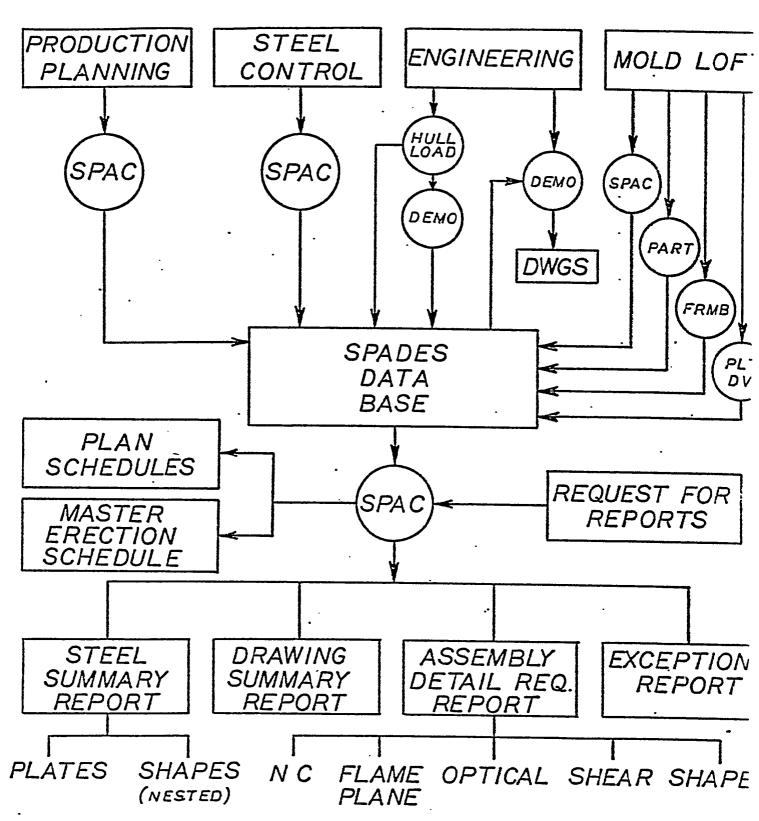
# SPADES SYSTEM

SHIP PRODUCTION & CONTROL MODULE

# **APPENDIX**

# SPADES SYSTEM

# DATA FLOW FOR SHIP PRODUCTION AND CONTROL MODL



# TYPES OF INPUT & RESPONSIBILITY FOR THEIR PREPARATION BY SHIP YARD FUNCTION

## 1. Production Planning

- a) Limiting boundaries of planned assemblies (units) and sub-assembly breakdown, if any. The system will always assume that ship's surface, such as deck, webs or shell will constitute a sub-assembly,
- b) Planned start date for processing each assembly.

## 2. Steel Control

- a) Final steel bill. This is intended to mean the steel take-off bill as modified for utilization of stock and/or standardization of plate size, The various items in the various steel bills will carry a unique stock number compatible with the shipyard system.
- b) Storage location of various items in the steel bill will be given to the system upon receipt of the steel.

#### 3. Engineering

- a) Loading of the data base. Through the detail engineering module, the data baseloading capabilities will be expanded, allowing at the same time the easy generation of detail drawings, As part of this activity, engineering will also update, as needed, database libraries of standards (brackets, chocks, etc.), shapes, characteristics, and associated cut-outs.
- b) Drawing list and associated range of pc. mks. used in each drawing. This will allow the system to generate exception reports calling attention to pieces not generated at any one point in time;

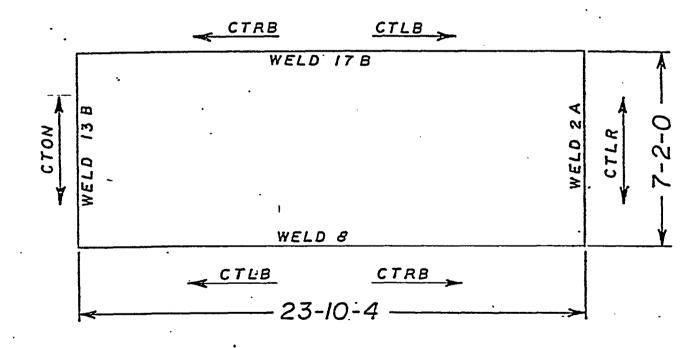
#### 4. Mold Loft

a) Through the use of 'PARTGEN', 'PARTSEP' and 'PLATDV', the loft will enable the system to allocate the pieces thusly generated to the various assemblies and sub-assemblies. Provision will be made for identifying drawings, pc. mk. and beveling detail, and also applicability of a part to another area of the ship.

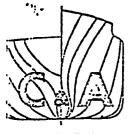
- b) Through the use of 'MANF AID' (frame bending), all shapes, whether straight or curved, will be identified and allocated to the proper assembly, The Frame Bending Program will be modified to easily do that for all flat surfaces.
- c) Through the use of the Ship Production and Control (SPAC) Module, the loft will input to the System all the miscellaneous pieces not otherwise identified.

# <u>SHIPYARD NAME</u> EXACTOGRAPH SKETCH

SHIP —	HULL NO	
JOB ORDER NO. —		
ASSEMBLY —		
SUB ASSEMBLY —		
DWG. NO. ———		
PLATE SLZE		— QTY REQD.
PLATE STOCK NO		



TAPE NO. 301376 -307-1 EXACTOGRAPH



insultants for Engineering and Marina Industrial Computer Applications

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Phone (504) 835-2641

. APPENDIX 2

# SPA-DES SYSTEM

# ENGINEERING DETAILING MODULE

(Preliminary Description) (DEMO)

All information contained in this document is considered proprietary of Cali & Associates, Inc.





## **OBJECTIVES**

The main purpose of this module is utilize the time and effort spent during. the detail design phase for numerical description of the ship structure. During this phase, all. structural details are defined, and if these definitions can be recorded on the database, interpretation of the drawing and the possibility of errors downstreams during part generation can be greatly reduced, Greatly expanded database loading loading capabilities will provide information over and above the geometrical part generation requirements which can be used by the planning and control module or other ship's systems.

As the volume of information. the database increases and the base becomes more comprehensive verification of loaded data becomes more and more difficult. 'The quickest way of verificationi by drawing. Therefore, a simple and easyway of accessing the data base with a few commasds is needed to automatically output all loaded data of a particular surface in to a composite drawing,

If this drawing capability is achieved; only a few options are needed to extract partial drawings for all kinds of purposes. Structural drawings can be complete, with the exception of lettering and dimensioning. Background drawings for arrangements and composites can be produced with just a few commands.

# **PREREQUISITES**

In order to make this module an officient tool for detailing, the loading capabilities of the data base will be expanded. The 'H'JLLOAD' Module will be capable of loading traces and details in transverse, plan and elevation views. Additional information on all surfaces will include:

- a) Stiffeners and their end connections
- b) Seams and plate thickness associated
- c) Brackets and chocks
- d) All access holes, including face bars
- e) Inside contours, as defined by web frames.

All through members affecting other surfaces must be handled by 'HUILOAD' Local details will be defined by 'DEMO'.

### OPERATING PROCEDURE

Although the module's primary task is to aid in loading the data base, direct loading capability is not conceived. The actual loading of the data base is reserved for the group of people responsible for loading the data base through 'HULLOAD'. This is to preserve the integrity of the data base by concentrating the rersponsibility onto one person, or one group of people. However, to avoid having the 'HULLOAD' people recode all the definitions Module 'HULLOAD' will have the capability of executing the same input deck, ignoring irrelevent commonds. but executing and loading the detail specifications,

The application of the module within the ship design effort is seen as follows:

- 1. Fairing and loading of the major structure through 'HULLOAD'
- 2. Extract a drawing of the surface containing outlines and through members through 'DEMO'.
- 3. Load repetitive patterns of stiffeners and seams through 'HULLOAD',
- 4. Extract a new drawing through 'DEMO' containing all loaded details.
- 5. With 'DEMO', add and modify details of stiffeners, seams, holes and-brackets, resulting in:
  - new drawing, complete with the exception of lettering and dimensioning
  - An input deck defining the details executable by 'HULLOAD'
  - An entry in a data base record which contains all input decks -that are generated by 'DEMO' and must be executed by 'HULLOAD'
- 6. When the design is completed, control is transferred to 'HULLOAD' The input deck is executed, loading the details. The entry of the final step above is deleted.
- 7. Revisions:
  - a) If the drawing is not released as yet, the revision may be added to the 'DEMO' input deck executing '5' and '6'
  - b) if the drawing is released and lettering and dimensioning has been added, revisions are effected through 'HULLOAD' only,
- 8. After the structural details have been loaded, drawings for other disciplines such as arrangements and composites may be called.

# INFORMATION DEFINED BY 'DEMO'

Only local details are defined through 'DEMO'. Details are defined as follows:

1, Stiffeners: Symbolic name s ABC P/S

Contour definition Shape code number

Orientation (near side or far side)

End connections (lap, snipes, knuckles).

2. Seams: Symbolic name J ABC P/S

Contour definition

Welding detail (bevel and gap)

Thickness on both sides.

3. Holes: Symbolic name H 123 P/S

Contour definition

Thickness, width and offset of face bar..

4. Brackets: Symbolic name B 123 P/S

Contour definition or standard detail identification

**Thickness** 

Width and thickness of flange.

Width and thickness of face bar.

5. Inner Lines: Accessible only as a contour

Identified by 'INNL'

Contour definition

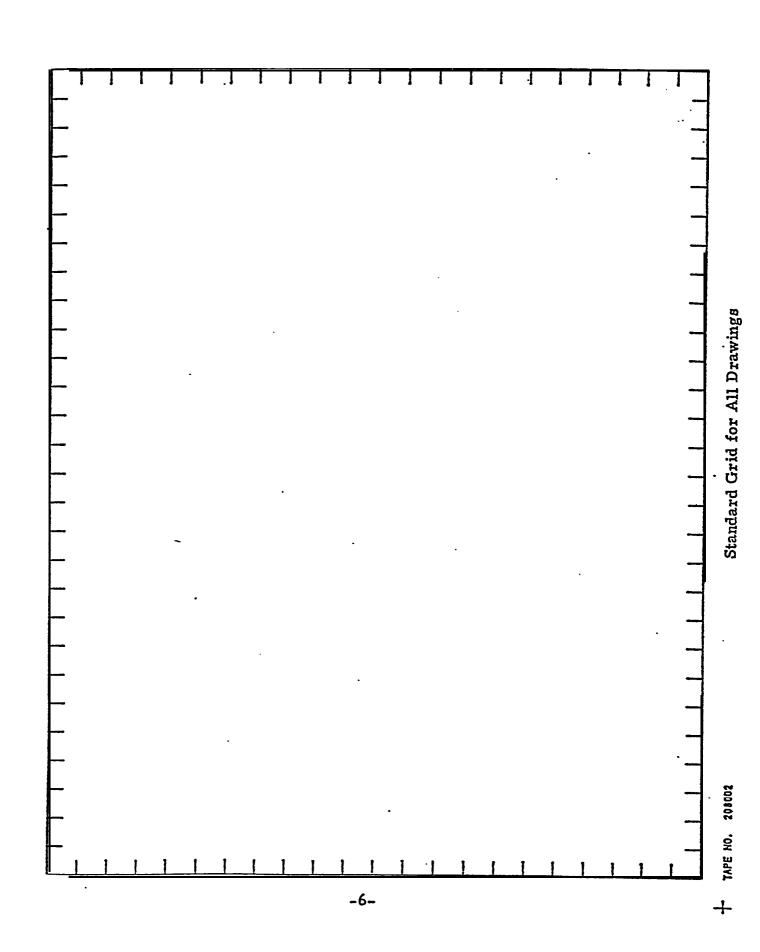
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### PROGRAM CAPABILITIES

Options with automatic drawing of data base contents:

- 4) Scales
- b) Windowing
- $_{c}$  ) with or without shapes ('T' 'L' etc.)
- d) With or without cut-outs and snipes
- e) With or without stiffeners and seams on the surface
- f) Include background frame or deck
- g) Pen selection for turret machines
- h) Line selections of different types of dashed lines.
- 2. Automatically included as drawing standard:
  - a) A standard grid surrounding the entire drawing
  - b) Center line and/or base line, if part of the drawing.
- 3. <u>Programming capabilities</u> and language 'as close to 'PARTGEN' as passible, so that people programming 'PARTGEN' and 'DEMO' are interchangeable. All 'PARTGEN' tools such as Math, Contours, Symbolic Calls, Loops and Reps will be available.
- 4. Added Commands for detail definition:
  - a) STIF
  - b) SEAM
  - c) HOLD
  - d) BRKT
  - e) INNL
- 5. Looping capability:

programming of similar surfaces by modification to typicalcsurface such that only changes have to be redefined.



### **EXAMPLES**

- . Floors 46 through 53 are similar, with the exception of the location of the holes. The sample input deck draws two floors as typical configurations and loads all other floors by continuously modifying the location of the holes-
- 2. Bulkhead 31 is preloaded by 'HULLOAD' with most 'of the vertical stiffeners.
  - Stiffeners T9, TIO and Tll are modified by 'DEMO'. Access door and' seams are added.
- 3. Web frames 49 through 52 involved quite extensive calculations and developments showing the method of programming details.

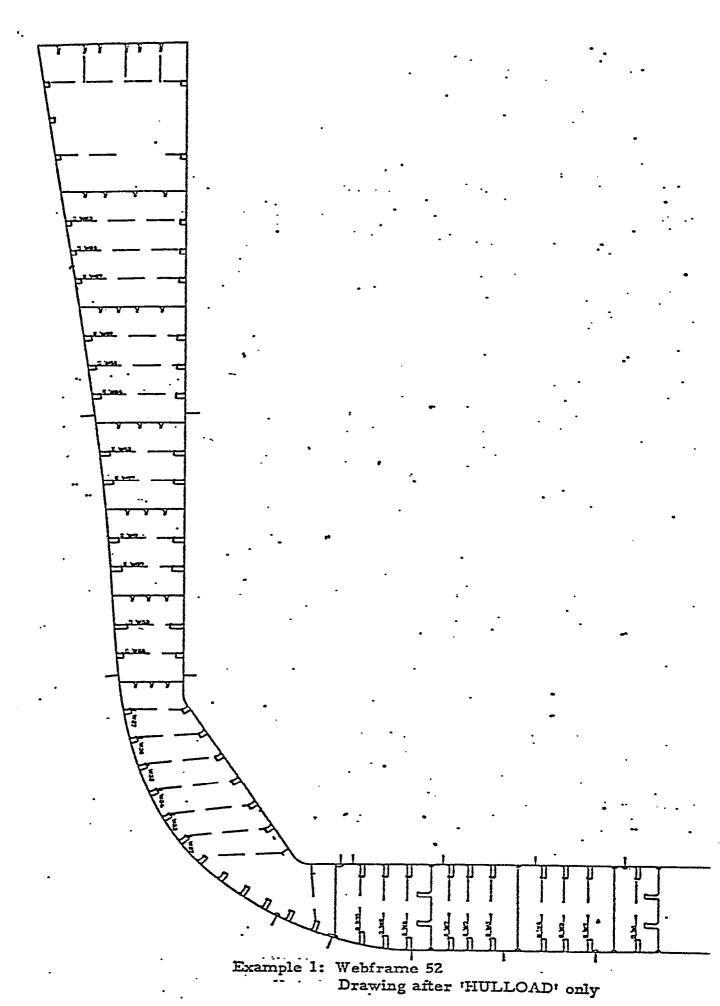
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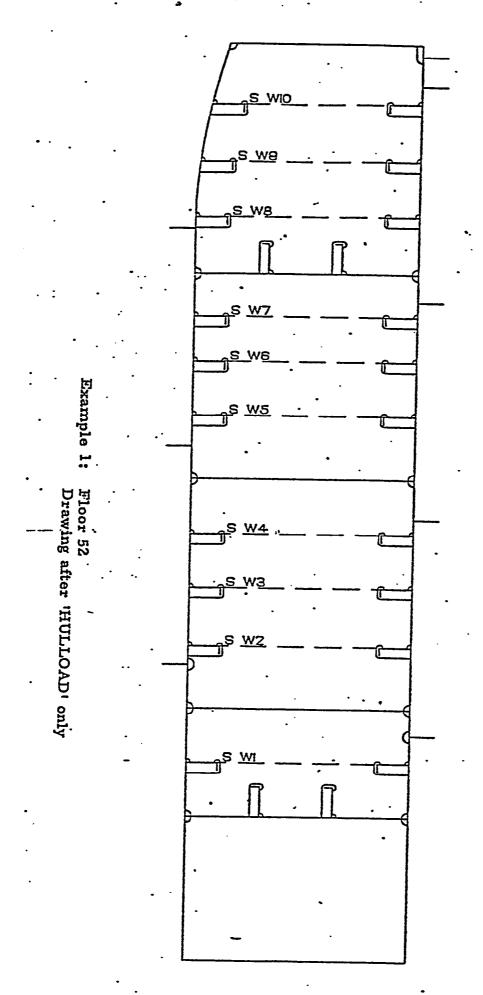
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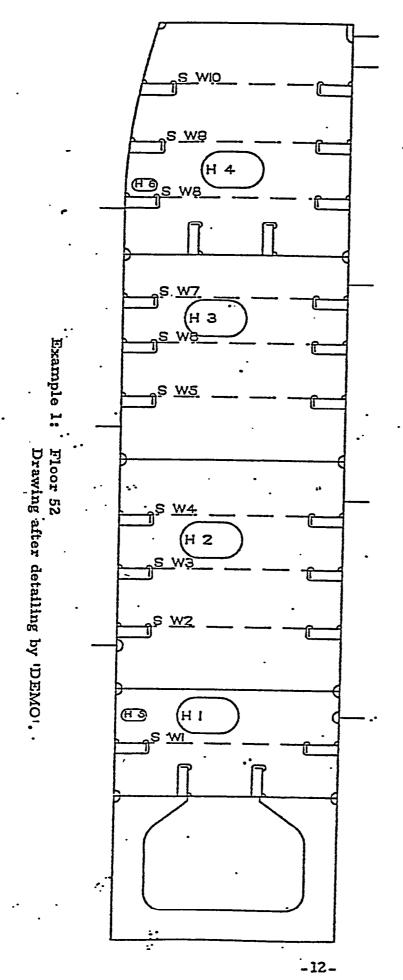
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P     15 P     15 B     H 6     3 12900       DAD     F 51 -     12900       PLT     \$ 10 S     \$ 9     12 12900       DAD     221 Y     12 H 4 2 12900       DAD     F 52 -     12900       LET     H 6 12900       DAD     F 53 -     12900       DAD     F 53 -     12900       NPE     12900		•			•	•	1290020
P     15 P     15 P     15 P     15 P     16 S     3 12900       OAD     F 51 S     12900       P     S 10 S     S 9 12 12900       OLD     221 Y     12 H 4 2 12900       OAD     F 52 S     12900       OLD     3 0 9 34 9 4 H 5 2 12900       OAD     F 53 S     12900       OAD     F 53 S     12900       OAD     F 53 S     12900	OP		S :	15	9		1290020
P 15 P 15 H 7 3 12900  F 51	DLD		9 Y	13	H	6 3	1290020
F 51.  PLT S 10 S 9 12 12900  PLD Z21 Y 12 H 4 2 12900  3 0 9 3211 8 H 5 2 12900  ET H 6 12900  PLD S 10 S 12 S 12 S 12 S 12 S 12 S 12 S 12		P	15 P	15	н		1290020
S 10 S 9 12 12900  S 10 S 9 12 12900  S 10 S 9 12 12900  S 10 S 9 3211 8 H 5 2 12900  S 10 S 9 3211 8 H 5 2 12900  S 10 S 9 3211 8 H 5 2 12900  S 10 S 9 3211 8 H 5 2 12900  S 10 S 10 S 10 S 10 S 10 S 10 S 10 S	DAD		F-:	51.			129002
DLD     221 Y 12     H 4 2 12900       3 0 9 3211 8     H 5 2 12900       DAD     F 52 12900       DLD     3 0 9 34 9 4     H 5 2 12900       DAD     F 53 12900       IPE     12900		and the contraction of the contr			· S · · · · · · · · · · · · · · · · · ·	. 9. 15.	
309.32118 H 5 2 12900 F 52. 12900 ET H 7 12900 DLD 309 3494 H 5 2 12900 DAD F 53. 12900				·	E R		
F 52. 12900  ET		• • :•				.T 6	
ET		g or more or equipment				E.	
DLD 309 3494 H 5 2 12900 DAD F 53. 12900 NPE 12900				)	·	· ,	
DLD 309 3494 H 5 2 12900 DAD F 53. 12900 NPE 12900	LT.					5	1290020
TAD F 53. 12900						7	1290020
12900 NPE 12900		•			H	5 2	1290020
NPE	DAD	•	, F 5	53.	•		1290020
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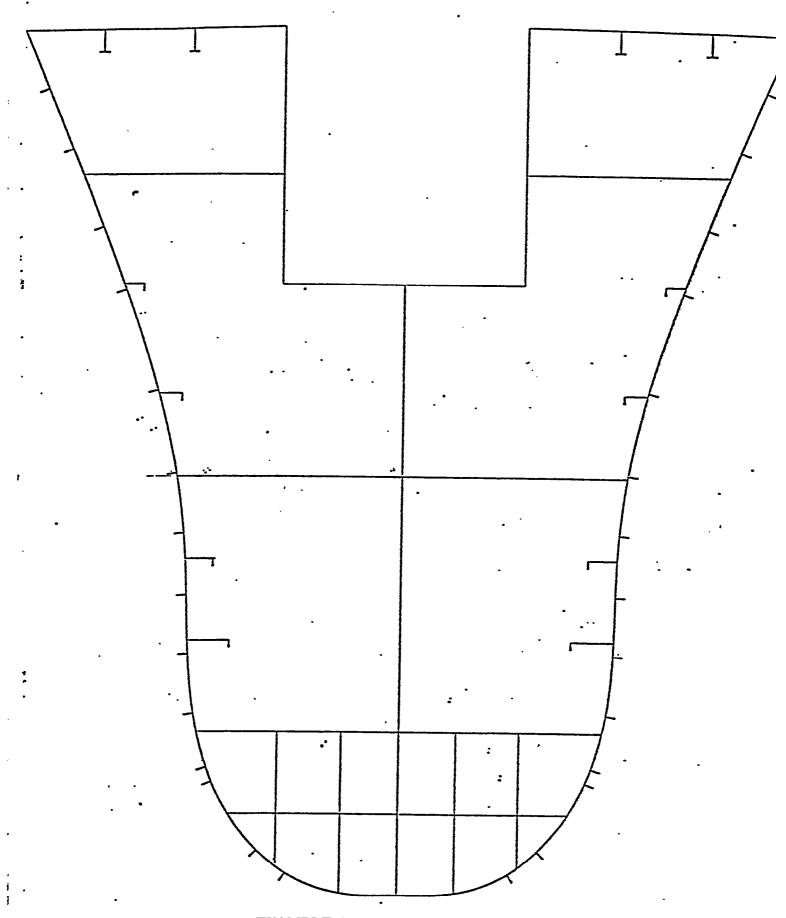




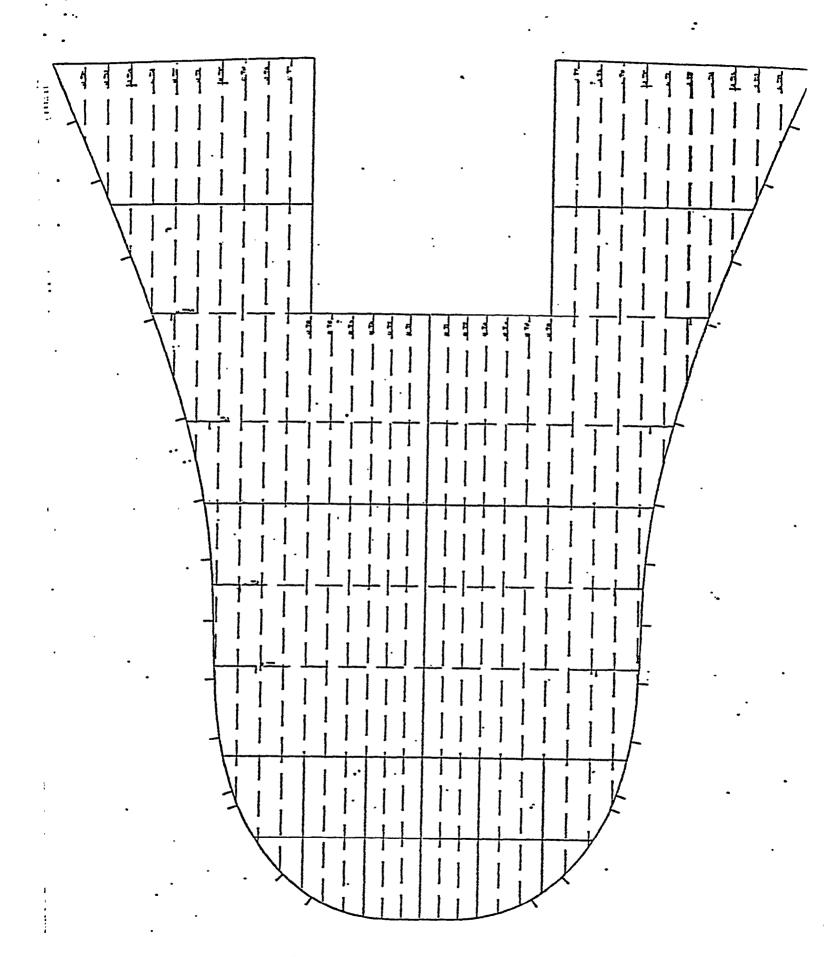


*****	EXAMPLE	2 : INP	UT DECK F	UR DETAIL	LING OF	F BULKH	EAD 31		
,									
				•					
*JOB L INPS	NG DE:10	N N							12901000 12901000
1460									15401000
•	AUTOMATIO	DRAWIN	G FROM TH	E DATA B	ASE				
TRSV		FWD	F 31.						12901000
OPTN	SCALNTE	E		48.				•	12901000
	START DE	AIL DEF	INIT.IONS						
CNTR .	CALC								12901000
SHEL		•	M						12901000
SLPE2	•	Y	2436 1823					10 1	12901000
CTRE	NOCT	•	1053	•• • • • • • • • • • • • • • • • • • • •	••••	•		11 1	12901000 12901000
CNTR	CALC	•							12901000
.DECK			MDK					•	12901000
SLPEZ		Y	1821		•			21 1	12901000
5		Y.	2026 2436	· · · ·				22 1 20 1	12901000
CTRE	NOCT	•	. 2436	•	• •			20 1	12901000 12901000
ADDP			D 46F	P ORG				13	12901000
			D 28F	P ORG				14	12901000
			D SWD		1		2436	1	12901000
		·	1				-1313 1313		12901000
	<u></u>	P	31		4	52021	1313	33 ·	12901000 12901000
		Р	32			52021		34	12901000
	•	P	1		1:	1 9	2 5	5	12901000
	ACCESS DOC	IR -	~~~~						•
CNTR									12901001
CORN		P	32						12901001
		P	34	•		13 8			12901001
		P P P	33 31			13 8	•		12901001
ÇTRE	HOLD	r	6		5		H D1	•	12901001 12901001
	MODIFY ST	IFFENER.	S						
PNCH3		P	20		P	2			12901001
STIF	502 LAF	CSNPE			-		ST22 · P		12901001
PNCH3		P	1		P	10			12901001.
STIF	503 SNF	PESNPE	<b>7</b> A .	-v -	<b>.</b> •		ST12 P		12901001
PNCH3 STIF	503 LAP	CSNPE	20-	٠	0 X	10-Y	10 ST12 S		12901001
CNTR		50·11 L					-115 3		12901001 12901001
MANU		P	22						12901001
LINE		x	13	2026					12901001

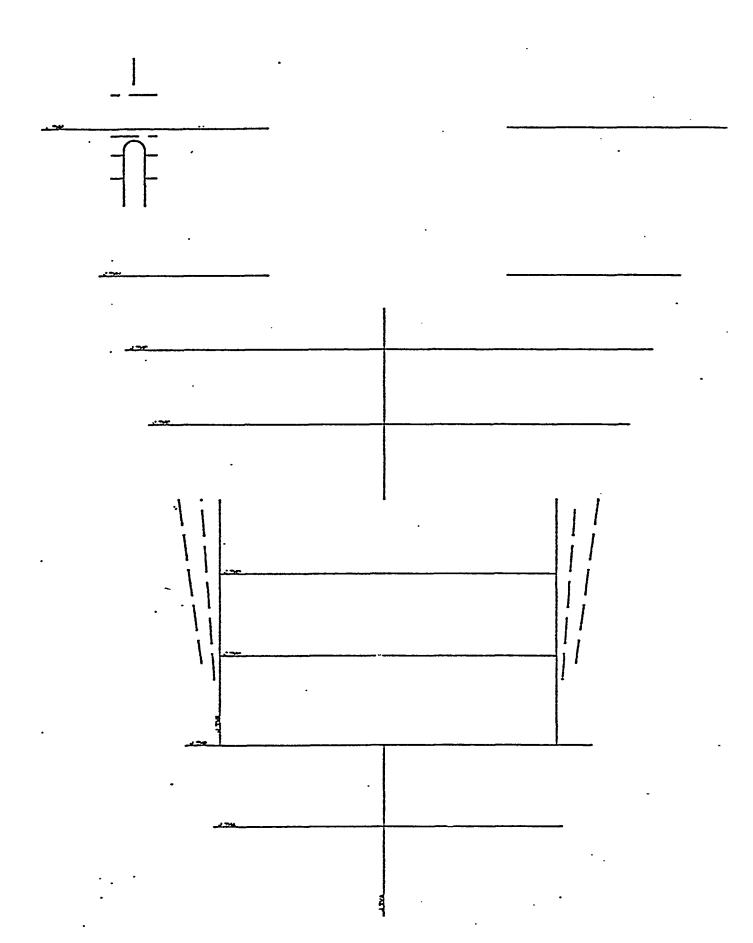
					14		182	<del></del>				<del></del>			1290100160
· : •			X		11	•	102	2							1290100164
:		077FL 1000MQ	-		11							ST10			1290100166
TRE		STIFLAPCSNPE	•								•	2110			
NTR			_												1290100.172
UNAL			P		21										1290100176
.INE			X		13		182								1290100180
; .			X		14		18	8							1290100184
TRE		STIFLAPCSNPE	=									STO			1290100188
		SEAMS													
·							- ~ -								
DDP				91									51		1290100200
			P		51	-		•					52		1290100204
•		•	P		52					911			53		1290100208
•			P		53					9			54		1290100212
•			P		54					8 2 8 3			55		1290100216
		<b></b>	Р		55			•		8 3			56		1290100220
		*	P		<sup>2</sup> 56	• . •••	•			83			57		1290100224
		•	P		57					8 3 8 3 8 3 8 3			58		1290100228
			P		58					8 3 8 3 8			59		1290100232
			P		59					8			60		1290100236
:NTR			•		•					•					1290100240
HEL		P+	M					• •				. • •			1290100244
OOP		• •	N		1	N	•	61	N	. 7	0				1290100248
TNO		•	"L		i	.,		-	••	•	•				1290100252
LPE	<b>-</b>	•	χX	•	-10									*	
NDL	=		N	•	1	•	•							_	1290100250
		NOCT .	. "		4			13							1290100264
TRE		NOCI	P		51	•		•	P	4	•				
EAM			์ J	L. A	31			11		.6	1				1290100268
C 1.1/			p	HA	= ~			11	P		5				1290100272
EAM			-	··in	52			11	r						1290100276
~			J P	HB	<b>=</b> 7			11	v		0	4744			1290100280
EAM		_			53			• •	X		3	1711			1290100284
			្ម្រ	HC				10			0	. ~			1290100288
EAM			Ρ.		54				X	6	4	1711			1290100292
			J	HĐ	<i></i>			10	_		9				1290100300
EAM			P		56	_			P	5	6				1290100304
				HE		D		17	_	_	8				1290100308
EAM		:	P		57				P	6	7 -		٠.		1290100312
_			J	нF			_	8	_		7				1290100316
EAM			X		58		12	8	P	. 6	8				1290100320
				HG				`7		•	6			•	1290100324
EAM			X		60		12		P	7	0				1290100328
			T					5			5				1290100332
DDP			D	67F		P	ORG			=		6	71		1290100336
EAM			P		71				X	, <b>5</b>	5	6			1290100340
		•	J	٧B	P			8		-	6				1290100344
EAM			X		52	•		6		0		6			1290100348
				VA	P			11		1	1		•		1290100352
EAM			χ	-	55	•	171		X	. 5		1711			1290100356
				VC		•	<b>_</b>	10			0				1290100360
NPE			•	_			•			-					1290100364



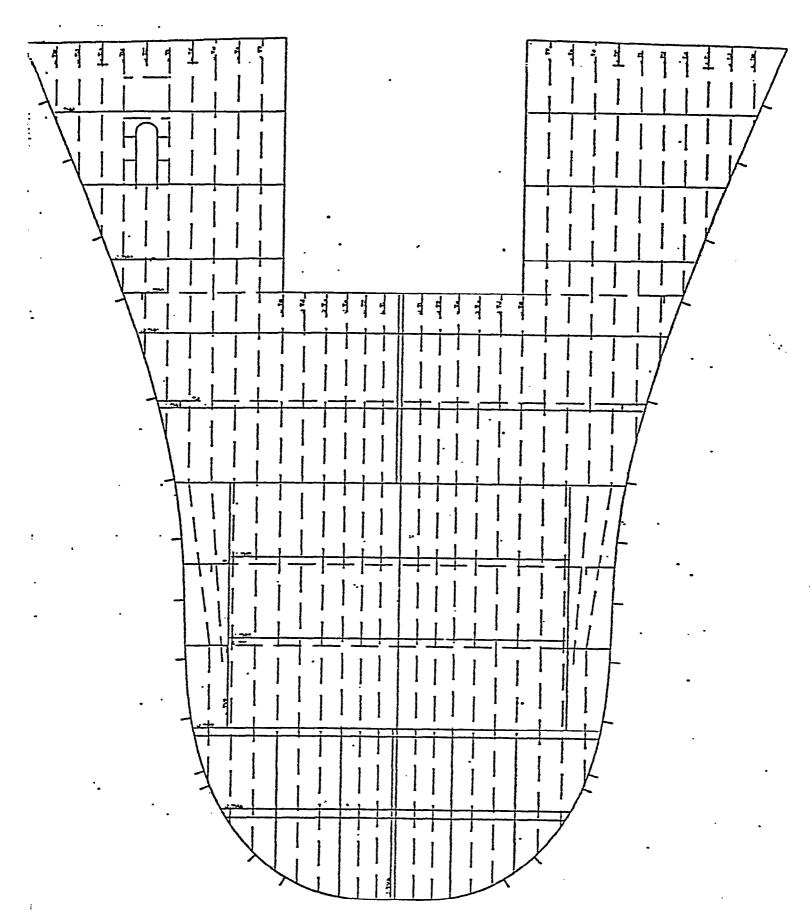
EXAMPLE 2: Bulkhead 31 Drawing of major structure after 'HULLOAD'



Example 2: Bulkhead 31 Drawing of loaded details after 'HTLLOAD'



Example 2: Bulkhead 31 Details programmed in 'DEMO'



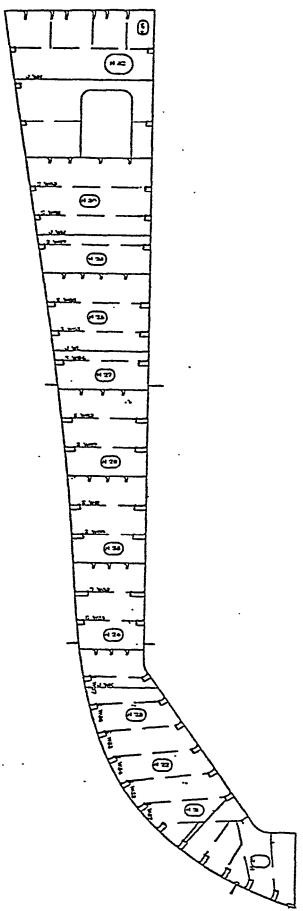
Example 2: Bulkhead 31 Final drawing after detailing by 'DEMO'

****	EXAMPLE	3: INPUT	DECK FOR	R DETAI	LING OF	WEBFRAMES 52	THROUGH	H 50
							~~~~	
*JOB L	NG DEMO	. N N	9003 9003	w es 10 10 es se se	w m w w w w w =	******	~~~~~	12900301 12900301
	AUTOMATIC	-DRAWING F	ROM THE	DATA BA	SE			•
·1RSV LIMT OPTN	SCÁL	FWD F S M CUTS	S	50. 14 48.	D MOK	P END		12900301 12900301 12900301
	START DET COMMAND L	AILING APC - DEFI	INING STAN	NDARD E	ND CONNE	CTIONS	-	
DEFH LAPC			23	15 1 8	1	8	1	12900301 12900301
	DEFINE SE	AMS .						
ADDP		D 6 D 6 D 2	7F P	END END END END	-4 8 -5 8 0	. 9 3	1 2 3 4	12900301 12900301 12900301
CNTR SHEL SLPE2 2 2	CALC	P+ Y D XX XX XX		END.		7 , 3	520 5 1 5 2	12900301 12900301 12900301 12900301 12900301 12900301 12900301
LINK LBHO SLPEZ 2	NEM	XX P+ L L XX XX XX XX XX	4				5 4 511	12900301 12900301 12900301 12900301 12900301
CTRE SEAM	NOCT	P	1 2 1 2 1 2 1 2 1 2 1 2 1 2 1 2 1 2 1 2	10 9 9	P P P L LB6	11 12 12 9 13 9 14 . 9 SS 18		12900302 12900302 12900302 12900302 12900302 12900302 12900302 12900302 12900302 12900302

	DEFINE HOLES													
ADDP		P		20					3 6				20	12900302
		P		50				A	0.		В	20	1	12900302:
HOLD		Ĺ	LB6					^	•••		•		•	12900302
MIDN		IS	17					M			IS	24	18	12900302
SPLT		15	18					M			IS	25	19	12900302
		P	10	13				Р		19	10	C 2	21	12900302
		IS	• •	10				, k			IS	26	18	
•	•	IS	20 U					P M			15	27	19	12900302
			EV.	18				P		19	13	21		12900302
		P	~ .	10						17	7.0	20	22	12900302
•		18	21					M.			15	28	18	12900302
:		IS	29	4.0				, M			IS	29	19	129003021
		P		18				P		19	_		23	12900302
		D	28F	.•				N			D	28F	18	129003030
		IS	32	٠. ـ				, _ <b>j</b> .			18	32	19	129003030
	•	P		16				P		19	_		24	129003030
		D	37F					M			Đ	37F	18	12900303:
		IS	35					M.			18	35	19	12900303:
		P		18			•	P		19			25	12900303
		Đ	46F					. h			D	46F	18	12900303
		IS	38					M			IS	38	19	12900303
		P		18				F		19			26	12900303:
	•	D	55F					۲			D	55F	18	12900303:
	-	18	41					۲			15	41	19	129003034
		P		18				P		19			27	129003034
		IS	42			. 41	•	M			IS	42	18	129003034
,		IS	43	•		•	•	þ			IS	43	19	129003039
•		P		18				P		19			28	129003039
		S	.67F			-		¥	•		D	67F	18	129003036
	=	IS	45					M			IS	45	19	129003036
		P		18				P		19			29	12900303
		IS	46					N			15	JL	18	12900303
		IS	47					K	•		IS	47	19	129003037
		P	• •	18				P		19		•	30	129003038
LOOP		N		1	N		21	N		30				129003038
PTNO		¨L		ī	••		h- 4	••						129003038
HOLD		Ρ̈	*	•				A	90.		н	100P 1	1	129003030
ENDL		Ŋ	-	1				^	,,,		•••	200. 1	• .	129003040
ADDP		. "L	L56	. *	S	50			2 1			3 4 8	35 .	129003040
		- ٩	L00	32	J	20			909		u	32	JE _	
ROLD		•	3:			2	n	۸.	707		••	26		129003040
						2	12							12900304:
4055	•			6	. 1	LB6			~ D =	>		4 2	77 .	12900304
ADDP		ם	WDK	77	٠ ـ	200			-2,2	•	ப	13	33	129003042
HOLD		P		23			_				r	33		129003048
			10	Þ	•	2000				•			4. *-	129003043
ADDP		ير	LB6	<i>n</i> A	Ũ	5%D						11015		12900304
		P		40					-	_		5 18		12900304
•		P		40					7	9			41	12900304

_												
		P		44		************		7 9			43	12900304
CNTR												12900304.
CORN		P		40								12900304
	_	P		41			_	12 9				12900304
	•	þ		42			•	12 9				12900304
		P		43								12900304
CTRE	HOLD	•		5	٠.		9		Н	31		12900304
~~~~	DEFINE STIFFEN	EKS			,	:						
4000					 						~~~~~	
ADDP VECT	•	M L	L86			24 · 17	Ŕ	49			49	12900304
		P	LDD	50	10	11	P P			6	50	12900304
LINT		•		20	IS	22		. 49				12900304
* •	•	M M			IS	50	A* P	E.,			51	12900304
•		m M			15	21		50			53	12900304
3		P		49	15	< I	A* P	· 50			52	12900304
•	• • • •	M		47	IS	23		50			E 7	12900305
PNCH3	•	M			IS	50.	Á* P.	50			53	12900305
STIF	501NSIDLAPCSNPE	14			13	20	٠٢.,		S	101	-	12900305
PNCH3	JUINSTULAPCONFE	M			7 0	21	P	1 8 52	3	101		12900305
STIF	501NSIDLAPOSNPE	1-1			10	. 21	F	1 8	s	102		12900305
PNCH3	DOINGIDEAR DOM E	М		-	75	55	· Р.	51	3	102		12900305
STIF	501NSIDLAPCSNPE	• •			10	2.2	• •	. 1 8	S	103	•	12900305
PNCH3		М			IS	23	Ρ.	53	J	103		
STIF	501NSIDLAPCSNPE	•••			10	<b>5 .</b>	•	1 8	\$	104		12900305 12900305
LINT	201110102211 00111 2	Ł	LB6		S	51	Ŋ		Š	51		12900305
P4111		Ď	NOK		Š	23	_		•	31	60	12900305
•		Ĺ	LB6		S	51	, P		s	51	80	12900305
		D	MDK	•	S	238	A *		•	21	61	12900305
		ĭ	LB6		S	51	, F		s	51	01	12900305
		ā	MDK		S	244	A*	•	·	<b>J</b> 1	62	12900305
		L	LB6		S	51	Ϋ́P		5	51		12900305
•	~	D	MDK		S	25	Α <b>*</b>		U	21	63	12900305
LAPC				2 .	s	23	7 ^^				63	12900305
PNCH2		D		_		23	P	60				
STIF	502NSIDLAPCSNPE	•			10	22	•	7	8	<b>%73</b>		12900305 12900305
PNCH3		n	MDK		TS	238	P	61	U			
STIF	502NSIDLAPCSNPE	v	MUK		10	230	•	7.	9	<b>%74</b>		12900305
PNCHS		n	MDK		1 9	244	Р	65	J	11.74		12900305
ŞTIF	502NSIDLAPCSNPE		1-011		.* •		. '	. 7	s	1175		12900305 12900306
PNCH3			MDK		TS	25	Ρ.	62	•	****		12900306
STIF	502NSIDLAPCSNPE		1.01		10		• •	7	S	₩ 76		12900306
ADDP	JULNOIDEN COM. E		LB6		s	49	_	•.	·	11015	6.71	12900306
		M			• S		=				65	12900306
PNCH3		L	LB6			49	·P	64.				12900306
STIF	503MSIDLAPCSNPE	-				~ •	•	.7	S	105		12900306
PNCH3		М			IS	49	x		Y	44		12900306
STIF	503NSIDLAPCSNPE	-•			_ ~	- •	••	7		N70		12900306
								•	_			

		~~~.			 -	52									1 ጋርላለንሴ4
LOAD															12900306
	REDEFINE STIFF		(5 F)				A !	NO :							
HOLD		P		29				A	90.			29		1	12900306
		P		30	_			Α.	90.		H			1 .	12900306
INT	r	L				51		M			S	51			12900306
	•	Ð	MUK			234		<b>A</b> *			_			66	12900306
		L	LB6		5	51		Ŋ			5	51		. =	12900306
	•	D	MDK		IS	248		<b>A</b> *			_			67	12900306
		Ļ	LB6		S	51		M			\$	51			12900306
		D	MDK		I			DO	SND		S	<b>2B</b>		68	12900306
	-	L	LB6		S	50		k			S	50			12900306
		D	MDK		IS	24		D	SND		S	28		69	1290030ė
_	•	L	LB6		5	49		M			5	49			12900306
•		D	MDK		IS	24		D	SND		S	<b>5</b> B		70	12900306
		D	5/10		IS	2 A	٠	D	67F		S	8 A			12900306
	•	L	L86		S	47		k			S	47		71	12900306
		D	SND		IS	24		D	67F		F	24			12900307
		L	L66		S	46		K			S	46		72	12900307
		D	SND		IS	2 A		D	67F		S	24			12900307
	•	L	LB6			45		K			Ş	45		73	12900307
NCH3		D	WDK		IS	234		F	66						12900307
STIF	502NSIDLAPCSNPE									7	S	<b>1</b> 77			12900307
NCH3	• •		MDK		IS	248-		P	67						12900307
STIF	502NSIDLAPCSMPE									7	S	W78			12900307
NCH3			MDK		IS	24		P	68						12900307
STIF	502NSIDLAPCSNPE	,	•							7	S	N79			12900307
DNCH3		P		68	•			P		69			•		12900307
STIF	502NSIDSNPESNEE						7			7	S	W60			12900307
DNCH3		P		69				. P		70					12900307
STIF	502NSIDSNPESNPE						7			7	S	181			12900307
NCH.		P		70				D	GNS		S	28			12900307
STIF	502NSIDSNPESNPE	•					7			7	S	w82			12900307
NCH3		D	DNS		IS	<b>2</b> A		P		71					12900307
STIF	502NSIDLAPCSNPE									7	S	W83			12900307
NCH3		P	•	71				P		72					12900307
STIF	SO2NSIDSNPESNEF	)		-	••	-	7			7	S	N84			12900307
EH3N <sup>c</sup>		P		72	• •			P	•	73					12900307
STIF	502NSIDSNPESNPE						7	•		7	S	<b>N85</b>			12900307
NCH3		P		73			•	D	67F		S	AS			12900307
STIF	502NSID SNPESNP	E			•		7			. 7	S	N86			12900307
	LUAD FRAMES 51	AN	D 50		-					•					
DAD.					<b>-</b>	51.		F	50.			1.			1290030
NPE	•														1290030
						_	22-								



Example 3: Window of Webframe 52 after detailing by 'DEMO'

## APPENDIX 1

# HULLOAD INPUT REQUIRED TO LOAD STIFFENERS 'TO DATA BASE

## \*\*\*\*\* EXAMPLE 1: INPUT DECK TO DEFINE AND LOAD VERTICAL STIFFENERS ON BULKHEAD 31.

			•								
*JOB LNG HULDDISK	N	3001									17300100
· INPS 16TH	N	3001							•		17300100
TRAC TBHD	T	31.								•	17300100
TOLR		100		100							17,300100
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•		T17	S	T18	S	T19		S	T20		17300100:
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## \*\*\*\*\* EXAMPLE 2: INPUT DECK TO DEFINE AND LOAD VERTICAL STIFFENERS ON TRANSVERSE FLOORS AT FRAMES 46 THROUGH 52.

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## \*\*\*\*\* EXAMPLE 3: INPUT DECK TO DEFINE AND LOAD STIFFENERS ON WEBFRAMES 46, 50 AND 52.

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	S #58		N59 S	N60 5		173003002
•	S #62		463 S	464 S		173003002
•	S #66		167 S	168 S		173003002
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Puci	•••	••	• •	-	16	173003003
DEDT.	L LB6		17 M	<b>S</b>	24	173003004
REPT .	·L LB6		51 M	. \$	51	173003004
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## APPENDIX K

LSCO FINAL REPORT - SUB-TASK 2.1 COMPUTER-AIDED DESIGN SYSTEMS

## FINAL REPORT--SUB-TASK 2.1 COMPUTER AIDED DESIGN SYSTEMS

## CONTENTS

- I. SUMMARY
- II. COMPARISON ANALYSIS CONCLUSIONS
- III. IHI PRODUCTION IMPROVEMENT SUGGESTIONS
- IV. CHANGE ANALYSIS CONCLUSIONS
- V. RECOMMENDATIONS
- VI. REFERENCES

#### I. Summary

Sub-task 2.1 began in November 1978 with a meeting of IHI and Levingston representatives. This first meeting was held primarily to determine the method of approach which would provide accurate descriptions of both systems from which detailed comparisons could be made. From such comparisons, deficiencies in the Levingston system could be isolated and recommendations for change could be developed.

Thus, IHI began its investigation of the Levingston system which continued through February of 1979. The result of this investigation was detailed report prepared by Masumi Hatake of IHI. This report consists of three sections: 1. Examine and Study the SPADES System; 2. IHI System; and 3. Comparison of the Capabilities between Levingston's SPADES System and IHI System. Each section is described as follows:

The purpose of Section 1 was to determine if full utilization and benefit is being realized from the use of the system by Levingston. All available SPADES modules were studied (even though some have not been installed for use at Levingston). (Appendix 1, P. 1.3-1).

Section 2, the IHI System, adequately describes the characteristics and functions of each IHI system (module). (Appendix 1, p. 2.1-l through 2.1-9). The relationships between the systems is shown in a flow chart. (Appendix 1; p-2-0-3).

Section 3, which compares the capabilities of the IHI system and the SPADES system, concludes that the capabilities of the two are almost equal. (Appendix 1, p.3.1.2). In addition analysis is provided on the adaptability of the IHI software to the SPADES system. (Appendix 1, p.312-1).

A meeting was held for the presentation of the report and to allow for explanations and discussions. (Technical memoranda-Appendix 12).

#### I. Continued

In conjunction with the above report, IHI also delivered to Levingston descriptions of the IHI computer design modules. The descriptions were used in the establishment of a detailed comparison table. This table is the basis for Levingston's report entitled "Study and Comparison of SPADES vs. IHI System". (Appendix 2). The comparison table was established according to the specific application of each module. The report consists mainly of a detailed explanation of the comparison table followed by conclusions.

A detailed deficiency analysis and trade-off study was not prepared due to the fact that IHI and Levingston both concluded that the SPADES system is adequate for Levingston's present and future needs. A memorandum to that effect was prepared and is attached as Appendix 3.

## II. Comparison Analysis Conclusions

(RE: Appendix 2)

In general, the IHI N/C system and the SPADES N/C system are both very good. Whereas, IHI utilizes its system to its maximum capabilities, Levingston's use of SPADES is not so comprehensive due to lack of implementation and facilities. For example, the absence of an N/C drafting machine is partly the reason for not perfoming lines fairing in-plant, and for not installing the new DEMO-module of SPADES. The functions of HULLCAL, another SPADES module not in use at Levingston, are duplicated through the use of SHCP which is obtained from another source. HULLCAL directly references the data base but needs some improvements in capabilities, whereas SHCP requires a manual loading of data but adequately performs the necessary functions.

In each case, there is some justification for non-use of certain SPADES modules. Benefits of the additional machine capabilities from the installtion of an N/C drafting machine may very well be offset by costly additional personnel requirements. Contracting lines fairing from outside sources may be less expensive than performing the same work in-plant, and definitely draws from the experience of the contractor in fairing many different hull types and shapes.

SPADES may have the advantage over the IHI system when comparing the modules that actually duplicate the old style lofting and template making methods due to easier input coding and full utilization of the data base set up with the Fairing and Hulload modules. (All SPADES modules access the data base, whereas some of the IHI modules are standalone systems.)

Implementation of the IHI system at Levingston is not necessaryz- The SPADES system currently available to Levingston is more than adequate for present and future needs.

The capability of SPADES is considered to be almost equal to the IHI system. Full configuration of the SPADES system covers much the same area as the IHI system, except for structural analysis and vibrational analysis (which are obtained by Levingston from other sources).

## III. Production Improvement Suggestions - IHI

IHI strongly recommends that Levingston make full use of the SPADES system. (p.1.3-6, Appendix). Specifically, it is recommended that Fairing can be performed in-plant as opposed to the present practice of contracting this work, thus creating a time delay. Fairing the hull inplant would enable a faster start on all design work. (p.1.3-5, Appenix 1).

In order to directly utilize the Fairing module, (as well as some other modules not currently used), IHI recommends the installation of a large drafting machine. (p.1.3-7, Appendix 1). This will also help eliminate duplicated work between the Mold Loft and Engineering. (p.1.3-5, Appendix 1). The drawings from the drafting machine can be used as the base drawings from which more detailed design can be performed as well as checking the drawing for input. (p.1.3-6, Appendix 1)

The DEMO module of SPADES is also facilitated through the installation of an N/C drafting machine. (p. 1.3-7, Appendix 1). According to the IHI report DEMO will provide the following:

\*Fast issue of working drawings

\*Easy and quick follow-up to design change

\*Exclusion of duplicated work

\*Manpower savings

As part of Sub-task 2.2, a justification analysis will examine these opinions in detail as to their validity.

### VI. Change Analysis Conclusions

It was finally concluded by IHI and Levingston that SPADES offers enough depth, flexibility and future growth potential so that no changes involving the IHI system will be necessary. For this reason a brief explanation of this conclusion was submitted in the form of a memorandum in lieu of a formal change analysis document. (See Appendix 3.)

Some recommendations have been received to install additional N/C hardware. In relation to IHI's recommendations, this activity is directly related to Sub-task 2.2-Numerical Control Steel Fabrication. Upon actual installation of recommended equipment, expansion of Levingston's use of SPADES is to be expected. Monitoring and reporting of the installation of such hardware will be reported in the sub-task 2.2 Final Report. As Levingston's use of SPADES increases, reports to that effect can be submitted in addendum to the sub-task 2.1 final report.

### V• Recommendations By IHI

IHI recommends, through Mr. Hatake's report, that Levingston should attempt to make full use of the SPADES system. (p.l.3-6 Appendix 1).

By installing a large N/C directed drafting machine, it would become possible to install two very helpful SPADES modules, Fairing and DEMO. Fairing is now done by Cali and Associates which creates additional expense and time delays. (p.1.3-5, Appendix 1). DEMO would contribute significantly to 1.) fast issue of working drawings, 2.) easy and quick follow-up to design change, 3.) exclusion of duplicated work (p.1.3-5, Appendix A) and 4.) manpower savings (in Engineering office p.1.3-6, Appendix A). (p.1.3-7, Appendix A).

As stated in Section III of this report, the benefits of the above recommendations will be weighed as part of sub-task 2.2.

## VI. <u>References</u> (Appendices)

- 1. Study Of SPADES and LSCo Utilization IHI
- 2. Study and Comparison of SPADES vs. IHI System LSCo
- 3. Memorandum: Deficiency Analysis and Trade-Off Studies LSCo
- 4. Brief Explanation of IHICS
- 5. IHICS Input/Output Examples
- 6. SPECS
- 7. SPECS Actual OUtput Example
- 8. Plate Users Manual and Output Sample
- 9. SHELL
- 10. LODACS
- 11. SPAC
- 12. Technical Memorandum: Meeting--1HI 2.1 Final Report (IHI)

### APPENDIX L

LSCO FINAL REPORT - SUB-TASK 2.2 NUMERICAL CONTROL STEEL FABRICATION

### FINAL REPORT--SUB-TASK 2-2 NUMERICAL CONTROL STEEL FABRICATION

### CONTENTS

- II. COMPARISON ANALYSIS CONCLUSIONS
- III. IHI PRODUCTION IMPROVEMENT SUGGESTIONS
- IV. CHANGE ANALYSIS CONCLUSIONS
- V. RECOMMENDATIONS
- VI. REFERENCES

### SUB-TASK 2.2 FINAL REPORT

### 1. SUMMARY

Work began on Sub-task 2.2 of the Technology Transfer Program in November 1978 when Levingston and IHI representatives met to discuss scope, approach and methods. As in Sub-task 2.1, the first step was to study and describe the IHI and Levingston system in order to quantify the present use of current system as compared to its potential capabilities. IHI's M. Hatake began a thorough study and investigation of the Levingston N/C system. At the same time, a documentation of the IHI N/C system was prepared. In March of 1979, IHI delivered a detailed report of their findings concerning Levingston's N/C system as well as a general description of the IHI N/C system. As the information was presented, iterms of comparison and contrast were identified.

A meeting was held for the formal presentation of this IHI report and to facilitate a question and answer session. Representatives of the Mold Loft and Engineering were afforded the opportunity to get a cl ear explanation of any item in the report they considered unclear or erroneous.

Levingston began a complete study and comparison of the two systems. The result of this report was a general comparison of the philosphies of N/C steel fabrication. A brief description of equipment used and the flow of events involved beginning with contract signing through actual fabrication was to be the basis of the resulting report, Consideration of IHI recommendations also became a major part of the t a s k .

Deficiencies, or area; in need of improvements in methods, facilities or equipment, were identified in Levingston's deficiency analysis and trade-off studies report. Only the items considered by Levingston to be deficient may be so because of restrictions not prevalent in the IHI yards. However, for the most part IHI's recommendations were followed up with justification analyses which helped in -the decision making of whether or not to implement change recommendations.

### II. COMPARISON ANALYSIS CONCLUSIONS

### A. Philosophies on N/C Steel Fabrication

Two different philosophies of N/C steel fabrication have developed at IHI and Levingston. These differences are perhaps rooted in differing conditions which affect the shipyards. The abundance of skilled manpower, the close relationships between yards and subcontractors, industry-wide standards and highly developed facilities, among other reasons, have all contributed to the present policies and methods utilized by IHI. At Levingston, in many cases, conditions are opposite of those at IHI. In turn, these conditions have had impact on Levingston's methods and policies regarding N/C steel fabrication.

In Japan, IHI is able to recruit mold loft and layout personnel from an abundant market of job seekers. They can also be relatively certain that once hired and trained, these employees 'Will remain at IHI until retirement, except in unusual circumstances.

At Levingston, and in the U.S. shipbuilding complex as a whole, there is a shortage of skilled lofting and layout personnel.

It follows that IHI's philosophy of N/C utilization for steel fabrication may be different than at Levingston, and it is. IHI tends to process through N/C only that steel which requires high precision and repetition of shape such as web frames, inner-bottom floors and curved shell plates. Flat, straight pieces are processed through the Panel Shop by a flame planer. Levingston, on the other hand, processes as much steel as possible through its N/C system, including flat and straight pieces.

At Levingston, plate marking for shape, structure locations, etc. is performed as much as possible by the N/C burning machine which is equipped with centerpunch marking heads. The necessary material marks and instructions are added manually after burning. Such marking is usually done before the plate is moved, but other pieces are burned simultaneously on other areas of the burning table.

On the other hand, IHI is equipped to perform a variety of N/C plate marking methods including zinc or plastic powder to form continuous line marks, manual marking of structure locations using steel tapes prepared from N/C generated data, and electro-photo marking .

At IHI-Aioi, a total of 25% of all plates for each Future 32 vessel was fabricated by the N/C equipment. By contrast, at Levingston almost all plates for a vessel are produced by N/C equipment.

\*Line marking-zinc powder burning

### II. COMPARISON ANALYSIS CONCLUSIONS, CONTINUED.

Equipment And Personnel

### Levingston-Orange

IHI-Aioi

\*Punch marking on same machine

\*13 Loft personnel \*1500 to 1800 tons/month

\*50 Loft personnel \*6000 to 8000 tons/month

One (1) Burning Machine \*Direct Control

Three (3) Burning Machines Tape Control

\*IHI System

\*SPADES

- C. Levingston Deficiencies Identified By IHI.
  - No alternate method of automatic burning for times when N/C burning machine is out of service.
  - 2. No panel shops for rapid fabrication and assembly of flat and curved panels, complete with stiffeners, using a flame planer.
  - 3. No numerically controlled drafting system for visual checking of data base for hull parts, template making for fabrication and bending, producing drawings for use with optical tracer type burning director and possible applications for producing working drawings for Engineering .
- Results of Deficiencies Indentification
  - 1. As an alternative to cutting flat straight panels on the N/C burning machine, an optical 1:1 burner director has been installed to direct automatic burning machine when direct numerical control system is not in operation.
  - 2. Work has begun or setting up a panel shop with a flame planer for flat panel fabrication.
  - 3. A system change analysis was conducted with the result being a recommendation to implement the installation of a complete N/C drafting system in the mold loft.

Page 4

### III. IHI PRODUCTION IMPROVEMENT SUGGESTIONS

Since the volume of Levingston's steel fabrication is expected to increase due to the construction of the 36,000 D.W.T. dry bulk carriers, IHI has offered some suggestions such as levelling of the work load, rearrangement of shops and supporting work areas, etc.

Related to sub-task 2.2, IHI recommends consideration of an alternate method of automatic burning (to compensate for times when the present N/C burner is broken down) and installation of a large N/C drafting machine in the mold loft to facilitate a scaled plan system of lofting for template making, plate validation before burning, etc.

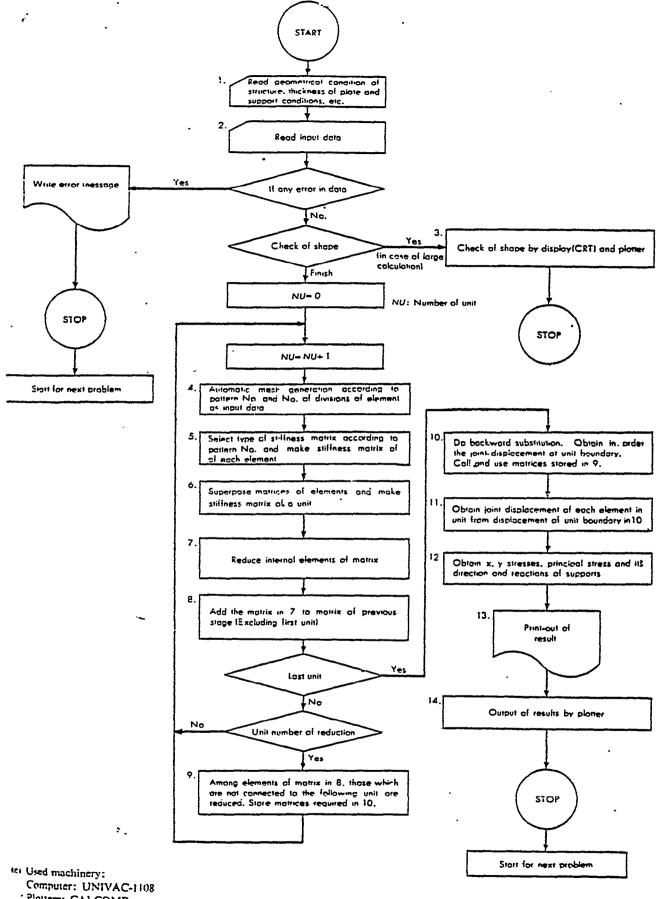
An alternate burning method of automatic burning has been installed as a result of IHI's suggestion. It is in the form of an optical 1:1 burner director. Thus, the main activity of this study is in the area of installing the N/C drafter in the mold loft.

### IV. CHANGE ANALYSIS CONCLUSIONS

In almost all of IHI's reports concerning Levingston's N/C steel fabrication, it has been suggested that a major deficiency in the mold lofting and Engineering systems is the absence of a scaled body plan system utilizing an N/C drafting machine. As a result, the attached system change analysis (reference 5) was conducted. Supporting investigative and statistical data was received from IHI (reference 6) and from Levingston's Industrial Engineering Department (reference 3). This data is the basis for determining the results of changing to an N/C drafting system in the mold loft.

In general, it is concluded that the mold loft could successfully utilize a large table N/C drafting machine with the necessary related hardware for interface with the present system. The cost vs. estimated benefit study indicates paybacks in several areas including highly accurate data bank verification, part validation before burning, template making, and providing drawings for use with Levingston's optical tracer burning director.

It may be further concluded that a monetary cost payback due to high accuracy and saved lofting manhours is the obvious benefit of using such a system. But another benefit will accrue in the form of taking more time away from the total amount of time between contract signing and delivery date. These are the kinds of savings that can benefit Levingston now as well as making the U.S.A. shipbuilding market more attractive in the near future on a worldwide basis.



Plotterr: CALCOMP
Display: ADAGE

Fig. 1 Flow chart of ZPLATE program

method item (2) is distinctively useful for structures with periodic identical patterns like ship hull structures. As to item (3), the program must be balanced with items (1) and (2) for saving labour of users. Otherwise, it is not meaningful no matter how refined the program is.

In item (5), an error message can be printed out and a check of structural geometrical data is to be done by plotter or graphic display. For item (6), automatic drawing of calculated results is done by plotter or graphic display. It is also devised to print out Calculated results in most convenient form. For item (7), it is not possible in the present stage to calculate the inverse of a stiffness matrix within the core memory of even a super large computer although it depends on the faculty of the computer. Therefore, some memory devices such as magnetic tape, disc or drum are required. Of course, it is truly useful in this case to use a super large computer with large core memory, but such a large computer is usually open to various jobs by time sharing system. This is because it is not economical to provide a large core memory only for a single small job. Therefore, we must make use of some external memory device effectively although a certain core memory is still needed.

If item (7) is solved, items (3) and (4) can be automatically worked out. If items (1) to (7) are all satisfied, it will be a very refined general purpose program finite cicment method.

### 3. Method of analyzation

ZPLATE is made to analyze by finite element method the elastic stress of plate structures subjected to static loads. Plate structures are composed of thin plates in the same plan: or of a three-dimensional combination of plates. The plate as an element of structures does not have stiffness against out-of-plane deformation but retains only in-plane stiffness. Therefore, stresses are assumbled to be in the plane stress equilibrium. However, the program can also analyze a plate with line membrers (created as truss elements because of no flexible stiffness). The input data are overall structural formulation, dimensions of members, material modulus, structural conditions of locations of supports and loads (including forced displacement) and the outputs are displacement and stress of each element and reactions of supports.

The structure to be analyzed is first divided into a remain number of units. Each unit is a **group** of **several** or several tens of elements and is functional as an element in a wide sense. Each input is associated with units and joints which are vertices of units. The unit has two types of geometry and is selected such that any structure can be easily idealized. The unit plays an important role in saving labour in preparing input data. We use "unit division method" to solve simultaneous equations. However, the unit of the unit division has a slightly different meaning from the unit defined here, **as** will be described later (section 3.1).

3.1 Composition of program

The flow chart of the program is shown in Fig. 1 (The processes 1 to 14 are a set of subprograms specified into units of functional characteristics of operation). For the substructure method, the main program shown in Fig. 1 wiI1 be used as a subroutine as described later-

In fig. 1, the discrimination "Unit number of reduction" between processes 8 and 9 indicates that reduction is to be performed when the reduced elements of the matrix in process 9 become large by superposing several units (process 8). In other words, the reduction is not applied to each unit. This is the reason why, in the unit division method, we call one unit for these groups of units put together.

#### 3.2 Stiffness matrix of element

Out of various stiffness matrices in plane stress problems, the following expressions can be used at present

(1) Line element

Geometric function

$$u = \alpha_1 + \alpha_2 x \dots (1)$$

(2) Triangular element (uniform stress)
Geometric functions

$$u = \alpha_1 + \alpha_2 x + \alpha_2 y$$

$$v = \alpha_4 + \alpha_5 x + \alpha_6 y$$

$$(2)$$

(3) Rectangular element<sup>(2)</sup> Geometric functions

$$u = \alpha_1 + \alpha_2 x + \alpha_3 y + \alpha_4 x y$$
  

$$v = \alpha_5 + \alpha_6 x + \alpha_7 y + \alpha_5 x y$$
(3)

(4) Arbitraray quadrilateral and triangular element by Hybrid Method<sup>(3)</sup>

Assuming linear displacement along each side of the element, stresses can be put in the forms;

$$\sigma_x = \beta_1 + \beta_2 y + \beta_6 y + \beta_4 y^2 + \beta_{10} x^2 + \cdots$$

$$\sigma_r = \beta_4 + \beta_4 x + \beta_7 y + \beta_9 x^2 + \beta_{10} y^2 + \cdots$$

$$\tau_{xy} = \beta_5 - \beta_6 y - \beta_7 x - 2\beta_{10} xy + \cdots$$

$$(4)$$

Equation (4) satisfies the equilibrium equations of stresses of the element as can be seen from the forms (4). Employed in the program are two types which include the terms up to  $\beta_5$  and  $\beta_7$  respectively.

Since each term of eqs. (1) to (4) satisfies the compatibility conditions for the adjacent elements due to the linear displacement along the sides of elements, mixed use of these elements. is possible.

Fig. 2 shows a comparison of stresses by his method and by beam theory applied to abeam with upper and lower face plates and with fixed end conditions. The line element (1) is used for the face plates. As is clear in the figure, stresses of the rectangular element with  $\beta_5$  by the hybrid method are closest to those of the beam theory and in fact almost identical.

This is because the stress distribution is almost identical if the terms up to  $\hat{\beta}_5$  are used, therefore, although we cannot conclude which is the best element, an accurate result might be expected for structures like frames of a slup hull if quadrilateral elements up to  $\hat{\beta}_5$  of the hybrid method are used even if the division of elements is coarse

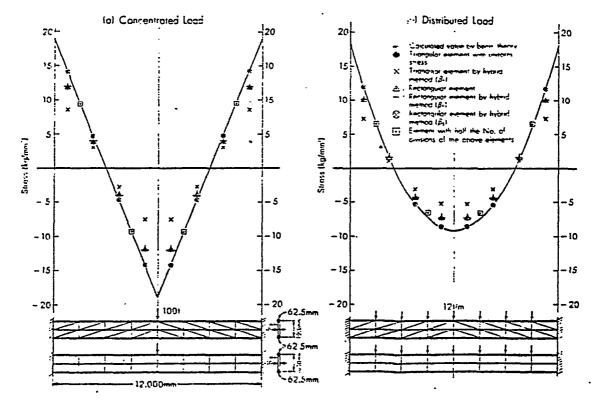


Fig. 2 Comparison between beam theory and P.E.M. with various elements

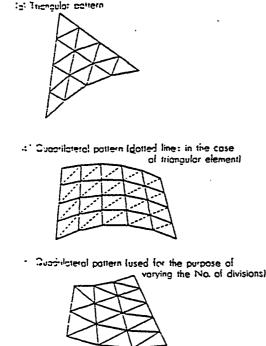


Fig. 3 Method of automatic mesh generation

general the rectangular element (3) seems to be greater than the rectangular element of (4). Also the rectangular element (2) and (4) give almost identical with However the latter is more useful for the stress shown of parts with stress concentration where vary distinctively since the stress distribution element can be obtained by this element.

When this program is used for structural analysis, first divide it into rectangular elements wherever it is possible. Then arbitrary quadrilateral elements are used for the parts where rectangular elements cannot be applied. Finally use triangular elements for only the part where other elements are not possible. Since automatic mesh generation within units is possible in this program, stiffness matrix or stress matrix of an element can be constructed in a single processing if the unit is rectangular. Since the same process is applied to other elements too the computer time will be shortened.

### 3.3 Automatic mesh generation

The fundamental patterns of units are of two types. arbitrary triangle and arbitrary quadrilateral as shown in Fig. 3. where one side of the triangle and the opposing two sides of the quadrilateral can be replaced with a circular arc. Each side is divided with equi-distance by using the number of divisions assigned in input data, and make elements as shown in the figure. The number of divisions of all sides of a triangle and of opposing two sides of a quadrilateral must be the same. Furthermore, since the number of divisions of the -boundary line between adjacent units must be the same, the determination of number of divisions is restrained. To solve this, a mesh generation patterns as in (c) of Fig. 3 is introduced.

The method of division is not only effective in the difference of labour between preparing input data of each element and preparing input data of each unit. but also convenient since it allows us to make fine mesh

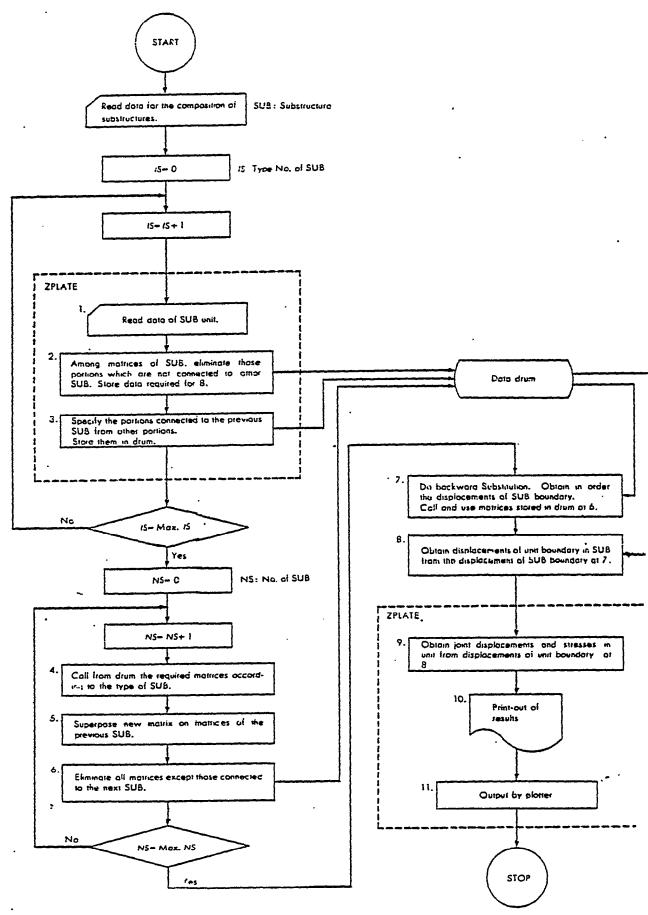


Fig. 4 Flow chart of program by substructure method

where stresses vary rapidly and to make coarse mesh wherever we have no significant change Of stresses in a similar manner to the current mesh generation when structure is idealized.

The shape of circular arc is assigned by given coordinates of an arbitrary point on a radius or circular arc. The latter method is convenient for preparing input by approximation of an arbitrary curve as a circular arc. It can be considered that this method retains the adaptablity of shape since any shape can be expressed by combination of these fundamental patterns.

### 3.4 input and output

In order to prepare input data accurately within a short time, the automatic mesh generation of elements is carried out from the viewpoint of reducing the total number of data. However more important is the faculty of the mechanism of error check. Note that the input error does not simply mean the consumptions Of computer time but means bringing about the risk of using croneous results without noticing. As in Fig. 1. data are examined for grammatical error or contradiction of geometry and mesh generation through the error check process of the program and the error messages are printed out. Data corrected in this way proceed to the next page where coordinates of joints and arrangement of and arc examined by graphic display. The examples in Figs. 6 and **10** arc photocopies of graphic display. The program is made in such a way that a view of structures from any angle is produced on the screen of the display device by simple operation of rotating the dial, and-even some complicated structures can be examined relatively only for error of coordinates. Since the plotter draws various mesh generations (Fig. 7 and others). the shape of divided elements can be affirmed at the same time as geographical error check.

As output the displacements and stresses are printed out for each unit and displacement of joints and stresses of elements within a unit are arranged such that they correspond to the arrangement of actual joints and elements. Therefore reading of calculated results is very easy. Outputs by plotter include figures of mesh general\*% and displacements of joints (Fig. 7) and of principal distribution (Figs. 7 and 12) in which maximum absolute principal stresses are shown by numbers and the directions by arrow signs.

#### 3.5 Treatment of large scale calculation

One of the problems for a program with large capacity how to handle large matrices. One advantage of INVAC-110S computer **is** the ability to use a high need magnetic drum. The random access of this magnetic drum is easy due to the treatment program and the stress time is in the order of one tenth of that of a disk.

advantage **is** made use of as an extension of the core The limitation of band width is a barrier to the treatment of a large matrix in the unit division method. In this program that limitation is avoided by using a drum. This will be explained in the following

paragraphs.

The process 9 of the flow chart in Fig. 1 is given as follows:

$$P = K \cdot D \qquad (5)$$

$$P = \text{Load matrix}$$

K = Stiffness matrix

D = Displacement matrix

This equation is divided into two parts; one is the part of joints connected to the front unit (Subscript-A) and the other is the part of joints to the back unit (Subscript-B) as follows:

$$\begin{bmatrix} P_A \\ P_B \end{bmatrix} = \begin{bmatrix} K_A & K_{AB} \\ K_{BA} & K_B \end{bmatrix} \cdot \begin{bmatrix} D_A \\ D_B \end{bmatrix} \dots (6)$$

Eliminating  $D_{A}$  from eq. (6)

$$P_R = K_R \cdot D_B \cdot \dots \cdot (7)$$

In eq. (7),

$$K_R = K_B - K_{BA} \cdot K_A^{-1} \cdot K_{AB} \qquad (8)$$

$$P_R = P_B - K_{BA} \cdot K_A^{-1} \cdot P_A \cdot \dots (9)$$

 $K_A$  is controlled within a certain size as described in section 3.1 to calculate accurately the inverse matrix in the core. The elements associated with the subscript B vary depending upon the case or each stage of calculation. Thus, this is divided equally as follows:

$$K_{B} = \begin{pmatrix} K_{B,11} & K_{B,12} & K_{B,12} & \cdots \\ K_{B,21} & K_{B,22} & K_{B,22} & \cdots \\ K_{B,21} & K_{B,22} & K_{B,33} & \cdots \\ \vdots & \vdots & \vdots & \vdots \\ K_{AB} = (K_{AB,1} K_{AB,2} K_{AB,2} & \cdots ) & \cdots \end{pmatrix}$$
(10)  

$$K_{B'} = (P_{B,2} P_{B,2} P_{B,3} & \cdots ) \qquad (11)$$

An area sufficient for the treatment of one matrix thus obtained is reserved in the core and each matrix is memorized in the drum. Note that since  $K_{BA} = K_{AB}'$ ,  $K_{B,ij} = K_{B,ij}'$ , these are not memorized. Since eqs. (7) and (8) are put in the forms

$$K_{R,ij} = K_{B,ij} - K_{RA,j} \cdot K_{A}^{-i} \cdot K_{AB,i}$$
  
 $(i = 1, 2, 3 \cdot \cdots j = 1, 2, 3 \cdot \cdots)$  (13)

$$P_{R,i} = P_{R,i} - K_{RA,i} \cdot K_A^{-1} \cdot P_A (i = 1, 2, 3 \cdot \cdot \cdot) ... (14)$$

the calculation proceeds by calling one at a time the matrices with subscripts i and j and the results are again stored in the drum. The process 10 repeats the following calculation.

$$D_{A} = K_{A}^{-1} \cdot P_{A} - \sum_{i=1}^{n} K_{A}^{-1} \cdot K_{AB,i} D_{B,i} \dots (15)$$

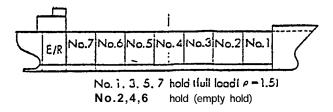
Following the above procedure any size of matrix can be treated **if the** capacity of the magnetic drum (external memory device) is sufficient In practice. however, it is wise to make **effective** use of magnetic drum since any drum capacity is limited. Consequently the required number of words must be figured out before the calculation and a process is **assigned** to put data in the drum without any blank. Especially  $K_{AB}$  and  $K_{A}^{-1}$  in the introduction process elimination require large

capacity of memory. As to  $K_{AB}$  only non-zero element is memorized since many elements of  $K_{AB}$  are zero. **Also, since K\_{A}^{-1}** is symmetric, only the upper half of the elements are memorized. As a result, even the largest matrix in the examples of application could be less than the drum capacity of 1,000,000 words. By the way, the core capacity is 65K words at present.

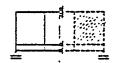
### 3.6 Substructure method

As described in the foregoing section, the computer time in the analysis of ship hull structures with periodic identical pattern can be largely reduced by common use of data and stiffness matrices by taking into consideration the periodic characteristics. In this case, the identical pattern is called substructure. Actual structures are composed of several types of substructures and in many cases they repeat periodically. The flow chart of the program by the substructure method is illustrated in *Fig.* 4, where the parts designated by ZPLATE imply using the *necessary* part of the program of *Fig.* 1. The other parts of the processes are also the same as those of the previous section in the basic consideration.

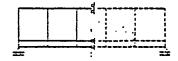
Regarding the substructure method, consider the problem of idealization of structures. As to ship hull structures, saving labour in preparation of data can be furthered by standardization of the frame patterns dependent upon the ship hull such as on tankers or ore carriers. In the example of a bulk carrier described later, the substructures are constructed quite arbitrarily. In this case, the time needed Lo prepare the input data increases, but it does not require a certain type of ship



(a) Calculated object (case of 1.5 holds)



1b) Calculated object (case of 2.5 holds)



(c) Cross section (calculated object: half portion)

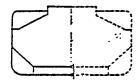


Fig. 5 Loading and bound

conditions of bulk carrier

hull structure and can be applied to any other structure. If they can be respectively called specific and general purpose programs, then the natural order is Lo first develop the general purpose and then extend it Lo the specific purpose. However, it is noted that the larger the increase in efficiency of the specific purpose, the more restrained is the idealization of structures. Important to designers are the idealization of structures and the interpretation of results by finite element method. In the case of finite element method, it is required LO introduce some approximations based on various assumptions although the idealization is comparatively easy. It seems that these problems should be determined by taking into consideration the characteristics of each structure. At the same time, it is instructive for designers to reevaluate the structural composition through the idealization process. in the future, the specification is expected to progress during the stage where the finite element method is introduced to the design process. However, at present which is still the transient stage, it seems necessary Lo make full use of the general purpose program.

### 4. Applications

As applications of the ZPLATE program, the results of three-dimensional stress analysis of a bulk carrier and longitudinal stress analysis of a destroyer are presented below.

4.1 Three-dimensional stress analysis of hull: carrier The test hull is a bulk carrier of 44,500 DWT with the following principal dimensions.

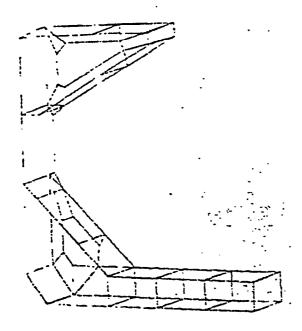
### $L \times B \times D \times d = 190 \times 30 \times 16 \times 11.55 m$

The models used for the calculations are two types with 1.5 and 2.5 holds respectively (corresponding to 3 and 5 holds if the symmetric property is considered). As to the 1.5 hold, two types of coarse mesh and fine mesh, or a total of three types, are analyzed (*Fig. 7*). Alternate loading at full draft is employed. The sizes of the three types in the analysis are shown in *Table 1*.

## 4.1.1 Assumption of calculation and preparation of input data

For the preparation of input data, the substructure method described above is employed. *Fig.* 6 shows copies of the shapes of substructures checked by graphic display. For instance, two types of substructure ((a) and (b) of *Fig.* 6) are prepared and then are orderly connected into the structure shown in (c) of the same figure. The analysis *was* carried out on the final structure. The figure shows the profile of the units where the elements are further divided into fine mesh. The details of the mesh of each part of the structure are shown in (a) and (b) of *Fig.* 7.

Although each member **is** treated as a plate, the face plates of the transverse rings and the bulkhead **stiffeners are** assumed to be line elements and the plate with longitudinal beams is treated as an orthotropic plate with increased thickness. The adjustment of **size** is made



g. 6 (2) Idealization of bulk carrier (substructure of transverse ring)

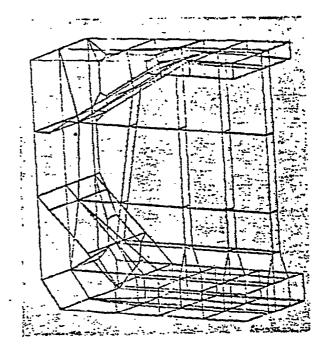
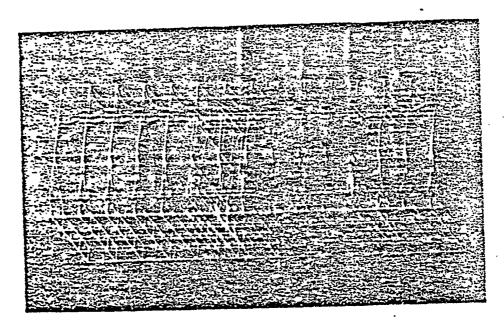


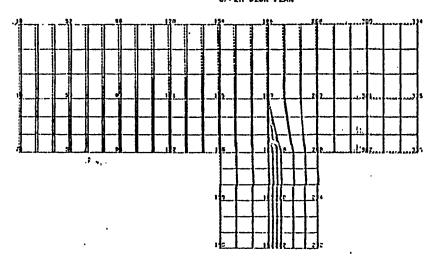
Fig. 6 (b) Idealization of bulk carrier (substructure of bulkhead

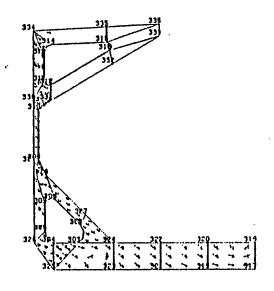


 $Fig. \ 6 \ (c) \ Idealization \ of \ bulk \ carrier \ (total \ structure)$ 

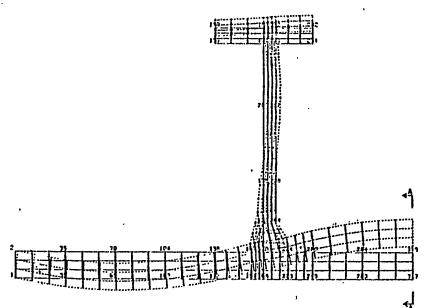
Table 1	Sizo	۸f	claculation	۸f	hulk	carriar
Table I	Size	oı	стаситатіоп	OI.	DIIIK	carrier

liems Calculation	Mesh generation	No. of unit	No.	of element	No. of joint	Degrees of freedom
1.5 hold	i Coarse mesh	444	<del>- ; .</del>	1,776	1,511	4,347
"	Fine mesh	"	:	3 <b>,9</b> 96	3,534	10,323
2.5 hold	Coarse mesh	798	i	3,192	2,640	7,680

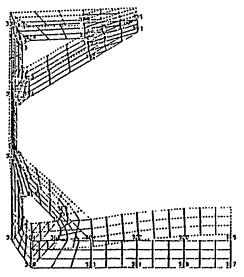












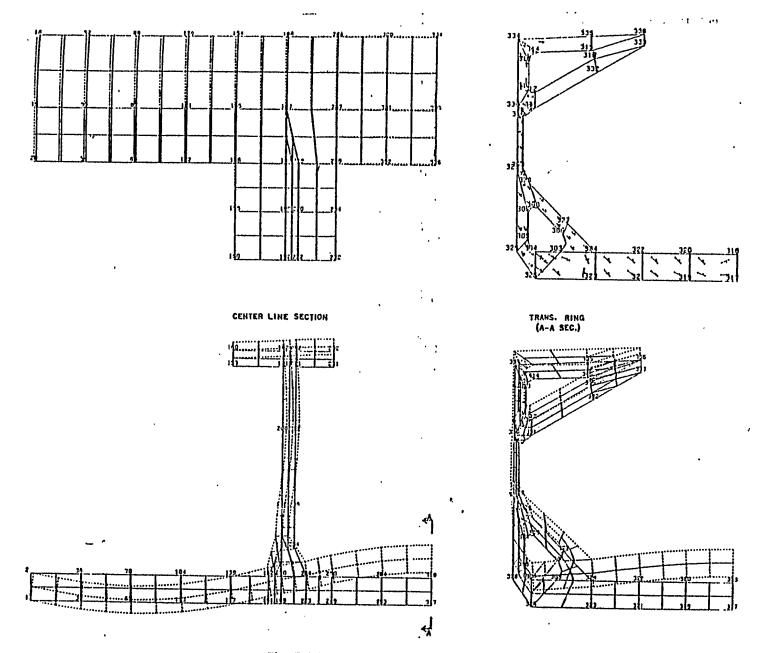


Fig. 7 (b) Results of calculation (coarse mesh)

**asuming** transverse rings as one unified ring and four frames as a unit. The objectives of the analysis are first to make a rough estimation of displacement of each part such as double bottom floor and hopper and then, with the results, to recalculate stresses of the transverse rings or the frames as the local stress analysis. The idealization is responsible to the above objectives.

In this analysis, we used mostly quadrilateral elements by the hybrid method, together with partial triangular elements of uniform stress. Most of the input data are common to the analysis of the three types. It took about five days for one person to prepare the data.

### 4.1.2 Results of calculation

Described in the following arc the way of mesh generation and the effect of adjacent holds on the threedimensional stress analysis of a bulk carrier.

Fig. 8 shows a comparison of stresses at the section of No. 4 hold center for two cases of coarse and fine mesh of 1.5 hold models. As seen in the figure, almost no difference can be recognized but the stresses at the transverse ring and frame are slightly different. However, this is mainly due to the idealization described in the section "Assumption of Calculation". In order to evaluate the local stresses, the zooming method" may be applied in which the part in question is taken out into small mesh. Therefore, the error of this order may be negligible in reality. The displacement is not different at all for the two cases arc seen in Fig. 7.

Fig. 9 shows a comparison of stresses at the same section of No. 4 hold center for the two casts of 1.5 and 2.5 hold models, to examine the effect of the adjacent holds. Again, no difference can be seen. The end conditions of each case are both simple supports at the front of the bulkhead. For the alternate loading,, the assumption does not affect the adjacent hold. Since the effect of longitudinal stresses of the front and rear holds is not taken into consideration, the stresses of longitudinal members art quite different. However, it does not affect stresses on the cross section as seen in the results.

The following conclusions are drawn from the analysis.

- The effect of the end boundary condition of the hold adjacent to the inspection hold is comparatively Small
- 2. Even the coarse mesh of this example gives very good-results as far as the hybrid mesh is used.

A bulk carrier has a complicated **structure** compared with a tanker and there are many unknown problems as to the effect of the shape of hopper tank and the type of hull, etc., on the overall strength. Therefore it is not *permissible* to draw. from the example, conclusions on the general analytical method. It is necessary to establish the three-dimensional stress **analysis of finite** element method by *further* studying bulk carriers of various types and boundary conditions.

4.2 Example of stress analyses of ship hull with long deck house

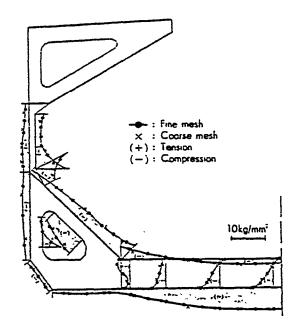


Fig. 8 Comparison of stress at section of No. 4 hold center between the mesh and coarse mesh of 1.5 hole models

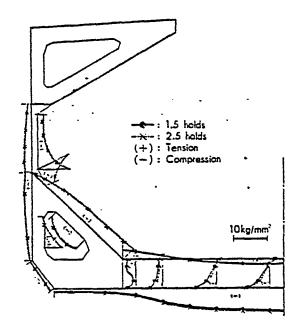


Fig. 9 Comparison of stress at section of No. 4 hold center between I.5 hold model and 2.5 hold model

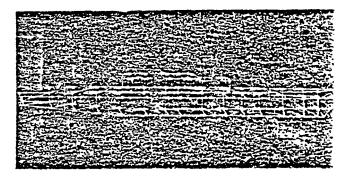


Fig. 10 idealization of destroyer

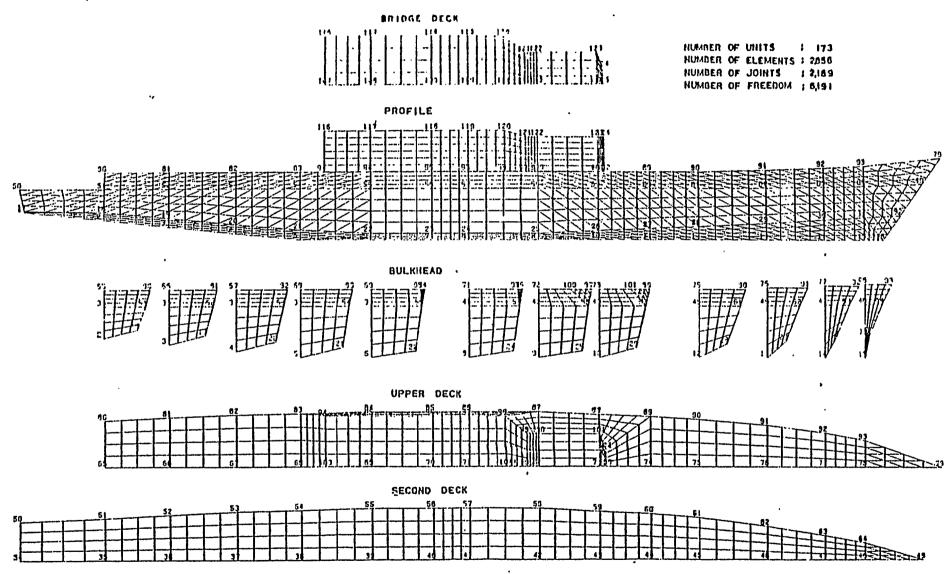
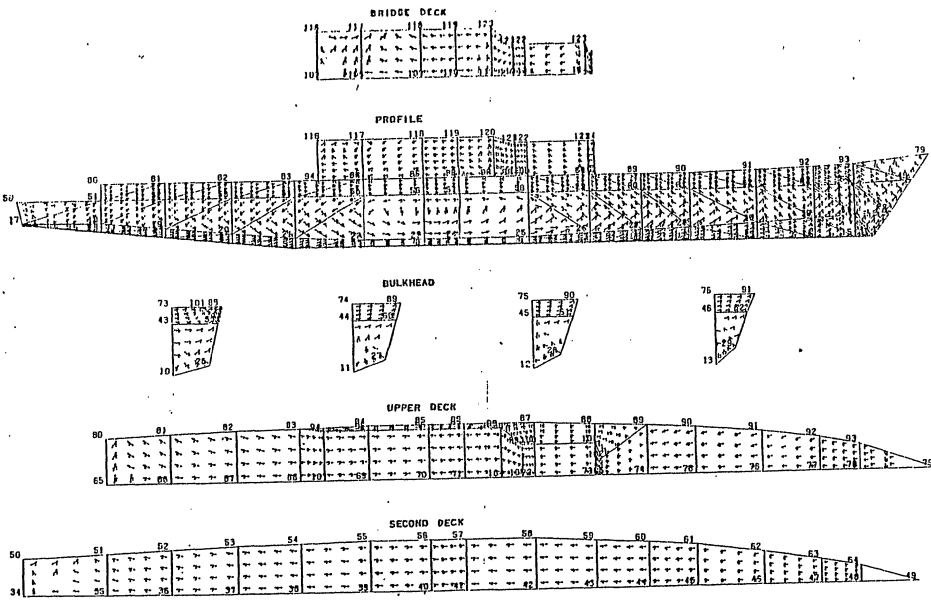


Fig. 11 Mesh layout



(Note) Arrow: Direction of principal stress
. ←→: Tension side
. →←: Compression side

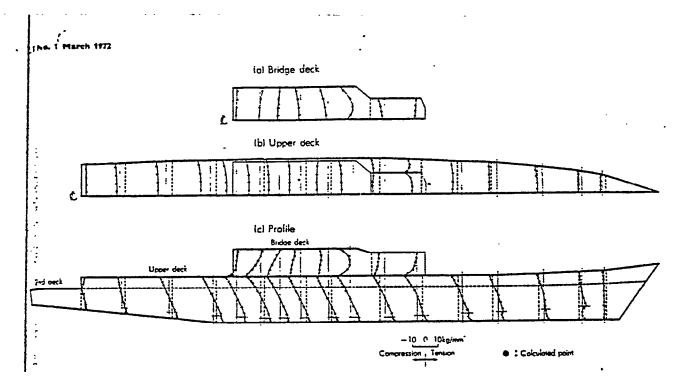


Fig. 13 Distribution of longitudinal stress of hull structure (hogging condition)

Colongitudinal stresses of a ship hull are currently tired by assuming the hull as a beam with both free. Because of shear lag, the results may be different from the actual stress distribution if the hull with a long deck house is simply treated as m of varying section. Although methods Of anare studied on the interaction of deck and ship hull, they cannot be applied directly to ship hull structures because of the simplified ideaiii)n of deck house and ship hull. However, ideaiizaclose to the real structures is possible by the finite ant method<sup>(7)</sup>. Presented in this section is the isis on a destroyer with a long deck house. Since fect of the deck house is great on the longitudinal sses of the ship hull structure in this case. stress entration is produced at the ends of the deck and quite large stresses can be seen. Therefore. saccessary to reinforce the part or to provide movable to the deck house.

### 421 Assumption of calculation

In this analysis, only a single side of the structure can be considered since both the structure and loading conditions are *symmetric* with respectit both sides. As in *Fig.* 10, the ship hull is divided it to a structure composed of flat plates so that the idealization is as close to the actual structure as possible. Treating these places as units. and applying the automatic mesh generation to the units. The mesh layout shown in Fig. 11 is obtained.

Tine longitudinal members attached to the side shell deck, deck plate and side wall of the deck house are assumed as orthotropic plates. The stiffeners of the transverse bulkheads are also treated as orthotropic plates and the in-plane

stiffness is taken into account.

- 3. As to loading, the load in the vertical direction of the ship hull is applied and distributed upon the lower end of the side shell under hogging condition. Where the crest of the wave of one twentieth the ship length locates at the midship.
- 4. As boundary conditions, the *lower* end of side shell is supported vertically at both ends. One of the ends is also supported longitudinally. Since the actual reactions at the two points are close to zero, we can examine the loading conditions. Furthermore taking the symmetry of the structure and loading into consideration, we assumed that the plate is supported in the transverse direction at the center line of the ship hull and is free in the vertical direction
- 5. Used in the analysis are quadrilateral element for the hybrid method and triangular element for the uniform stress.

### 42.2 Results of calcualtion

Fig. 12 shows the distribution of principal stresses of each part of the ship hull structure. *Fig. 23* shows the distribution of longitudinal stresses of the ship hull structure. As can be *seen* in *Fig. 13*, the deck house affects very much the longitudinal stresses of the ship hull and the distribution is quite complicated since the side wall of the deck house and the deck plate are not on the same plane. For analysis of the stress concentration at the ends of deck plate, the mesh layout of this example is too coarse to obtain accurate results. Accordingly. the zooming method may be applicable in this case. The study is not given in this paper. However, we can see some stress concentration at the ends of the deck house even in the given mesh layout.

### 5. Conclusions

A general purpose computer program designated ZPLATE was described for the plane stress analysis of plane and three-dimensional structures through finite element method As applications, the three dimensional stress analysis of a bulk carrier and the longitudinal stress analysis of a destroyer with a long deck house were presented. In the development of the program, we intended to obtain a program with large capacity which is sufficient for practical use and whose scale is balanced with the faculty of saving labour in preparation of input data and in analysis of calculated results. In this paper, examples with 4,000 to more than 10,000 degrees of freedom in calculation are presented. However, this program can solve these examples within a few days. We believe therefore that our initial objectives have been attained satisfactorily. The future problem is to extend the program to those which include analysis of out-of-plane bending, buckling, vibration and elasto-plasticity.

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### APPENDIX P

Z VIBRA - MATRIX-METHOD OF VIBRATIONAL ANALYSIS OF FRAMED STRUCTURES, AND ITS APPLICATION

# Matrix Method of Vibrational Analysis of framed Structures and its Application

Isao Neki\*

Recently, with the matrix method, analysis of elastic, plastic, vibrational or buckling problems of framed structures widely diffused in the various field of structural design, plus analysis of complex three-dimentional frames structures are becoming daily routine work. A computer program designated ZVIBRA for vibrational analysis of three-dimensional framed structures, has been devcloped, using the stiffness matrix method of the author. The present paper outlines the ZVIBRA program with some examples of its application. The program aims at analysis of elastic, vibrational responses of sinusoidal forced-loaded frames, including the influence of shear rigidity and rotatory inertia, which should be particularly useful for analysing ship hull structures.

### 1. Introduction

In recent years, with the matrix method, analysis of clastic. plastic, vibrational or buckling problems of framed structures which are broadly diffused in the various fields has come into wide use. and-by the use of this method the analysis of complex three-dimensional framed structures are becoming daily routine work. The author has developed a computer program ZVIBRA. for vibrational analysis of three-dimensional framed structures, using the stiffness matrix method. Therefore its outline and examples of its application will be presented here with an intention to place it at the service **of** the interested public.

ZVIBRA has been prepared for analysis of vibrational responses in elastic range of three-dimensional framed structures under sinusoidal forced vibration. Because the of effects of shear rigidity and rotatory inertia are taken into consideration. this *method* is particularly effective in the analysis of structures such as the ship hull structures where the effects of those factors cannot be ignored.

For vibrational analysis method of framed structures, a method in general use obtains natural frequencies in free vibration by replacing the structure to be analysed with a structure made up of multiple masses and spring system. However in framed structures where shearing deformation and rotatory inertia are taken into consideration. it is extremely difficult to obtain natural values by the use of the above method. Moreover, in replacing the structure with that of a multiple masses and spring system, adequate precision can not be obtained unless the members are approximated by as many masses and spring as possible. In consideration of the above fat-

tors. the author has assumed the framed structures to be a conglomerate of beams. which have infinite degree of freedom. and then their stiffness matrices were obtained, Using such matrices, their vibrational responses were calculated and as a result their vibrational modes and amplitudes were obtained. Accordingly, since the natural frequencies can not be obtained under this method, vibrational responses corresponding to the number of vibrations at the necessary number of points were obtained, then by reviewing the increase or decrease of vibrational amplitudes as well as the phase change of vibrational modes, the natural values can be obtained.

### 2. Solving method

2.1 Application and assumptions

The application and assumptions used in this program are outlined as follows:

- 1. The objective is the vibrational analysis of framed structures in elastic range under varying loads (including forced displacements) of sinusoidal form with a constant amplitude. The framed structures defined here are structures consisting of joints and straight members between joints, with uniform cross sectional areas. The members are assumed as ideal lings, neglesting the thickness, and the joints are assumed as ideal points. Therefore, the structures consisting of plates, curved bums, and tepered beams have to be analysed as struttures consisting of *respectively* equivalent beams by using proper approximation method, following the above assumptions.
- 2. The ends of the members of the framed structures can be pins, rollers, spring-supported, built-in connections, rigid attachments or free-ends. For loading conditions, the concentrated loads applied on joints

and the forced displacements at joints will be considered.

- 3. For vibrational mass, the dead weight of beams, the added mass uniformly distributed over the beams and the concentrated mass at joints will be, considered.
- 4. Damping factor of structures is not considered. Therefore, the amplitudes near the resonance point will become considerably larger than the actual values. However, at points 10% apart from the resonance frequency, it is safe to think that the amplitudes in the ordinary structures will be quite close to the actual values, even if damping factor is not considered.
- 5. The results of analysis calculated will be: The displacements and angular deflections at all joints, the displacements and angular deflections at quarter points of each member. stresses at ends each member. support reactions. etc. When necessary, vibrational mode curves of each structure can be plotted by the use of a plotter.
- 6. As for the analysis method, the stiffness matrix method derived from the displacement method is used, and by analysis of vibrational responses (but not taking into account of damping factor) the vibrational modes at the desired vibrational frequency are examined, and also by calculating at several levels of vibrational frequency the resonance frequency can be found.
  - 2.2 Stiffness matrix for vibration of beams having uniform cross section

Stiffness matrix of beams having uniform cross section is obtained. when sinusoidal forced load is applied at each end. where bending, shearing, torsion, elongation and rotatory inertia are taken into account. The symbols used in the equations will have the following meaning:

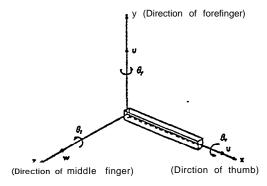


Fig. 1 Right handed coordinate system adapted in present study

- X: Coordinate in the direction along the length of beam.
- **Y. 2:** Coordinates perpendicular to the beam, in accordance with the right handed coordinate system (See in Fig. 1)
  - U: Displacement in x direction
  - T: Displacement in y direction

w: Displacement in : direction

 $v_B$ : Displacement only due to bending in y direction

 $w_B$ : Displacement only due to bending in : direction

 $\theta_x$ : Angle of rotation around x axis

 $\theta_{\mathbf{r}}$ : Angle of rotation around y axis

 $\theta_z$ : Angie of rotation around z axis

**GJ:** Torsional stiffness of beam

EI,: Bending stiffness of beam around y axis

 $EI_z$ : Bending stiffness of beam around z axis

 $A_x$ : Cross sectional area of beam

GA.: Shearing stiffness of beam in y direction

GA:: Shearing stiffness of beam in z direction

 $\mu$ : Weight of beam per unit length

 $I_{mx}$ : Rotatory inertia of beam per unit length around x axis

Rotatory-inertia of beam per unit length around v axis

Rotatory inertia of beam per unit length around

1: Total length

w: Forced circular frequency

F: Axial force in x axis direction

Fy: Shearing force in y axis direction

Fz: Shearing force in z axis direction

Mx: Torsional moment around x axis

My: Bending moment around y axis

Mz: Bending moment around z axis

m: Mass of concentrated material paint at joint.

J<sub>x</sub>: Rotatory inertia of concentrated mass at joint around x axis

Jy: Rotatory inertia of concentrated mass at joint around y axis

J<sub>z</sub>: Rotatory inertia of concentrated mass at joint around z axis

xG: Distance from joint to Center of Gravity of concentrated mass in x axis direction

JG: Distance from joint to Center of Gravity Of concentrated mass in y axis direction

**2G**: Distance from joint to Center of Gravity of concentrated mass in z axis direction

k<sub>x</sub>: Spring constant of support joint in x axis direction

**ky:** Spring constant of support joint in y axis direction

**k**<sub>z</sub>: Spring constant of support joint in z axis direction

 $k_{mx}$ : Torsional spring constant of support joint around x axis

 $k_{my}$ : Torsional spring constant of support joint around y axis

**k:** Torsional spring constant of support joint around z axis

(1) Longitudinal vibration of beam.

$$EA_x \frac{\partial^2 u}{\partial x^2} - \frac{\mu}{g} \frac{\partial^2 u}{\partial t^2} = 0 \qquad (1)$$

$$u = u_0(x) \sin \omega t$$
 .....(2)

and substitute equation (2) into (I)

$$\frac{d^2u_0}{dx^2} - a^2u_0 = 0 \quad \cdots \qquad (3)$$

where.

$$a^2 = \omega^2 \mu / E A_{\pi} g$$

Therefore, the solution of equation (3) can be given by the next equation

$$u_0(x) = A_1 \cos \alpha x - A_2 \sin \alpha x \cdots (4)$$

- A1, A2: constant of integration.
- (2) Lateral vibration of beam

The lateral vibration of beam having uniform cross section, where bending, shearing and rotatory inertia are taken into account, can be expressed as follows, according to the well known equation" of Timoshenko Firstly, the lateral vibration in y direction can be expressed by the next equation

$$\frac{\partial^{4}v}{\partial z^{4}} - \left(\frac{b_{z}}{p_{x}} + \frac{a}{q_{y}}\right) \frac{\partial^{4}v}{\partial x^{2}\partial t^{2}} + \left(\frac{a}{p_{z}} + \frac{\partial^{2}v}{\partial z^{2}} + \frac{b_{z}}{q_{y}} + \frac{\partial^{4}v}{\partial t^{4}}\right) = 0$$
 (5)

Here, since

$$a = \frac{ft}{g}$$
,  $b_t = \frac{I_{mt}}{g}$ ,  $p_t = EI_t$ ,  $q_t = GA_t$ 

$$|et \qquad r = v_0(x) \sin \omega t \qquad (6)$$

and substitute in equation (5)

$$\frac{d^4 v_0}{dx^4} + 2k_{1y} \frac{d^2 v_0}{dx^2} + k_{2y} v_0 = 0$$
 (7)

where:

$$2k_{1y} = \left(i\frac{b_y}{p_z} + \frac{a}{q_y}\right) \cdot \omega^2 \qquad (8)$$

$$k_{zy} = \frac{a}{p_z} \left( -\omega^2 + \frac{b_z}{q_y} \omega^4 \right) \qquad (9)$$

Hence. equation (7) can be expressed by the next equation

$$v_{\bullet}(x) = C_1 \cosh m_1 x + C_1 \cos \sinh m_1 x$$

$$+ C_2 \cos m_2 x + C_4 \sin m_2 x \cdots (10)$$

$$C_1, C_2, C_4, C_4: \text{ constant of integration}$$

$$m_1 = \sqrt{\sqrt{k_{1z}^2 - k_{2y} - k_{1y}}},$$

$$m_2 = \sqrt{\sqrt{k^2}_{1y} - k_{2x} + k_{1x}}$$

In the similar manner, the lateral vibration in z direction to be obtained as below.

$$\omega = \omega_0(x) \sin \omega t \qquad (11)$$

$$\omega_0(x) = D_1 \cosh n_1 x + D_2 \sinh n_1 x + D_3 + D_4 \cos n_2 x \sin n_2 x \qquad (12)$$

 $D_1, D_2, D_5, D_6$ : constant of integration  $n_1 = \sqrt{\sqrt{k_{1x}^2 - k_{2x} - k_{1x}}},$   $n_2 = \sqrt{\sqrt{k_{1x}^2 - k_{2x} + k_{1x}}}$ 

$$2k_{1z} = \left(\frac{b_z}{p_y} + \frac{a}{q_z}\right) \cdot \omega^2$$

$$k_{2z} = \frac{a}{p_z} \left(-\omega^2 + \frac{b_z}{q_z}\omega^4\right)$$

$$b_y = \frac{I_{my}}{\sigma}, \ p_y = EI_y, \ q_z = GA_z$$

(3) Torsional vibration of beam

$$GJ\frac{\sigma^2\sigma_x}{2} - \frac{J_{mx}}{g}\frac{\sigma^2\sigma_x}{\partial t^2} = 0 \qquad (13)$$

and substitute (14) into equation (13)

$$\frac{d^2\theta_{x0}}{dz^2} - \beta^2\theta_{x0} = 0$$
 (15)

where;

$$\hat{\beta}_2 = ar^2 I_{mx}/GJg$$

Therefore the solution of equation (is) can be given by the next equation

$$\theta_{x0}(x) = B_1 \cos \beta x + B_2 \sin \beta x \qquad (16)$$

### $B_1$ , $B_1$ : Constant of integration

(4) Stiffness matrix

Stiffness matrix is obtained by using equations (4). (10), (12). (16). Firstly, let

$$F_x = F_{x0}(x) \sin \omega t \quad M_x = M_{x0}(x) \sin \omega t$$

$$F_y = F_{y0}(x) \sin \omega t \quad M_y = M_{y0}(x) \sin \omega t$$

$$F_z = F_{x0}(x) \sin \omega t \quad M_z = M_{x0}(x) \sin \omega t$$

Here, considering that

$$\theta_{y} = -\frac{dw_{B}}{dx}, \quad M_{y} = EL_{z} \frac{d^{2}w_{B}}{dx^{2}},$$

$$F_{y} = -GA_{y} \frac{d^{2}(v - v_{B})}{dx^{3}}$$

$$\theta_{z} = \frac{dv_{y}}{dx}, \quad M_{z} = -EL_{y} \frac{d^{2}v_{B}}{dx^{2}},$$

$$F_{z} = -GA_{z} \frac{d^{2}(w - w_{B})}{dx^{3}}$$
(18)

where the related equations [33] J<sup>41</sup>, obtained by using compatibility conditions of the displacement and equibrium of forces, are expressed in matrix form, it will become as follows:

$$\begin{vmatrix} u_{4} \\ v_{0} \\ w_{6} \\ \theta_{x0} \\ \theta_{y0} \\ \theta_{x0} \\ \theta_{y0} \\ \theta_{y0} \\ \theta_{x0} \\ \theta_{y0} \\ \theta_{x0} \\ \theta_{y0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta_{x0} \\ \theta$$

where 
$$C_y = \left(1 - \frac{b_z}{q_y}\omega^2\right)$$
,  $C_z = \left(1 - \frac{b_y}{q_z}\omega^2\right)$ 

The equation (19) shall be rewritten in the form of next equation

$$\begin{bmatrix} D(\dot{x}) \\ F(x) \end{bmatrix} = B \begin{bmatrix} H_1(x) \\ H_2(x) \end{bmatrix} \qquad (20)$$

Where D(x), H1(x) are from the top to the 6th row. and F(x), H2(x) are the last 6 rows. Then, by using quations (4), (IO), (12), (16), the next equation can be 0btained.

Equation (21) will be expressed as in the next equation

$$\begin{bmatrix} H_1(x) \\ H_2(x) \end{bmatrix} = L(x) \cdot C \qquad (22)$$

By substituting equation (22) into equation (20), the following can be obtained:

$$\begin{bmatrix} D(x) \\ F(x) \end{bmatrix} = B \cdot L(x) \cdot C \qquad (23)$$

Therefore

$$C = (B \cdot L(x))^{-1} \begin{bmatrix} D(x) \\ F(x) \end{bmatrix} \qquad (24)$$

BY using equations (23), (24), the following can be obtained:

$$\begin{bmatrix} D(l) \\ F(l) \end{bmatrix} = B \cdot L(l) \cdot C = B \cdot L(l)(B \cdot L(0))^{-1} \begin{bmatrix} D(0) \\ F(0) \end{bmatrix}$$

$$= \begin{bmatrix} M_{11}, & M_{12} \\ M_{21}, & M_{22} \end{bmatrix} \begin{bmatrix} D(0) \\ F(0) \end{bmatrix} \dots (25)$$

Where the above takes the form of

$$\begin{bmatrix} M_{11}, & M_{12} \\ M_{21}, & M_{22} \end{bmatrix} B \cdot L(l) (B \cdot L(0))^{-1} \cdots (26)$$

then,  $M_{12}$ ,  $M_{21}$ ,  $M_{12}$ ,  $M_{22}$  are each a matrix of  $6 \times 6$ . From equation (25),

$$D(I) = M_{11} \cdot D(0) + M_{12} \cdot F(0) \cdot \cdots (27)$$

$$F(l) = M_{11} \cdot D(0) + M_{22} \cdot F(0) \cdots (28)$$

By using equations (27), (28) and applying the symmetry of stiffness matrix. the following equation can be obta in e d:

$$\begin{bmatrix} F(0) \\ F(l) \end{bmatrix} = \begin{bmatrix} K_{11}, K_{12} \\ K_{21}, K_{22} \end{bmatrix} \begin{bmatrix} D(0) \\ D(l) \end{bmatrix} = K \begin{bmatrix} D(0) \\ D(l) \end{bmatrix} \dots \dots (29)$$

where

$$K_{12} = (M_{12}M_{12}^{-1})^{-1}$$

$$K_{12} = K_{21} = M_{22}^{-1}K_{22}$$

$$K_{11} = M_{21}^{-1}(K_{21} - M_{21})$$
(30)

K in equation (29) is the stiffness matrix for vibration of beam having uniform cross section. Furthermore, when there is a concentrated mass m at the starting point or ending point of beam, the following matrix should be added to  $K_{11}$  or  $K_{22}$ .

Moreover, when the joint is spring supported,  $k_x$ ,  $k_y$ ,  $k_z$ ,  $k_{mx}$ ,  $k_{my}$ ,  $k_{my}$ ,  $k_{mz}$ , etc. should be added to the terms in diagonal line of the matrix for that joint.

2.3 Solution of vibrational stiffness equation

Using the equation (29) derived in the preceding section. the stiffness matrix for each member will be prepared. These matrixes and external loads are placed in absolute coordinates, by carrying out transformation of Wrdinates. Then the stiffness matrix for the entire structure will be prepared by adding the matrices of ail joints. When the inversed matrix of this stiffness matrix

for the entire structure is obtained and when it is multiplied by the external loads, the vibrational displacement and vibrational modes for each joint in that structure, in the case of circular frequency  $\omega_{\bullet}$  can be obtained. By changing this  $\omega$  little by little and repeating the above calculations. it is possible to obtain the resonance curve of that structure. In the actual computational programs, calculations for several vibrational frequencies art carried out simultaneously.

Because the stiffness matrix of the entire structure, in the case of three-dimensional framed structures, will become a matrix, of a size 6 times the number of general joints, it is not possible to directly calculate& this inversed matrix, from the standpoint of computer capacity, when a structure contains very many joints. Therefore, the author has adopted the unit splitting method, which is generally used. The author. however, invented a method which will eliminate the need of splitting the structure into several units by the user himself, which is very troublesome work and is considered as shortcomings of unit splitting method. Since the details of this system had already been presented121, its explanation will be omitted here. Furthermore, assumptions concerning the relation between the structural member coordinate system and the absolute coordinate system. and the releases at joints and at ends of members are the same as the above reference(z).

### 3. Program

The program was prepared in accordance with the contents as explained above. The points which were specially considered in the preparation of this program are as follows:

- 1. The sectional properties and dead weight of each member will be automatically calculated by the input of in dimensions and specific gravity.
- 2. The vibrational modes can be drawn by the plotter, thereby the results of analytical work can be quickly and appropriately grasped by appealing to vision.
- Attempts have been made to minimize the user's efforts by providing the error messages at each check point, thereby permitting early detection of erroneous data.
- 4. Considerations have been made to permit computational capacity of structures having large number of joints and members. At present, the maximum computational capacity is as follows:

Number of joints: 800 Number of members: 1,600 Number of loading conditions: 16

the computer employed in this system is UNIVAC-1108, and the plotter is CALCOMP.

- 4. Example of applications
- 4.1 Example of vibrational analysis of radar mast An example of analysis of radar mast in an escort ship, as shown in Fig. 2, will be introduced here. This

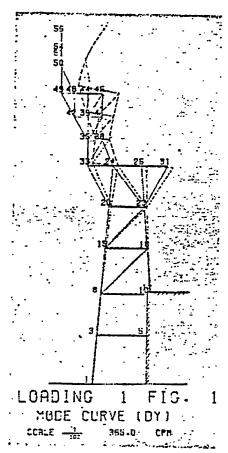


Fig. 2-(a) Vibration mode of radar mast (first order in fongitudinal direction)

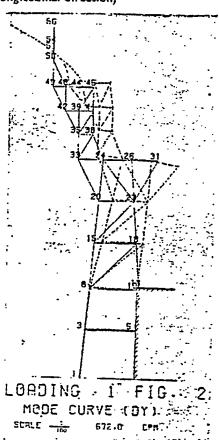


Fig. 2-(b) Vibration mode of radar mast (second order in iongitudinal direction)

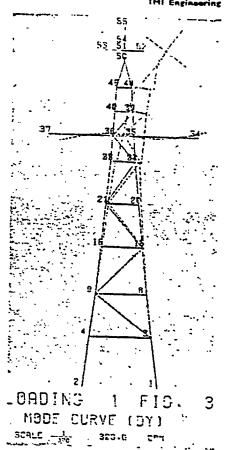


Fig. 2-(c) Vibration mode of radar mast (first order in port and starbord direction)

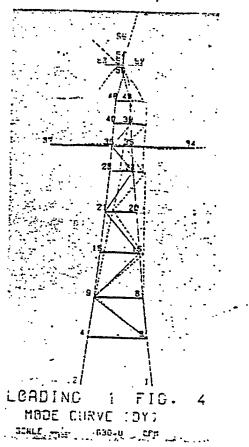


Fig. 2-(d) Vibration mode of radar mast (second order in port and starbord direction)

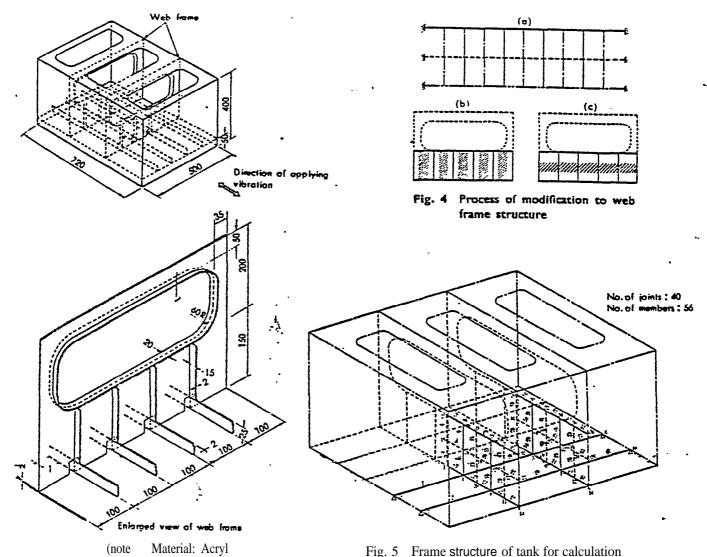


Fig. 3 Model of experiment (mm)

Elastic modulus: 460 kg/cm<sup>2</sup>

radar mast is a three-dimensional framed structure made up of steel pipes. The radar and other equipment are attached to 14 joints near the top of the mast. and they jic considered to constitute a concentrated masses.

The dead weight of steel pipes, which constitute the framed structure will be computed as distributed mass in the program by inputting pipe diameter, thickness and specific gravity, and the distributed mass will be applied to each member. More over, since the dimensions of each member are given the sectional properties of each member will be automatically computed in the Program. In addition, the coordinate values and boundary conditions of each joint will be input. As for the hud, forced displacements, consisting of sinusoidal waves of same amplitude, were applied to 10 support wants: at the base of radar, in the directions along the length of ship as well as toward port and starboard.

The 'computational results of vibrational modes are Drawn by the plotter as shown in Fig. 2 The resonance Points in port and starboard directions were 320 cpm rig. 5 Traine structure of talk for calculation

for first order and 630 cpm for second order. and in Iongitudinal direction 365 cpm for first order and 672 cpm for second order. Moreover, because the weight distribution of this mast is not symmetrical longitudinally, a small amount of torsion will be generated in lateral vibrations.

## 4.2 Example of vibrational analysis of web frame in t a n k

Recently. with the increase in ship size, there are sometimes damages of web frames in tank near the stem, due to vibrational forces. In order to investigate the behavior of vibration on web frames in tank, a model test was done by the technical research laboratory of our company. A comparison between the test results and the analytical results by this program ZVIBRA has been made.

### (I) Model test

A test model shown in Fig. 3 consists of tank, made of acrylic resin. in which 2 web frames were installed. The model was mounted on the vibrator base, then vibration was applied on to the model in longitudinal direction (perpendicular to surface of web frame), and

vibrational responses and vibration modes were measured for the cases of with and without water in the tank.

Measurements were made in the following manner, namely. 8 accelerating type pick-ups (weight: 1.5 g) were attached to the web frames, and pick-up was fixed onto the vibrator base. By changing the vibrational frequency both acceleration and phase were measured for each vibrational frequency. The positions where the pick-ups were attached arc shown in Fig. 6.

### (2) Computation by this program

Because this program is applicable only to framed structures, it is necessary to modify the type of structure where it is close to a plate, as in the case of web frame, into a model of frame-type structure. According to test results (Fig. 6), it is clear that the panel itserf between stiffeners also vibrates considerably. Therefore, the conventional method of solving the web frame stiffness problems. in assuming the web frame to be substituted by a grillage structure made up of stiffeners with appropriate effective width, as generally used, is not useful. Therefore, the panel between **stiffeners** was also assumed lo constitute a part of the grillages, and the fixing condition of web frame ends was also taken into consideration. Then computations were carried out with the assumptions as below:

- 1. The portion shown by the dotted lines in Fig. 4-(b) will be ignored. and it is assumed that the web frame is made up of panels and stiffeners shown by solid lines only.
- 2 The panel between stiffeners is assumed 'to constitute a part of grid. thereby the web frame is assumed to be a grillage structure in Fig. 4-(a). Further, lines in Fig. 4-(a) show the following items:

Heavy continuous line:

Stiffness and weight will be determined by assuming a member made up of a vertical stiffener plus an effective width equal to a half of the panel between stiffeners.

Alternate long and short dash line:

The shaded portion in Fig. 4-(b), namely, stiffness and weight will be computed by taking up the central tral half of the panel between stiffeners. Dotted line:

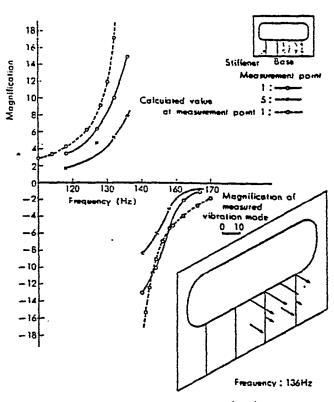
An area having a width equal to 1/2 of panel in horizontal direction is taken up in the middle of panel (shaded portion in Fig. 4-(c)), and it is assumed that there is a member which has stiffness equivalent to the above dimensions but has no weight.

Light continuous line:

The face plate only on upper edge, and its stiffness and weight will be determined.

Alternate long and two short dashes line:

The lower plate of web frame and 1/2 of tank bottom plate are assumed to contribute to the stiffness and only the latter item will be accounted for weight.



(Note) Magnification= Measured acceleration
Acceleration on bibrator base
(Negative sign if in inverse phase)

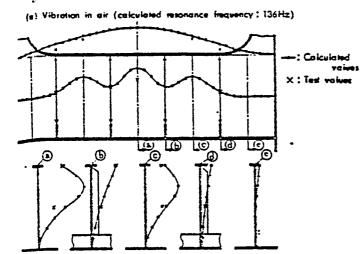
Fig. 6 Resonance curves and vibration mode of web frame in air

- 3. Both left and right ends of the grillage shown in *Fig.* 4-(a) are supposed to be rigidly fixed to both walls of sufficiently rigid model tank.
- 4. The lower end of web frame which is assumed to be of grillage structure is rigidly fixed to the tank bottom stiffeners at the bottom of vertical stiffeners. and the stiffness and weight of the tank bottom stiffeners is computed by including the entire effecive width of the panel.
- 5. The entire model is handles as a framed structure shown in *Fig. 5*. In this case since the model is symmetrical about the center of tank, only a hail of the above model will be considered and at the center only a half of the stiffness and weight of a member is taken. and its rotatory displacement was ssumed to be zero.

As for the loads used in computation. an attempt was made to produce the same conditions as in the case of test by applying forced displacements of sinusoidal type onto joints 3, 6, 7, 10, 13, 14. 25,26, 37.38, in *Fig. 5*.

(3) Comparison of test and computational results

Frequency response curves and vibration modes of the model tested in **air** arc shown in *Fig.* 6, and the frequency response curve was plotted from the computations at measurement point-i. Further, in Fig. 7 a comparison is made of measured and computed



(b) Vibration in water (calculated resonance frequency: 19Hz)

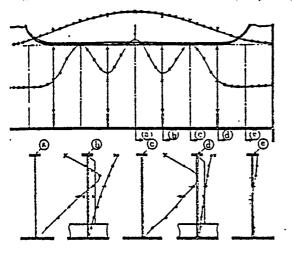


Fig. 7 Measured and computed vibration mode of web frame

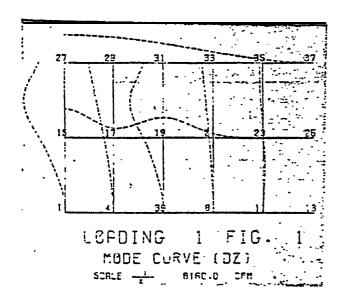


Fig.8.(a) Output from CALCOMP plotter of vibration mode curve of web frame in air (1)

vibration modes. **both in air and in water.** However. in the case of computation of vibrations in **water**. the virtual mass of water was estimated from the frequency of both in air and in water according to tests. and those values were used in the computation.

As can be seen from Fig. 6, the calculated value of resonance frequency is 136Hz which agrees fairly well with the test value. However, the magnification of vibrations shown to be of large value near the resonance point. because damping factor is not taken into account, and the calculated values approach to the test values as the frequency moves away from the resonance point. In Fig. 7, when the test and computative values are taken equal at measurement point-l, a comparison of the two values at other points is shown. As can be seen from the above, the form of vibration mode agrees fairly well with test results. and the mode from computation clearly shows that the panel between stiffeners is in coupled vibration. Examples of output mode by the plotter in drawing the vibration mode curves are shown in Fig. 8.

## 43 Example of analysis of coupled vibration between hull and bottom

As one method of analyzing the vibration of hull, the' hull **can** be considered as a beam with varying cross section, and it has been ascertained that computation based on the above assumption produces adequately precise results, if the degree of vibration is rather low. However it is known that in high degree vibration having more than 4 nodes or so, unless the coupled vibration with local vibrations arising from bottom, etc. is-not taken into account, it is difficult to carry out accurate analysis.

As an example of analysis of coupled vibration between hull and bottom (double bottom), the case of Ship A (a bulk carrier) will be explained below:

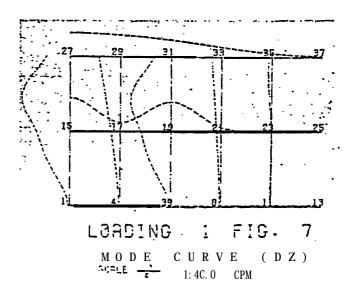


Fig. 8-(b) Output from CALCOMP plotter of biration mode curve of web frame in air (2)

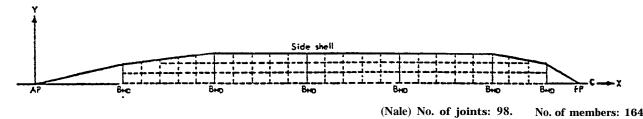


Fig. 9 Modification to plane grid of side shell. transverse bulk had and double bottom

### (1) Computative assumption

In **carrying** out the analysis the following assumptions have been made:

- 1. The hull will be approximated by a plane grid as shown in Fig. 9. **Since the hull** is a symmetrical structure at port and starbord, of the hull need be considered. Namely, by letting the vertical displacements of all joints on the centerline of ship as well as the rotatory displacements around Y axis free, and also by letting other components displacements constrained, the results would be the same as if both sides were computed. As for the means of approximating the structure by plane grids, the double bottom structure is shown by a plane grid with dotted lines, and both side structure and transverse bulkheads by a grid with continuous lines.
- 2. The stiffness of each structure was-taken as below: Double bottom:

The double bottom was split into 4 transverse girders and 3 longitudinal girders, then their stiffness was computed. The torsional stiffness was computed by using a value assuming that the double bottom **is** made of plates with anisotropic p l a t e .

### Side shell:

One half of the longitudinal stiffness of ship {moment of inertia and the effective shearing area used in the so-called longitudinal strength of ship) was used. For the torsional stiffness of double bottom, the torsional stiffness of hopper, which is connected with the double bottom floor was used.

### Transverse bulkhead:

The bending stiffness was taken to be very large, and computing the shearing stiffness the cross sectional area of bulkhead plate was used.

3. The distribution of hull weight has been allocated as below:

### Double bottom:

The dud weight of double bottom and the ballast within double bottom were uniformly distributed to transverse girders.

#### Side shell:

Both hull weight and ballast other than those of the double bottom were distributed, according to the respective distribution in longitudinal din tion. so that the resultant distribution in o member would be uniform.

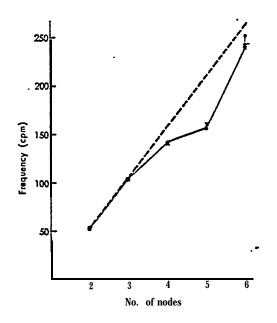
### Transverse bulkhead:

The dead weight of transverse bulkhead is to included in the side shell.

4. Virtual mass of the water, as obtained using t method of Lewis<sup>(6)</sup> was distributed to the ion tudinal girders of double bottom, and where the is no longitudinal girder. it was distributed to t side shell.

Table: Comparison between calculated and measured natural frequencies of ship-A

Node (Category	2	3	4	5	6
measured values	:33	103	140	160	(252
Calculated values	52	104	141	156	(23° 25°



(Note) - - -: calculated values

X : Measured
-----: calculated values uunder assumption
that hull is abeam with varying
cross section.

Fig. 10 Natural frequency versus number of nodes c ship-A

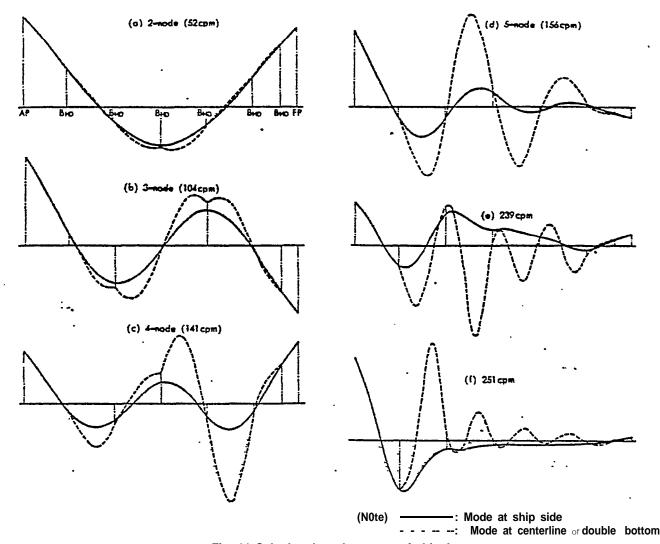


Fig. 11 Calculated mode curves of ship-A

- 5. As for the load, vertical forces of sinusoidal form were applied onto joint at **stern** end.
- (2) Results of analysis and comparison with actual measurement results

Under the above assumptions. the vertical hull vibration of Ship A was computed. Table I contains the comparison between calculated and measured values of natural frequencies, and Fig. 10 shows the comparison between calculated and measured results of natural frequency curves. The dotted line in the above figure represents the calculated values when the hull is assumed of a beam having varying cross section. Moreover. the frequency with 6 nodes in the above table is actually not of a mode with 6 nodes. and because only the stern is vibrating, its frequency is equivalent to 6 nodes but not of 6 nodes in strict sense.

As can be seen from Fig. 10, the values obtained from calculations agree very well with measured values. when calculations were made by assuming a beam with varyi ng cross section. as a matter of course the coupling effects between hull and double bottom can not be accounted

for. and because the resonance frequency varies linearly as the number of nodes increases. the calculated results will differ considerably from the measured values starting from around the 4th node.

Vibration modes of side shell and centerline of double bottom at the resonance point. according to the calculation of this system is shown in Fig. II. It can be seen that the vibration of double bottom increases starting from around the 4th node. Moreover. the fact that at resonance points of 239 cpm and 251 cpm the nodes on side shell disappear and only the stern portion vibrates will provede a good explanation on the actual cases where stem vibrations frequently occur in higher degree vibrations.

### 5. Conclusions

With the stiffness matrix method, a general **purpose** program ZVIBRA for vibrational analysis of three-dimensional and two-dimensional framed **structures** has been prepared. Its outline and several examples of its application have been presented, showing that

there have been good agreements between theory and test results. As a target for future studies, it is necessary to continue research which takes into account of the damping factor of vibration.

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- (2) I. Neki: Matrix Method of Structural Anaiysis of Framed Structures and its Application, Ishi-kawajima-Harima Engineering Review Vol. 9 No. **3**, May 1969, p. 266.
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- tion. IHI Engineering Review Vol. 3 No. 4 March 1970 p. 7
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- (5) H. **Nagai: Natural** Vibration of Beam with Varying Cross Section, IHI Engineering **Review** Vol. 8 No. 40 March 1968 p. 183
- (6) K. Suetsugu: On Vibration of Ship Bottom. Kansai Shipbuilding Association, No. I I I October 1963
- (7) F. M. Lewis: The inertia of the water surrounding the vibrating ship, Trans. S. N. A. M. E. Vol.37 (1929)

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### V. RECOMMENDATIONS

The current upgrading of facilities and equipment related to Numerical Control Steel Fabrication, Sub-task 2.2 of the T.T.P., has produced a system which is capable of increasing the tons per month of fabricated steel. The related N/C software package, SPADES, currently provides more data than the system is utilizing. It is the common opinion of IHI and key Levingston Production and Engineering personnel that an N/C system of scaled body plan mold lofting be implemented through the installation of a numerically controlled drafting machine in the mold loft.

Please see Reference 1 which documents the rationale for this decision.

### VI. REFERENCES

- 1. Study of LSCo N/C System and Description of IHI N/C System-IHI
- 2. Comparison Analysis: LSCo vs. IHI N/C Systems-Levingston
- 3. Memo: RW Taylor to Clyde LaRue; April 24, 1979; N/C Drafting Machine
- 4. Study of SPADES and LSCo Utilization-IHI
- 5. System Change Analysis
- 6. Memo: K. Honda to Bob. Peterson, June 20, 1979; Merits for Installation of N/C Drafter Machine.

# APPENDIX M IHI WORKING FLOW AND SCHEME FOR HULL STRUCTURE DESIGN

# <u>APPENDIX N</u> <u>EXPLANATION OF IHI'S DESIGN FLOW (PIPING)</u>

